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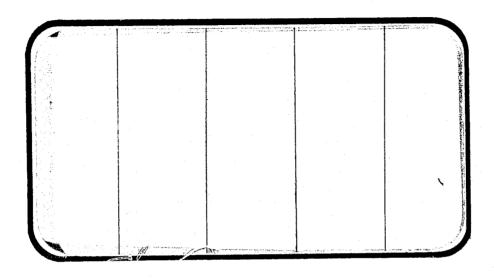
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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT



JOHNSON SPACE CENTER HOUSTON, TEXAS

DATA MAN agement services

SPACE DIVISION CHRYSLER

DMS-DR-2217 NASA CR-141,844

VOLUME 1 of 3

AERODYNAMIC RESULTS OF A SEPARATION TEST (CA20)

CONDUCTED AT THE BOEING TRANSONIC WIND TUNNEL USING 0.030-SCALE MODELS OF THE CONFIGURATION 140A/B (MODIFIED) SSY ORBITER (MODEL NO. 45-0) AND THE BOEING 747 CARRIER (MODEL NO. AX 1319 I-1)

by

T. Dziubala, V. Esparza, R. L. Gillins and M. Petrozzi Shuttle Aero Sciences Rockwell International Space Division

Prepared under NASA Contract Number NAS9-13247

by

Data Management Services Chrysler Corporation Space Division New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center
National Aeronautics and Space Administration
Houston, Texas

WIND TUNNEL TEST SPECIFICS:

Test Number:

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NASA Series Number:

CA20

Model Number:

45-0 Mod/747 Carrier AX 1319 I-1

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Occupancy Hours:

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FACILITY COORDINATOR

AERODYNAMICS ANALYSIS ENGINEERS:

B. Sendek The Boeing Company

Orgn. B-8342 MSIW-82

Seattle, Washington 98007

Phone: (206) 655-3037

W. L. Osborn and J. F. Kerswell

Rockwell International Mail Code ACO7

12214 Lakewood Blvd.

Downey, California 90241

Phone: (213) 922-5049

PROJECT ENGINEERS:

T. Dziubala, V. Esparza R. L. Gillins, M. Petrozzi Rockwell International

Space Division

12214 Lakewood Blvd. Mail Code ACO7

Downey, California 90241

Phone: (213) 922-4898

C. R. Mullen

Boeing Aerospace Company

M. S. OT-55

P. O. Box 3999

Seattle, Washington 98124

Phone: (206) 342-1220

DATA MANAGEMENT SERVICES:

Prepared by:

Liaison--D. A. Sarver

Operations--R. H. Lindahl

Reviewed by:

D. E. Poucher

Approved:

J. L. Glynn, Manager

Data Operations

Tomela Concurrence:

N. D. Kemp, Manager

Data Management Services

Chrysler Corporation Space Division assumes no responsibility for the data presented other than display characteristics.

AERODYNAMIC RESULTS OF A SEPARATION TEST (CA20)

CONDUCTED AT THE BOEING TRANSONIC WIND TUNNEL

USING 0.030-SCALE MODELS OF THE CONFIGURATION

140A/B (MODIFIED) SSV ORBITER (MODEL NO. 45-0) AND

THE BOEING 747 CARRIER (MODEL NO. AX 1319 I-1)

by

T. Dziubala, V. Esparza, R. L. Gillins and M. Petrozzi Rockwell International Space Division

ABSTRACT

An experimental aerodynamic investigation (CA20) was conducted in the Boeing Transonic Wind Tunnel from October 9 through October 16, 1974. A Rockwell built 0.030-scale 45-0 modified SSV Orbiter Configuration 140A/B model and a Boeing built 0.030-scale 747 carrier model were tested to provide six component force and moment data for each vehicle in proximity to the other at a matrix of relative positions, attitudes and test conditions. Orbiter model support system tare effects were determined for corrections to obtain support-free aerodynamics.

In addition to the balance force data, pressures were measured. Pressure orifices were located at the base of the Orbiter, on either side of the vertical blade strut, and at the mid-root chord on either side of the vertical tail. Strain gages were installed on the Boeing 747 vertical tail to indicate buffet onset.

The 747 carrier was varied through angles of attack (measured with respect to its FRL) of 0° , 2° , 4° , 6° , 8° , and 10° and varied through sideslip

angles of 0° , $+5^{\circ}$, and -5° . Elevator settings were also varied.

The SSV Orbiter model was varied through angles of attack of 6° , 8° , 10° , 12° , 14° , 16° , and 18° and varied through sideslip angles of 2.5°, 0° , -2.5° , -5° , -7.5° , -10° , and -15° .

Vertical displacements of 0", 1", 2", 3", 5", 7", 9", 11", 13", 15", 18", and 21.6" (model scale) were tested. Longitudinal movements of 0", 3.6", and 7.2" (model scale) and lateral displacements of 0" and 3.6" (model scale) were tested to simulate various separation positions. Orbiter elevon deflections were also varied.

Orbiter support system tare and interference effects were determined utilizing various support and image support strut configurations. Carrier support system tare and interference effects were determined during test CA5.

The Orbiter tail cone and carrier models were provided by The Boeing Company. The Orbiter model was provided by Rockwell. These were the same models used earlier in test CA5.

This report for CA20 consists of three volumes: Volume 1 - data figures 1 through 25; Volume 2 - data figures 26 through 39; Volume 3 - tabulated source data.

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- (B) CN, CLM, CA, CL, CD, CY, CYN, CBL versus DZ
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NOMENCLATURE

Symbol	Plot Symbol	Description
b ·	BREF	reference span, in
BSTA	XC ,	longitudinal carrier station, in
BWL	ZC	vertical carrier station, in
c	LREF	mean aerodynamic chord, in
c_{A}	CA	axial force coefficient
c_D	CD	drag coefficient
C _{LB}	CBL	body axis rolling moment coefficient
C _{ls}	CSL	stability axis rolling moment coefficient
c_{L}	CL	lift coefficient
C _m	CLM	pitching moment coefficient
c_{η_B}	CYN	body axis yawing moment coefficient
c_{η_S}	CLN	stability axis yawing moment coefficient
CN	CN	normal force coefficient
С _{РВ1}	PB1	Orbiter base pressure coefficient for orifice no. 1, see Figure 2c
$c_{P_{B_2}}$	PB2	Orbiter base pressure coefficient for orifice no. 2, see Figure 2c
$c_{P_{B_4}}$	PB4	Orbiter base pressure coefficient for orifice no. 4, see Figure 2c
CPCAV	PCAV	Orbiter cavity pressure coefficient

Symbol	Plot Symbol	<u>Definition</u>
c _{PEB1}	LHLS	coefficient of pressure measured on fuselage at left side of vertical tail
c _{PEB2}	RHLS	coefficient of pressure measured on fuselage at right side of vertical tail
c _p sc	PSC	carrier cavity pressure coefficient
c _p s ₁	LHVERT	coefficient of pressure measured on left side of Orbiter strut
c _p s ₂	RHVERT	coefficient of pressure measured on right side of Orbiter strut
CY	CY	side force coefficient
C.G.		center of gravity
C.R.	•	center of rotation
FRL		fuselage reference line
^L o	IORB	Orbiter incidence relative to carrier FRL, deg.
LB	LREF	reference body length, in
MACH	MACH	Mach number
M.R.C.	XMRP,YMRP ZMRP	moment reference center, in
MS		model station, in
P _{Bi}		base pressure measured at station i, i=1,2,4, psia
P_{EB_1}		pressure measured on Orbiter fuselage surface on left side vertical tail/fuselage juncture, psia

Symbol	Plot Symbol	<u>Definition</u>
PEB ₂		pressure measured on Orbiter fuselage surface on right side vertical tail/fuselage juncture, psia
PS ₁		pressure measured on left side of Orbiter strut S1, psia
P _{S2}		pressure measured on right side of Orbiter strut S ₁ , psia
q	Q(PSF)	freestream dynamic pressure, psf
RN/ET	RN/L	freestream unit Reynolds no., 10 ⁶ per foot
V .		mean freestream velocity, ft/sec
S	SREF	wing area or reference area, ft ²
WL	Z	water line, in
X		longitudinal Orbiter separation distance, measured from nominal mated position, ft
Х _с	хс	carrier longitudinal station, in
X _{MRP}	XMRP	longitudinal location of MRC, in
Χo	хо	Orbiter longitudinal station, in
Y		Orbiter lateral separation distance, measured from nominal mated position, ft
YC	YC	carrier lateral station, in
YMRP	YMRP	lateral location of MRC, in
Yo	YO	Orbiter lateral station, in
Z		Orbiter vertical separation distance, measured from nominal mated position, ft
ZC	ZC	carrier vertical station, in

Symbol	Plot Symbol	<u>Definition</u>
Z _{MRP}	ZMRP ,	vertical location of MRC, in
Z _o	ZO	Orbiter vertical station, in
1/(∆Z+10)	1/Z+10	separation parameter, inverse of vertical separation distance plus 10 ft, per foot
α	ALPHA	angle of attack, deg.
αC	ALPHAC	carrier fuselage angle of attack, $\alpha_W^2-2^\circ$, deg.
α0	ALPHAO	Orbiter angle of attack, deg.
α _W	ALPHAW	carrier wing angle of attack, α_c + 2°, deg.
^α Wall	ALPWAL	wind tunnel wall correction to carrier angle of attack, deg.
β	ВЕТА	angle of sideslip, deg.
βC	BETAC	carrier sideslip angle, deg.
^β 0	BETA0	Orbiter sideslip angle, deg.
δ _a	AILRON	aileron deflection angle, deg.
δe	ELEVON	Orbiter elevon deflection angle, deg.
⁵eI	ELV-IB	inboard carrier elevator panel deflection angle, deg.
်e φ	ELV-OB	outboard carrier elevator panel deflection angle, deg.
δeγ	ELEVTR	carrier elevator deflection angle, deg.
s _r	RUDDER	carrier rudder deflection angle, deg.
δrL	RUD-L	carrier lower rudder panel deflection angle, deg.

Symbol	Plot Symbol	Definition
$^{\delta}\mathbf{r_{u}}$	RUD-U	carrier upper rudder panel deflection angle, deg.
δS		spoiler deflection angle, deg.
$^{\Delta C}$ A	DCA	incremental axial force coefficient
ΔCD	DCD	incremental drag coefficient
$^{\Delta C}$ _{&B}	DCBL	incremental body axis rolling moment coefficient
ΔC _L S	DCSL	incremental stability axis rolling moment coefficient
ΔCL	DCL	incremental lift coefficient
ΔC _m	DCLM	incremental pitching moment coefficient
ΔC _{ηB}	DCYN	incremental body axis yawing moment coefficient
ΔC _n S	DCLN	incremental stability axis yawing moment coefficient
ΔCN	DCN	incremental normal force coefficient
ΔCγ	DCY	incremental side force coefficient
ΔΧ	DX	Orbiter longitudinal separation distance from nominal mated position, ft
Δ Y	DY	Orbiter lateral separation distance from nominal mated position, ft
ΔΖ	DZ	Orbiter vertical separation distance from nominal mated position, ft
Δα	DALFA	incremental angle of attack between Orbiter and carrier FRL, α_0 - α_c , deg.

NOMENCLATURE (Concluded)

Symbol .	Plot Symbol	Definition
Δβ	DBETA	incremental angle of sideslip between Orbiter and carrier, β_0 - β_c , deg.
Δφ	DPHI	incremental roll angle between Orbiter and carrier, deg.
ф	PHI	Orbiter roll angle, deg.
c_{L_C}	CL-C	carrier lift coefficient with test mounting system corrections
CDC	CD-C	carrier drag coefficient with test mounting system corrections
CmC	CLM-C	carrier pitching moment coefficient with test mounting system corrections
CYC	CY-C	carrier side force coefficient with test mounting system corrections
C _{nBC}	CYN-C	carrier body yaw moment coefficient with test mounting system corrections
c _{nSC}	CLN-C	carrier stability yaw moment coefficient with test mounting system corrections
c_{LSC}	CSL-C	carrier stability roll moment coefficient with test mounting system corrections
c _{rbc}	CBL-C	carrier body roll moment coefficient with test mounting system corrections
c _A c	CA-C	carrier axial force coefficient with test mounting system corrections
c _{NC}	CN-C	carrier normal force coefficient with test mounting system corrections

REMARKS

The Orbiter axial force, measured during this test, exhibits the following trend:

- at low angles of attack, axial force decreases with increasing angle of attack, as would normally be expected,
- 2) at high angles of attack, axial force increases with increasing angle of attack, contrary to normal expectations.

Extensive investigations and analysis, conducted during the test, indicated that trend number (2) was not caused by model fouling or other test problems and was, indeed, representative of aerodynamic characteristics.

Vertical tail pressure instrumentation (P_{EB_1} and P_{EB_2}) was disconnected during runs 588 through 599.

Configuration D (as described in figure 2f) was not at ϕ = 90°, as planned, because of support system deflections (caused by the Orbiter model touching strut S₃).

CONFIGURATIONS INVESTIGATED

The Orbiter model was an 0.030-scale representation of the Space Shuttle Orbiter VL70-000140A/B lines with modified OMS pods and elevons as shown in figure 2a. The basic Orbiter is a blended wing-body design with a double delta wing (75° and 45° leading edge sweeps). The Orbiter model was tested both with and without a tail cone fairing. The tail cone fairing covered the MPS nozzles, OMS nozzles, and base, as shown in Figure 3b. The Orbiter model was mounted in the tunnel using several blade atrut configurations as follows:

 S_1 = Orbiter support blade strut, upper entry position,

 S_2 = Orbiter support blade strut, lower entry position,

 S_3 = Orbiter dummy support blade strut.

Figure 2f shows the strut arrangements. Orbiter elevon and aileron deflection angles were varied. The Orbiter was tested both isolated and in the presence of the carrier at various separation locations. The following Orbiter configurations were tested:

$$O_1 = B_{26} C_9 E_{43} F_8 M_{16}$$
 $W_{116} T_{5.1}$
 $O_2 = B_{26} C_9 E_{43} F_8 M_{16} N_{28} N_{24}$ W_{116} (with strut S_1)
 $O_3 = B_{26} C_9 E_{43} F_8 M_{16}$ $R_5 V_8 W_{116} T_{5.1}$
 $O_4 = B_{26} C_9 E_{43} F_8 M_{16}$ $W_{116} T_{5.1}$
 $O_5 = B_{26} C_9 E_{43} F_8 M_{16} N_{28} N_{24} R_5 V_8 W_{116}$
 $O_6 = B_{26} C_9 E_{43} F_8 M_{16} N_{28} N_{24} R_5 V_8 W_{116}$ MPS cover plate off

CONFIGURATIONS INVESTIGATED (Continued)

⁰ 7	=	^B 26	С ₉	E ₄₃	F ₈	M ₁₆	N ₂₈	R ₅	٧8	^W 116	strut, S ₂ cover base plate off	plate	#1	off	MPS
08	=	^B 26	с ₉	E ₄₃	F ₈	M ₁₆	N ₂₈	R ₅	٧8	W ₁₁₆	strut, S ₂ cover base plate off	plate	#2	off	MPS
09	=	B ₂₆	c ₉	E ₄₃	F ₈	M ₁₆	N ₂₈	N ₂	4	W ₁₁₆	(with strut S_2)				

where:

Component	<u>Description</u>
^B 26	Orbiter fuselage per Rockwell lines VL70-000140A/B, model drawing SS-A01360
c ₉	Orbiter canopy per Rockwell lines VL70-000140A/B, model drawing SS-A01360
E ₄₃	Orbiter full-span, unswept hingeline, 6" gapped elevons per Rockwell lines VL70-000200, model drawing SS-A01360
F ₈	Orbiter body flap per Rockwell lines VL70-000200, model drawing SS-A01360
^M 16	Orbiter OMS/RCS pods per Rockwell lines VL70-000203A, VL70-008401, model drawing SS-A01360
N ₂₄	Orbiter main propulsion system (MPS) nozzles - VL70-000140A, VL70-005030A, model drawing SS-A01360
N ₂₈	Orbiter OMS nozzles - VL70-000140A model drawing SS- A01360
R ₅	Orbiter rudder per Rockwell lines VL70-000146A, model drawing SS-A01360
TC _{5.1}	Orbiter tail cone fairing which covers the MPS nozzles and the OMS nozzles and base, built by the Boeing Company, also used in CA5
٧8	Orbiter centerline vertical tail per Rockwell lines VL70-000146A, model drawing SS-A01360
W ₁₁₆	Orbiter double delta wing per Rockwell lines VL70-000200, model drawing SS-A01360

CONFIGURATIONS INVESTIGATED (Continued)

19

Effects of simulated attach hardware were investigated using the following model components attached to the carrier.

AT 38 Forward attach structure between the Orbiter and carrier model used for i_0 of 3 to 10 degrees for $\Delta Z = 0$ feet

AT 39 Aft attach structure between the Orbiter and carrier model for $\Delta 7 = 0$ feet

The carrier model was an 0.030-scale representation of the Boeing 737-100 aircraft with surface contours built to represent the 747 under loads it would experience with a 600,000 pound gross weight flying at Mach 0.86 at an altitude of 35,000 feet. The model also had a built in 0.64° leading edge up wing tip twist to compensate for model aeroelastic effects, which are estimated to produce a 0.64° leading edge down twist. The carrier had 200 square foot tip fins on its horizontal tail. Spoilers were deflected to 45° and flaps were retracted during most of the test. Several runs were made with spoilers retracted. Elevator and rudder deflections were varied during the test. The carrier was tested both isolated and in the presense of the Orbiter at various separation conditions. Carrier configurations investigated were:

$$747/0 = B_{27.8} F_0 H_{15.6} M_{26}^{25} N_{58}^{57} T_{19} V_{9.1} W_{44.1}$$

 $747/1 = B_{27.8} F_0 H_{15.6} M_{26}^{25} N_{58}^{57} S_{1-12} T_{19} V_{9.1} W_{44.1}, \delta_S = 45^{\circ}$

where:

 $\begin{array}{c} \underline{\text{Component}} & \underline{\text{Description}} \\ B_{27.8} & \text{fuse lage} \\ F_{0} & \text{all flaps retracted} \end{array}$

CONFIGURATIONS INVESTIGATED (Concluded)

H _{15.6}	horizontal tail (H ₁₅) with 200 ft ² tip fins
M ²⁵ M ²⁶	inboard (M_{25}) and outboard (M_{26}) nacelle struts
N ₅₈	inboard (N_{57}) and outboard (N_{58}) nacelles
s ₁₋₁₂	12 spoiler panels located on wing upper surface, all deflected 45°
T ₁₉	flap track fairing
V _{9.1}	vertical tail
W _{44.1}	wing

Orbiter base pressures were measured, for configurations without tail cone, at locations as shown by figure 2c. Pressures were measured on both sides of Orbiter support strut when S_{\parallel} was used and pressures were measured on the fuselage near the vertical tail when the vertical tail was installed as shown by figure 2d. Pressures were measured in the Orbiter and carrier balance cavity.

TEST FACILITY DESCRIPTION

J

The Boeing Transonic Wind Tunnel (BTWT) is a continuous flow, closed circuit, single return, atmospheric facility with the following characteristics:

Test Section F	low Parameters	Test Section Dimer	sions
Freestream Condition	Range	Description	Value
Mach number	0 thru 1.15	Cross-section (minus	
Dynamic pressure, psia	0 thru 6.3	corner fillets), ft.	8 x 10
Static pressure, psia	15 to 5.4	Length, ft.	14.5
Stagnation pressure	atmospheric	Area, ft. ²	88
Maximum unit Reynolds number, per foot	4 x 10 ⁶		
Maximum total temperature, °F	160		

The test section can be operated with either solid or slotted walls. The slotted wall configuration consists of 16 slots which can vary wall porosity from 3.5% to 11%.

Test data acquistion, recording, computations, and display are done by an XDS-9300 computer and Astro data sub-system.

DATA REDUCTION

Force and moment data were reduced in both body and stability axes using standard Boeing data reduction procedures. The following data reduction constants were used:

		Carri	er	<u>Orbiter</u>	<u> </u>
Symbol	Description	Model Scale	Full Scale	Model <u>Scale</u>	Full Scale
S	reference area, ft. ²	4.950	5500	2.421	2690.0
b	reference span, in	70.441	2348.04	28.100	936.68
Ē,	reference mac, in	9.833	327.78	14.244	474.81
MRC	moment reference center, in				•
	XC or XO	40.197	1339.90	33.270	1109.0
	YC or YO	0.0	0.0	0.0	0.0
	ZC or ZO	5.723	190.80	11.250	375.0

No base or cavity corrections were applied to the data.

Wind tunnel data were interpolated versus the applicable separation parameters (α_0 , ΔZ , ΔX , α_W , ΔY , β_0 , β_C , and ϕ) as summarized by Table VII. These interpolated data were used to compute interference increments by subtracting isolated data from interference data as summarized by Table VIII. A special interpolation routine was used for datasets with simulated attach hardware as summarized by Tables IX and X. Interpolated carrier data were corrected for support strut tare and interference using corrections obtained during test CA5 as summarized by Table XI. Basic data, interpolated data, incremental data, and carrier data with tare and interference corrections, are presented in this report. Tables IV through VI describe data presentation formats.

REFERENCES

Reports and Internal Letters

- Speed Letter, SAS/WTO/74-365, "Fabrication of a new 0.03-scale Orbiter Model," dated July 3, 1974
- IL, SAS/WTO/74-173, Addendum #1, "Updated Model Design Requirements for Model 45-0", dated July 24, 1974
- IL, SAS/WTO/74-173, Addendum #2, "Additional Requirements for Model 45-0," dated July 24, 1974
- IL, SAS/AERO/74-493, "Piggyback Separation Tests Orbiter Support Configurations and Corrections," dated August 9, 1974
- IL, SAS/AERO/74-552, "Orbiter Model Support and Instrumentation Requirements"
- IL, SAS/AERO/74-617, "Test Requirements for Separation Test CA20," dated August 20, 1974
- NA-74-541, "Structural Analysis of the 0.03-scale SSV model 45-0", dated July 23, 1974
- DMS-DR-2211, "Results of a 0.03-scale Aerodynamic Characteristics Investigation of a Boeing 747 Carrier (Model AX 1319 I-D) Mated with a Space Shuttle Orbiter (Model 45-0) conducted in the Boeing Transonic Wind Tunnel (CA5)", by 747 Aerodynamics, 747 Flight Controls, and Wind Tunnel Test Group, Boeing Aerospace Company

Drawings

Rockwell International - SSV Orbiter

- SS-A01360 Model Assy., 45-0, 0.03 Sc. SSV Orbiter (140A/B) Revision B, dated August 1, 1974
- SS-A01361- Model Instl. 45-0, 0.03 Sc. SSV Ferry Separation, Release 1, dated August 12, 1974
- SS-A01362 Blade Strut Assy., 0.03 Sc. 45-0 SSV Model, dated July 29, 1974

The Boeing Company - 747 Carrier

- 65-69716 Model Assy., TE 1007 I-1, dated August 23, 1973
- 65-89585 Wing W44.1 AX 1319 I-1, dated August 1, 1974

REFERENCES (Continued)

- 747-MD-572 Structural Arrangement Forward "A" Frame Support Orbiter 747 MOD, dated June 25, 1974
- 747-MD-461 General Arrangement 747 Space Shuttle Orbiter Carrier Aircraft (Piggyback Configuration), dated July 15, 1974
- 747-MD-576 Structural Arrangement Orbiter Aft Support, 747 MOD, dated August 1, 1974
- 1319-6, "Inbd Main Flap," dated 7-26-74
- 1319-15, "Wing Coves," dated 7-29-74
- 1319-24, "Outbd Fore-Flap," dated 8-5-74
- 1319-25, "Outbd Fore-Flap," dated 8-5-74
- 1319-33, "Inner Body Orbiter (Bal #660)" dated 8-13-74
- 1319-34, "Spoiler, dated 8-14-74
- 1319-35, "Balance Holder Orbiter (Bal #660)" dated 8-14-74
- 1319-36, "Rear Mtg. Parts Orbiter," dated 8-28-74
- 1319-37, "Aft Support and Balance Adapter Assy. Orbiter," dated 8-28-74
- 1319-38, "Inbd Flap Assy 20° F8.7," dated 8-17-74
- 1319-39, "Inbd Flap Brkts 20° F8.1," dated 8-19-74
- 1319-40, "Setting Temp L.E. Flaps," dated 8-17-74
- 1319-41, "Outbd Flap Brkts 20° F8.2," dated 8-20-74
- 1319-42, "Outbd Flap Assy 20° F_{8.2}," dated 8-20-74
- 1319-43, "Fwd Orbiter Support Parts & Assy," dated 8-21-74
- 1319-44, "L. E. Kruger & Flap Instl.," dated 8-21-74
- 1319-45, "BTWT Orbiter Alone Mtg Parts & Assy," dated 8-22-74
- 1319-47, "Template-Stabilizer Tip Fin," dated 8-22-74

REFERENCES (Concluded)

- 1319-55, "Stabilizer Fins," dated 8-23-74
- 1319-57, "Stabilizer Fin Brkts," dated 8-24-74
- 1319-60, "Stabilizer Fin Instl," dated 8-26-74
- 1319-63, "Orbiter Modif, & Inner Body Instl," dated 8-29-74
- 1319-64, "Model Support Mat'l," dated 9-3-74
- D6-25552, "Model Geometry Estimated Loads and Stress Analysis, Model AX13181-1," dated 9-11-74

TEST : CA20			DATE: 11-20-74
	TEST CON	DITIONS	
MACH NUMBER	REYNOLDS NUMBER	DYNAMIC PRESSURE	STAGNATION TEMPERATURE
MACH HUMBER	(per unit length)	(pounds/sq.ft.)	(degrees Rankine)
0.3	1.93 x 10 ⁶ /FT	126	548
0.48	2.81 x 106/FT	293	559
0.50	2.94 x 106/FT	315	555
0.60	3.30 x 10 ⁶ /FT	422	563
	The state of the s		,
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		1,000,000	1.2
	Orbiter: BTW Carrier: BTW	Γ #660F 2.074 inch Γ External Balance	ala.
BALANCE UTILIZED:		ACITY	
	Orbiter	Carrier	COEFFICIENT TOLERANCE:
ALE	1780 lb.	10,000 lb.	
NF SF	1335 lb.	5,000 lb.	
SF AF	301.5 lb.	1,000 lb.	
PM	4266 in1b.	100,000 in1b.	
RM	2014.5 in1b.	25,000 in1b.	
YM	2014.5 inlb.	25,000 in1b.	-
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TABLE II. (Continued)

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37	747	1/10,8	, 4	0	9/3	Ö	5	0		A	0	0	0	0	7,5				851
38			4	A	9/3	0	5	0		10	0	0	0	0	7.5				850
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TABLE II. (Continued)

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\top			S,AT38 AT39		0	0/3	0	5	0	.6		0	0	0	0		627			
	46			4	0	0/3	0	5	0	1.6		0	0	0	0			625		
\top	47			8	0	0/3	0	5	0	.6		0	0	0	0				62.6	
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55			8	0	9/3	0	5	0	0.6	0	0	0	0		633	645	644	
56			8	0	9/3	0	5	0	5.6	0	0	10	0		634	691	692	
57			8	0	9/3	0	5	0	0.6	0	0	20	0		643	674	675	
58			4	0	9/3	0	5	0	0.6	0	0	0	10			775	781	
59			4	0	9/3	0	5	0	0.6	0	0	10	10			735	738	
60			8	0	9/3	0	5	0	0.6	0	0	0	10			780	787	
61			8	0	9/3	0	5	0	0.6	0	0	10	10			736	737	
62			4	-5	9/3	0	5	0	0.6	0	0	0	0			649	648	
63			4	-5	%	0	5	0	0.6	0	0	10	0			687	688	
64-			4	-5	0/3	0	5	0	0.6	0	0	20	0			670	671	
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TABLE II. (Continued)

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88			14	0	0/3	0	5	0	0.6	0	7.5	0	10			790	793		1
89			4	0	0/3	0	5	0	0.6	0	7.5	10	10			748	751		1
90			18	0	9/3	0	5	0	0.6	0	7.5	0	10			799	† <u>†</u>		Second pressure
91			8	0	0/3	0	5	0	0.6	0	7.5	10	10			749	d-constant on the	te de s	
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94			8	5	0/3	0	5	0	0.6	0	7,5	0	10			800	795		-
95			8	5	0/3	0	5	0	0.6	0	7.5	10	10			757	758		1
96			4	-5	0/3	0	5	0	0.6	-5	7.5	0	10	\Box			CONTRACTOR STORES		1
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98			4	0	0/3	0	5	0	0.6	-5	7.5	0	10			903	806		
99			8	0	9/3	0	5	0	0.6	-5	7.5	0	10			812	809		1
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TABLE II. (Continued)

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	16		T		4	-5	0/3	0	5	0	0.6	-5	0	0	10			844	•		1
	77		T		4	-5	9/3	0	5	0	0.6	-5	0	10	10				820		1
	80				8	-5	0/3	0	5	0	0.6	-5	0	10	10	\Box			823		1
-	9		T		4	0	9/3	0	5	0	0.6	-5	0	10	0				834		۱
	0		T		8	0	9/3	0	5	0	0.6	-5	0	10	0			terminate control	837		١
	1				4	0	0/3	properties of particular deposits of	5	0	0.6	-5	O	0	10	\Box		843	principal executive and princi		١
1	2				4	0	9/3	0	5	0	0.6	-5	0	10	10				158		١
	3		T		8	0	0/3	0	5	0	0.6	-5	0	0	10				846		۱
	4		T		8	0	9/3	0	5	0	0.6	_5	0	10	10			827	824		ı
	5		T		4	5	93	0	5	0	0.6	-5	0	10	0			832	833		ı
	16		T		8	5	9/3	0	5	0	0.6	-5	0	10	0	\Box		839	838		ı
	7		T		4	5	9/3	0	5	0	0.6	-5	0	0	10			845			ı
	8		T		4	5	0/3	0	5	0	0.6	-5	0	10	10	十十		Management of the Control	822		-
	9		T		8	5	9/3	0	5	0	0.6		0	10	10	$\forall \exists$					ı
	0		Y		4	-5	0	0	5	0	0.6	-5	0	0	10	V	THE REAL PROPERTY AND ADDRESS OF THE PERSON NAMED IN	ennement de	THE RESERVE OF THE PERSON NAMED IN COLUMN		l
	7		13	19		2	5		31		37	43	49		6.6		c+				les E
•	oi ,	8 .			4	-5 -3	0	0	5	0	0.6	43	O 49		10	(SAME SAME SAME SAME SAME SAME SAME SAME	THE RESIDENCE OF THE	61	61	61 67	61 67 7

DA	TASET			1 (AI	RRIE	eR.	Π		C	RBIT	ER					41-X O	60	
	TIFIER	CONFI	GURATION			Sev		Se	Sa	THE WATER WHEN PROPERTY OF THE PARTY	€0	do	ΔX	AY	42	6	10	14-	
G1	1121	747/1	0,51	8	-5	0	0	5	0	0.6	-5	0	0	10	4		-	767	
T	122			4	0	0	0	5	0	0.6	-5	0	0	10			761	764	
T	123			8	0	0	0	5	0	0.6	-5	0	0	10			762	763	
T	124			4	5	0	0	5	0	0,6	-5	0	0	10			769	772	
T	125	1	1	8	5	0	0	5	0	0.6	-5	0	0	10			770	771	
T	126	747/1	0251	14	-5	0/3	0	5	0	0.6	0	0	0	0			656		
T	127			4	-5	9/3	0	5	0	0.6	0	0	10	0			657		
T	128			4	-5	9/3	0	5	0	0.6	0	0	20	0			669		
T	129			4	0	9/3	0	5	0	0.6	0	0	0	0			652	653	
T	130			4	0	0/3	O	5	0	0.6	0	0	10	0			661	659	
T	131			4	0	0/3	0	5	0	0.6	0	0	20	0			665		
T	132			8	0	0/3	0	5	0	0.6	0	0	0	0			655	Account of the Park of the Par	
Ť	133			8	0	0/3	0	5	0	0.6	0	0	10	0			658	660	
\dagger	134			8	0	0/3	0	5	0	0.6	0	0	20	0			668	667	
\dagger	135			4	-5	9/3	0	5	0	0,6	0	0	0	10			728		
Ť	136			4	-5	9/3	0	5	0	0.6	0	0	10	10			732		
Ť	137			4	0	9/3	0	5	0	0.6	0	0	٥	10			727		
¥	138		1	4	0	9/3	0	5	0	0.6	0	0	10	10	V		731		
	1000	Salah Sa	13 19	erikan ma		25		31		37	43	49		55		61		67	
_	a or									ICENTS		<u></u>				100	AR (1)	IE: AR	(21

TABLE II. (Continued)

EST: C			_) T_{		RRIE		I/KU	טא א	WREK C	ORBIT		IAKT			,		15/75	
DATA SET.	CONFIG	URATION	a.	-	-	-	Se	Sa	MACH	80	90	DX	AY	12	6	10	14-	
CN139	747/1	0251	14	5	9/3	0	5	0	0.6	0	0	0	10	A		729		
T 140	1		4	5	9/3	0	5	0	c.6	0	0	10	10			733		
141	747/1	0,51	4	0	10/13	O	5	0	0.6	0	0	0	0			707	708	
142			4	0	-1%-7	6	5	0	0.6	0	0	0	0			709	710	
143	4		4	0	43	15	5	O	0.6	0	0	0	0	Π		711	712	
144	747//	025,	4	0	9/3	15	5	0	0.6	0	0	0	0			725		
145	747/1	0,51	4	0	0/3	0	0	0	0.6	0	0	0	0			719	720	
146			4	0	0/3	0	10	0	0.6	O	0	0	0			714	715	
147			4-	0	0/3	0	10	0	0.3	0	0	0	0			717		
148			4	0	0/3	0	10	0	0.7	0	0	0	0			716		
¥ 149	-4		4	0	0/3	0	5	-10	0.6	0	0	0	0	¥		722	723	
	, ,	3 19			25		31		37	43 **	49		55		61		67	

TABLE II. - DATA SET/RUN NUMBER COLLATION SUMMARY (Continued)

Symbol Definition

Orbiter

01	=	Vertical tail off Tail cone on	(V ₈) (TC _{5.1})	05	=	Vertical tail on (v_8) Tail cone off
02	1 2	Vertical tail off Tail cone off with strut S ₁	(V ₈)	⁰ 6	=	MPS base plate off Vertical tail on (V_8)
03	**	Vertical tail on Tail cone on	,	07	=	MPS base plate off, S ₂ cover plate #1 off, Vertical tail on
04	=	Tail cone on Vertical Tail simulating dummy strut	(TC _{5.1})			MPS base plate off, S ₂ cover plate #2 off, Vertical tail on
				0 ₉	=	MPS base plate on, Vertical tail off with strut S ₂

Orbiter Support Strut

 S_1 = Orbiter support blade strut, upper entry position

 S_2 = Orbiter support blade strut, lower entry position

 S_3 = Orbiter dummy support blade strut

Carrier

747/0 = Carrier with spoilers and flaps retracted

747/1 = Carrier with spoilers deflected 45° and flaps retracted

α , β , and ΔZ Schedules

$$\triangle$$
 $\alpha_c = 0^{\circ}, 2^{\circ}, 4^{\circ}, 6^{\circ}, 8^{\circ}, 10^{\circ}$

$$\triangle$$
 $\beta_c = -10^{\circ}, -7^{\circ}, -5^{\circ}, -3^{\circ}, -2^{\circ}, -1^{\circ}, 0^{\circ}, +1^{\circ}, +2^{\circ}, +3^{\circ}, +5^{\circ}, +10^{\circ}$

$$\triangle \alpha_0 = 6^\circ$$
, 8°, 10°, 12°, 14°, 16°, 18°

TABLE II. (Concluded)

$$\triangle$$
 $\Delta Z = 0^*, 3^*, 7.5^*, 15^*, 30^*, 45^*, 60^*$

$$\triangle$$
 $\beta_0 = 2.5^{\circ}, 0^{\circ}, -2.5^{\circ}, -5^{\circ}, -7.5^{\circ}, -10^{\circ}, -15^{\circ}$

* minimum attainable

Table III. - MODEL DIMENSIONAL DATA A. Carrier Model

MODEL COMPONENT: BODY - B27.8		
GENERAL DESCRIPTION: Body 74-7 Project	with A.P.V.	()
Model Scale: 0.03		
Drawing Number: 65-69716		
Dimensions:	Full Scale	Model Scale
Length, in	2702	81.06
Max. Width, in		7.66
Area	•	
Wetted, ft ²	·	12.71

Table IIIA - Continued.

MODEL COMPONENT: F	0	·	
GENERAL DESCRIPTION	Clean Wing		
Flaps	Up		

Table IIIA - Continued

MODEL COMPONENT: Horizontal Tail H _l	5.6	
GENERAL DESCRIPTION: Horizontal Tail	with Vertical Fins on	each
Tip at Body B. L. 12.82	Access to the second se	
Model Scale 0.03		
Drawing Number 1319-55 1/2 - 60		
Dimension:	Full Scale	Model Scale
EXPOSED DATA (one side)		
Area-ft ²	200	

Table IIIA - Continued.

MODEL COMPONENT: M25		·
GENERAL DESCRIPTION: Inboard 747, JT9D	nacelle strut	
Model Scale: 0.03		
Dimensions	Full Scale	Model Scale
Wing B.L. of nacelle C_{l} , in.	470.0	14.100
Cont angle deg. inboard	2	2

MODEL COMPONENT: M26			
GENERAL DESCRIPTION: Outboard	747,	JT9D	
Strut			
Model Scale: 0.03			
Drawing Number: 937-590			
<u>Dimensions</u>		Full Scale	Model Scale
W L of C _L , in Cant angle, deg inboard		2	25.020

Table IIIA - Continued.

MODEL COMPONENT	•	N ₅₇								
GENERAL DESCRIF	TION_	Inboard	Fan	Cowl	and	Primary	747	Nacelle,	· · · · · · · · · · · · · · · · · · ·	
Flow Throu	ıgh Tyj	ре					· · · · · · · · · · · · · · · · · · ·	. · ·		_
Model Scale:	0.03	·								
Drawing Number:	s.0	. 1007-96	-97							

Table IIIA - Continued

MODEL COMPONENT:	N ₅₈				
GENERAL DESCRIPTION	: Outboard Fan	Cowl and	Primary	747 Nacelle,	
Flow Through Typ	e				
Model Scale: 0.	03	*			
Drawing Number S.O.	1.007-96,-97				

Table IIIA - Continued.

MODEL COMPONENT: Spoilers S1-12		
GENERAL DESCRIPTION: Multi-panel flig	ght spoilers. Four outboa	rd and
two inboard spoiler per side. Subscr	ript denotes spøiler pane	1 S _{1 is}
the most outboard L.H. panel and \$12	is most outboard R.H. pa	nel.
747 Model Scale: 0.03	Model: 1065	
Drawing No.: 65-71450, S.O. 1065-5	1, -59, -81, -173	
Dimensions: (One panel)	Full Scale Ft.	Model Scale IN.
Outboard S_{1-4} and S_{9-12} (Ft ²)	_21.48	0.019 ft ²
Span (equivalent)	6.25	2.25
Chord	3.44	1.238
Inboard S_{5-6} and S_{7-8} (Ft ²)	35.31	
5-0 /-8 Span (equivalent)	7.50	2.70
Chord	4.71	1.696
VII.01 56		

Table III A - Continued

MODEL COMPONENT: '19		
GENERAL DESCRIPTION: Flap Track Fairin	gs,	· .
4 on each side		
Model Scale: 0.03		
Drawing Number: S.O. 1007-403		
DIMENSIONS	Full Scale	Model Scale
WBL of Track no. 1, in.	235.3	7.06
2, in.	353.0	10.59
3, in.	652.0	19:56
4, in	743.6	22.31
Distance from wing	50.0	1.5

Table IIIA - Continued.

MODEL COMPONENT: Vertical V9.1		
GENERAL DESCRIPTION: Swept Vertical Tail		
Model Scale: 0.03		
Drawing Number: 65-6.9716; 1007-26,-610; 937-3	319	
Dimensions:	Full Scale	Model Scale
TOTAL DATA		
Area (Theo) Ft ² Span (theo) - In.	630.0 386.5	.567 11.595
Sweep-Back Angles, Degrees Leading Edge	50.12	50.12
Chords:		
Root (Theo) WP-in. Tip (Theo) WP-in. Cus. Sta. of .25 MAC	461.67 157.0 2529.6	13.85 4.71 75.888

Table IIIA - Continued.

MODEL COMPONENT: WING-W44.1		· · · · · · · · · · · · · · · · · · ·
GENERAL DESCRIPTION: Swept 747 Wing	· · · · · · · · · · · · · · · · · · ·	
Model Scale: 0.03	• · · · · · · · · · · · · · · · · · · ·	
Test No.	DWG. No. 65-89585	
Dimensions:	Full Scale	Model Scale
<u>Total Data</u>		
Area (Theo.) Ft ² Planform	5500	4.95
Span (Theo In.	2348.04	70.441
Aspect Ratio	6.96	6.96
Incidence Angle, degrees	7	7
Chords:		
MAC	327.78	9.83,3
Fus. Sta. of .25 MAC	1339.90	40.197
W.P. of .25 MAC	190.80	5.723

MODEL COMPONENT: ATTACH STRUCTURE - AT38

GENERAL DESCRIPTION: Orbiter to carrier forward attach

struts.

MODEL SCALE: 0.030

DRAWING NO.: BOEING 1319-43

	SC	ALE
DIMENSIONS:	FULL	MODEL
AT ₃₈	15.5	0.465
AT38.1	91.67	2.75
AT38.2	75.00	2.25
AT38.2A	75.0	2.25
AT38.3	ROD REMOVED	ROD REMOVED

TABLE IIIA - Concluded.

MODEL COMPONENT: ATTACH STRUCTURE - AT39

GENERAL DESCRIPTION: Orbiter to carrier aft attachment, pitch

adjustable from 0 to 10 deg.

MODEL SCALE: 0.030

DRAWING N(.: Boeing 50 1319-37.

DIMENSIONS:	FULL SCALE	MODEL SCALE
Pivot location:		
In., X _C	400.0	12.0
In., Z _C	160.7	4.821
Equivalent Span (At 0 deg iorb):		
Centerline orbiter	521.0	15.63

TABLE III MODEL DIMENSIONAL DATA B. Orbiter

MODEL COMPONENT : BODY - B26	
GENERAL DESCRIPTION : Configuration 140A/B	orbiter fuselage.
NOTE: B26 is identical to B24 except underside	of fuselage has been
refaired to accept W ₁₁₆ .	
MODEL SCALE: 0.030 MODEL DWG:	SS-A00147, Release 12
DRAWING NUMBER: VL70-000143B, 000200, - VL70-000140A, -000140B	
DIVENSIONS	
DIMENSIONS: FULL S Length (OML: Fwd Sta X ₀ =235), In. 129	CALE MODEL SCALE 3.3 38.799
· · · · · · · · · · · · · · · · · · ·	0.3 38.709
Max Width (At $X_0 = 1528.3$), In. 26	4.0 7.920
Max Depth (At $X_0 = 1464$), In. 25	7.500
Fineness Ratio	0.264 0.264
Area - Ft ²	
Max. Cross-Sectional 34	0.307
Planform	
Wetted	
Base	

MODEL COMPONENT : CANOPY - C9			
GENERAL DESCRIPTION :Configuration	3A. Canopy us	ed with fuselage B	2 (
		·	
MODEL SCALE: 0.030 MOI	DEL DWG: SS-A	A00147, Release 12	
DRAWING NUMBER: VL70-000143A			
DIMENSIONS:	FULL SCALE	MODEL SCALE	
Length ($X_0 = 434.643$ to 578), In.	143.357	4.301	
Max Width (At $X_0 = 513.127$), In.	152.412	4.572	
Max Depth (At $X_0 = 485.0$), In.	25.00	0.750	
Fineness Ratio			
Area			
Max. Cross—Sectional			
Planform			
Wetted	•	 	
Base		-	

MODEL COMPONENT: SLOTTE	D ELEVON (6	INCH GAP) - E ₄₃	
GENERAL DESCRIPTION: Config	guration 140A/	B orbiter elevon.	
NOTE: E ₄₃ is a slotted v	ersion of E ₂₆ .	Data are for one	side.
MODEL SCALE: 0.030			
DRAWING NUMBER:	VL70-000200,	-006089, -006092	
DIMENSIONS:		FULL-SCALE	MODEL SCALE
Area - Ft ²		_210,00	0, 189
Span (equivalent) , In.		349.2	10.476
Inb'd equivalent chord	, In.	118.004	3.540
Outb'd equivalent chor	d, In.	55. 192	1. 656
Ratio movable surface of total surface chord	chord/		
At Inb'd equiv. c	nord	0.2096	0. 2096
At Outb'd equiv.	chord	0.4004	0.4004
Sweep Back Angles, degr	rees		
Leading Edge		0,00	0.00
Tailing Edge		-10.056	-10.056
Hingeline (Product o	f Area & c)	0.00	0.00
Area Moment (अञ्चलक्षे ४३४)	*hingextined, Ft	³ 1587. 25	0,043
Mean Aerodynamic Ch	ord, In.	90.7	2, 721

MODEL COMPONENT : BODY FLAP - F	8			
GENERAL DESCRIPTION: Configuration 140A/B orbiter body flap				
NOTE: Hingeline located at X = 1528.3, Z = 284.3				
MODEL SCALE: 0.030 MO	DEL DWG: SS-	A00147, Release 12		
DRAWING NUMBER: VL70-000140A, -	000145			
DIMENSIONS :	FULL SCALE	MODEL SCALE		
Length ($X_0 = 1520 - 1613$) In.	93.00	2.79		
Max Width , IN.	262.00	7.86		
Max Depth ($X_0 = 1520$), In.	23.00	0.69		
Fineness Ratio Area - Ft ²				

Max. Cross-Sectional

Planform

Wetted

Base

o.13**5**5

0.0377

150.525

41.847

MODEL COMPONENT : OMS POD - M ₁₆			
GENERAL DESCRIPTION: Configuration 140C			
Orbiter OMS pod - Short pod			
MODEL SCALE: 0.030			
DRAWING NUMBER:	8410		
DIMENSIONS:	FULL SCALE	MODEL SCALE	
Length (OMS Fwd Sta. X _o =1310.	5) 258.50	7.755	
Max Width (At $X_0 = 1511$), In.	136.8	4.104	
Max Depth (At $X_0 = 1511$), In.	74. 70	2.241	
Fineness Ratio	2.484	2.484	
Area - Ft ²			
Max. Cross-Sectional	58.864	0.053	
Planform		<u> </u>	
Wetted			
Base		<u> </u>	

MODEL COMPONENT: MPS NOZZLES - N24	
GENERAL DESCRIPTION: Configuration 140A/B	orbiter MPS nozzles
MODEL SCALE: 0.030 MODE	L DWG: SS-A00147, Release 12
DRAWING NUMBER: VL70-005030A, -000140A	
DIMENSIONS:	FULL SCALE MODEL SCALE
MACH NO.	
Length - In. Gimbal Point to Exit Plane Throat to Exit Plane	$\begin{array}{c cc} 157.00 & 4.71 \\ \hline 99.2 & 2.976 \end{array}$
Diameter - In. Exit Throat Inlet	91.00 2.73
Area - ft ² Exit Throat	45.166 0.0407
Gimbal Point (Station) - In. Upper Nozzle X Y Z	$ \begin{array}{c cccc} 1445.00 & 43.35 \\ \hline 0.0 & 0.0 \\ \hline 443.00 & 13.29 \end{array} $
Lower Nozzles X Y Z	$ \begin{array}{c cccc} 1468.170 & 44.045 \\ $
Null Position - Deg. Upper Nozzlę Pitch Yaw	16 16 0
Lower Nozzle Pitch Yaw	10 10 3.5 3.5
56	

MODEL COMPONENT:	OMS NOZZLES - N	V ₂₈		
general descripti	ON: Configuration	140A/B orb	oiter OMS No	zzles
MODEL SCALE: 0.	030			
DRAWING NUMBER:	VL70-000140A (Lo	cation), SS	5-A00106, Re	lease 5 (Contou
dimensions:			FULL SCALE	MODEL SCALE
MACH NO.				
	oint to Exit Plane Exit Plane			
Diameter - Ir Exit Throat Inlet				
Area - ft ² Exit Throat				
Gimbal Point Left xpyer N or X Y Z	(Station) - In. zzle		1518.00 -88.0 492.0	45.54 -2.64 14.76
Right Example No. X o Yo Zo	zzles		1518.0 88.0 492.0	45.54 2.64 14.76
Null Positio Left XYSES No Pitc Y aw	zzle	13 ⁰ 17 Ou	±8 tb <u>'d, 2 ⁰30' In</u>	±8 b' <u>d Same</u>
Right Kacwer No Pitc Yaw		13 ⁰ 17' O	_±8 utb'd, 2°17'	<u>±8</u> Inb <u>'d</u>

MODEL COMPONENT: RUDDER - R5		
GENERAL DESCRIPTION: Configuration 140C of	orbiter rudder (ide	entical to
configuration 140A/B rudder)		
MODEL SCALE: 0.030		y
DRAWING NUMBER: VL70-000146B,	-000095	
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Area - Ft ²	100. 15	0. 090
Span (equivalent) , In.	201.00	6. 03
Inb'd equivalent chord, In.	91.585	2. 748
Outb'd equivalent chord, In.	50.833	1.525
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	0.400	0.400
At Outb'd equiv. chord	0.400	0.400
Sweep Back Angles, degrees		
Leading Edge		
Tailing Edge	26, 25	26. 25
Hingeline (Product of area and c)	34.83	34. 83
Area Moment (Movement x to Mindre), Ft3	610.92	0.0165
Mean Aerodynamic Chord, Inches	73.2	2. 196

MODEL COMPO	ONENT : ORBITER TA	AILCONE - TC5	.1
GENERAL DES	SCRIPTION: Fairing moun	ted on orbiter f	uselage base for
ferry mission	ons.		
MODEL SCA	ALE: 0.030		
DRAWING NUM	BER: Boeing Dwg No.: 1	319-71	
DIMENSIONS :		FULL SCALE	MODEL SCALE
Lengt	ih	445.83	13.375
Max V	Yidth	303.33	9.10
Max C	heptox Height	265.00	7.95
Finen	ess Ratio	· · · · · · · · · · · · · · · · · · ·	
Area	- Ft ²		
	Projected frontal area Max. Cross-Sectional	324. 105	0.2917
	Planform		
	Wetted		
	Base		

MODEL COMPONENT: VERTICAL - V8			
GENERAL DESCRIPTION: Configuration 140A/B orbiter vertical tail.			
MODEL SCALE: 0.030 MODEL DV	VG: SS-A001	48, Release	
DRAWING NUMBER: VL70-000146A			
dimensions:	FULL SCALE	MODEL SCALE	
TOTAL DATA			
Area (Theo) - Ft ² Planform Span (Theo) - In. Aspect Ratio Rate of Taper Taper Ratio	413, 253 315, 720 1, 675 0, 507 0, 404	0.372 9.472 1.675 0.507 0.404	
Sweep-Back Angles, Degrees. Leading Edge Trailing Edge 0.25 Element Line	45.000 26,25 41.13	45.000 26.25 41.13	
Chords: Root (Theo) WP Tip (Theo) WP MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC	268, 50 108, 47 199, 81 1463, 35 635, 52 0, 00	8.055 3.254 5.994 43.901 19.066 0,00	
Airfoil Section Leading Wedge Angle - Deg. Trailing Wedge Angle - Deg. Leading Edge Radius Void Area	10.00 14.92 2.00	10.00 14.92 0.060	
Blanketed Area	0.00	0,00	

trailing edge of wing. Geometric twist = MODEL SCALE: 0.030 EST NO. IMENSIONS: TOTAL DATA Area (.neo.) Ft2 Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees Leading Edge		MODEL SCALE 2.421 28.10 2.265 1.177 0.200 3.500 0.500 45.00 -10.056 35.209
TOTAL DATA Area (.neo.) Ft ² Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	2690.00 936.68 2.265 1.177 0.200 3.500 0.500	2. 421 28. 10 2. 265 1.177 0. 200 3.500 0.500
TOTAL DATA Area (Theo.) Ft2 Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	2690.00 936.68 2.265 1.177 0.200 3.500 0.500	2. 421 28. 10 2. 265 1.177 0. 200 3.500 0.500
TOTAL DATA Area (Theo.) Ft2 Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	2690.00 936.68 2.265 1.177 0.200 3.500 0.500 45.00 - 10.056	2. 421 28. 10 2. 265 1.177 0. 200 3.500 0.500 45. 00 = 10. 056
Area (Theo.) Ft ² Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	936.68 2.265 1.177 0.200 3.500 0.500 45.00 - 10.056	28, 10 2, 265 1,177 0, 200 3, 500 0, 500 45, 00 = 10, 056
Planform Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	936.68 2.265 1.177 0.200 3.500 0.500 45.00 - 10.056	28, 10 2, 265 1,177 0, 200 3, 500 0, 500 45, 00 = 10, 056
Span (Theo In. Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	936.68 2.265 1.177 0.200 3.500 0.500 45.00 - 10.056	28, 10 2, 265 1,177 0, 200 3, 500 0, 500 45, 00 = 10, 056
Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	2.265 1.177 0.200 3.500 0.500 45.00 - 10.056	2.265 1.177 0.200 3.500 0.500 45.00 - 10.056
Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	0,200 3,500 0,500 45,00 - 10,056	0.200 3.500 0.500 45.00 = 10.056
Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	3.500 0.500 45.00 - 10.056	0.200 3.500 0.500 45.00 = 10.056
Incidence Angle, degrees Aerodynamic Twist, degrees Sweep Back Angles, degrees	0.500 45.00 - 10.056	0.500 45.00 - 10.056
Aerodynamic Twist, degrees Sweep Back Angles, degrees	45,00 - 10,056	45.00 - 10.056
Sweep Back Angles, degrees	- 10.056	- 10.056
	- 10.056	- 10.056
	- 10.056	- 10.056
Trailing Edge		
0.25 Element Line		
Chords:		
Root (Theo) B.P.O.O.	689.24	20.677
Tip, (Theo) B.P.	137.85	4.136
MAC Fus. Sta. of .25 MAC	474.81	14.244
W.P. of .25 MAC	1136.83	34.105
B.L. of .25 MAC	290.58 182.13	8.717 5.464
EUDACED DATA		
Area (Theo) Ft ²	1751.50	1,576
Span, (Theo) In. BP108	720.68	21,620
Aspect Ratio	2.059	2.059
Taper Ratio	0.245	0.245
Chords		
Root BP108	562.09	16.863
Tip 1.00 <u>b</u>	137.85	4 136
MAC	392, 83	11.785
Fus. Sta. of .25 MAC	1185.98	35,579
W.P. of .25 MAC	294.30	8.829
B.L. of .25 MAC	251.77	7.553
Airfoil Section (Rockwell Mod NASA)		
XXXX-64 Root <u>b</u> =	0.113	0.113
7 Tip <u>b</u> =	0.120	0.120
2		
Data for (1) of (2) Sides		
Leading Edge Cuff 2 Planform Area Ft2	113.18	0. 102
Leading Edge Intersects Fue M ! @ Sta	500.00	15.00
Leading Edge Intersects Wing @ Sta	1024.00	30.72
Leading Edge Intersects Wing & Sta Op Poor Page 18		
OF DOWNA.		
100p P1		
Or ACE.		
· ALM B		

Table IV. CA20 DATASET DESCRIPTION (Raw Data)

DATASET TYPE	DESCRIPTION
RGNXXX	Longitudinal coefficient schedule for 747 carrier balance data which contain "standard" wind tunnel corrections.
AGNXXX	Lateral coefficient schedule for 747 carrier balance data which contain "standard" wind tunnel corrections.
BGNXXX	Longitudinal coefficient schedule for orbiter balance data which contain "standard" wind tunnel corrections.
CGNXXX	Lateral coefficient schedule for orbiter balance data which contain "standard" wind tunnel corrections.
DGNXXX	Pressure coefficient data as follows: Q(PSF) - dynamic pressure, psf PB1, PB2, PB3 - orbiter base pressure coefficients PCAV - orbiter cavity pressure coefficient PSC - carrier cavity pressure coefficient LHLS, RHLS - left and right hand pressure coefficients in proximity to orb. vert. tail for blade/sting support system. LHVERT, RHVERT - identical to LHLS and RHLS but for base sting support system.

Table V.
CA20 COEFFICIENT SCHEDULE
(Raw Data)

						<u></u>		Coeffic	cients				
Dataset Type	Dataset•Sequence	lst ID	2nd ID	1	2	3	4	5	6	7	8	9	10
RGNXXX	034-036	MACH	ALPHAW	ВЕТА	Q(PSF)	CL	CD	CLM , .	СҮ	CLN	CSL		
•	037	MACH	ALPHAO	ALPHAW	ВЕТА	DY	DZ	CL	CD	CLM	СҮ	CLN	CSL
	038 & 039	масн .	BETA	ALPHAW	ALPHAO	DY.	DZ	CL .	CD	CLM	СҮ	CLN	CSL
	040-149	ALPHA0	DZ	MACH	DX	DY	вета0	PHI	ALPHAW	ВЕТА	CL	CD	CLM
AGNXXX	040-149	ALPHAO	DZ	масн	ĎΧ	DY	ветао	PHI	ALPHAW	BETĀ	СУ	CLN	CSL
BGNXXX	001-011 & 015-033 & 037	MACH	ALPHAO	ветао	PHI	Q(PSF)	ĊL	CD	CLM	ĊY	CLN	CSL	
	012 & 013	MACH	DZ	ALPHA0	ветао	PHI	Q(PSF)	CL	CD	CLM	CY	CLN	CSL
	014	MACH	.BETAO	ALPHA0	PHI	Q(PSF)	CL	CD	CLM	CY	CLN	CSL	
	038 & 039	МАСН	вета	ALPHAW	ALPHA0	DΥ	DZ	CL	CD	CLM	СҮ	CLN	CSL
	040-149	ALPHAO	DZ	MACH	DX	DΥ	BETA0	PHI	ALPHAW	BETA	CL	CD	CLM
CGNXXX	040-149	ALPHA0	DZ	масн	DX	DY	BETA0	PHI	ALPHAW	BETA	СҮ	CLN	CSL
DGNXXX	001-011 & 015-019 & 037	MACH	ALPHAO	Q(PSF)	PB1	PB2	PB4	LHLS	RHLS	PCAV	1		
	012 & 013	MACH	DZ	Q(PSF)	PBI	PB2	PB4	LHLS	RHLS	PCAV			
	014	MACH	ветао	Q(PSF)	РВ1	PB2	PB4	LHLS	PHLS	PCAV			
	020-033	MACH	ALPHA0	Q(PSF)	PB1	FB2	PB4	LHVERT	RHVERT	PCAV			
:	034-036	МАСН	ALPHAW	PSC						·			
	038 & 039	MACH	вета	Q(PSF)	PB1	PB2	PB4	LHLS	RHLS	PCAV			
	040-149	ALPHAO	DZ	Q(PSF)	PB1	PB2	PB4	LHLS	PHLS	PCAV	<u></u>		<u> </u>

Note: ID--Independent variable

Table VI
CA20 DATASET DESCRIPTION
(INTERPOLATED/INCREMENTED DATASETS)

DATASET TYPE	DESCRIPTION
MGNXXX	Interpolated data for 747 carrier balance data in carrier reference dimensions.
NGNXXX	Interpolated data for orbiter balance data in orbiter reference dimensions.
UGNXXX	Incremental data - 747 carrier data in presence of orbiter (mated) minus 747 carrier alone data in 747 carrier reference dimensions.
VGNXXX	Incremental data - Orbiter data in presence of 747 carrier (mated) minus orbiter alone data in orbiter reference dimensions.

NOTE: Datasets M, N, U, and V contain the full ΔZ array of 0 3 7.5 15 30 40 and 60 ft. Therefore, the datasets reflect extrapolations for some individual test arrays of ΔZ . For subsequent data plotting, the full ΔZ arrays were truncated to the actual tested arrays.

Table VII. CA20 INTERFULATED DATASET SUMMARY

(M AND N DATASETS)

	(M AND N DATASETS)
DATASET(S)	INTERPOLATED VARIABLES (1) (2)
NGN001 → 011	MACH, ALPHAO
NGN012 → 013	MACH, DZ
NGNO14	MACH, BETAO
NGN015 + 033	MACH, ALPHÃO
MGN034 → 036	MACH, ALPHAW
MGN037 NGN037	MACH, ALPHAO
MGN038 NGN038 → 039	MACH, BETAC
MGN040 NGN040 → 048	ALPHAO, DZ (SEE NOTE 3)
MGN049 NGN049 → 119 MGN126 NGN126 → 140	ALPHAO, DZ, DX, ALPHAW, DY, BETAO, BETAC, PHI
MGN141 → 149 NGN141 → 149	ALPHAO, DZ
MGN120 NGN120 → 125	ALPHAO, DZ, BETAC, ALPHAW

NOTES:

(1) Interpolation procedure:

Number of Values Available for Interpolation	Interpolation Procedure
1	Substitute actual test value with a nominal test value (Note 3 below) Straight line
3	Parabolic spline fit Cubic spline fit

BETA = BETAC

(3)
Interpolation was versus DZ and ALPHAO; however, since each of these datasets (40 → 48) has only one ALPHAO there was therefore no ALPHAO interpolation. The recorded test ALPHAO was replaced with a nominal test ALPHAO (i.e., 8, 12, or 16) so that the only interpolation was versus DZ.

Table VII. Concluded.

(4) Interpolation on DX was not performed on all datasets due to large data fluxuations from the nominal condition.

Table VIII. CA20 INCREMENTAL DATASET SUMMARY

(INTERFERENCE) - (ISOLATED)

(U	AND	٧	DATASETS)

BASE DATASET	VEHICLE	BETAC, deg.	ALPHAW, deg.	BETAO, deg.	ELEVON, deg.	AILRON, deg.
MGN034	CARRIER (1)	- 5	0 4 8	NA 	NA 	NA
MGN035			0 4 8			
MGN036		5 ↓	0 4 8			\
NGN007 NGN010 NGN008 NGN011 NGN009 NGN018	ORBITER-0 ₁ S ₁ (2) ORBITER-0 ₂ S ₁ (2)	NA V2···	NA 	-5 0	.5 5 0 10 5	0 t -10 0

NOTES:

(1) ALPHAW Sweep (0, 4, 8°)

(2) ALPHAO Sweep (6, 8, 10, 12, 14, 16, 18)

(3) Procedure - (a) Interpolate base datasets to various nominal α and β combinations.

(b) Subtract appropriate interpolated base dataset from interpolated separation (mated) data, except for datasets 45 thru 48 which were utilized to provide the increment due to attach hardware as follows:

Resulting Dataset Number	First Dataset Number	Second Dataset Number
UGN045	MGNO45	MGNO49 @ ALPHAO = 8
VGN04 5	NGNO45	NGNO49 @ ALPHAO = 8
UGNO46	MGN046	MGN052 @ ALPHA0 = 12
VGNO46	NGN046	NGN052 @ ALPHA0 = 12
UGNO47	MGNO47	MGN055 @ ALPHA0 = 16
VGNO47	NGNO47	NGN055 @ ALPHA0 = 16
UGNO48	MGN048	MGN046
VGNO48	NGN048	NGN046

INCREMENT = (First Dataset) - (Second Dataset)

Datasets 45 thru 48 interpolated per note (3) on "Interpolated Dataset Summary". Datasets 49, 52, and 55 interpolated versus ALPHAO, DZ, DX, ALPHAW, DY, BETAO, BETAC, PHI.

Table IX. SPECIAL INTERPOLATION FOR CONFIGURATIONS WITH ATTACH HARDWARE

RESULTANT DATASET SGNO 1		CONFIGURATION	INPUT DATASETS βο = 0°, βc = 0°			
CARRIER	ORBITER		aw aw	8° 2°	12° 6°	16° 10°
А	В	747/0 0 ₁ S ₁ AT ₃₈ AT ₃₉		40	41	42
E	F	747/1 0 ₁ S ₁ AT ₃₈ AT ₃₉		45	46	47
I	J	747/1 0 ₁ S ₁		49		
К	L	747/1 0 ₁ S ₁			52	
M	N	747/1 0 ₁ S ₁				55
			a,	l o = 12°	, a _W =	5.83°
			β ₀ β _C	-5° -5°	0°	
С	D	747/0 0 ₁ S ₁ AT ₃₈ AT ₃₉		43	41	
G	H ,	747/1 0 ₁ S ₁ AT ₃₈ AT ₃₉		48	46	

NOTES:

- (1) Orbiter data were interpolated versus α_0 and ΔZ
- (2) Carrier data were interpolated versus α_W and ΔZ
- (3) The interpolation assumes a constant incidence angle between Orbiter and Carrier even though they were mounted on separate support systems (see Configuration A, in Figure 2F).
- (4) Resultant datasets SGNO 1 have both lateral and longitudinal coefficient data.

			INPUT_DATASETS				
RESULTANT DATASET		αο	$3\mathbf{o} = 0^{\circ}$	$\beta_{\rm C} = 0^{\circ}$	16 <i>°</i>		
יאכ	INOLI	a _W	2 °	6°	10°		
CARRIER	ORBITER						
WGNR45	XGNB45		45- 49				
WGNR46	XGNB46			46-52			
WGNR47	XGNB47				47-55		
			xo = 12°,	$\alpha_W = 5$.83°		
WGNR48	XGNB48		48-	46			
WGNR43	XGNB43		43-	41			
WGNRDB	XGNBDB	(4	18, 46) -	(43, 4	<u>)</u>		
			$\beta_0 = 0^\circ$,	$\beta_{\rm C} = 0^{\circ}$			
WGNRCA	XGNBCA	(45,	46, 47) -	(40, 4	1, 42)		

NOTE: Resultant datasets have incremental lateral and longitudinal coefficient data.

Table XI. CARRIER SUPPORT STRUT TARE AND INTERFERENCE CORRECTION PROCEDURE

ij

α, deg.	β, deg.	CA5 Run with Image Strut	CA5 Run without Image Strut
	0	15	99
2	2>	20	104
6	2>	23	108
8	2	27	112

$$\alpha_{W} = 1000 = 3^{\circ} \rightarrow 16^{\circ}, 1^{\circ} \text{ increments}$$

$$\beta = 1000 = -12^{\circ}, -10^{\circ}, -8^{\circ}, -6^{\circ}, -4^{\circ}, -3^{\circ}, -2^{\circ}, -1^{\circ}, 0^{\circ}, 1^{\circ}, 2^{\circ}, 3^{\circ}, 4^{\circ}, 6^{\circ}, 8^{\circ}, 10^{\circ}, 12^{\circ}$$

$$Correction = \left(\begin{array}{c} \text{Run with} \\ \text{Image Strut} \end{array}\right) - \left(\begin{array}{c} \text{Run without} \\ \text{Image Strut} \end{array}\right)$$

"Correction" datasets are 6GMDA4, 6GMDB4 and 6GMDC4, which were interpolated for α_W = 2° to 12° and β = -5°, 0°, +5°, respectively.

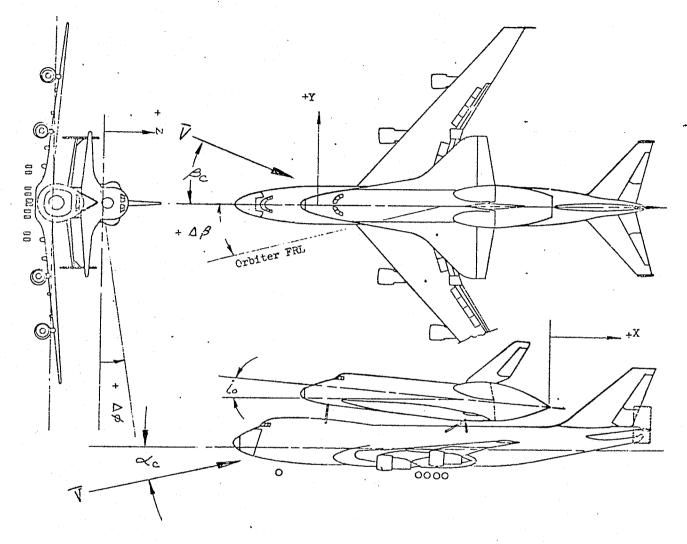
"Corrected" datasets are 5GN034-149. For the DZ and α_0 sweeps (2nd independent variable), the "correction" is a constant value for all coefficients. For the α_W and β sweeps (2nd independent variable), the "correction" is a function of α_W and β , respectively.

Note:

"Correction: and "corrected" data are shown in the Appendix.

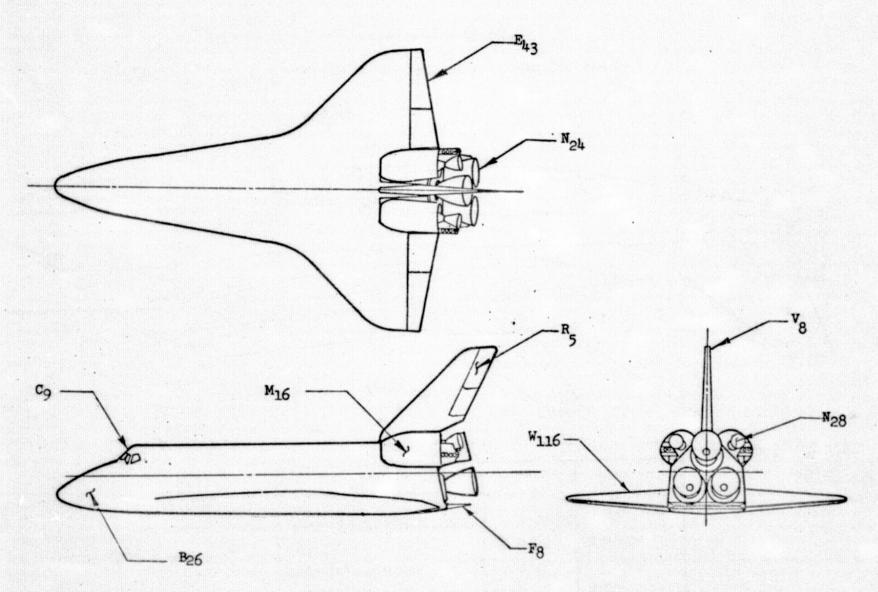
a. General

Figure 1. - Axis systems.



b. Orbiter/747 Axis System Definition

Figure 1. - Concluded.

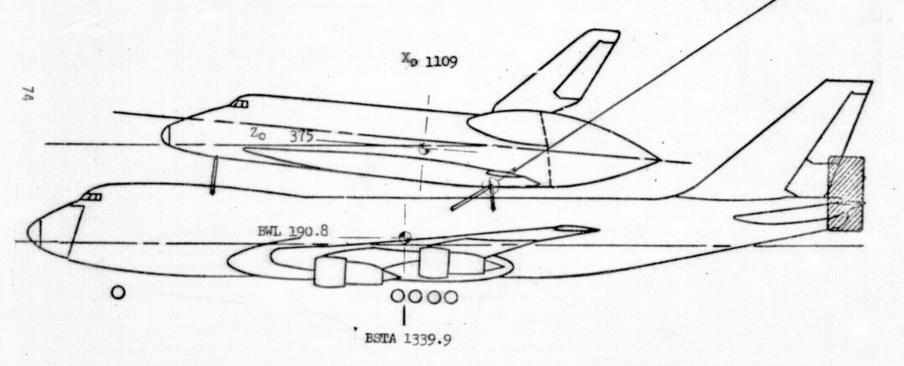


a. SSV Orbiter Configuration (VC70-000002)
 Figure 2. - Model sketches

REFERENCE DIMENSIONS (FS)

	ORBITER	747 CARRIER
WING AREA ∼ Ft ²	2690	5500
MAC (c) ~ INCHES	474.81	327.78
SPAN (b) ~ INCHES	936.68	2348.04
MOMENT REFERENCE CENTER	67.5% LB	25.0 % €
F.S. ~ INCHES	1109.0	1339.9
W.P. ~ INCHES	375.0	190.8

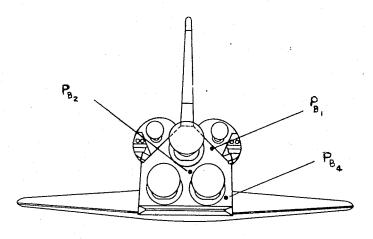
BWL 400 (\$\frac{2}{2}\circ 267.5\)
BSTA 1607 (\$\frac{2}{2}\circ 267.5\)



b. Orbiter/747 Flight Test Configurations

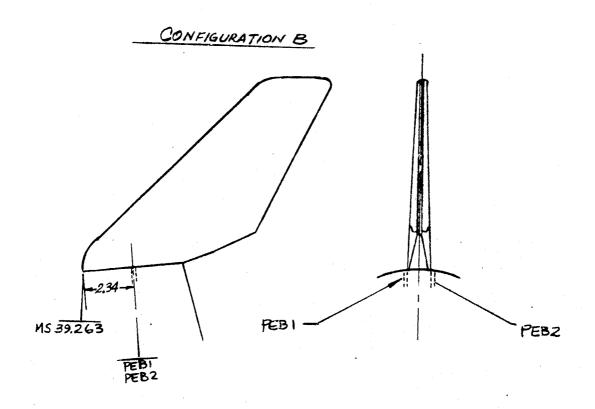
Figure 2. - Continued.

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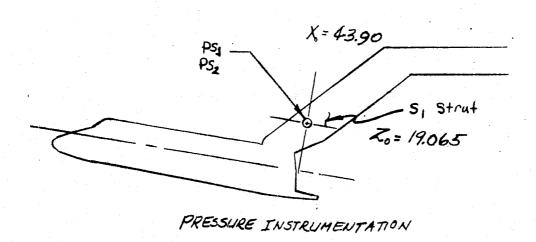


c. Base Pressure Locations

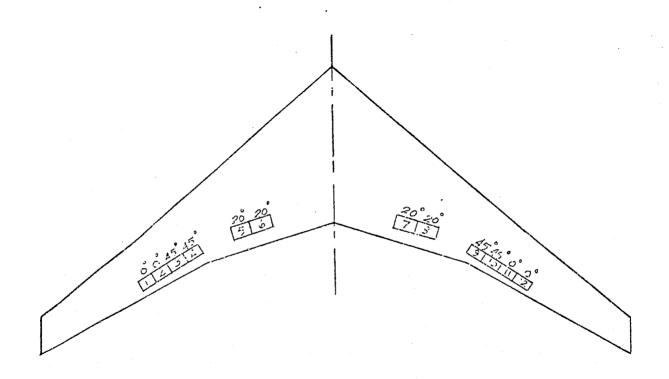
Figure 2. - Continued.



CONFIGURATION A

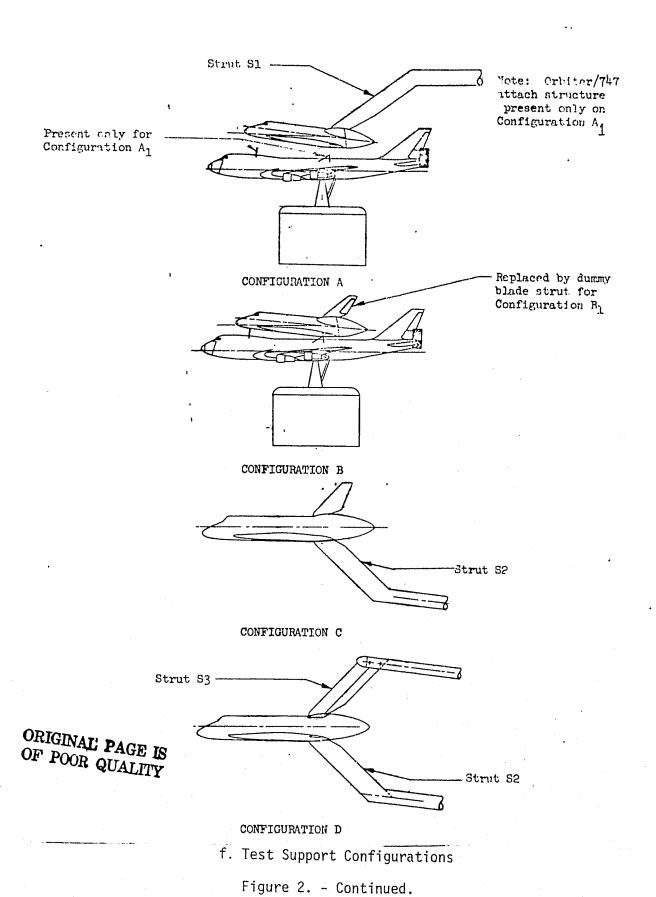


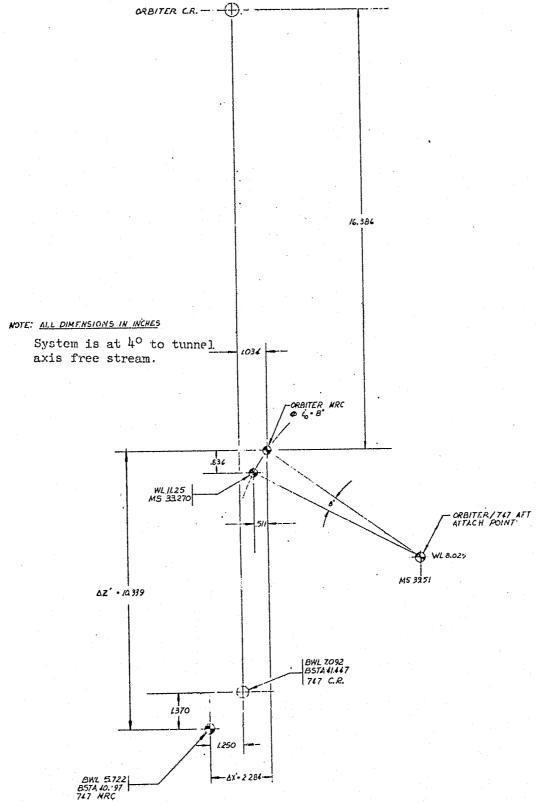
d. Blade Strut and Vertical Tail Pressure Locations
 Figure 2. - Continued.



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e. Standard In-Flight Speed-Brake
Figure 2. - Continued.

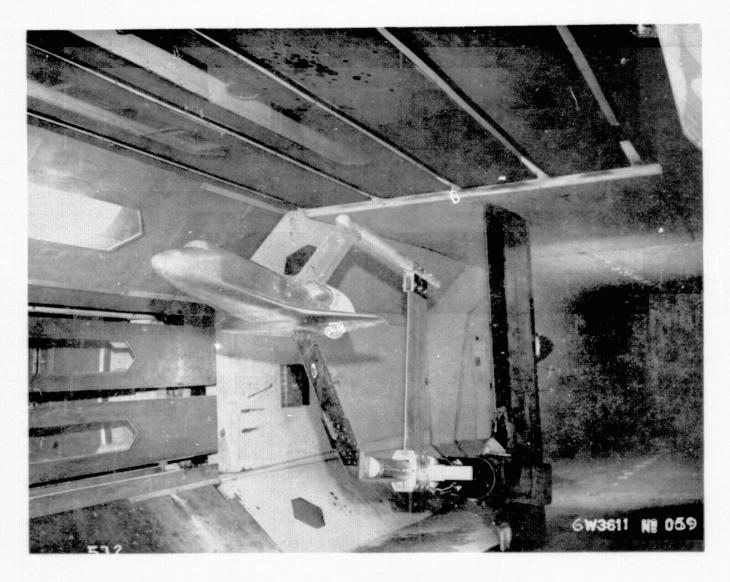




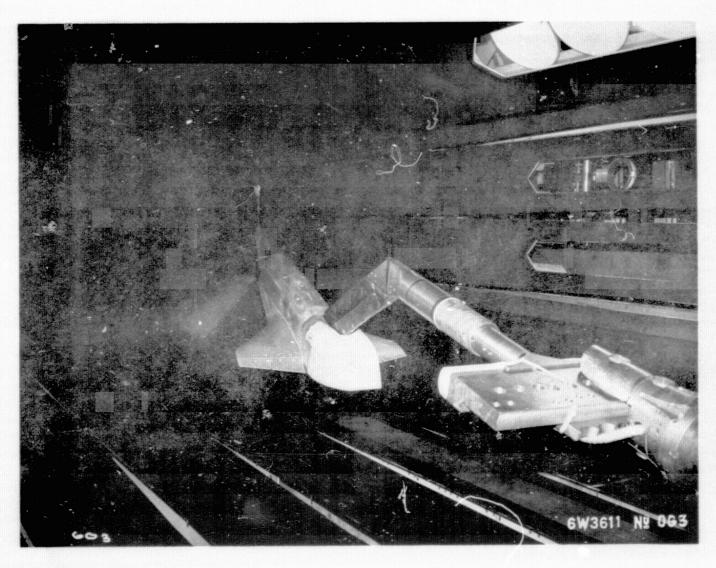
g. Orbiter/747 C.G. and C.R. Orientation

Figure 2. - Concluded.

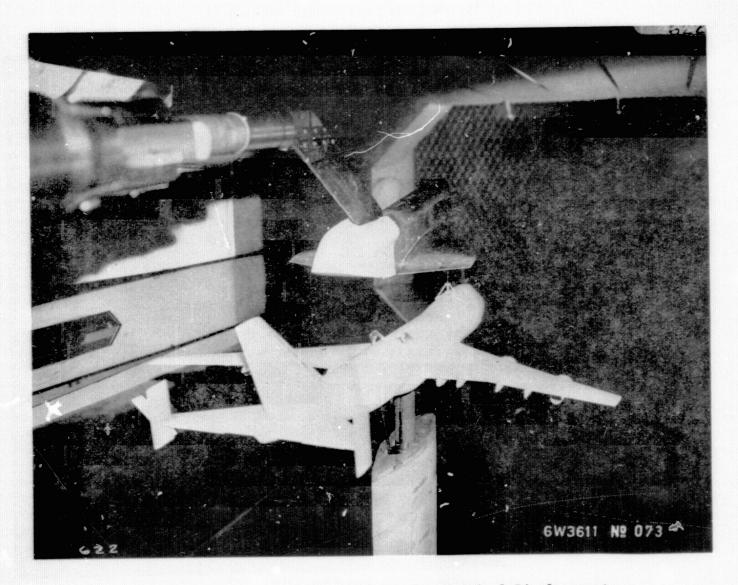
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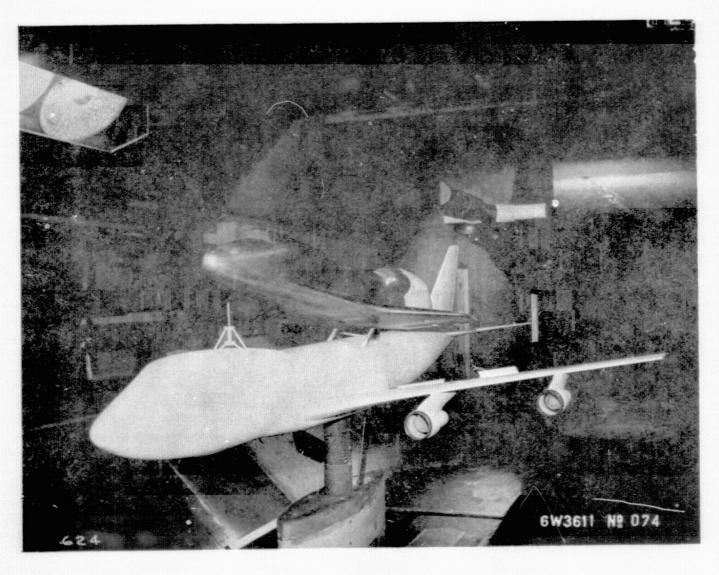
a. Orbiter Alone with Dummy Blade in Proximity for Sting Tare Effect Study Figure 3. - Model photographs.



b. Orbiter Alone with Tail Cone ${{\rm TC}_5.1}$ Figure 3. - Continued.



c. Aft View of the Orbiter/747 Showing Vertical Displacement Figure 3. - Continued.



d. Front View of the Orbiter at an Angle of Attack with Respect to the 747 Carrier Figure 3. - Concluded.

DATA FIGURES

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VOLUME 2 Figures 26-39 Pages 832-1863

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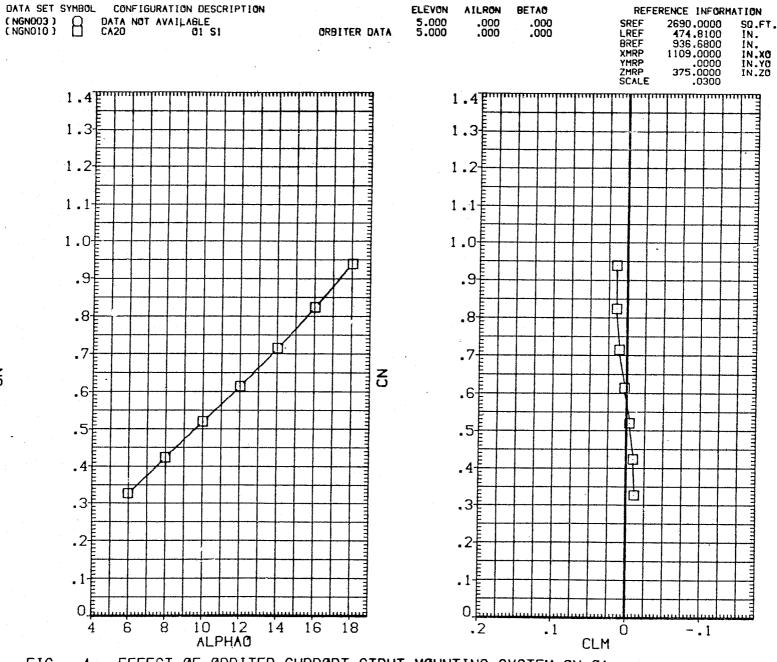
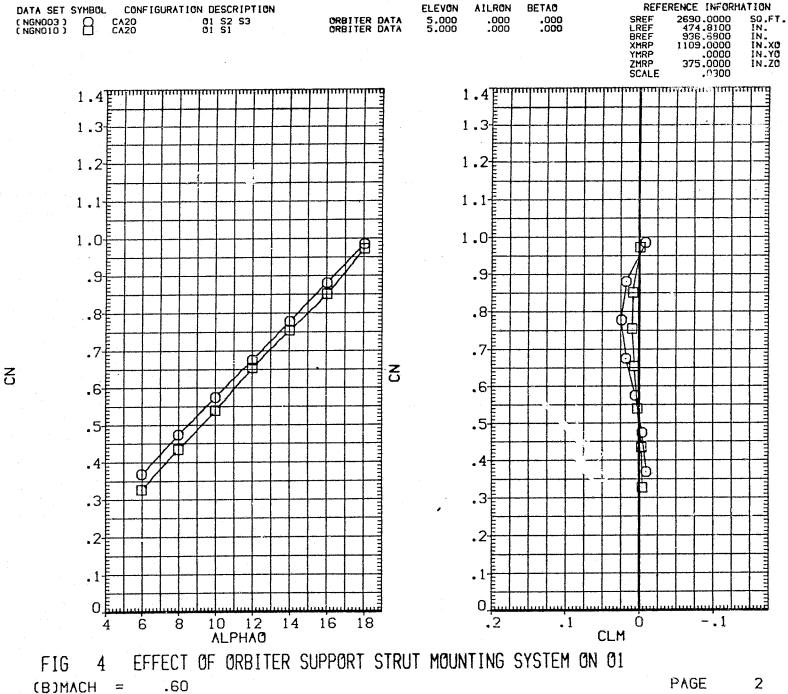


FIG 4 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 01

(A)MACH = .30



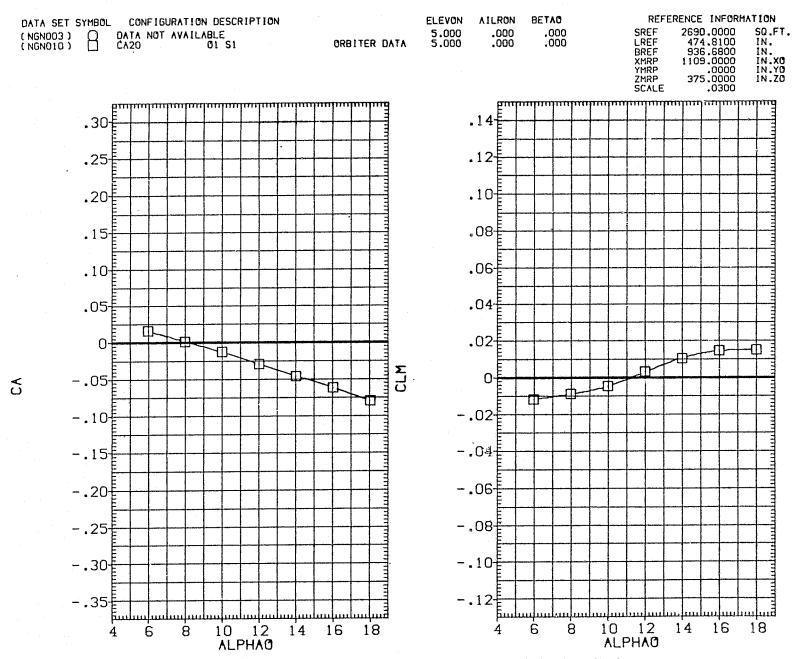


FIG 4 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 01

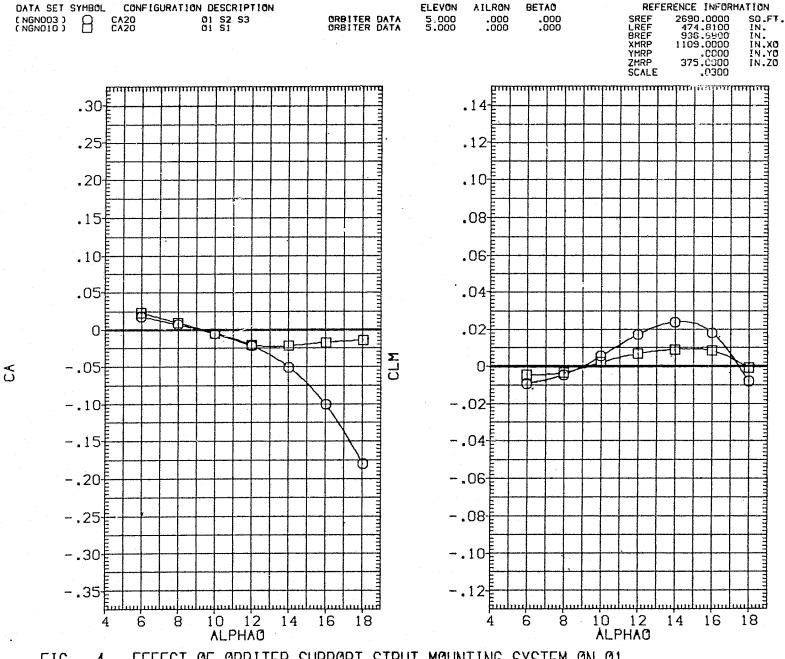


FIG 4 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 01
(B)MACH = .60

4

<u> </u>

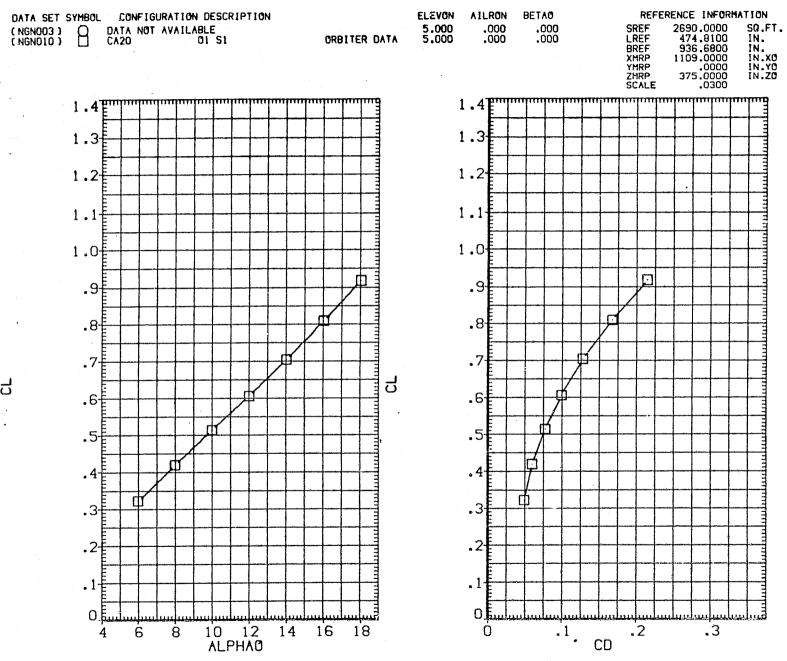
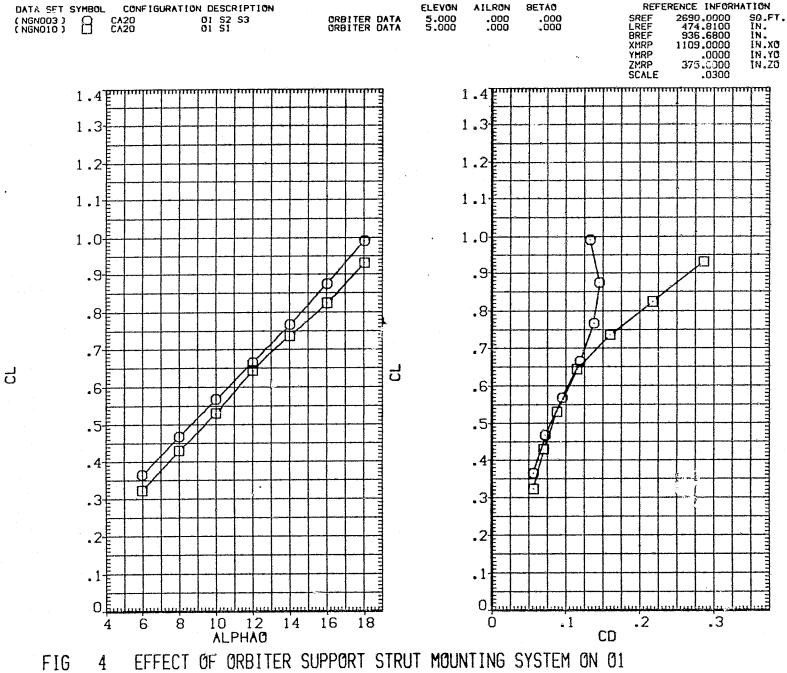


FIG 4 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 01



.60 (B)MACH

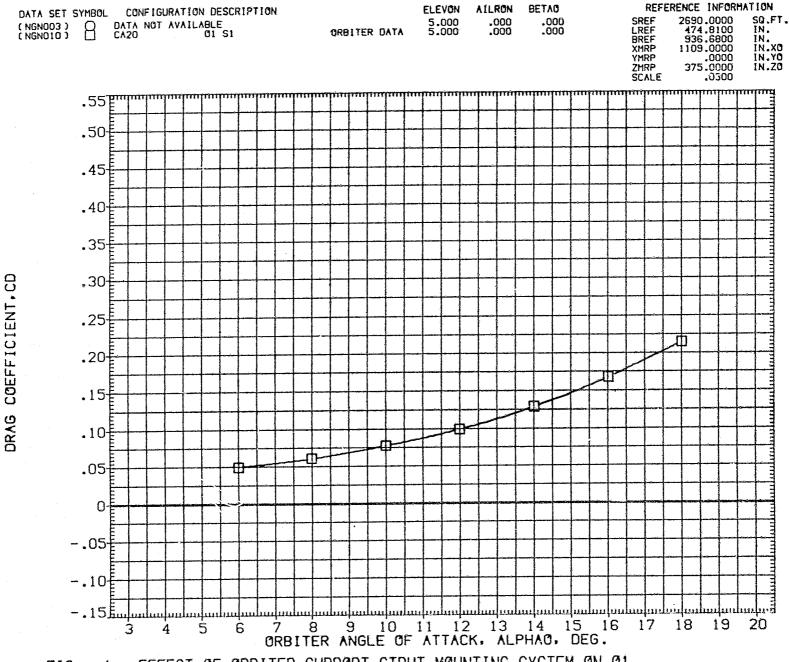
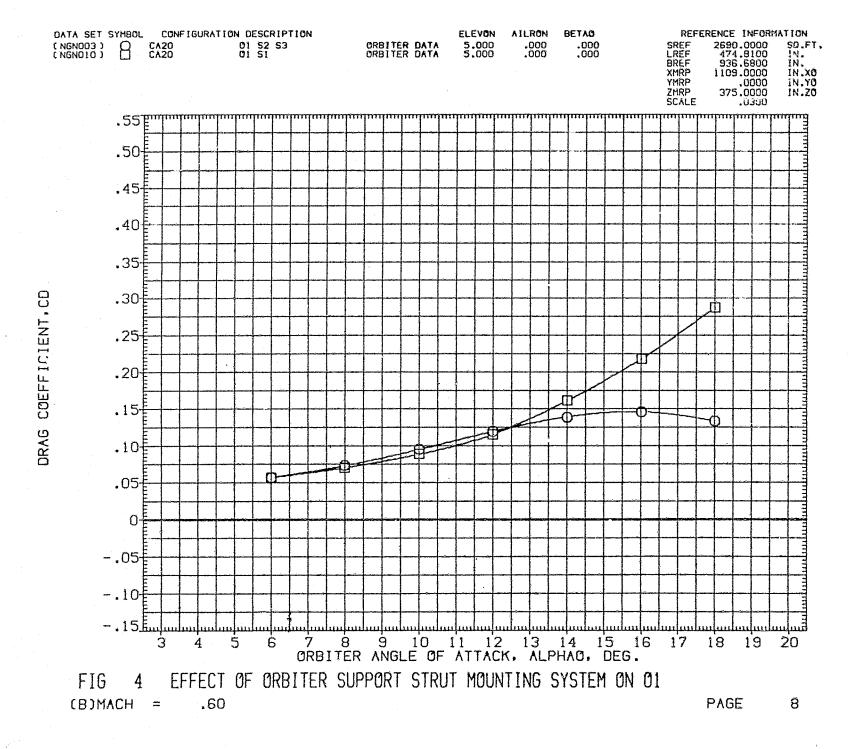
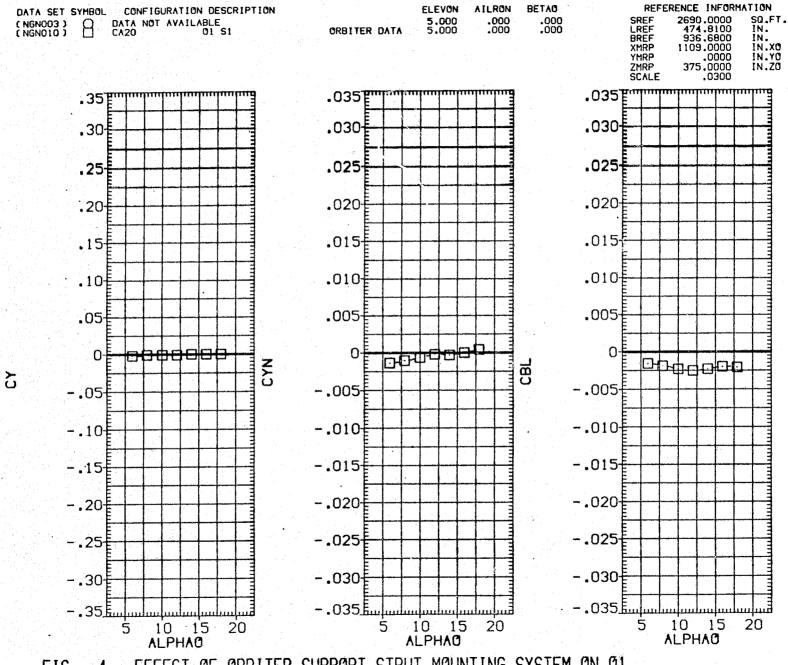


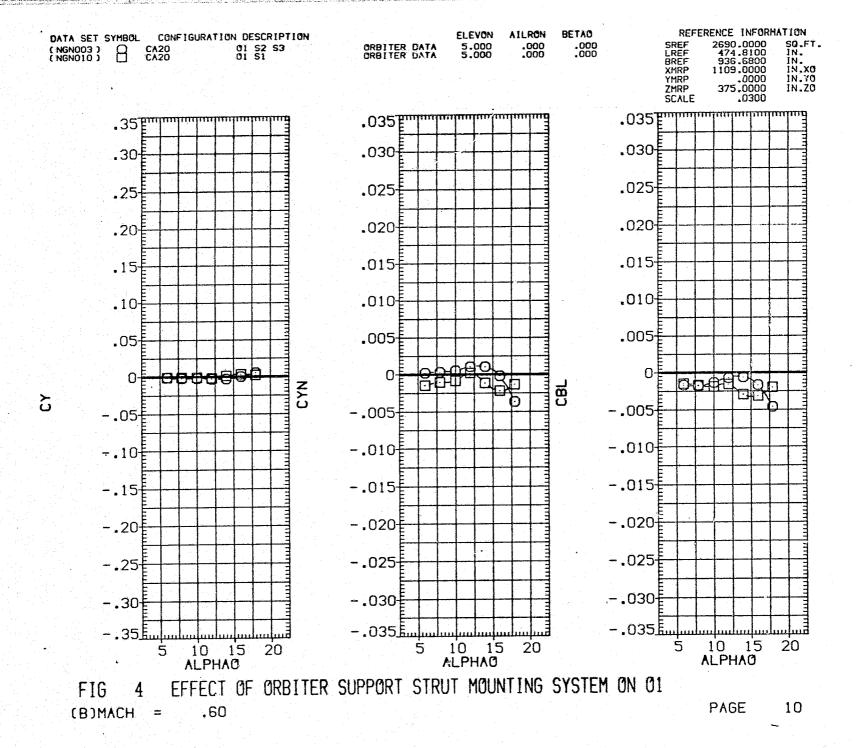
FIG 4 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 01
(A)MACH = .30

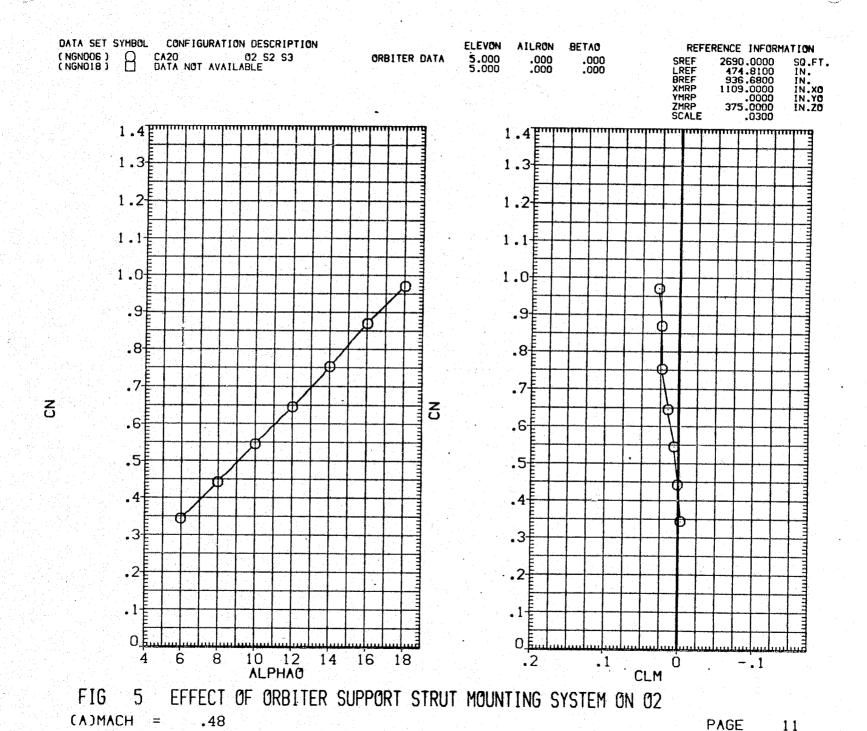


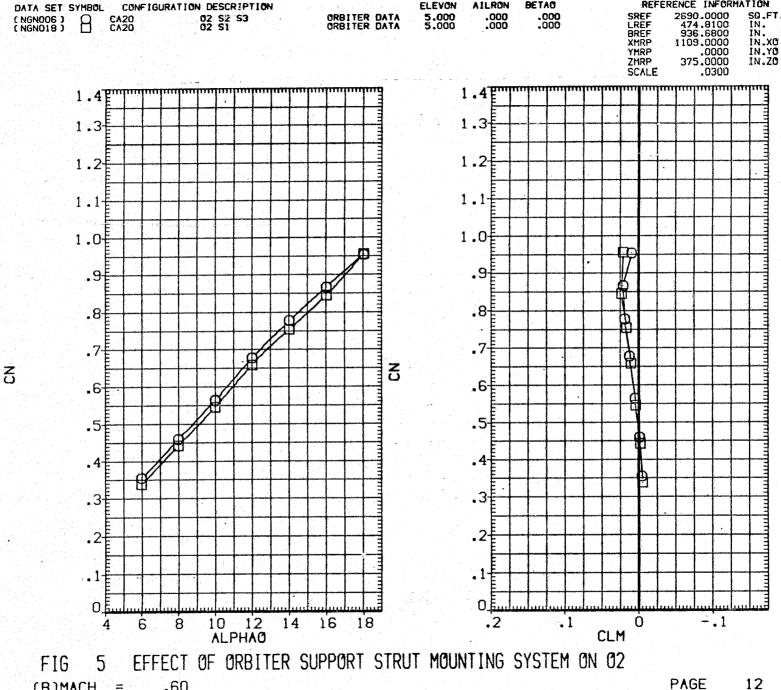


10

FIG 4 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 01







BETAO

AILRON

REFERENCE INFORMATION

PAGE (B)MACH .60 =

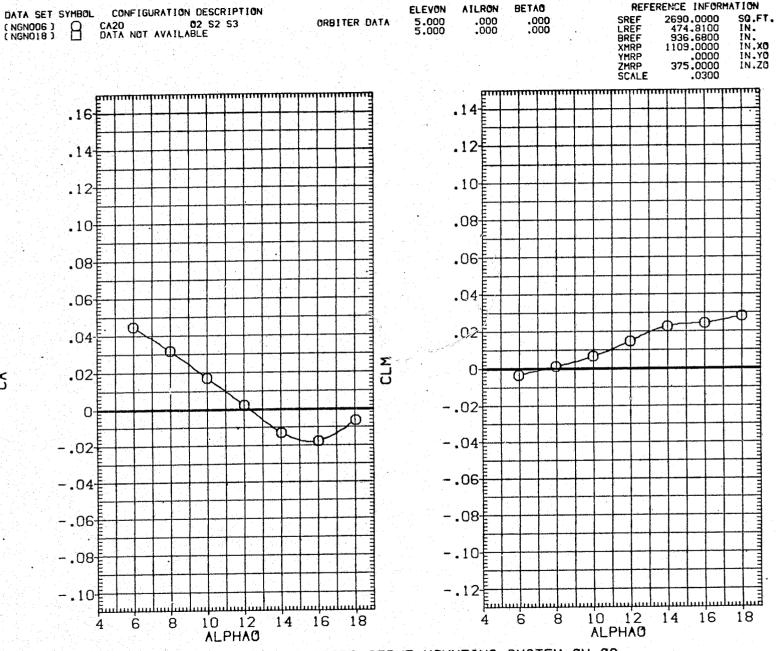


FIG 5 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 02

(A)MACH = .48

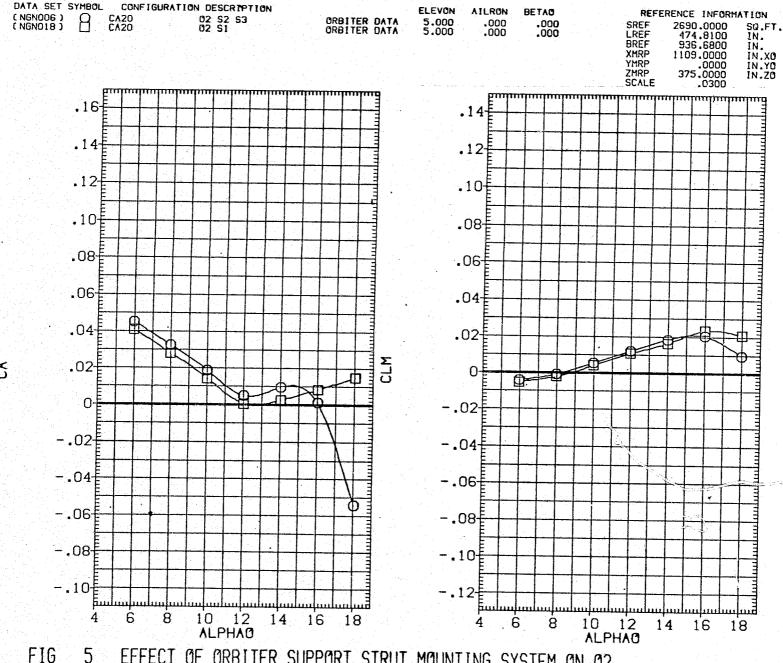
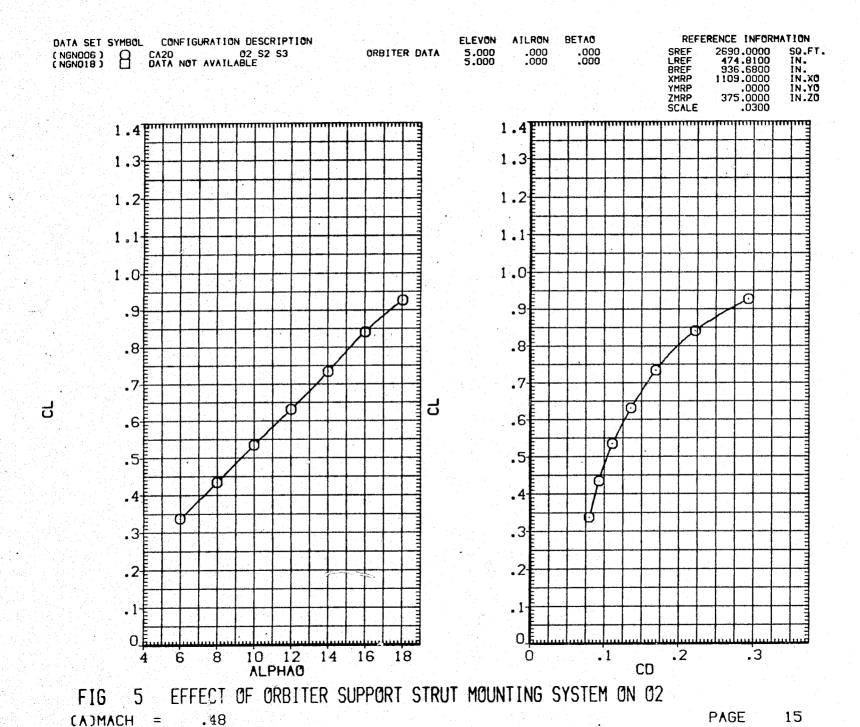
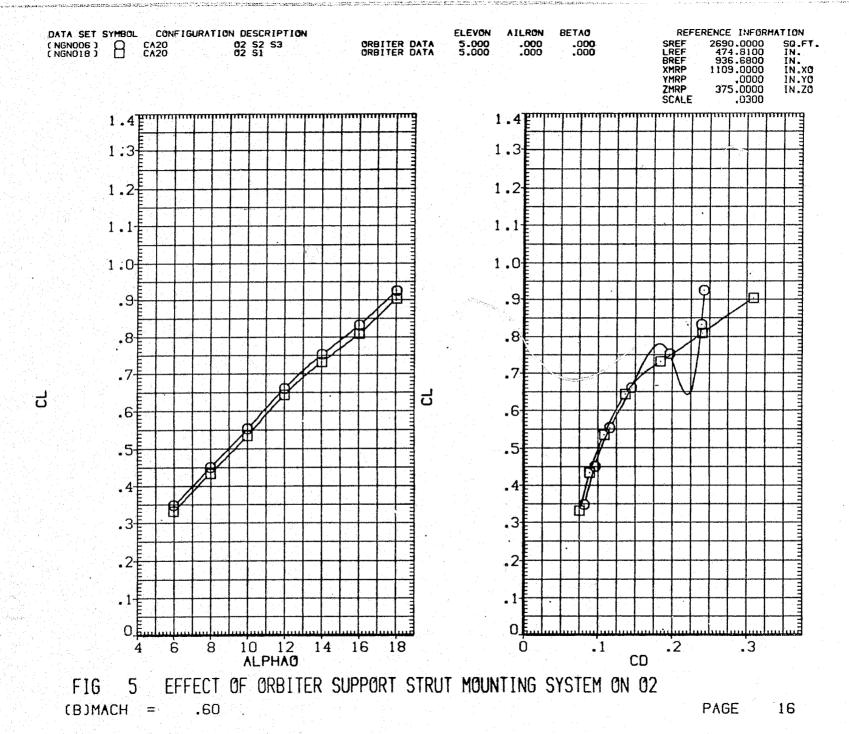
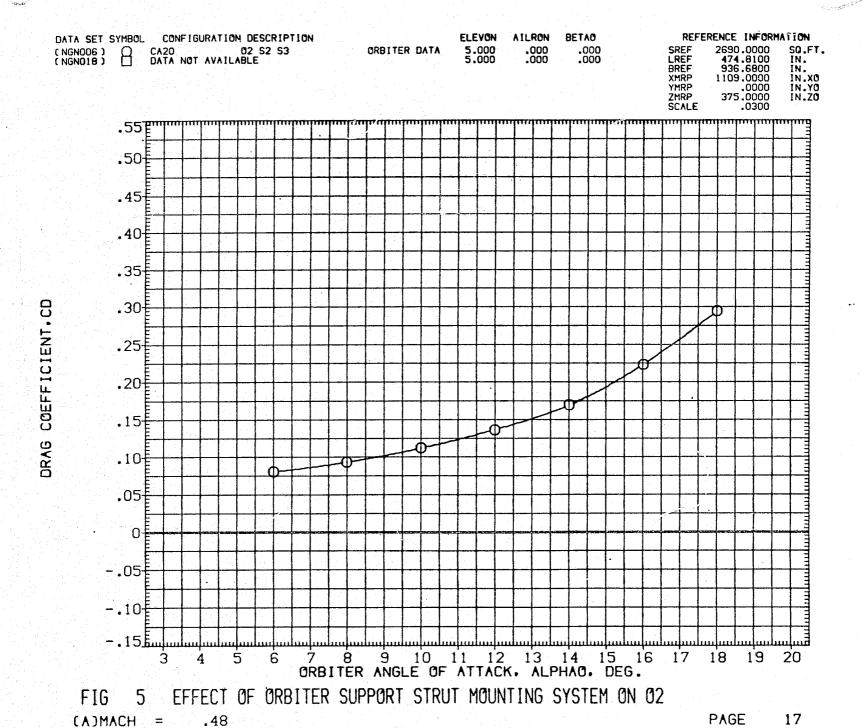
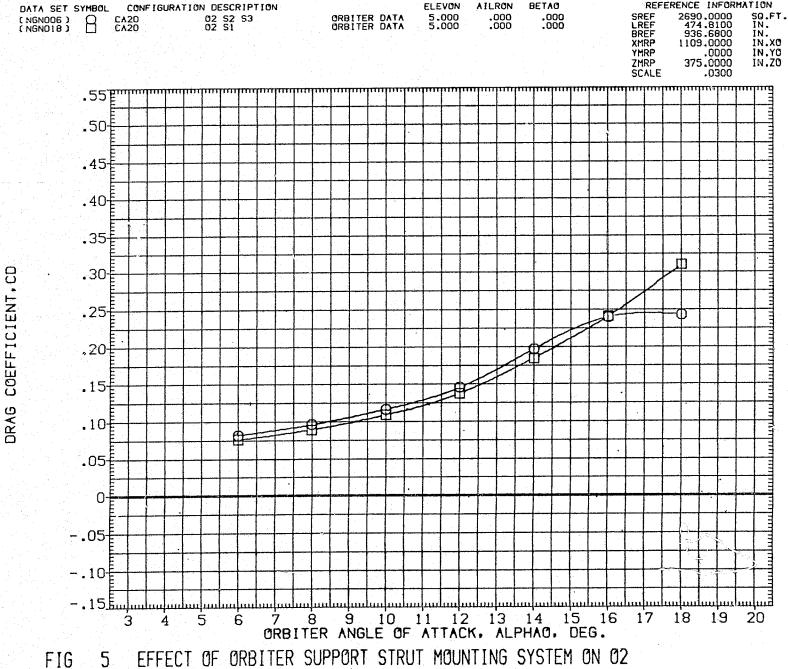


FIG 5 EFFECT OF ORBITER SUPPORT STRUT MOUNTING SYSTEM ON 02
(B)MACH = .60









ELEVON AILRON

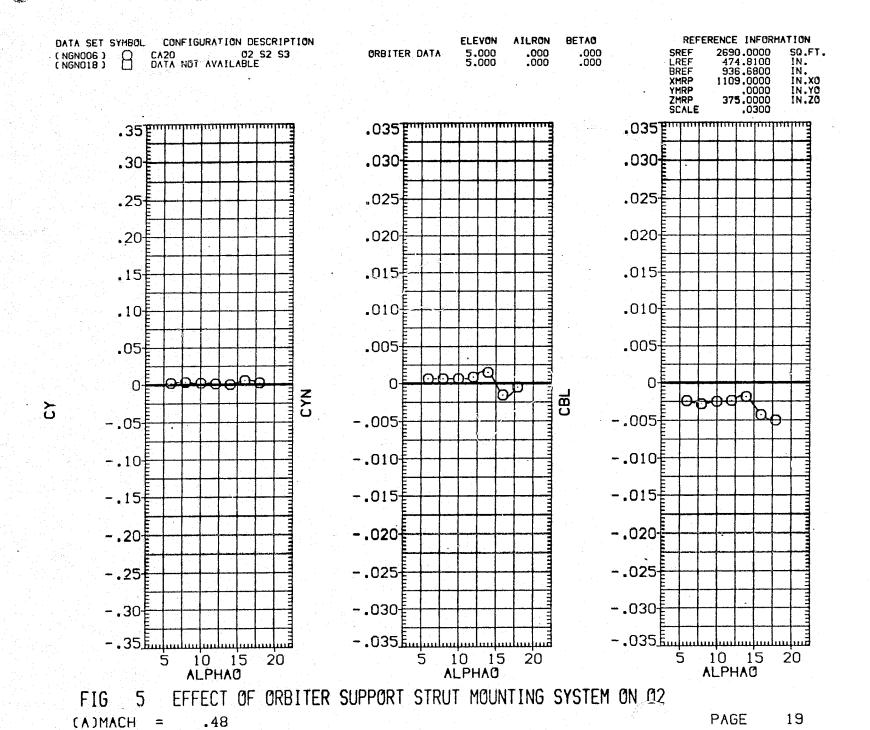
BETAO

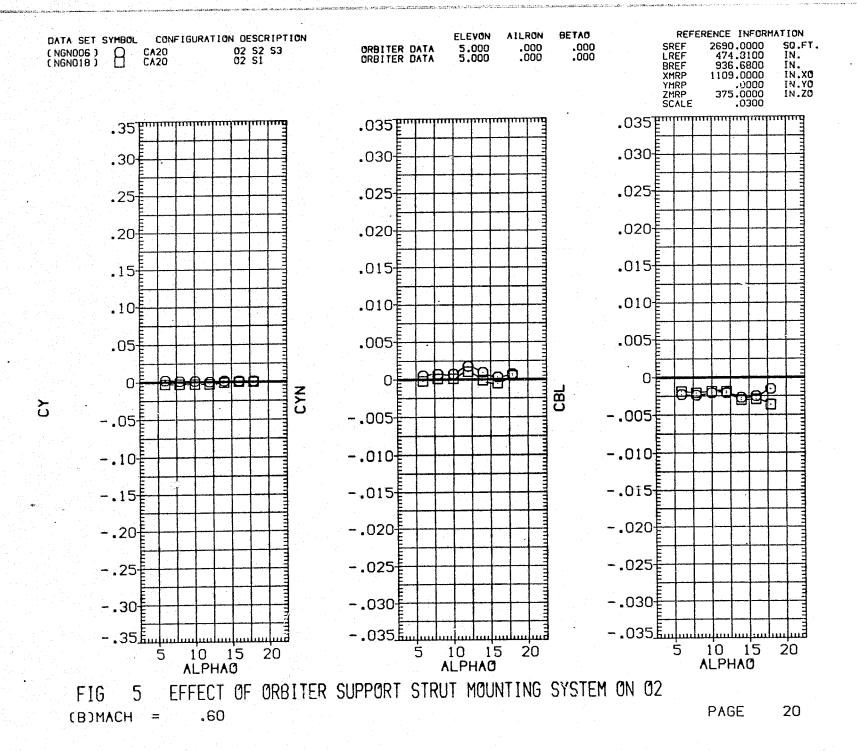
FIG (B)MACH = .60

PAGE

REFERENCE INFORMATION

18





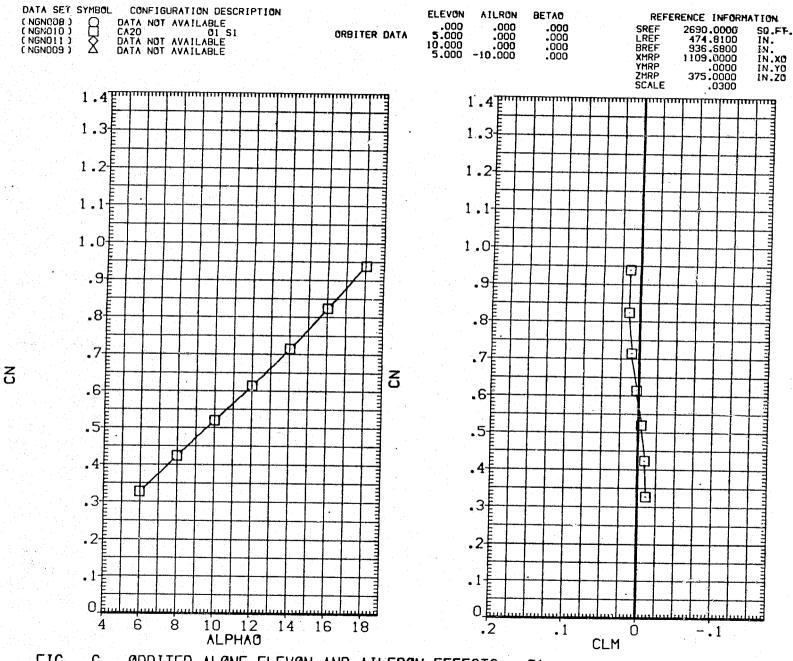
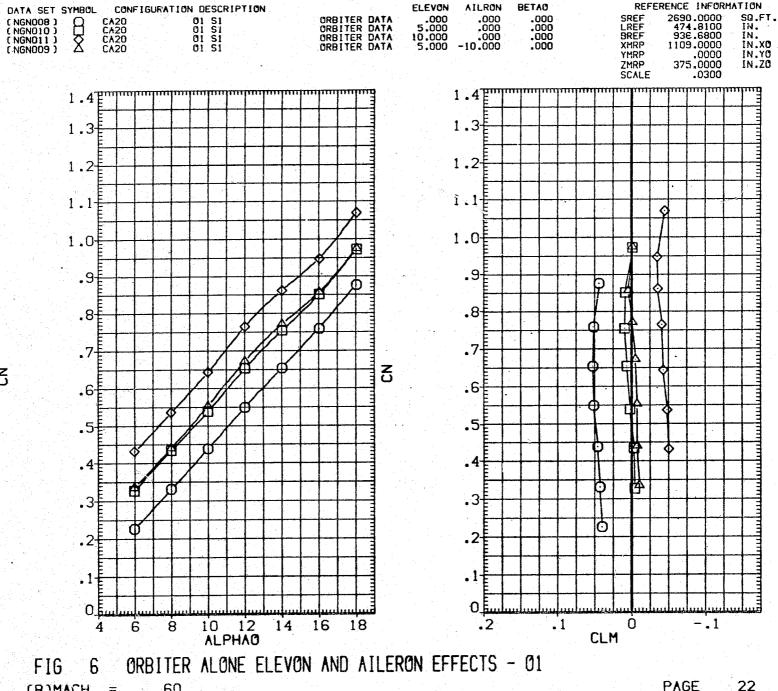


FIG 6 ORBITER ALONE ELEVON AND AILERON EFFECTS - 01
(A)MACH = .30



PAGE .60 CBOMACH

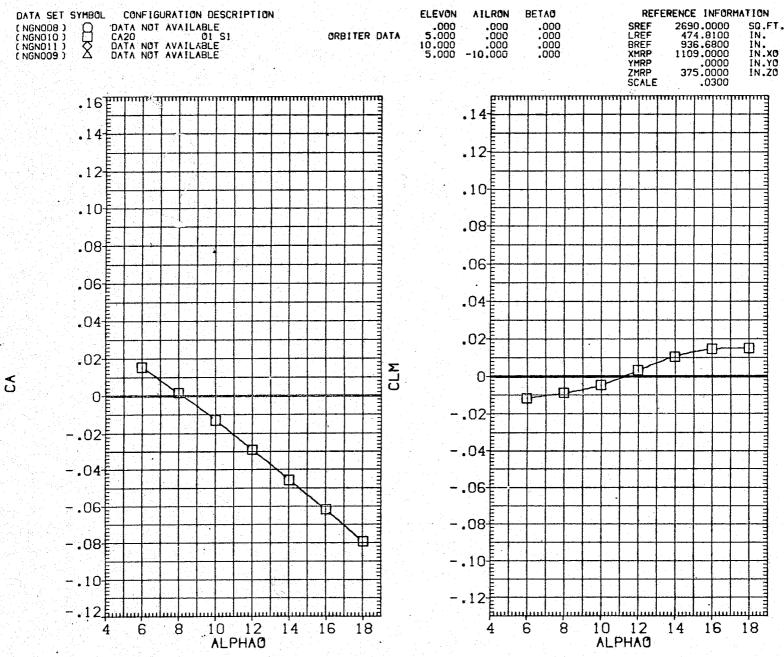
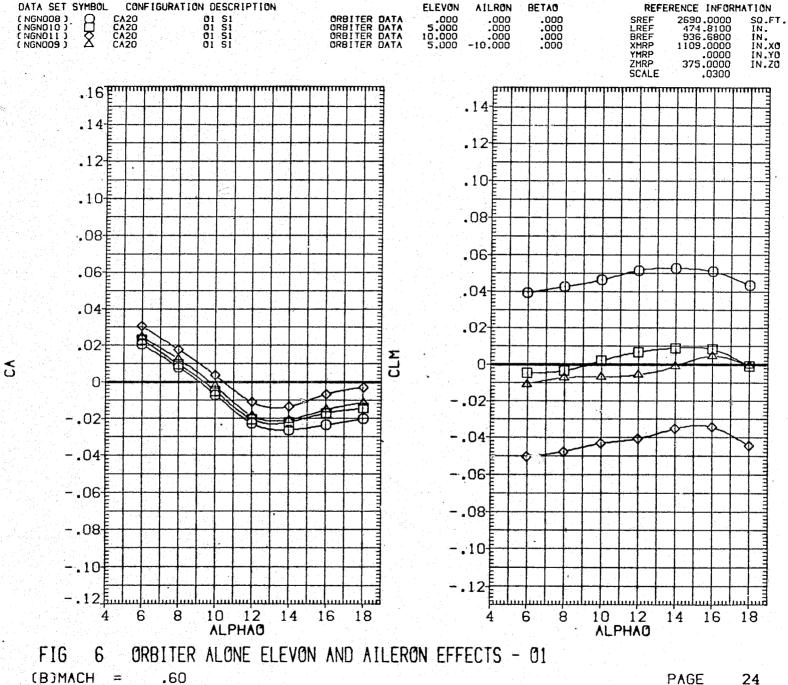


FIG 6 ORBITER ALONE ELEVON AND AILERON EFFECTS - 01
(A)MACH = .30



(B)MACH .60 PAGE

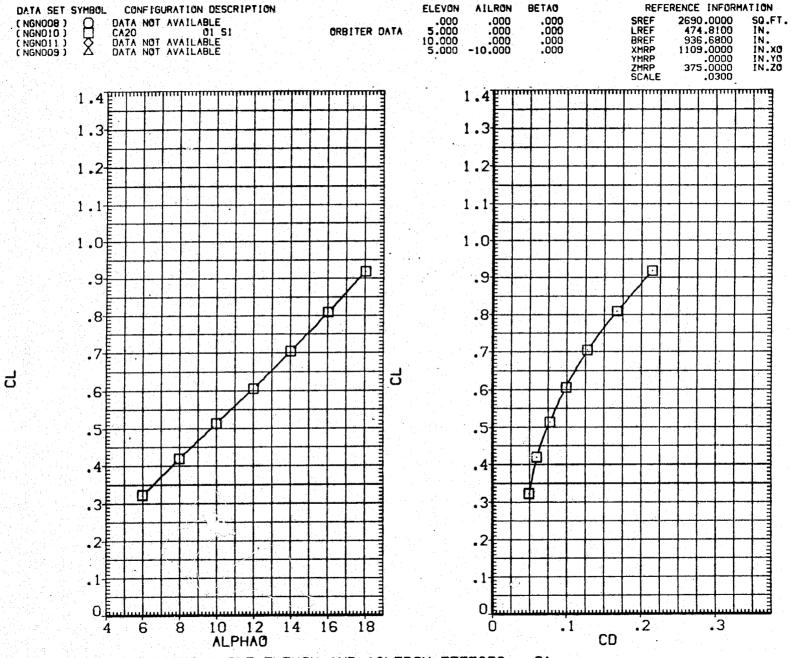
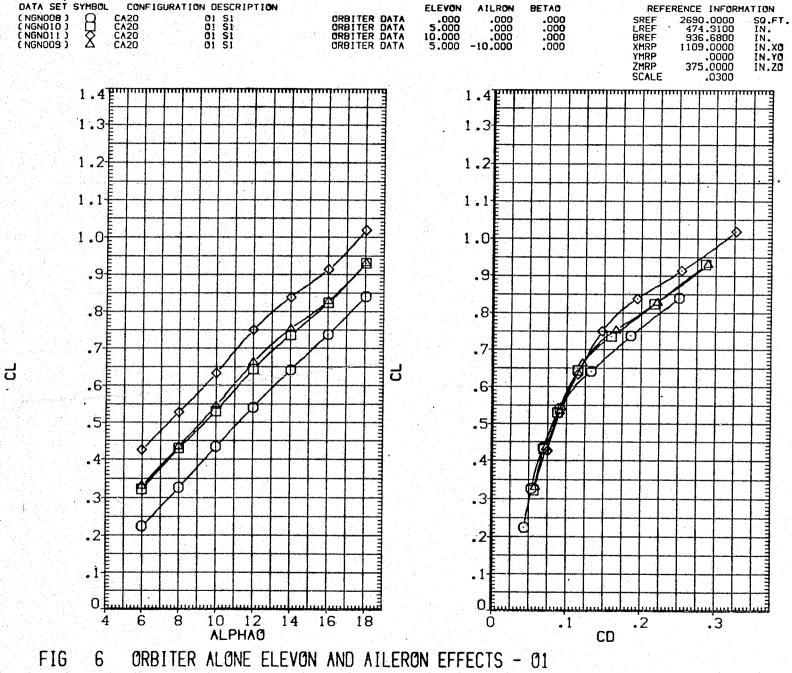


FIG 6 ORBITER ALONE ELEVON AND AILERON EFFECTS - 01

(A)MACH = .30



(B)MACH = .60

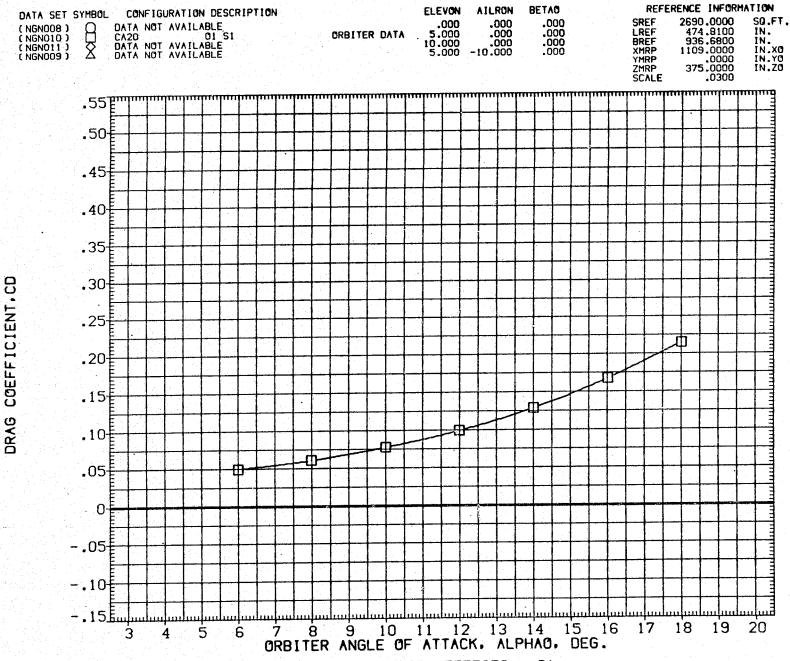
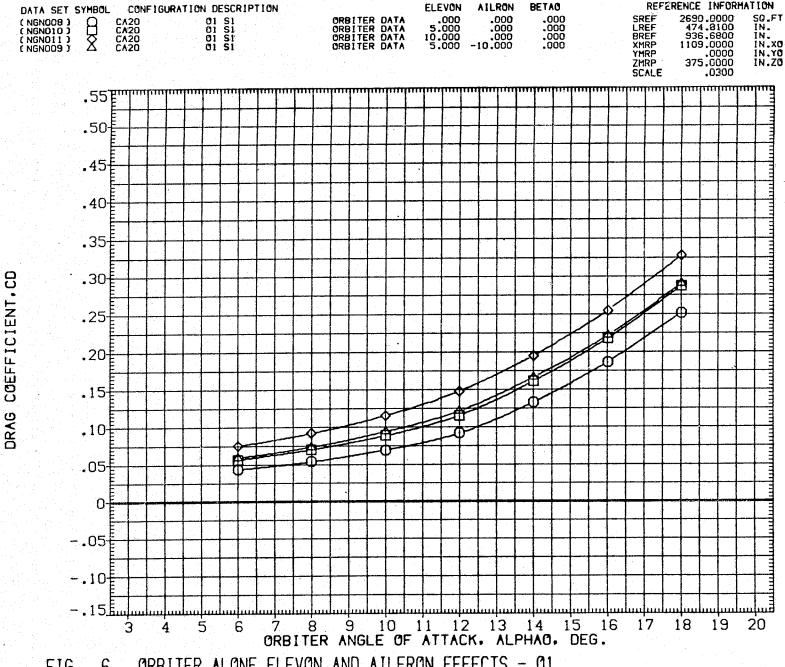


FIG 6 ORBITER ALONE ELEVON AND AILERON EFFECTS - 01



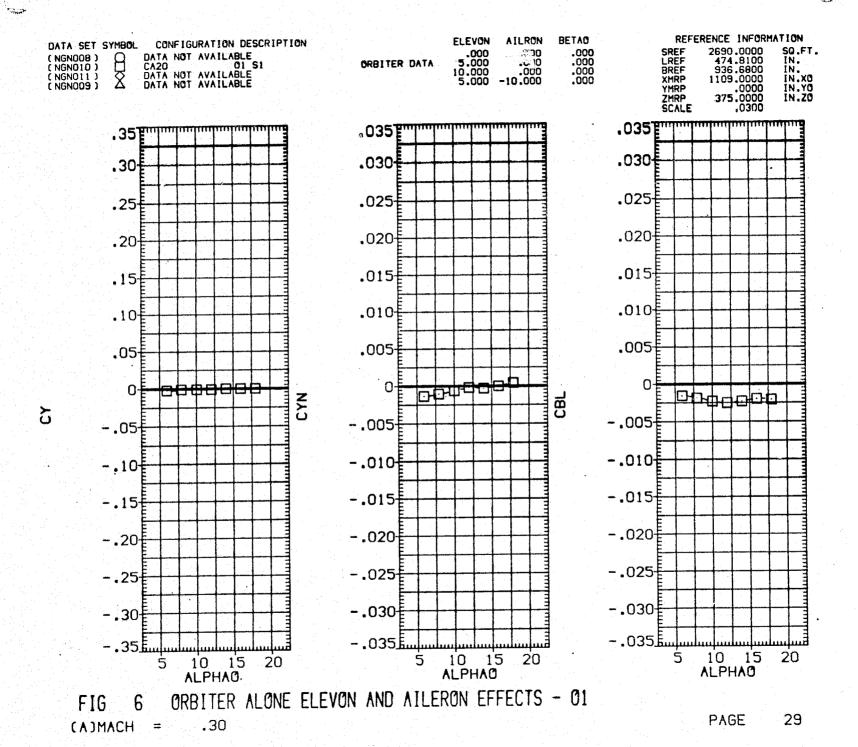
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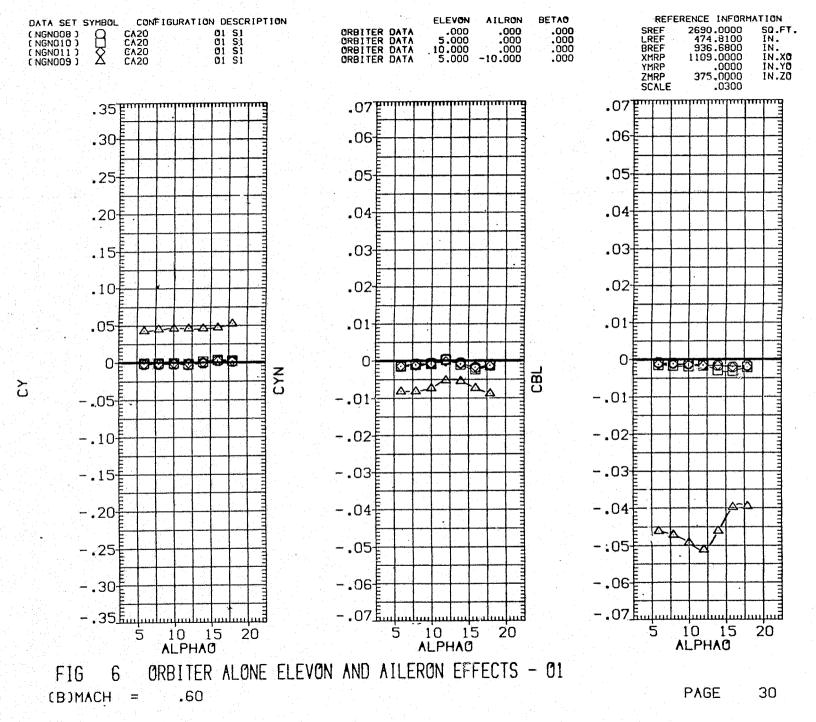
ORBITER ALONE ELEVON AND AILERON EFFECTS - 01 FIG .60 (B)MACH =

PAGE

28

REFERENCE INFORMATION





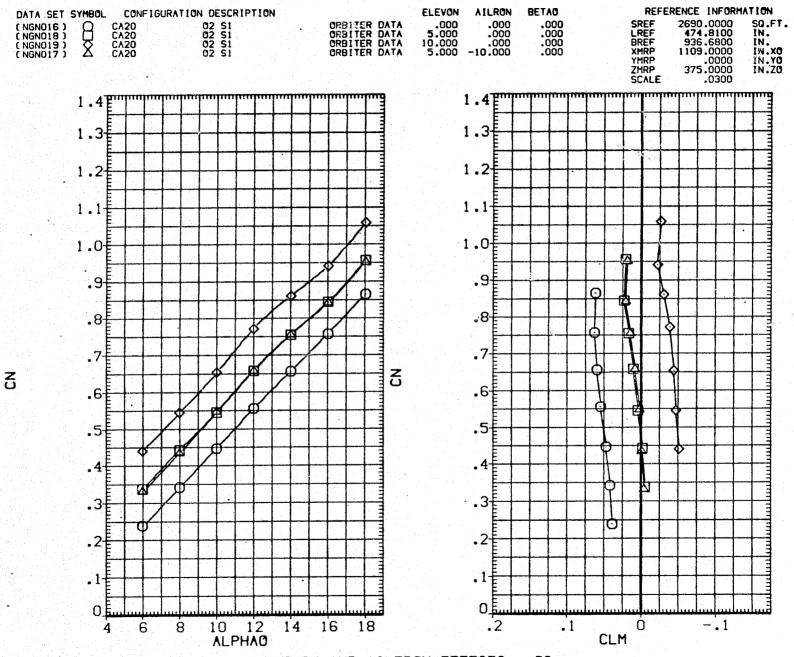


FIG 7 ORBITER ALONE ELEVON AND AILERON EFFECTS - 02

(A)MACH = .60

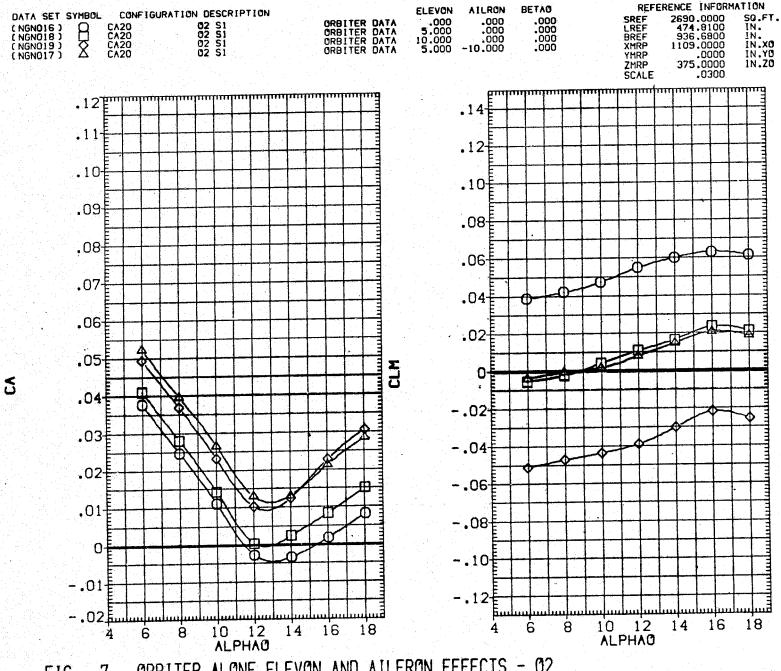


FIG 7 ORBITER ALONE ELEVON AND AILERON EFFECTS - 02

(A)MACH = .60

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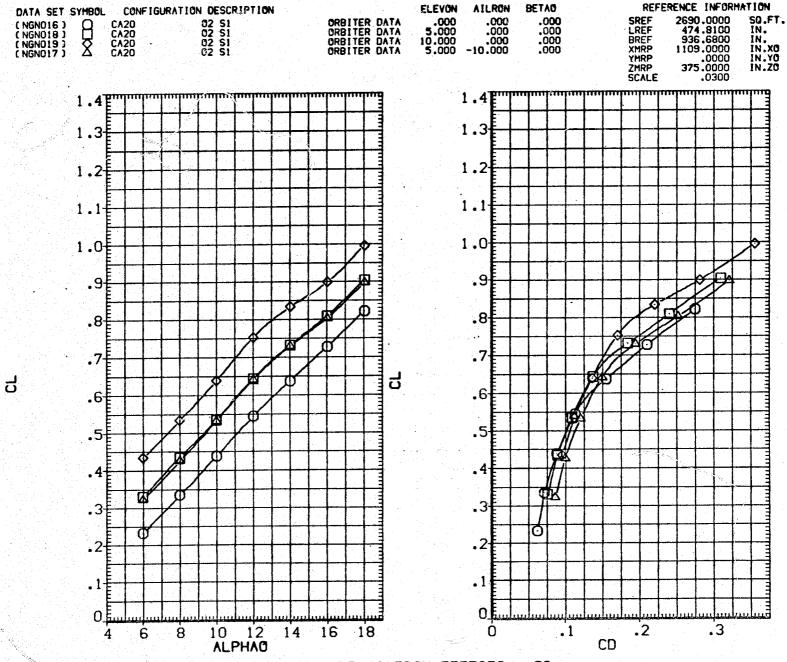


FIG 7 ORBITER ALONE ELEVON AND AILERON EFFECTS - 02

(A)MACH = .60

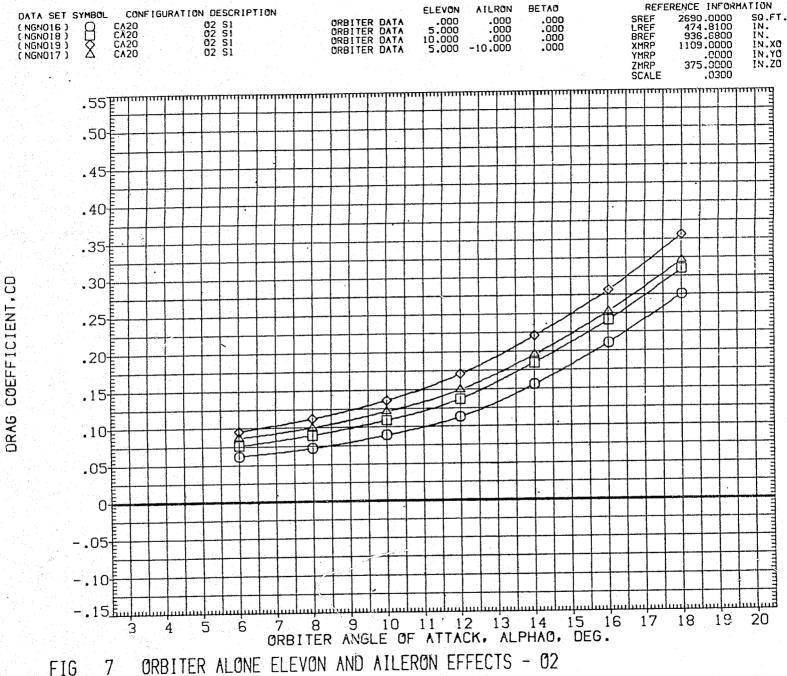


FIG 7 ORBITER ALONE ELEVON AND AILERON EFFECTS - U.

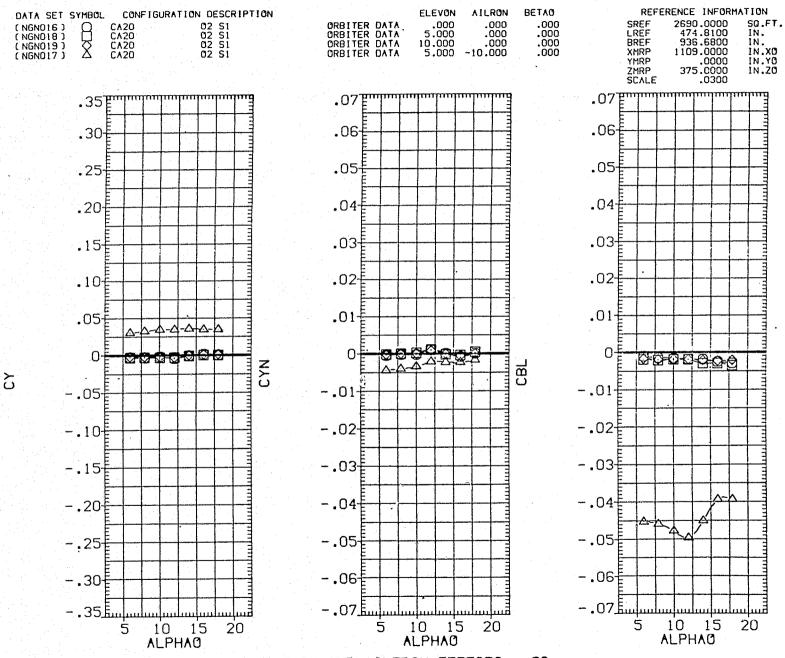
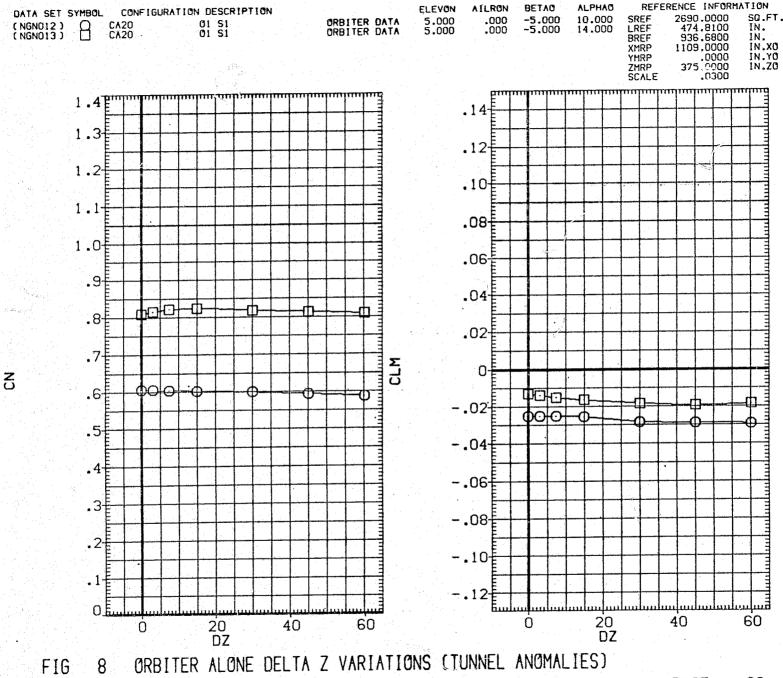
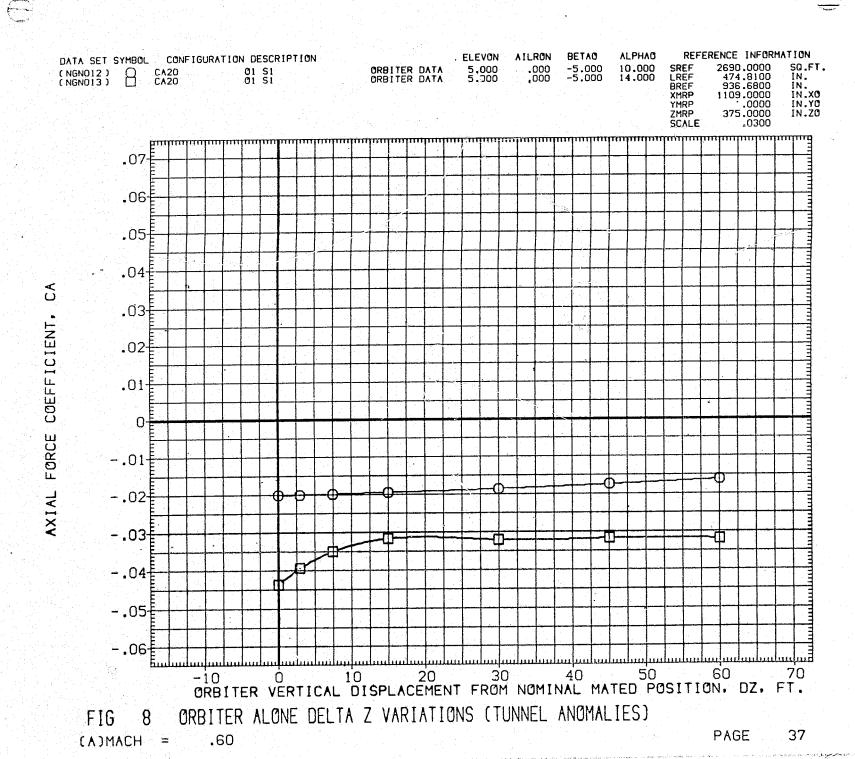


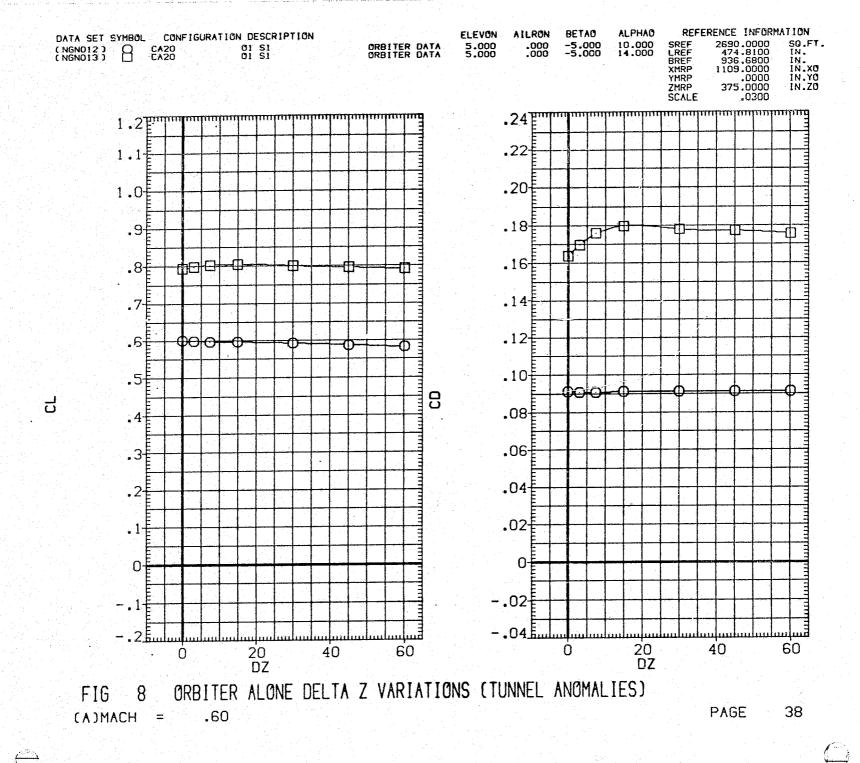
FIG 7 ORBITER ALONE ELEVON AND AILERON EFFECTS - 02

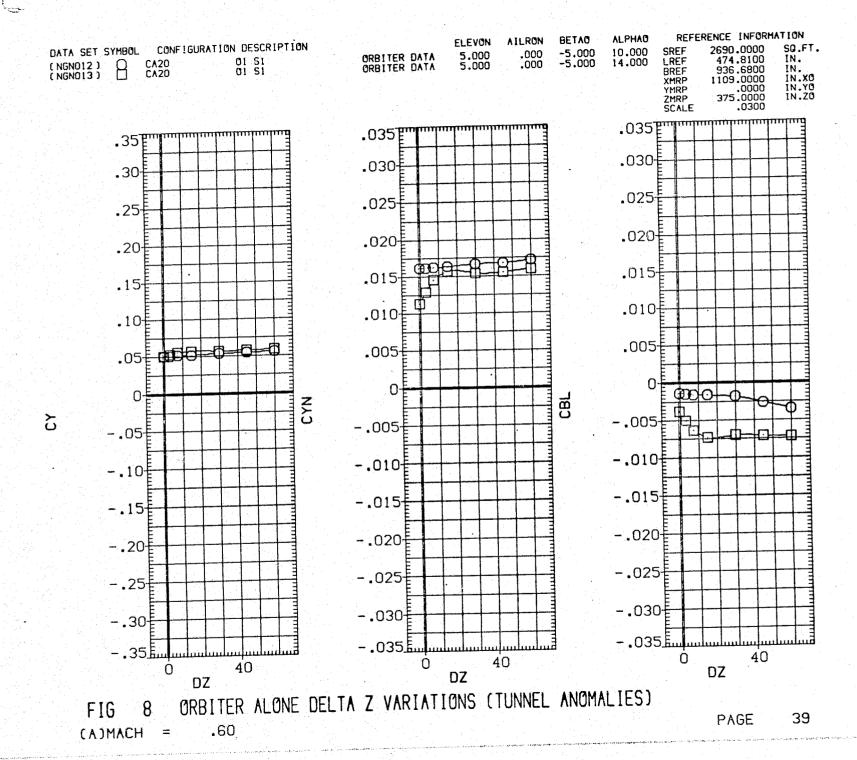
(A)MACH = .60

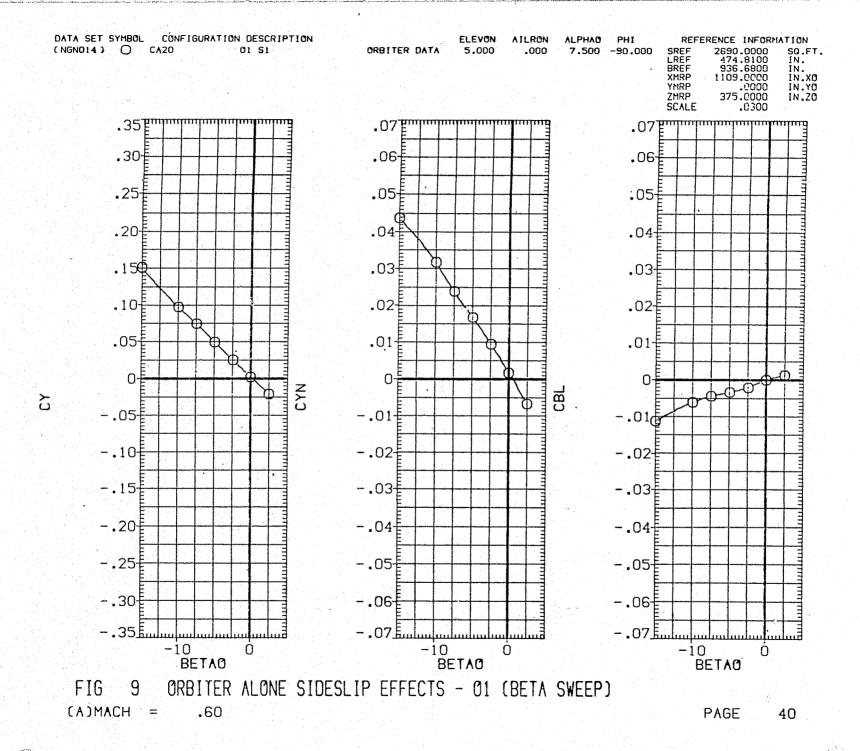


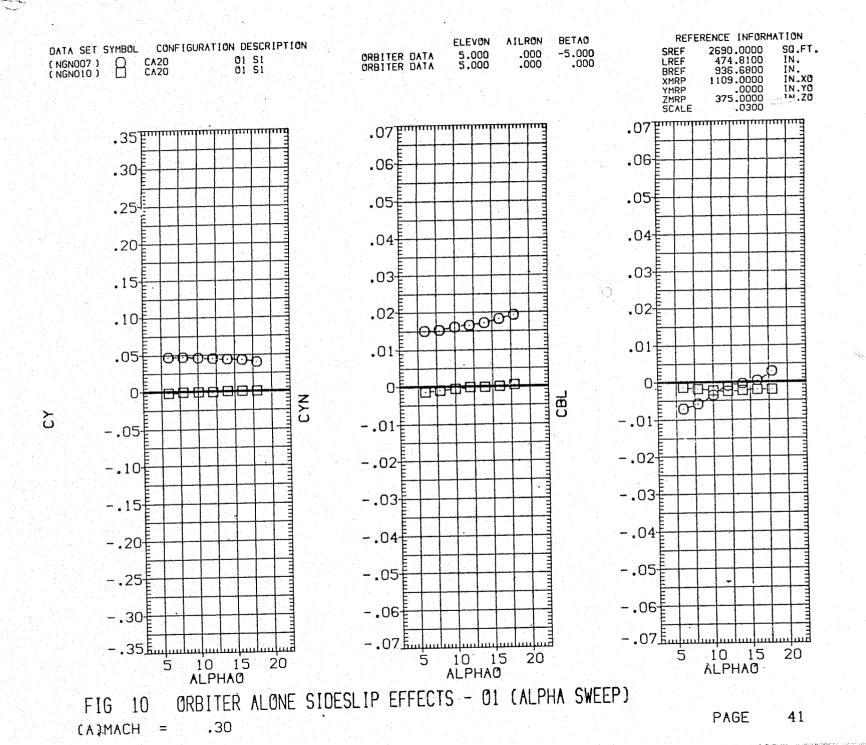
.60 (A)MACH =

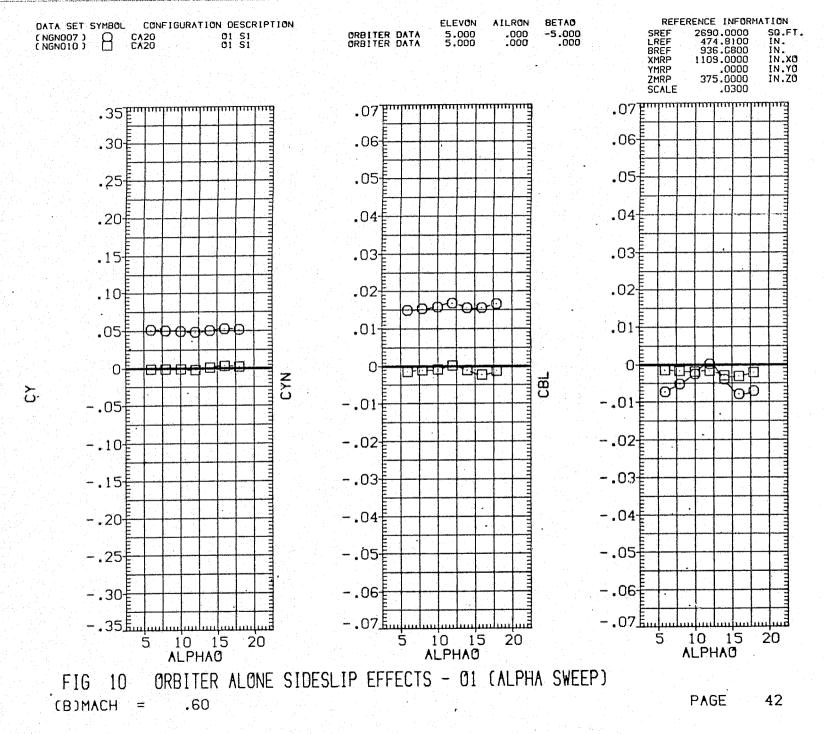












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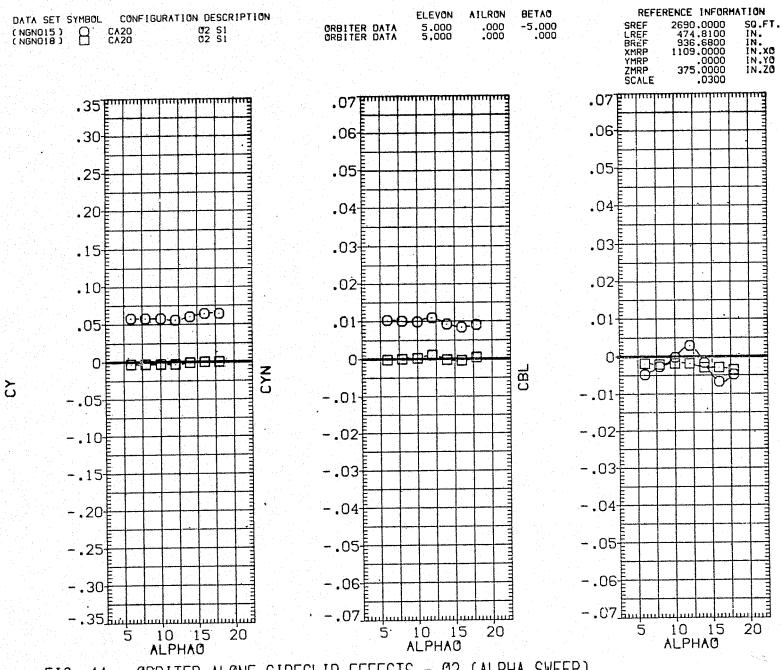
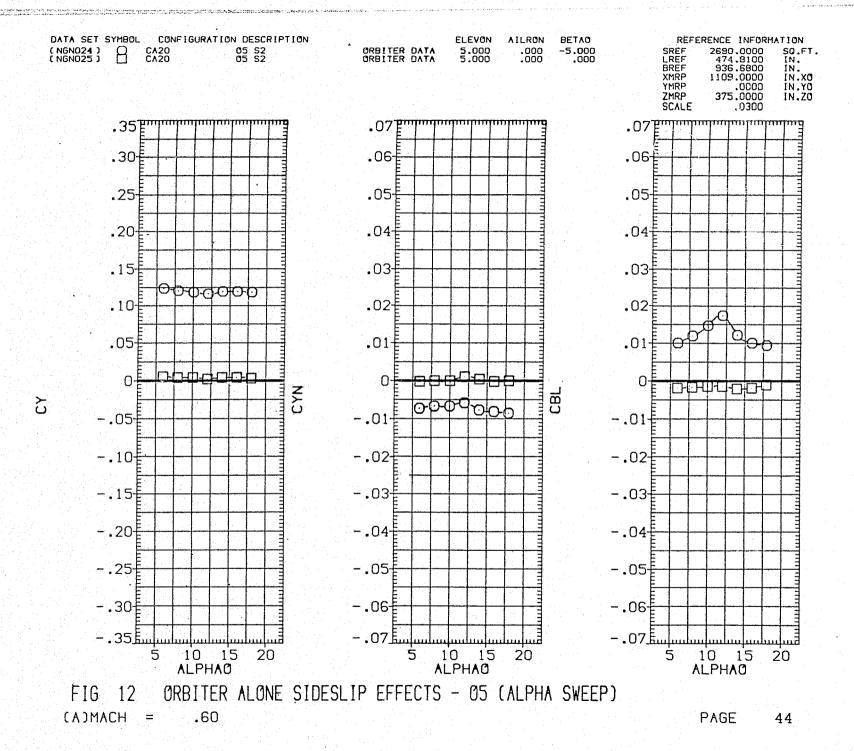


FIG 11 ORBITER ALONE SIDESLIP EFFECTS - 02 (ALPHA SWEEP)

(A)MACH = .60



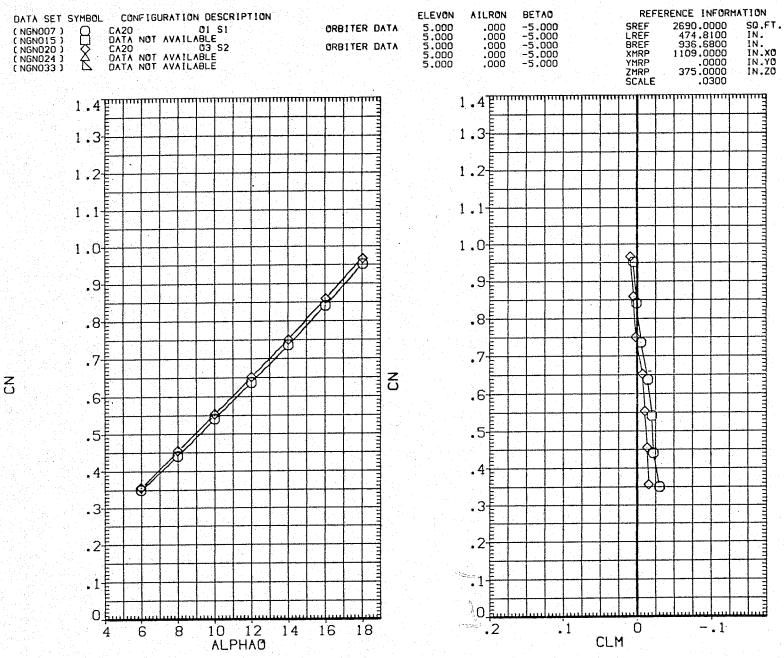
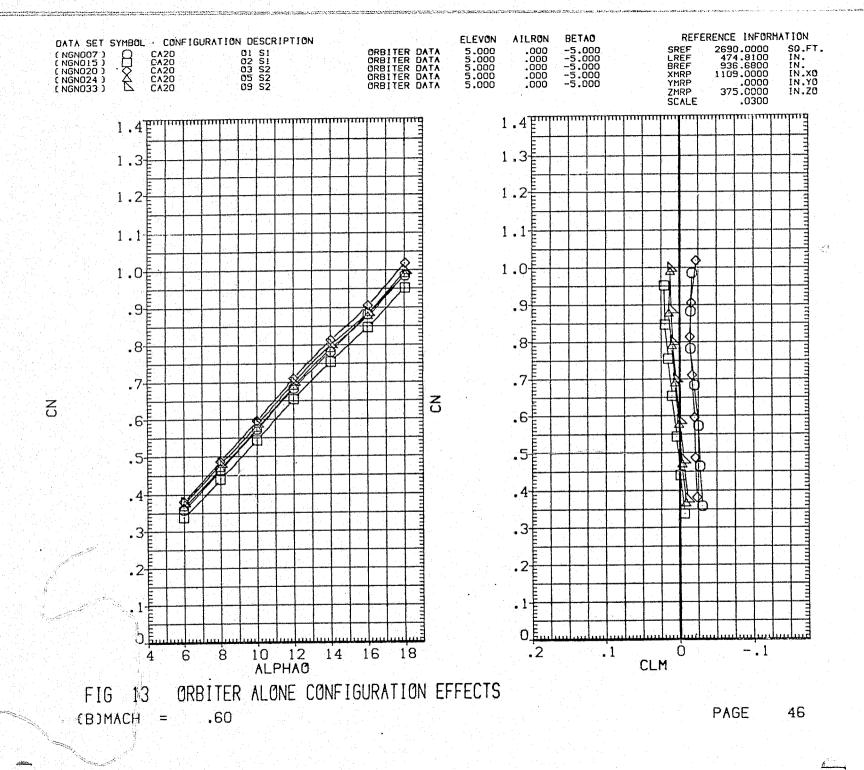


FIG 13 ORBITER ALONE CONFIGURATION EFFECTS

(A)MACH = .30



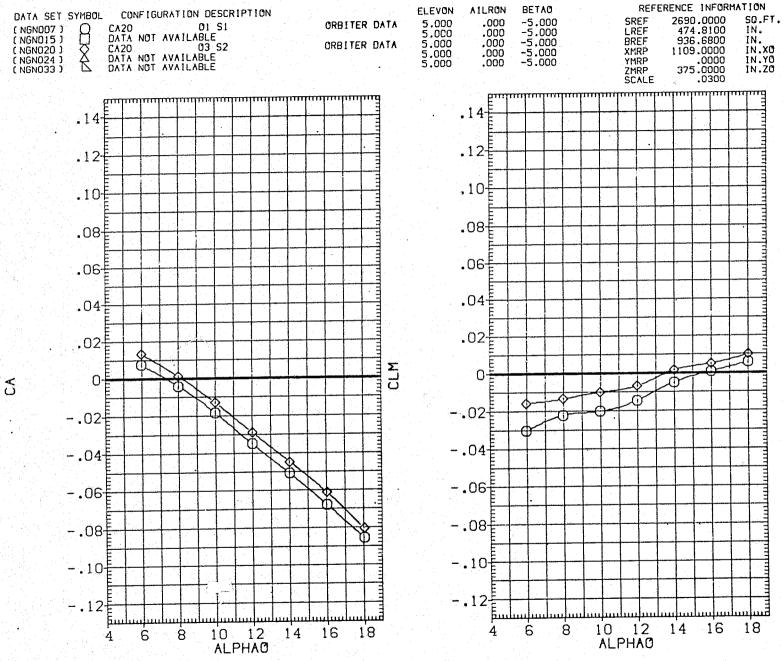


FIG 13 ORBITER ALONE CONFIGURATION EFFECTS

(A)MACH = .30

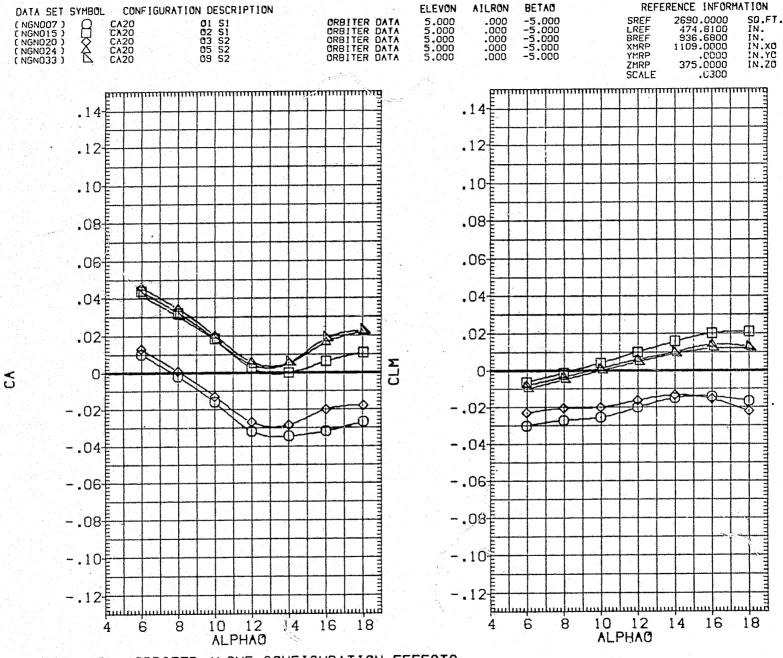


FIG 13 ORBITER ALONE CONFIGURATION EFFECTS

(B)MACH = .60

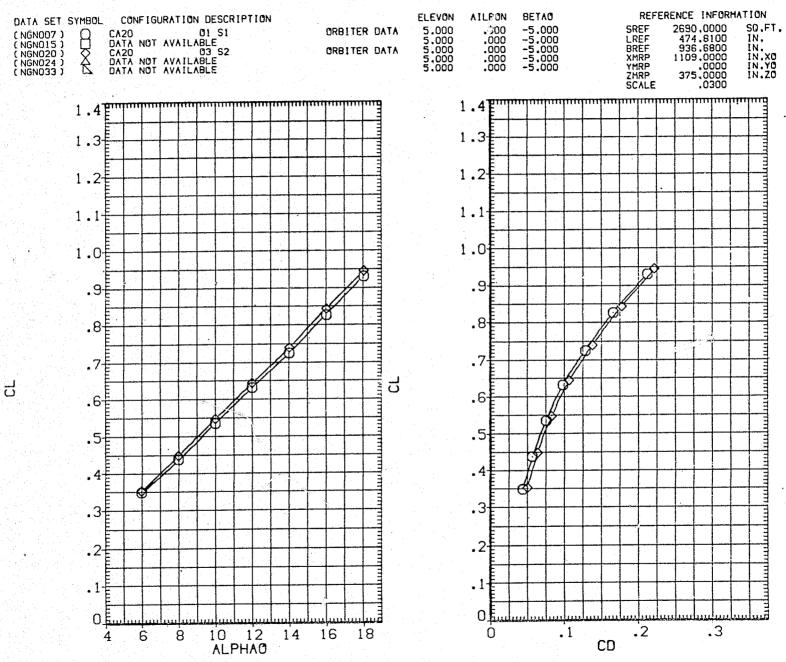
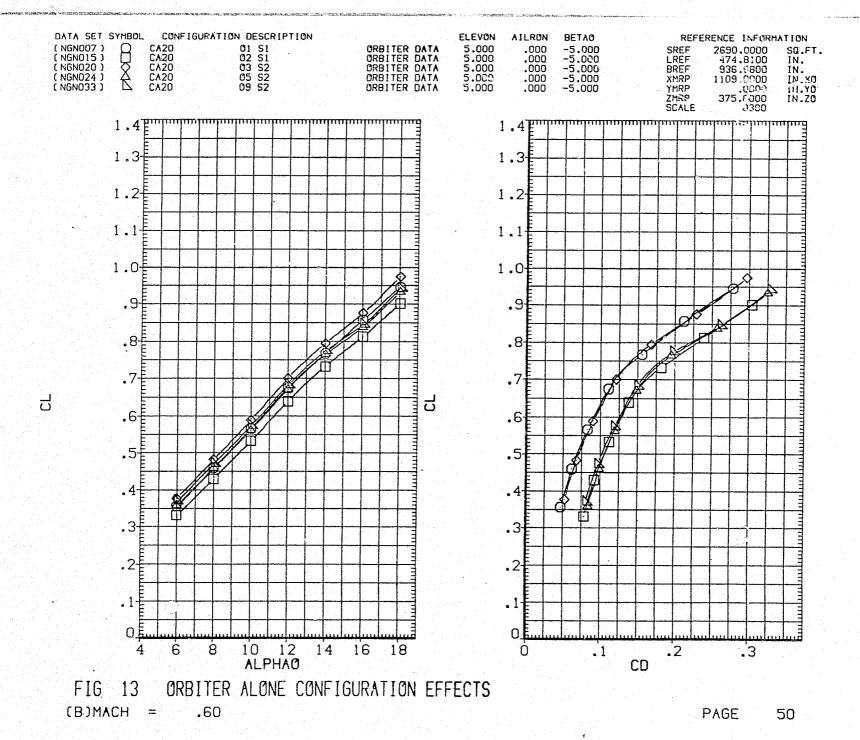


FIG 13 ORBITER ALONE CONFIGURATION EFFECTS

(A)MACH = .30



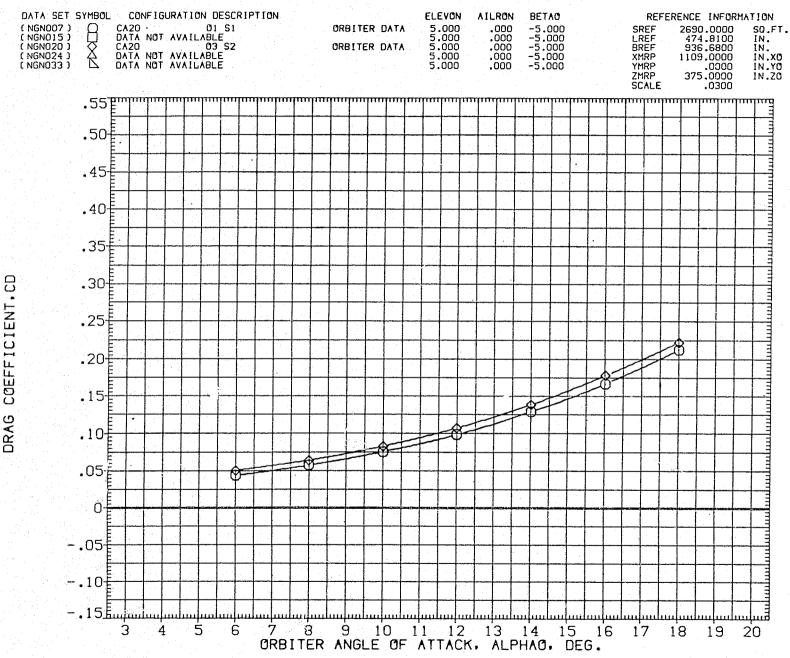
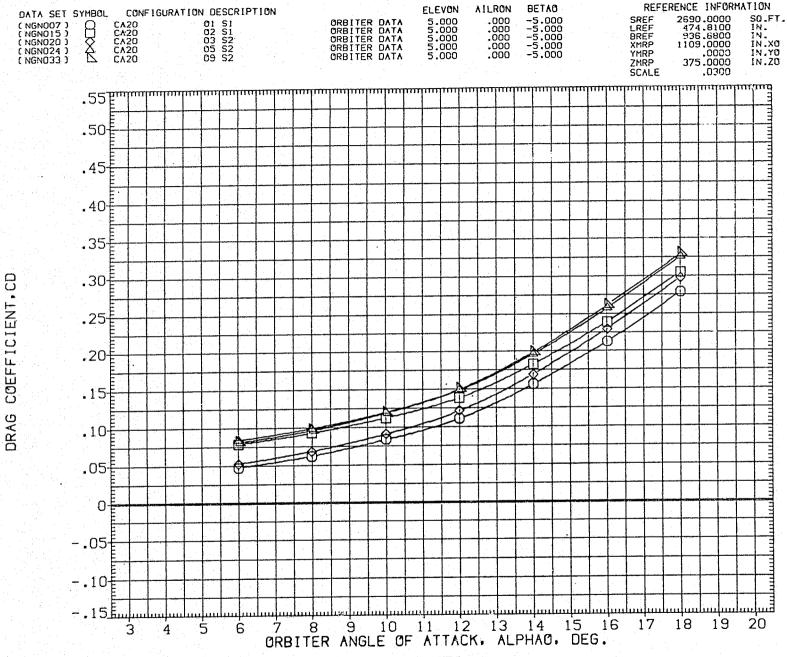


FIG 13 ORBITER ALONE CONFIGURATION EFFECTS

(A)MACH = .30



FIC 13 ORBITER ALONE CONFIGURATION EFFECTS

(B)MACH = .60

provide productive descriptive and the contractive providers of the contractive of the co

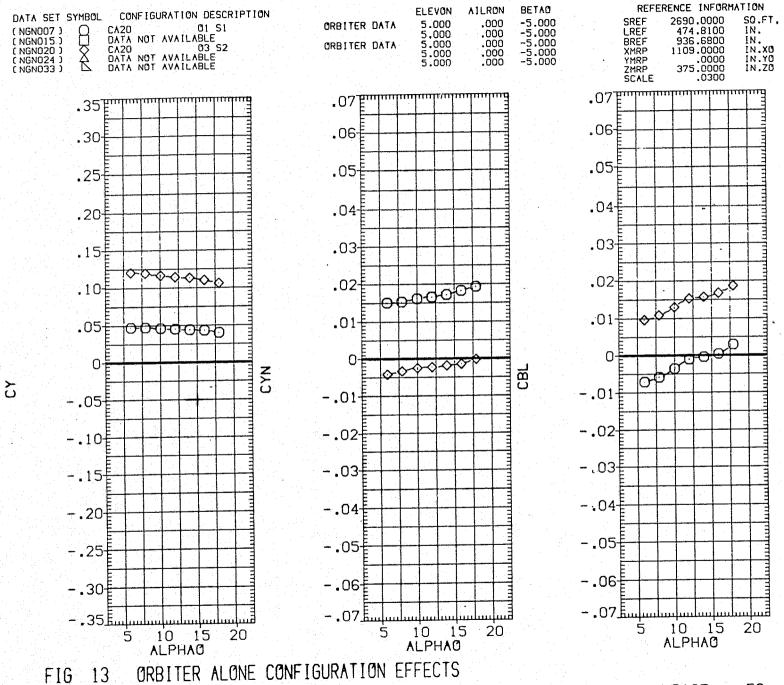
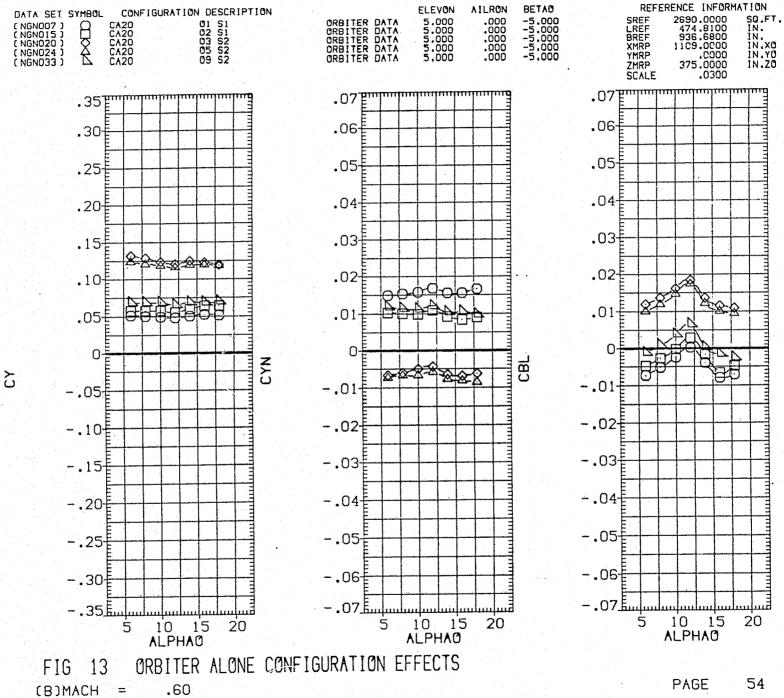


FIG 13 ORBITER ALONE CONFIGURATION EFFECT

PAGE



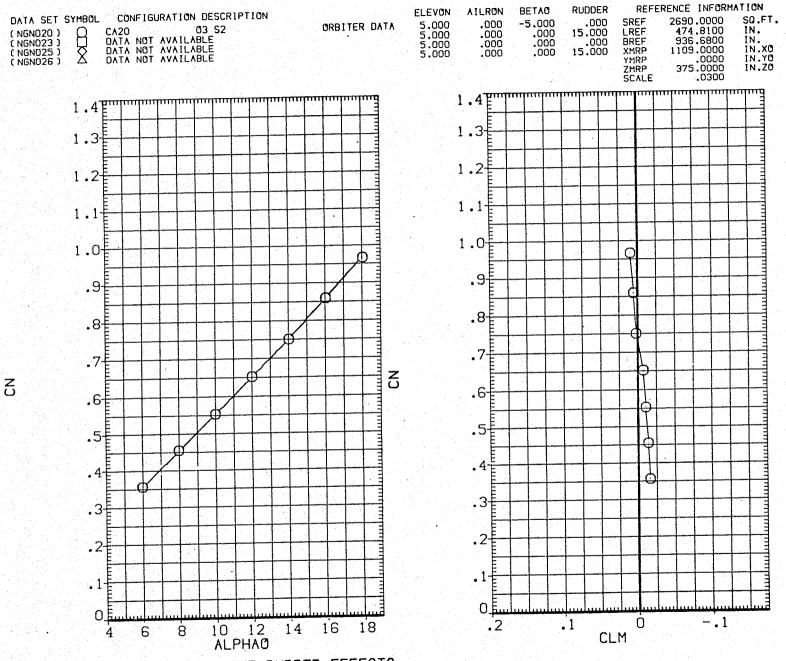
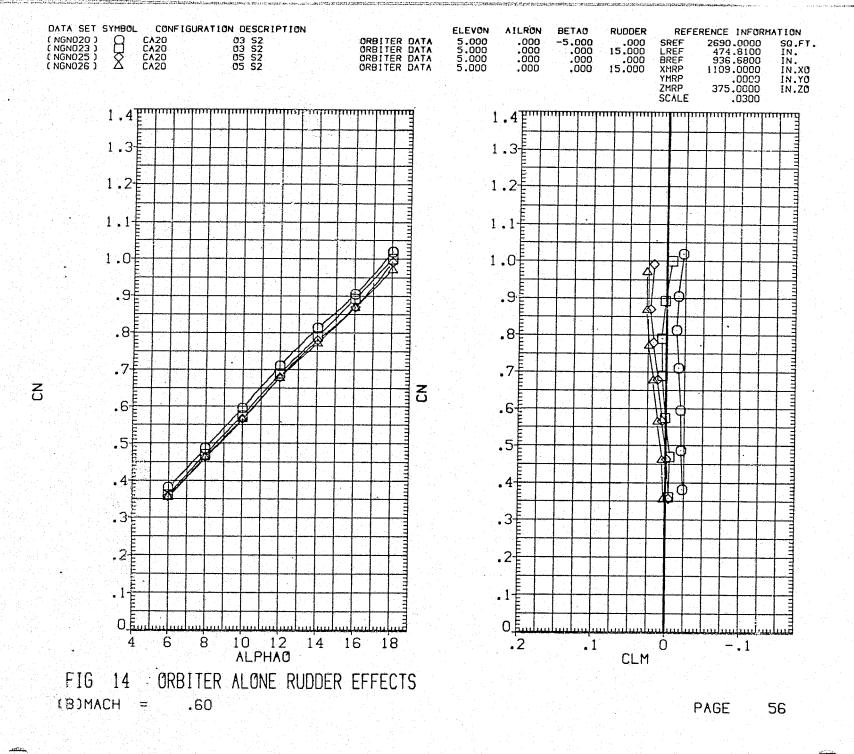


FIG 14 ORBITER ALONE RUDDER EFFECTS

(A)MACH = .30



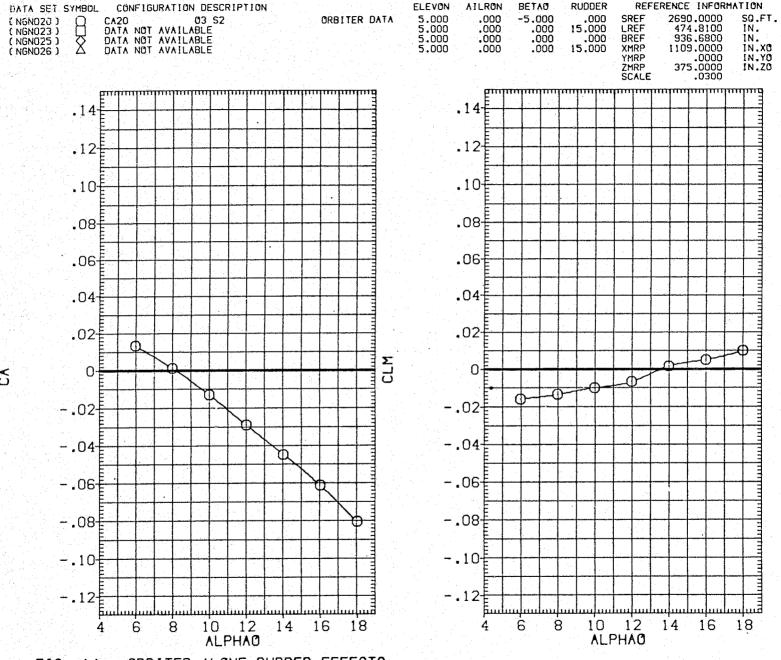
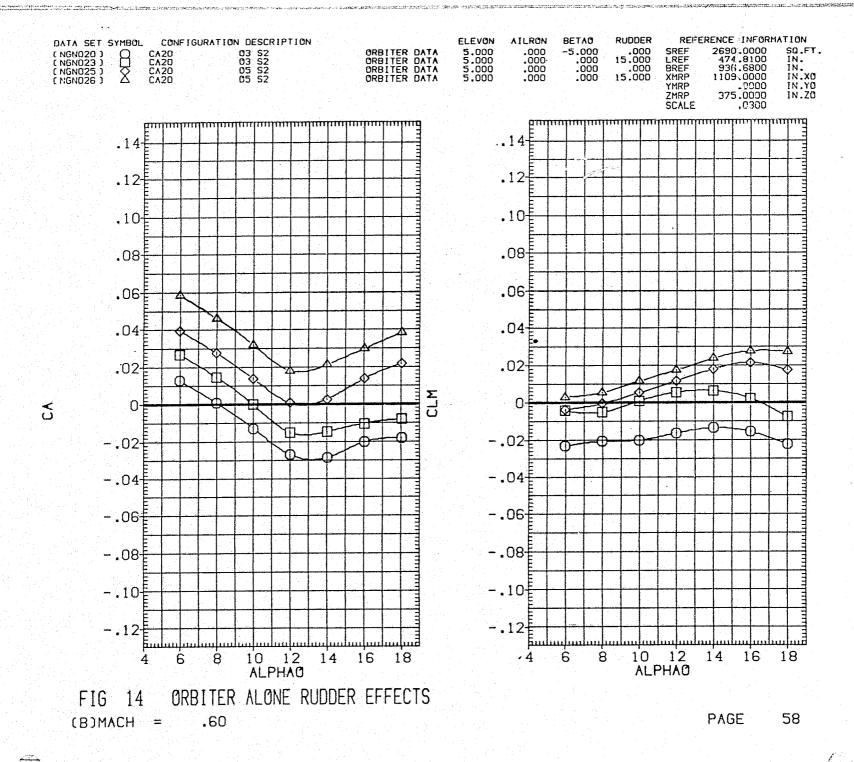


FIG 14 ORBITER ALONE RUDDER EFFECTS

(A)MACH = .30



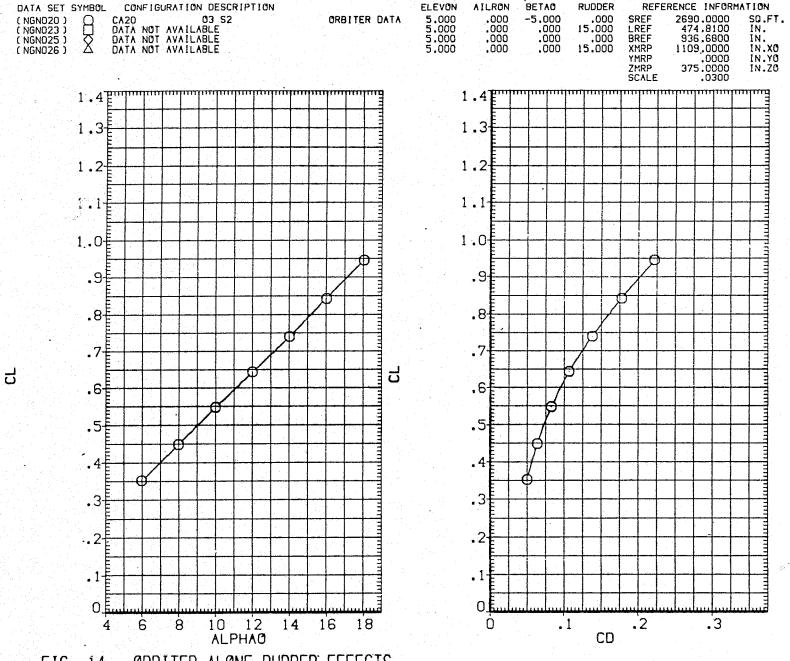
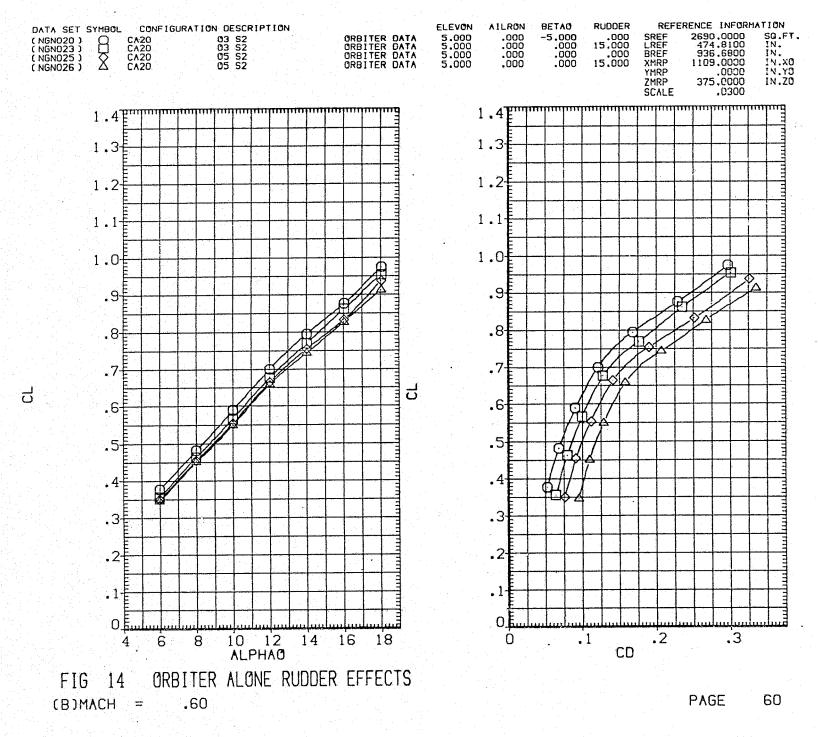


FIG 14 ORBITER ALONE RUDDER EFFECTS

(A)MACH = .30



A STATE OF THE STA

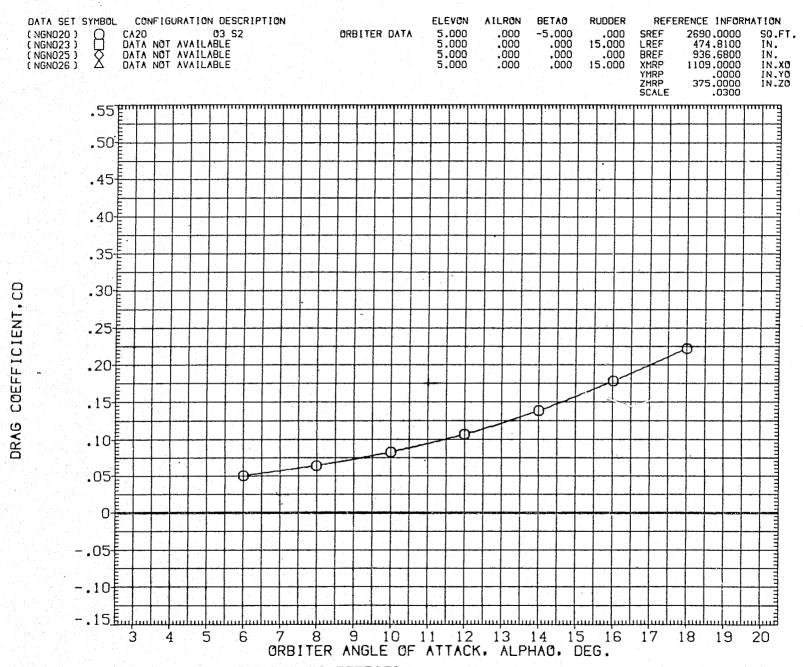


FIG 14 ORBITER ALONE RUDDER EFFECTS

(A)MACH = .30

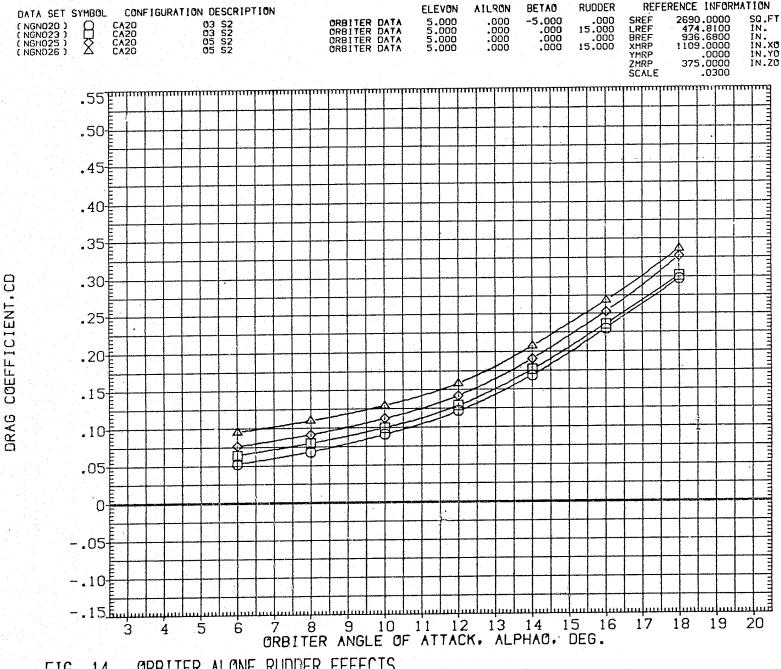


FIG 14 ORBITER ALONE RUDDER EFFECTS

(B)MACH = .60

PAGE 62

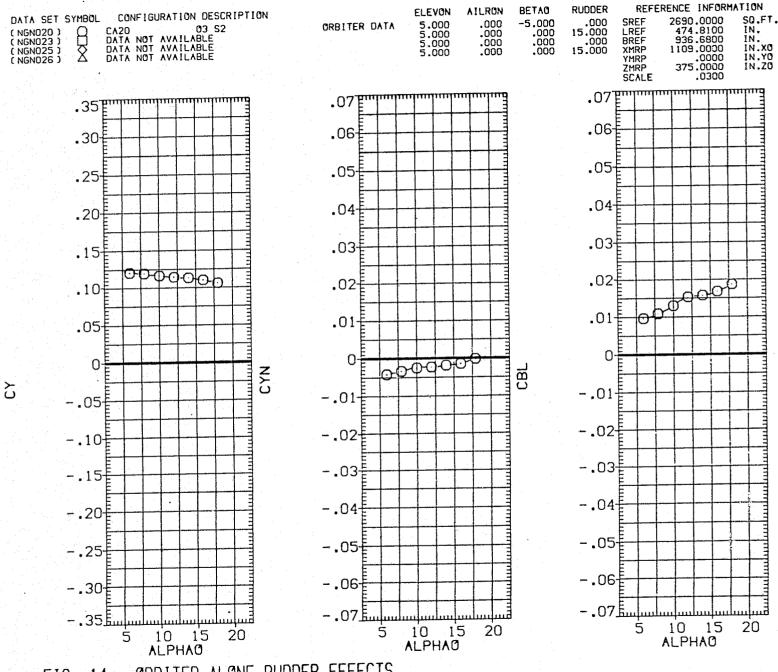
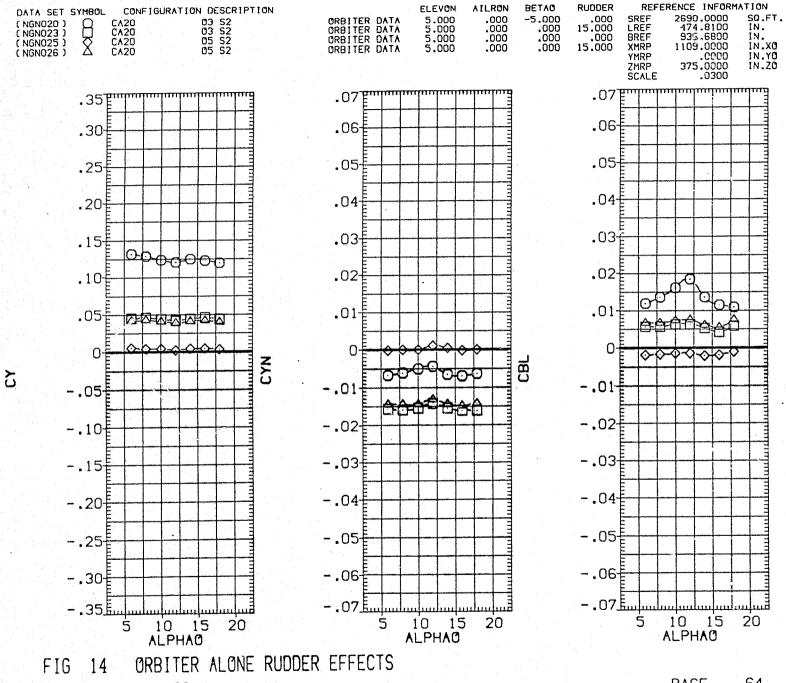


FIG 14 ORBITER ALONE RUDDER EFFECTS

(A)MACH = .30

PAGE



(B)MACH .60

PAGE 64

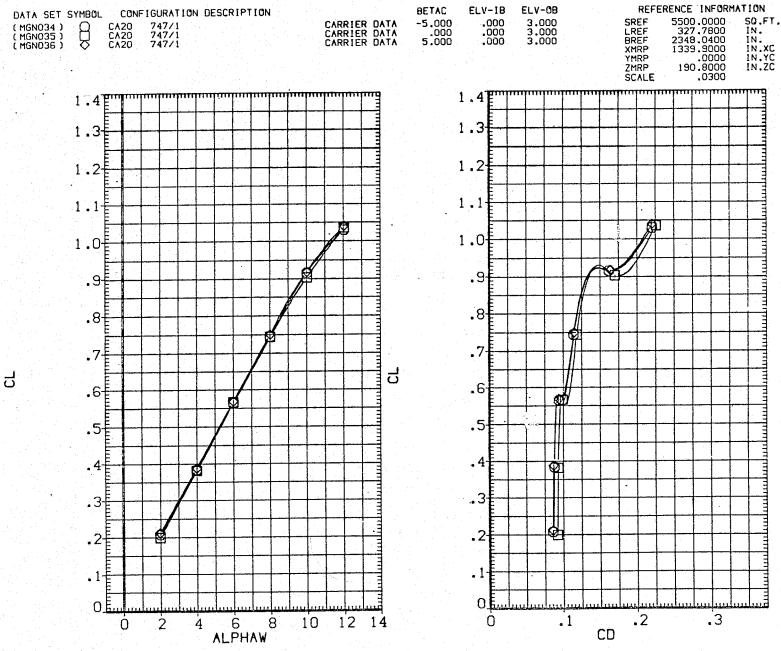


FIG 15 CARRIER ALONE BASIC AERODYNAMIC CHARACTERISTICS

(A)MACH = .60

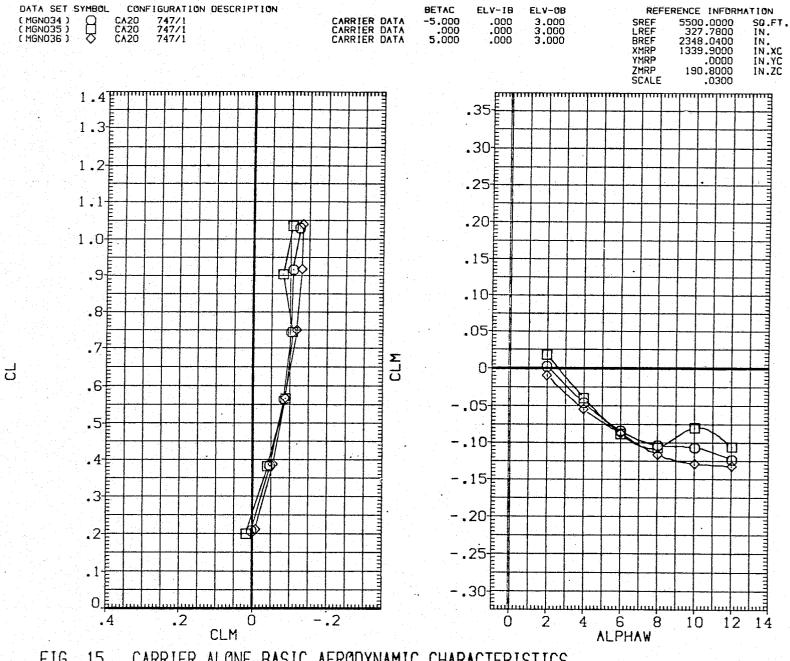
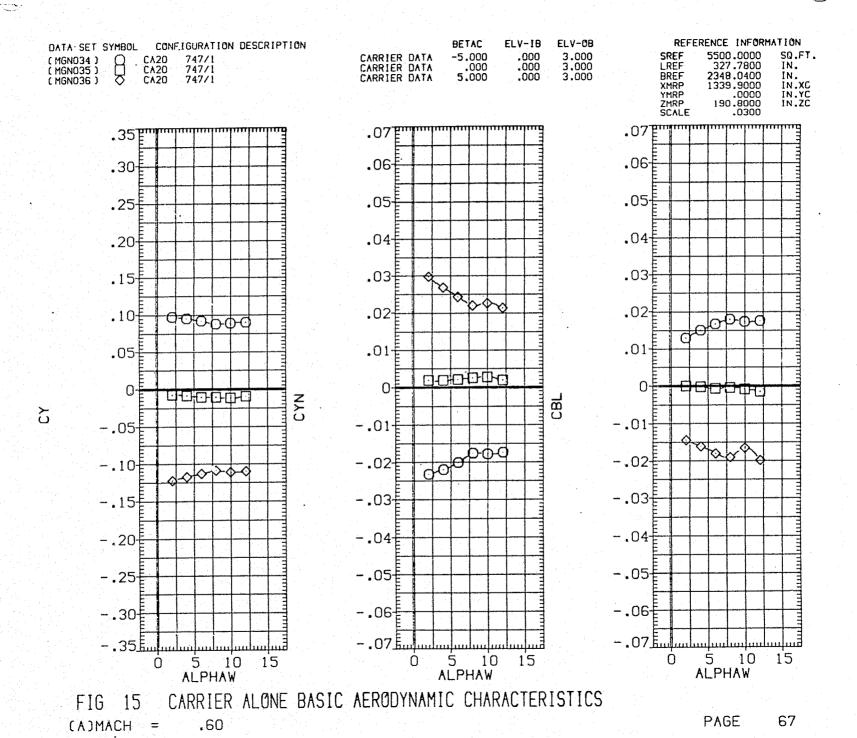
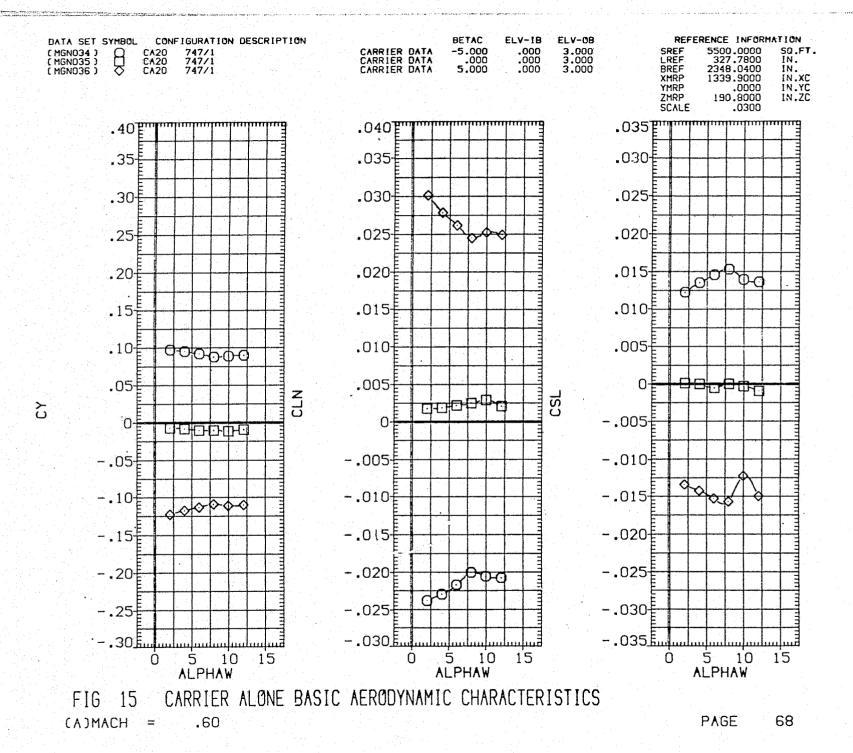
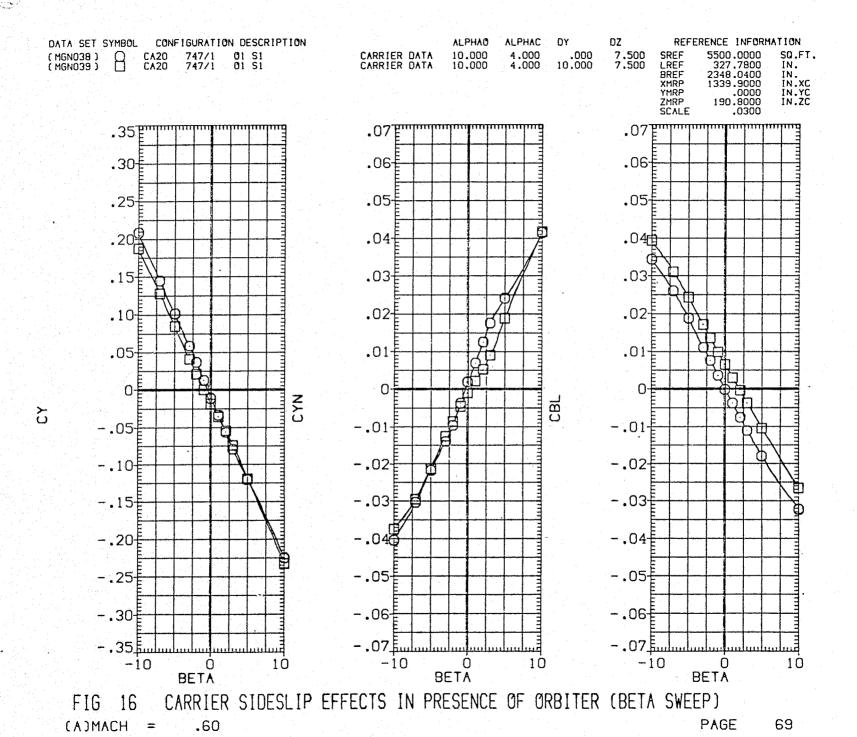


FIG 15 CARRIER ALONE BASIC AERODYNAMIC CHARACTERISTICS
(A)MACH = .60

PAGE 66







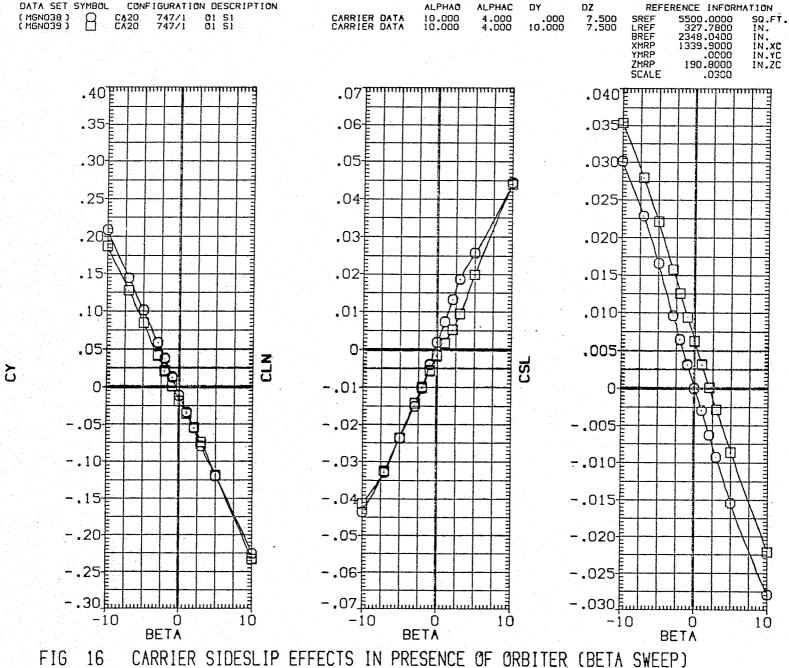
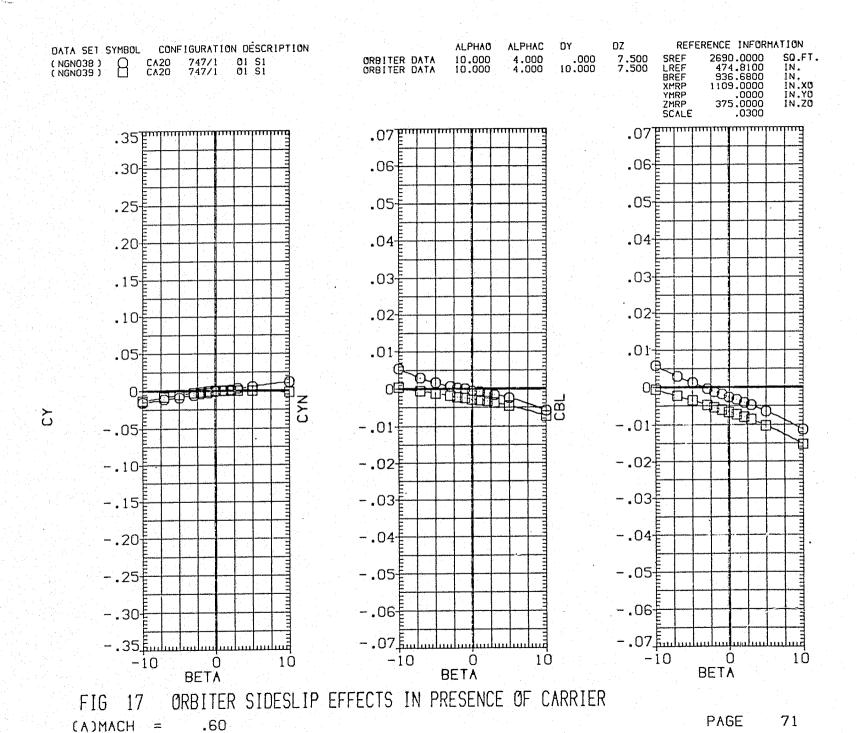


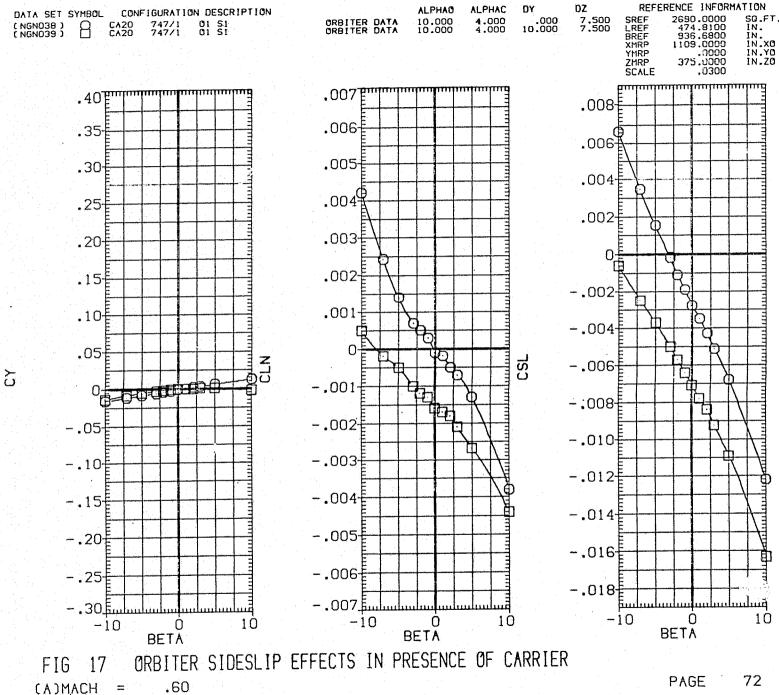
FIG 16 CARRIER SIDESLIP EFFECTS IN PRESENCE OF ORBITER (BETA SWEEP)

[A]MACH = .60

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- Comment





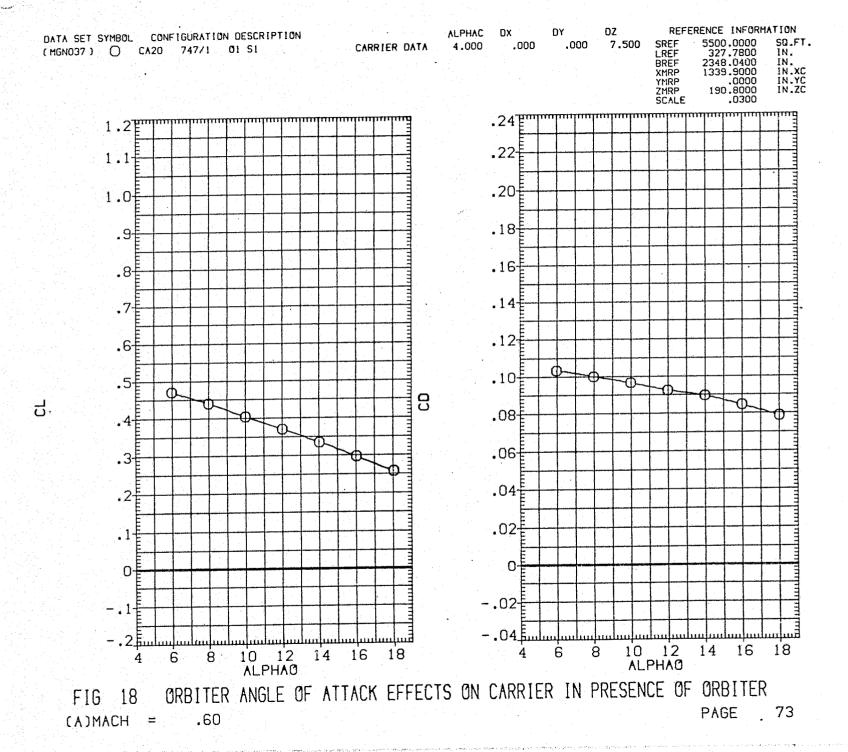
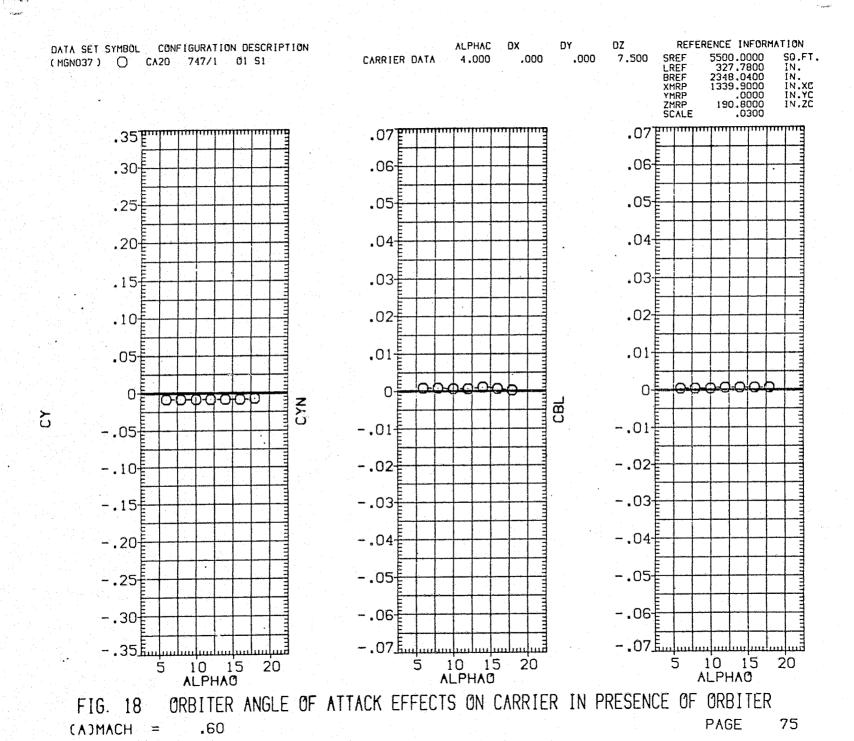
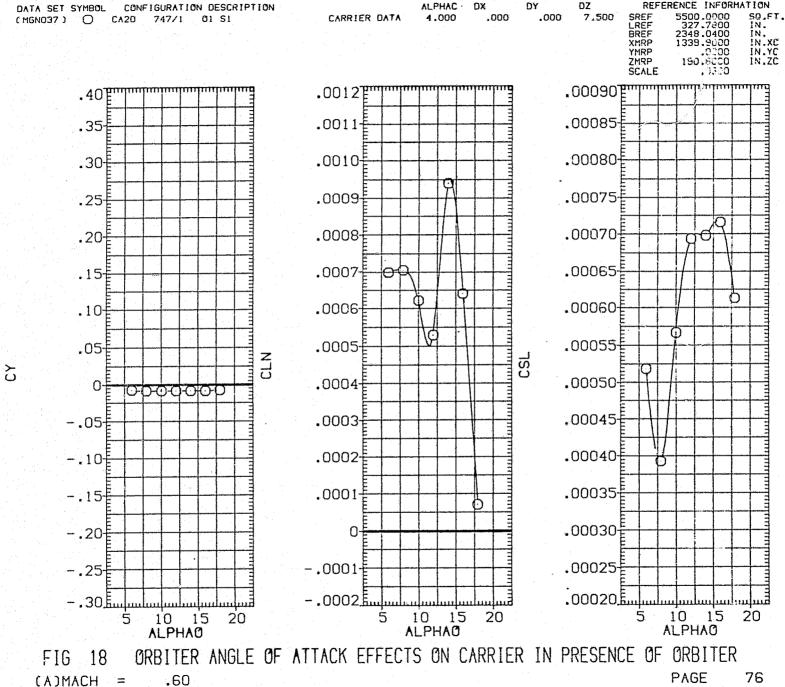


FIG 18 ORBITER ANGLE OF ATTACK EFFECTS ON CARRIER IN PRESENCE OF ORBITER

(A)MACH = .60

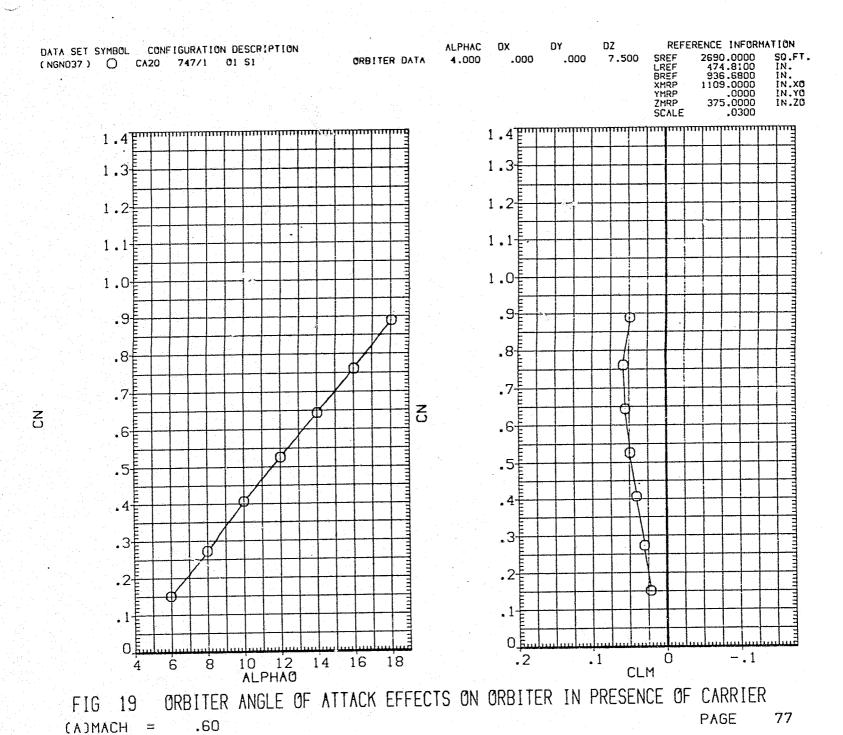
PAGE 74





.60

PAGE



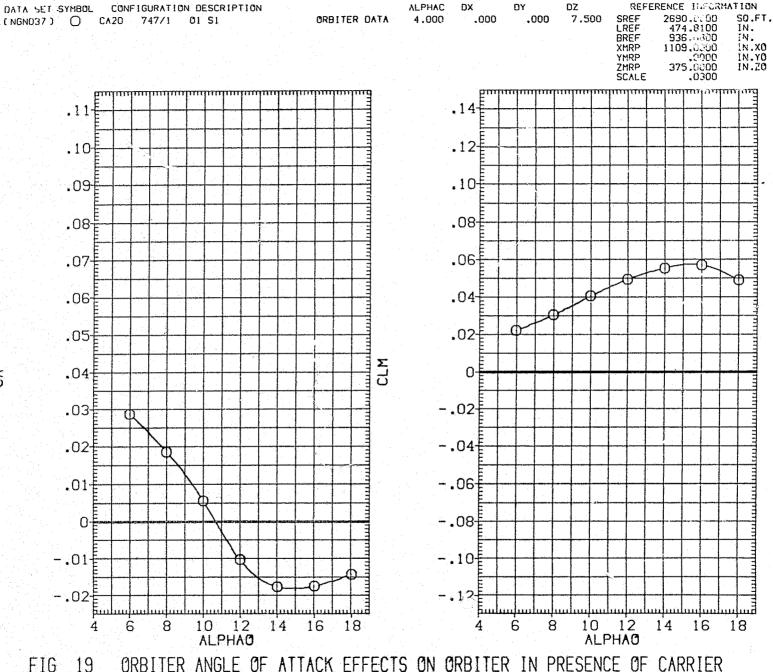
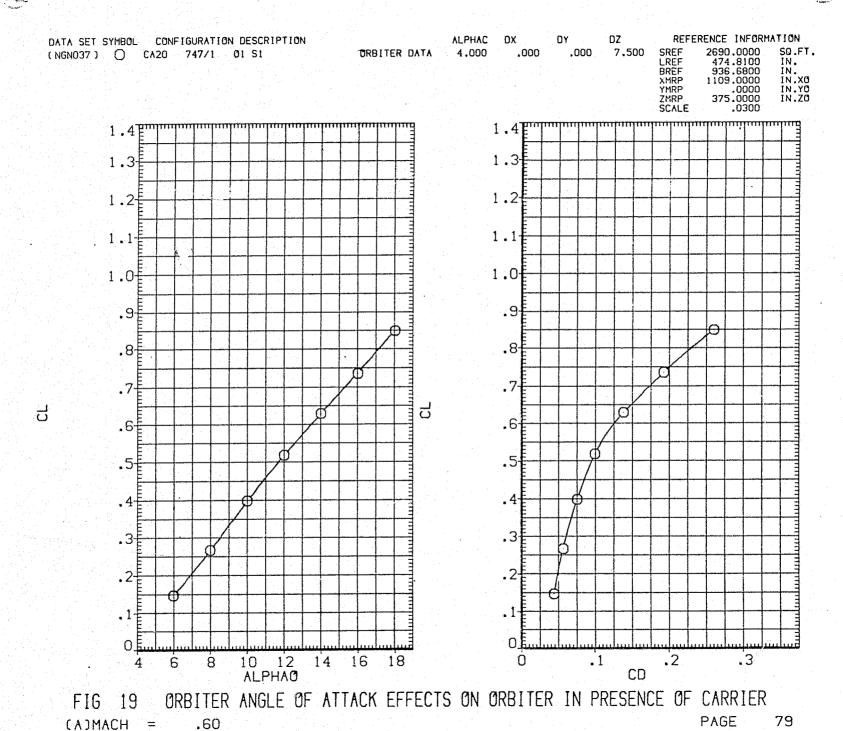
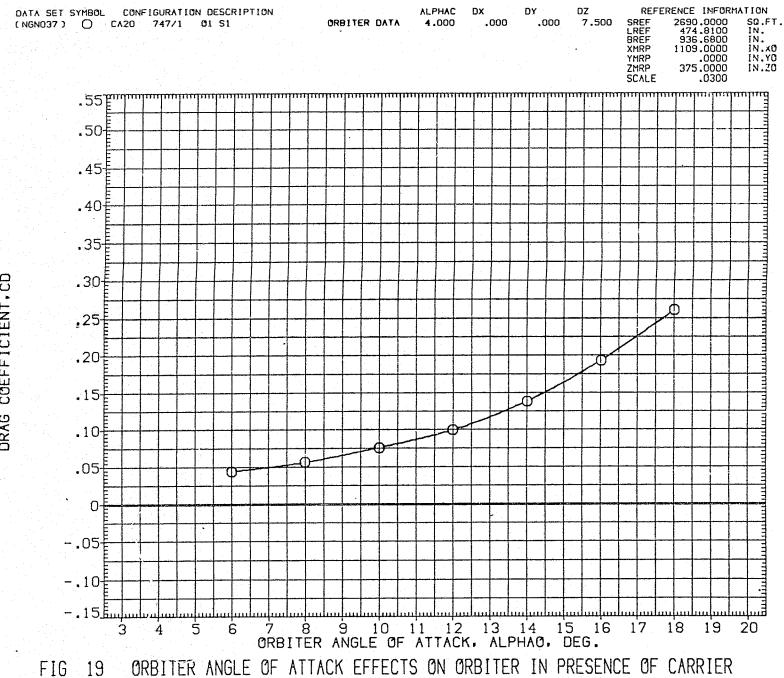


FIG 19 ORBITER ANGLE OF ATTACK EFFECTS ON ORBITER IN PRESENCE OF CARRIER

(A)MACH = .60

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FIG 19 ORBITER ANGLE OF ATTACK EFFECTS ON ORBITER IN PRESENCE OF CARRIER

(A)MACH = .60

PAGE 80

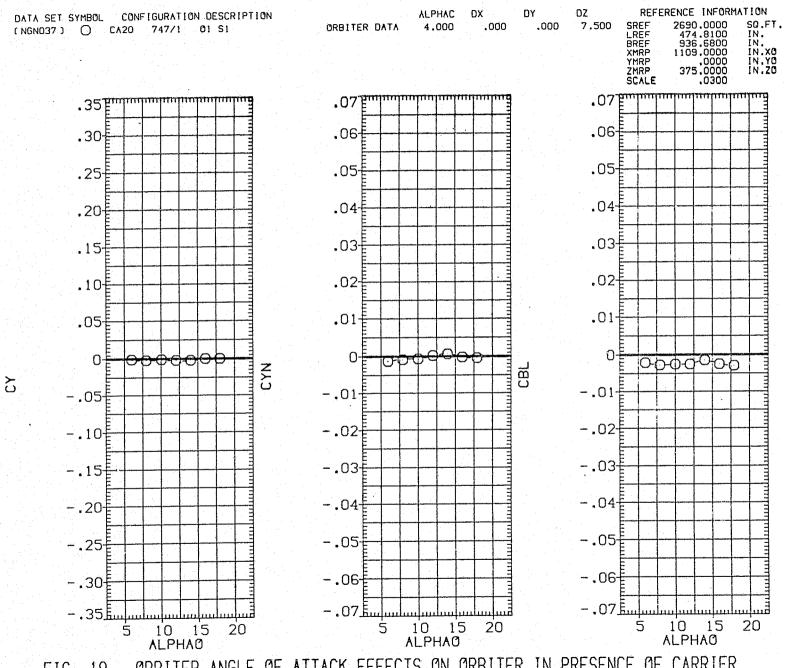


FIG 19 ORBITER ANGLE OF ATTACK EFFECTS ON ORBITER IN PRESENCE OF CARRIER

(A)MACH = .60

PAGE 81

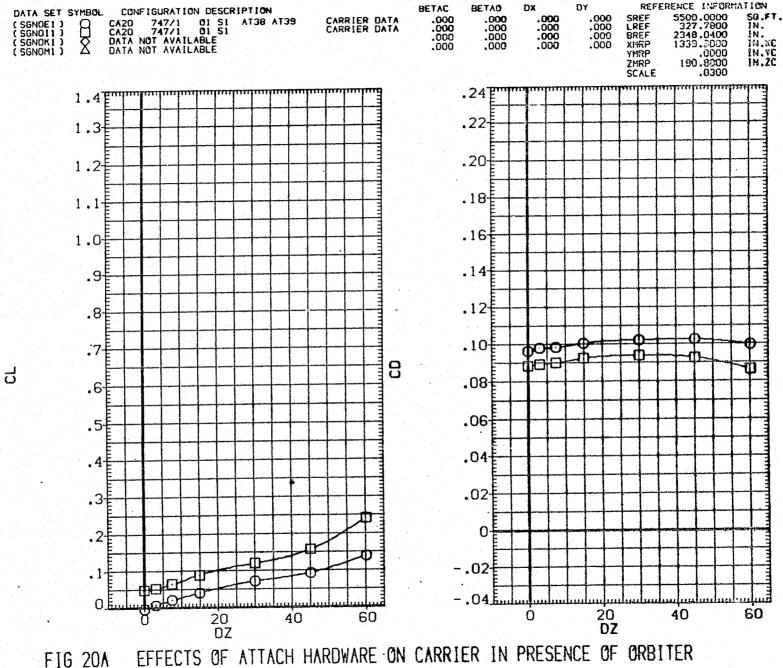


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF URBITER

(A) ALPHAO = 8.00

PAGE

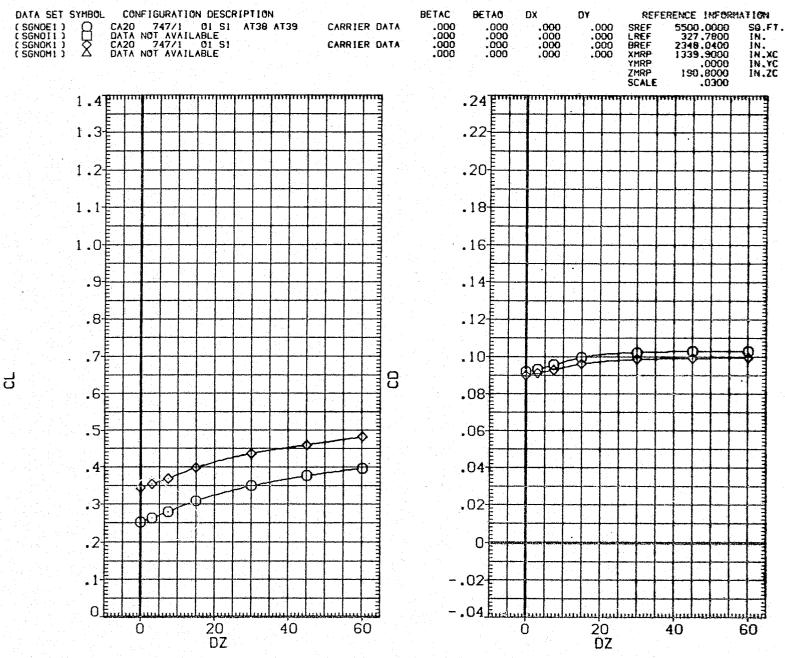


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(B) ALPHAO = 12.00

PAGE

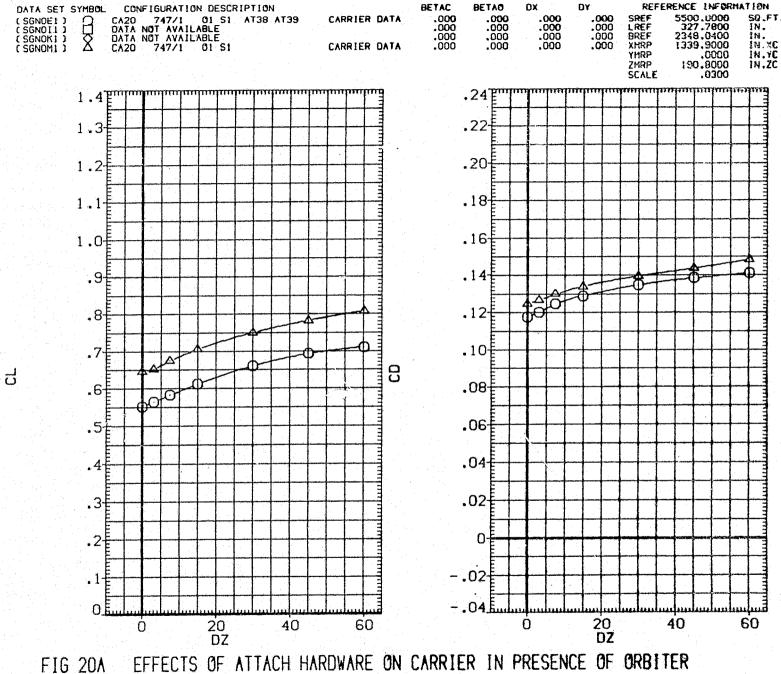
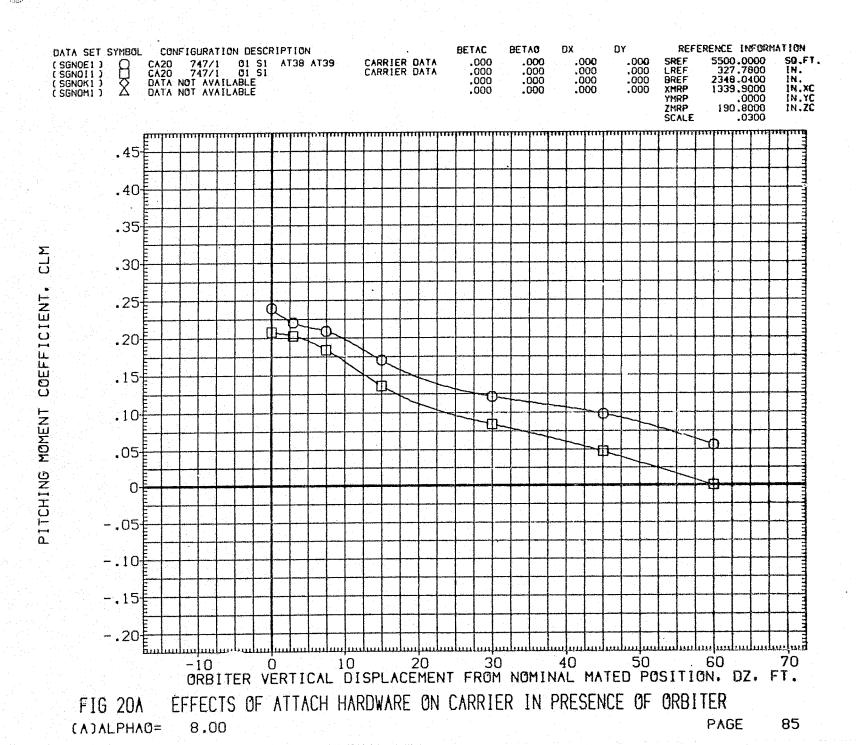
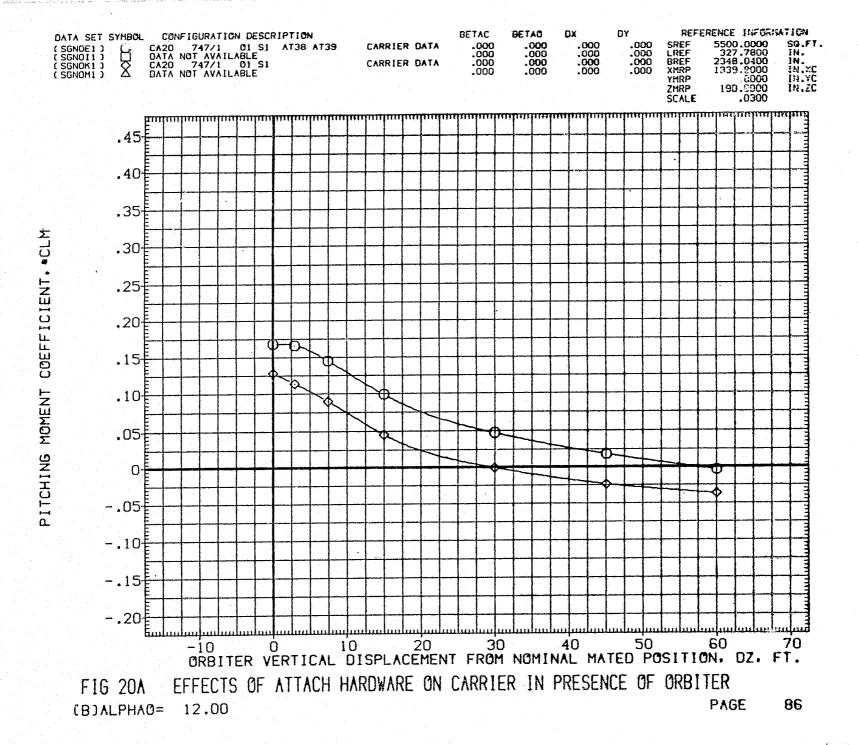
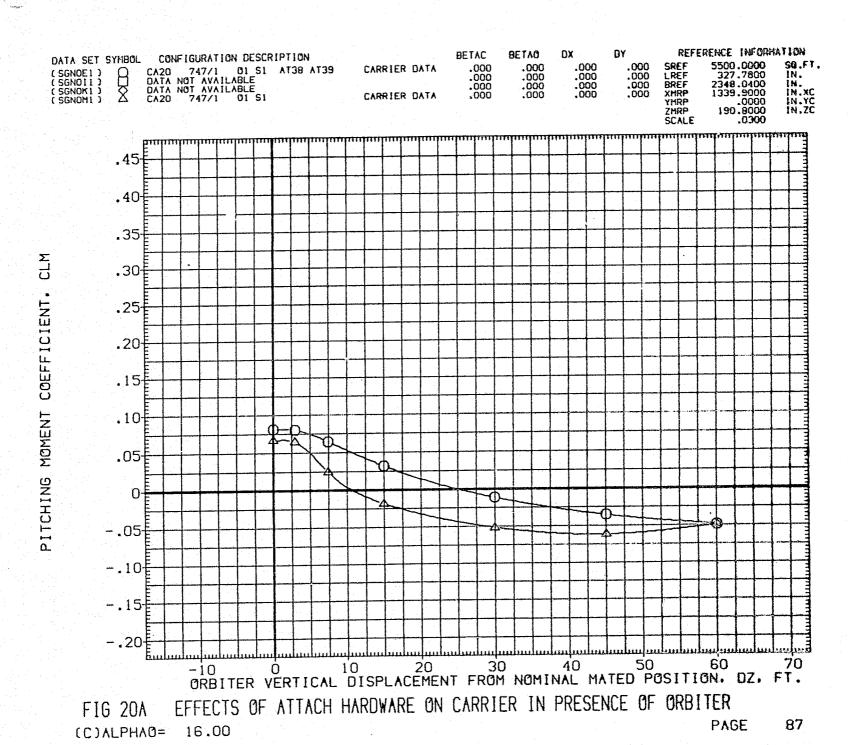


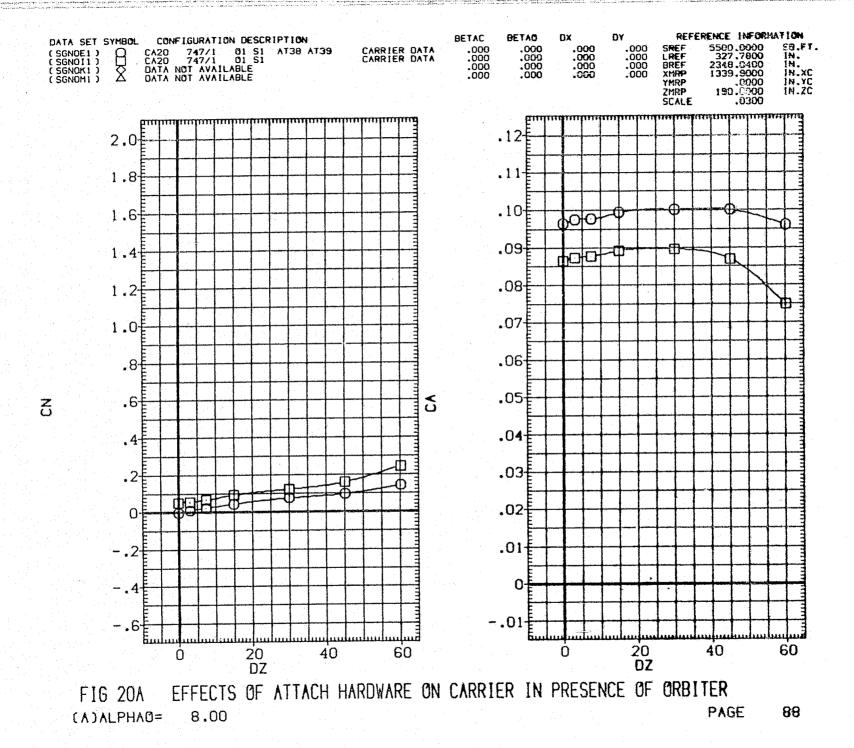
FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(C)ALPHAO= 16.00 PAGE









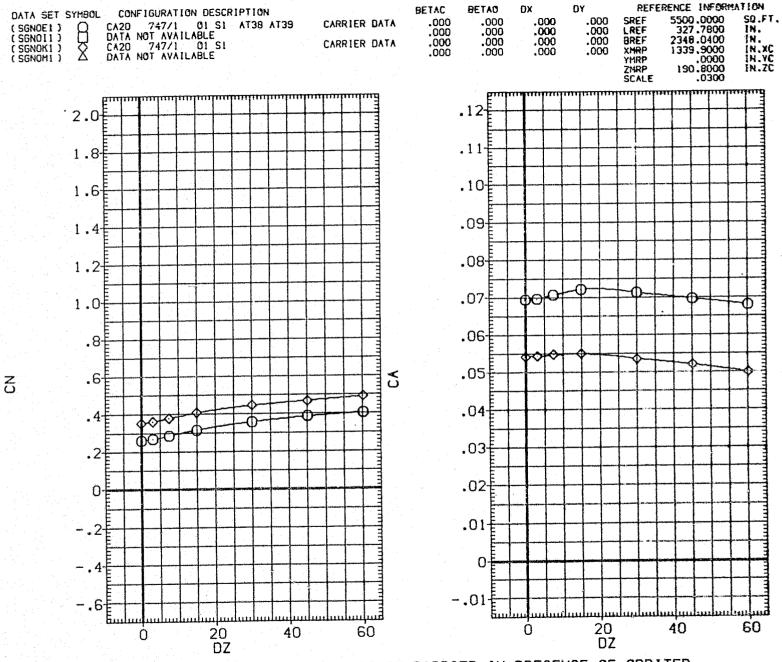


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(B) ALPHAO= 12.00 PAGE

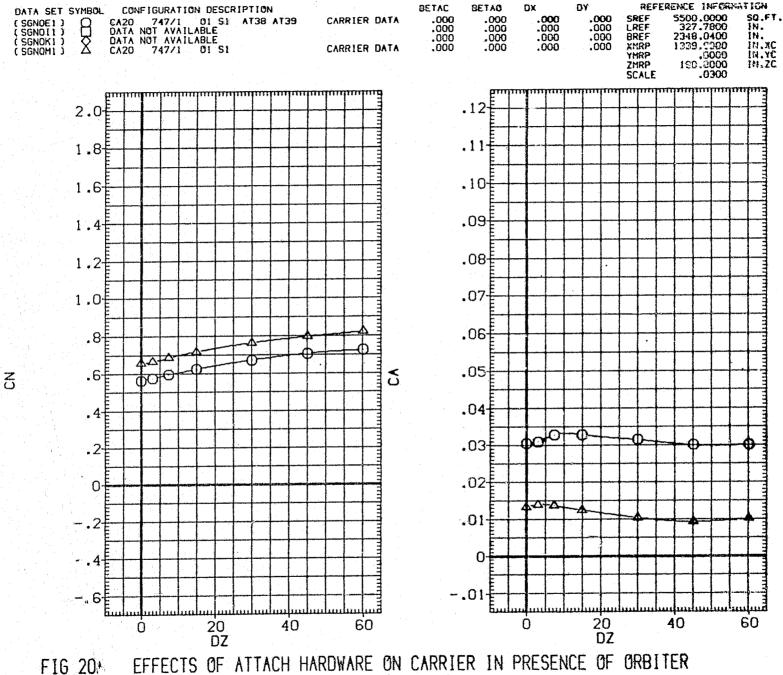


FIG 20* EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(C)ALPHAD= 16.00

PAGE 90

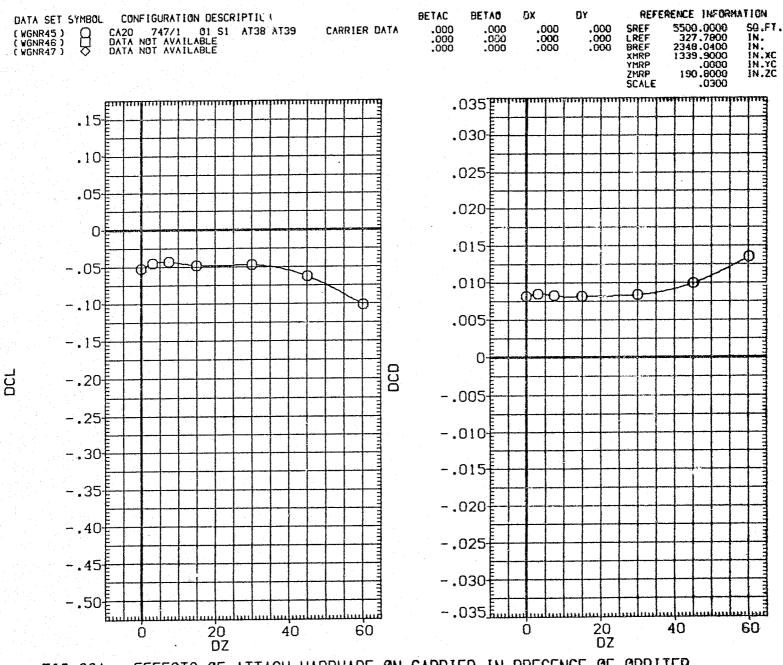


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(A)ALPHAO= 8.00

PAGE

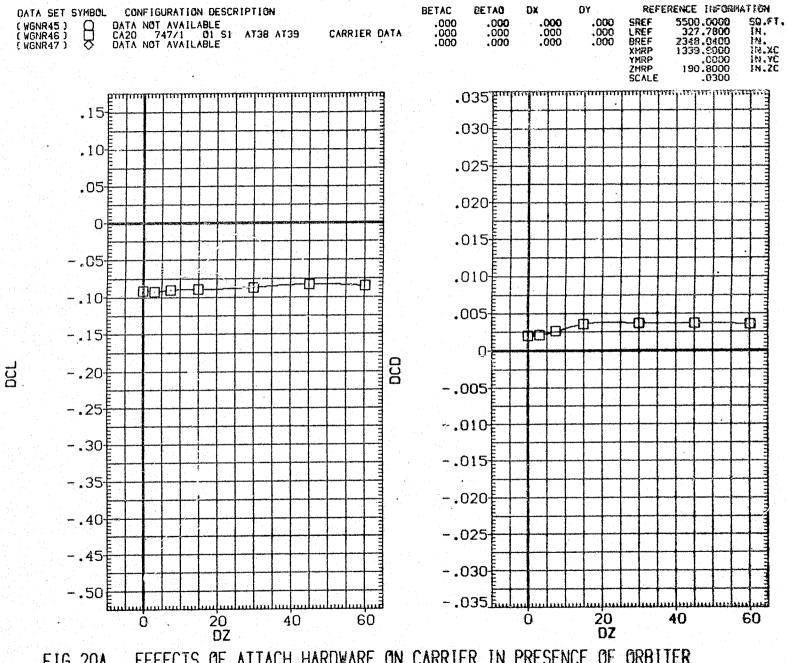


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(B)ALPHAO= 12.00

PAGE

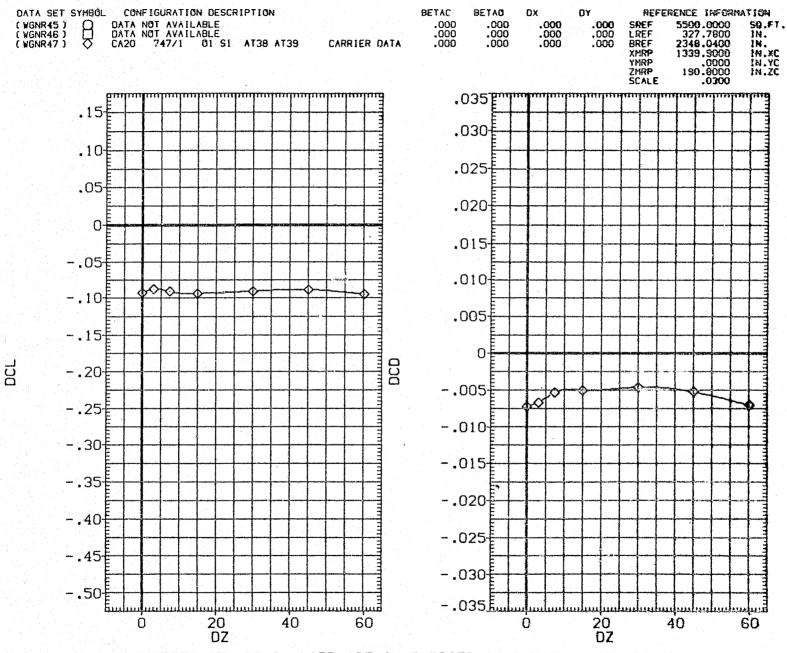
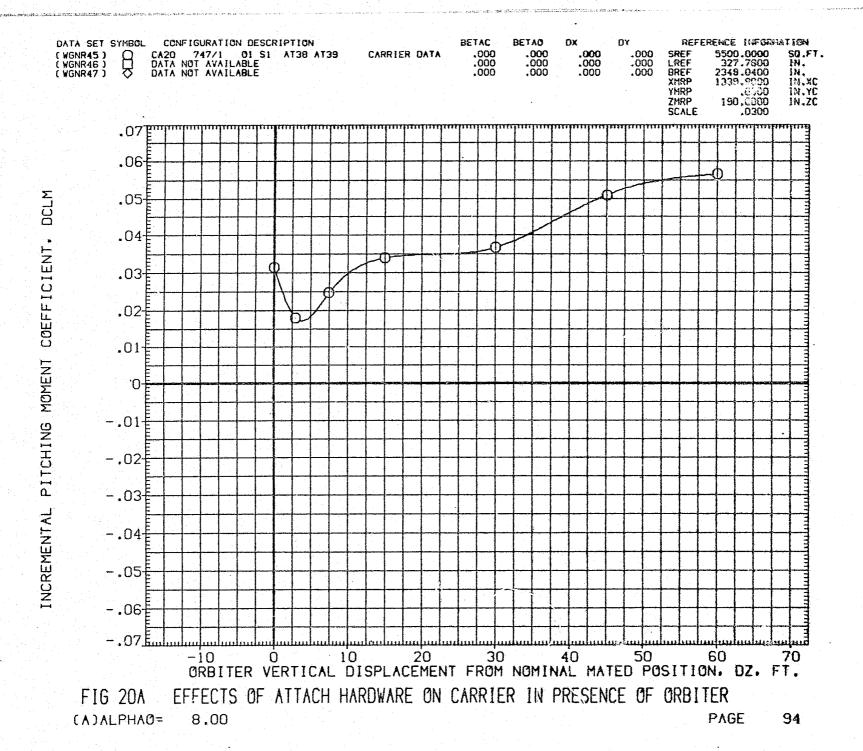
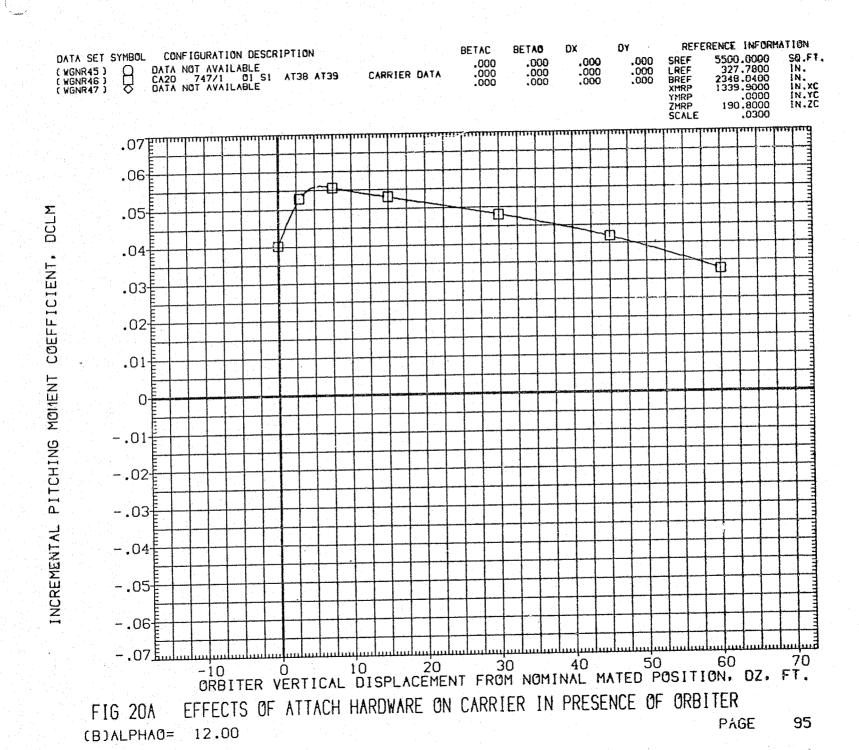


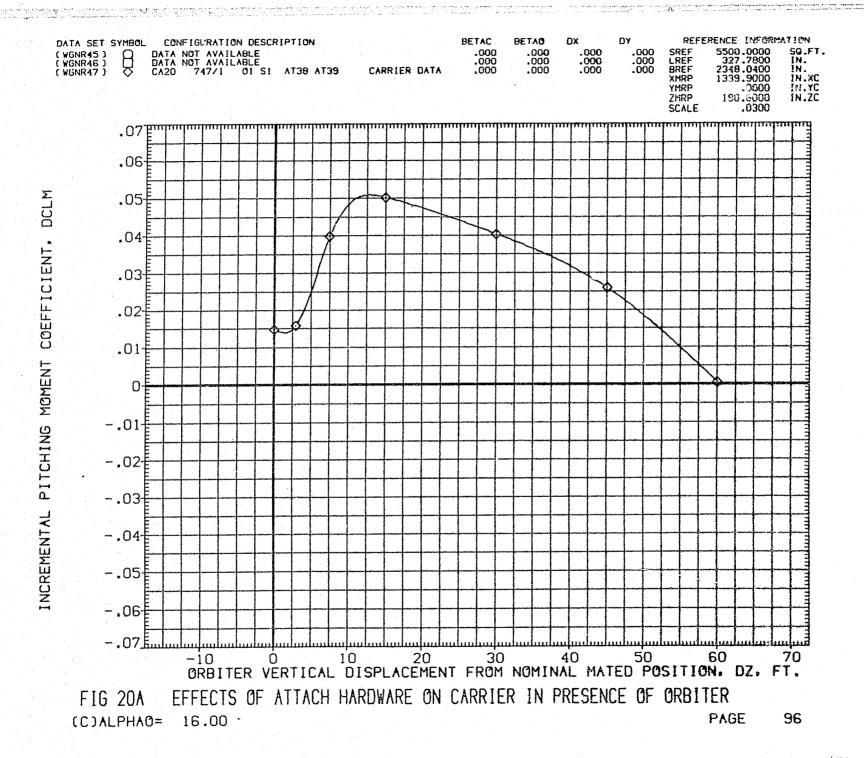
FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(C)ALPHAO = 16.00

PAGE







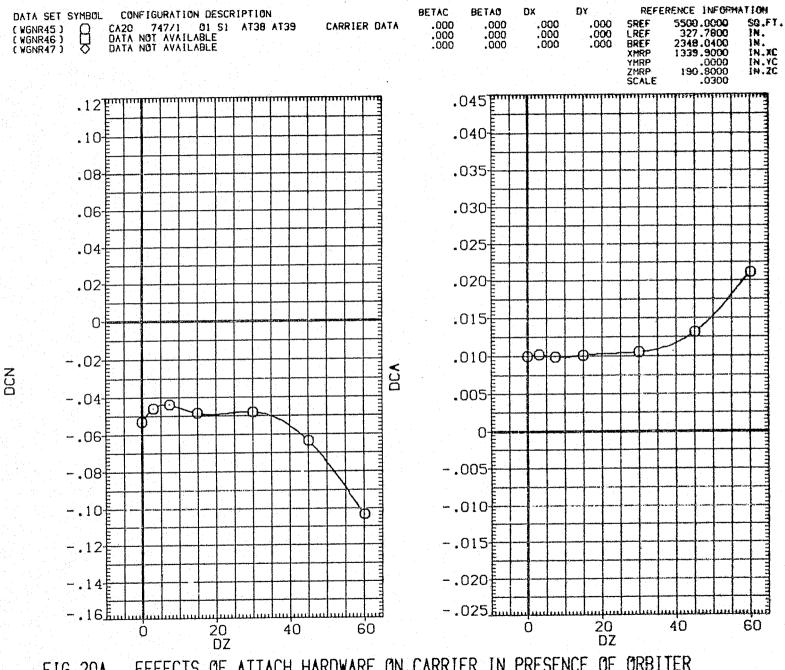


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(A)ALPHAO= 8.00

PAGE

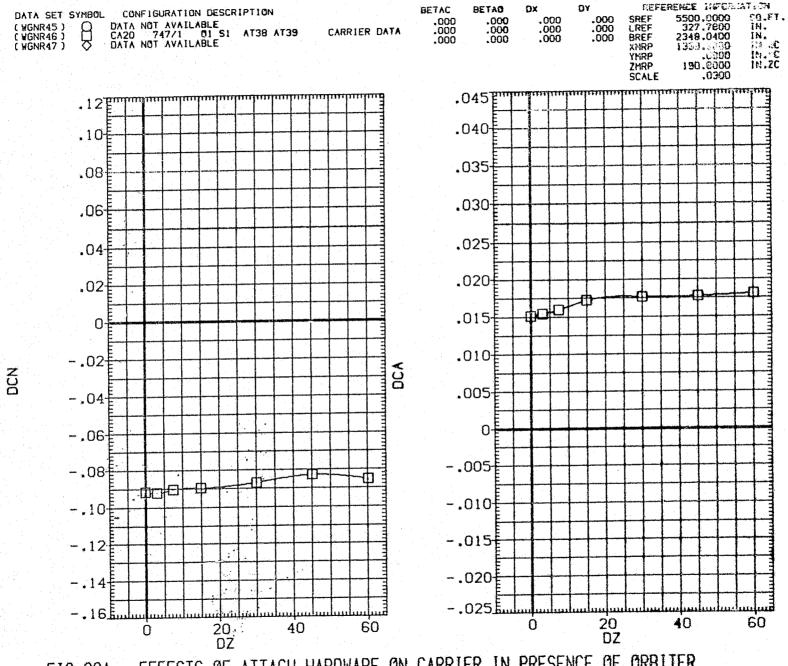


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(B) ALPHAO = 12.00

PAGE

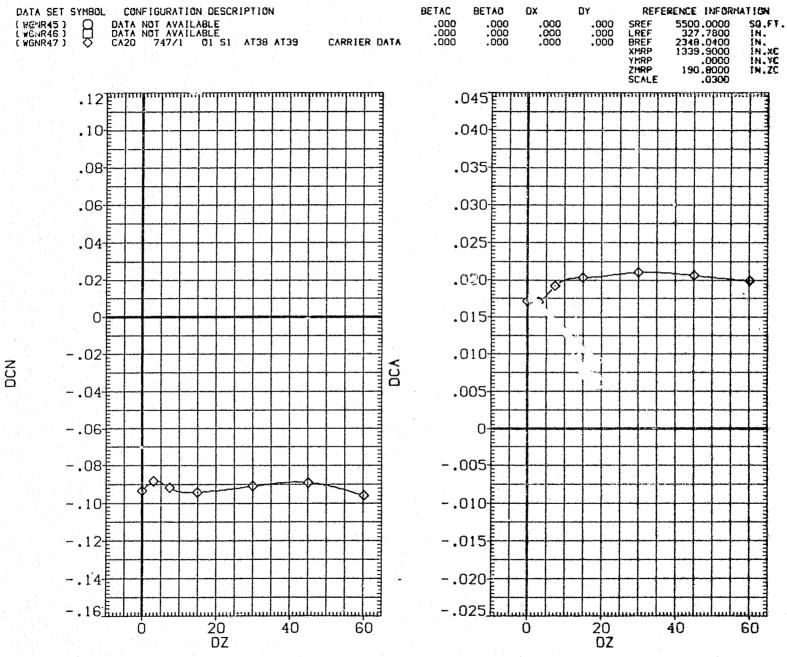


FIG 20A EFFECTS OF ATTACH HARDWARE ON CARRIER IN PRESENCE OF ORBITER

(C)ALPHAO= 16.00 PAGE

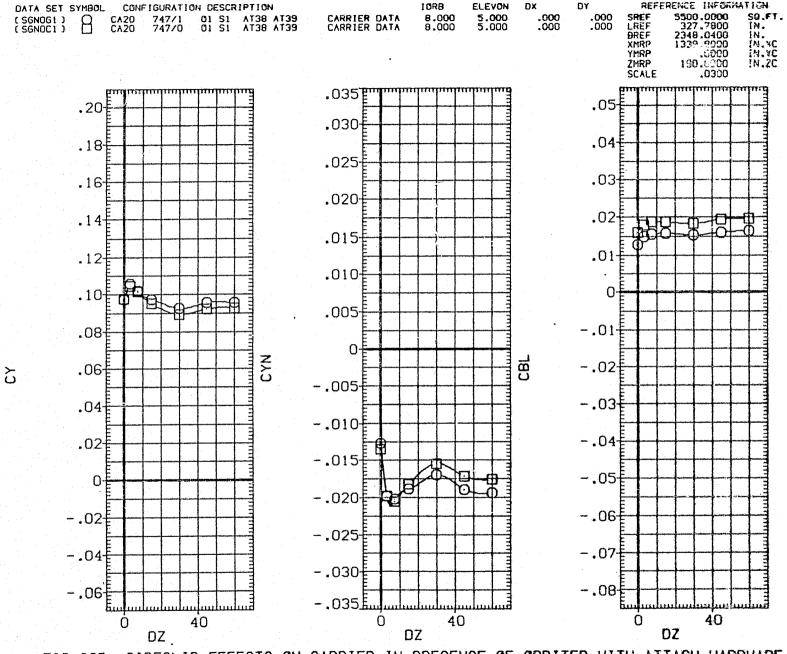


FIG 20B SIDESLIP EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)BETA = -5.00

PAGE 100

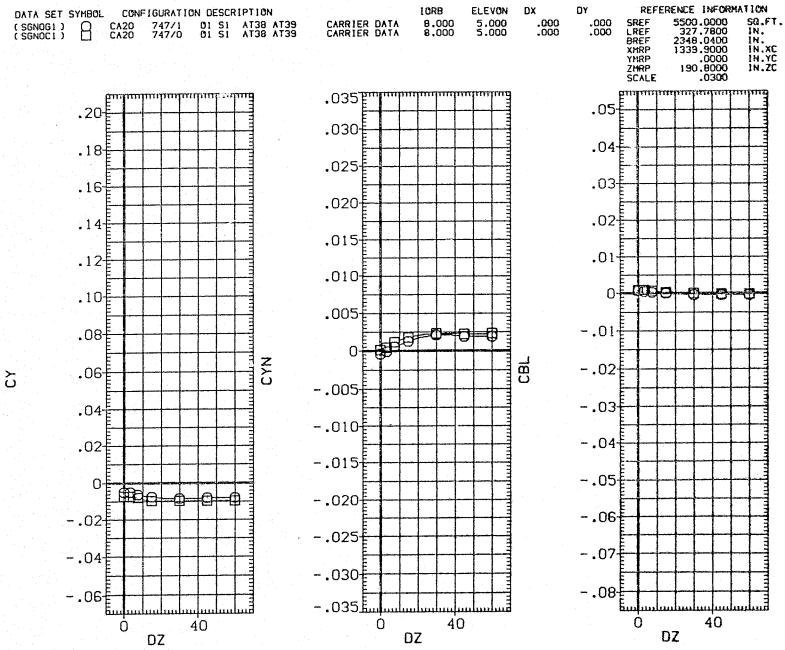


FIG 20B SIDESLIP EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)BETA = .00

PAGE 101

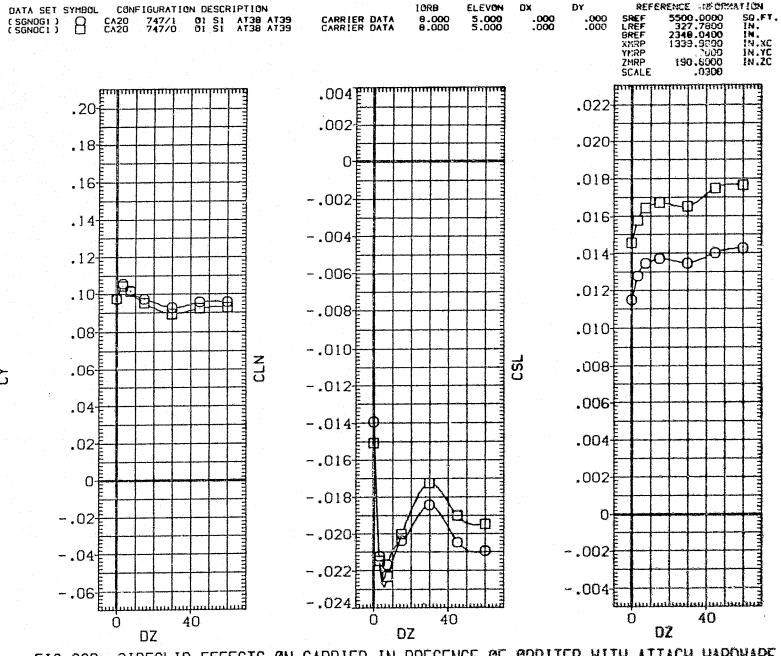
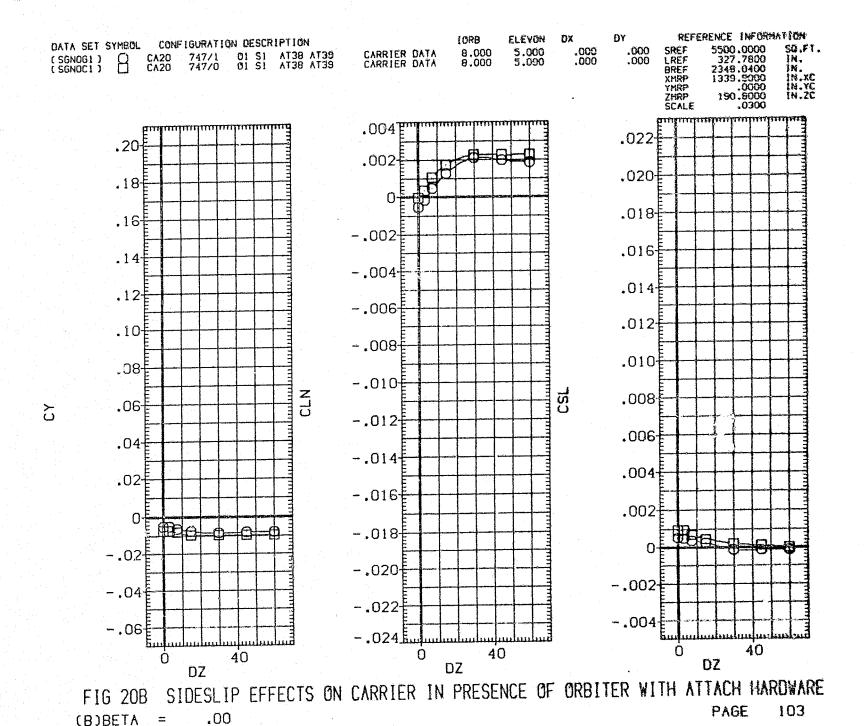


FIG 20B SIDESLIP EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)BETA = -5.00

PAGE 102

£



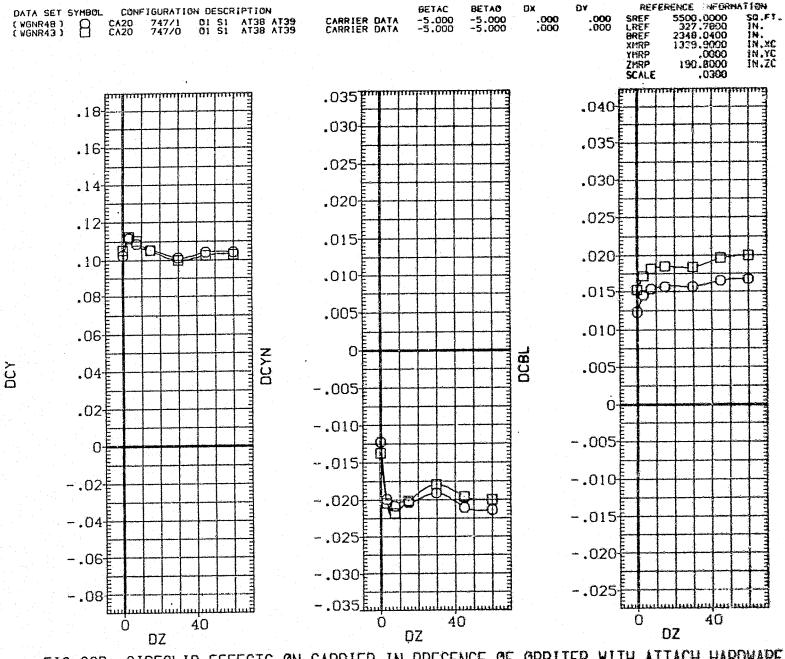


FIG 20B SIDESLIP EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A) ALPHAO = 12.00

PAGE 104

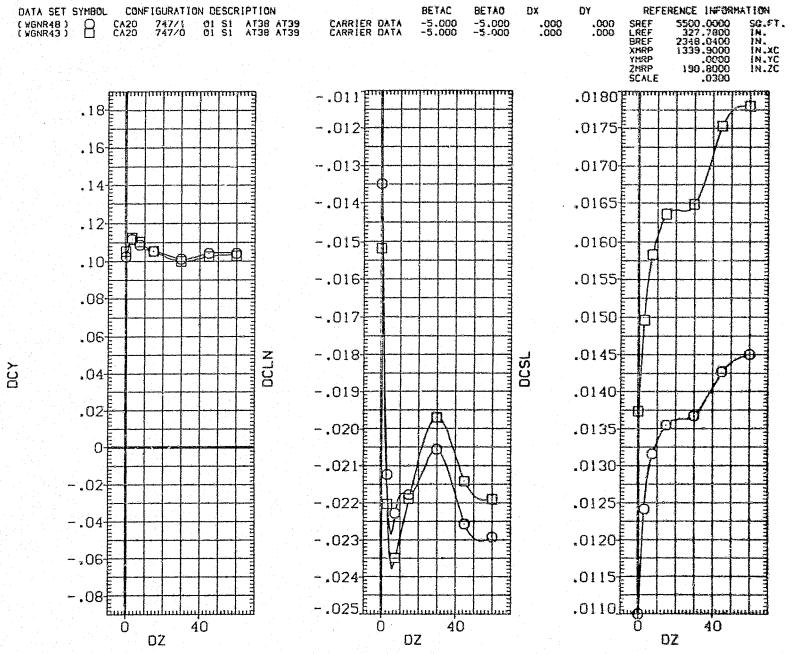


FIG 20B SIDESLIP EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)ALPHAO= 12.00 PAGE 105

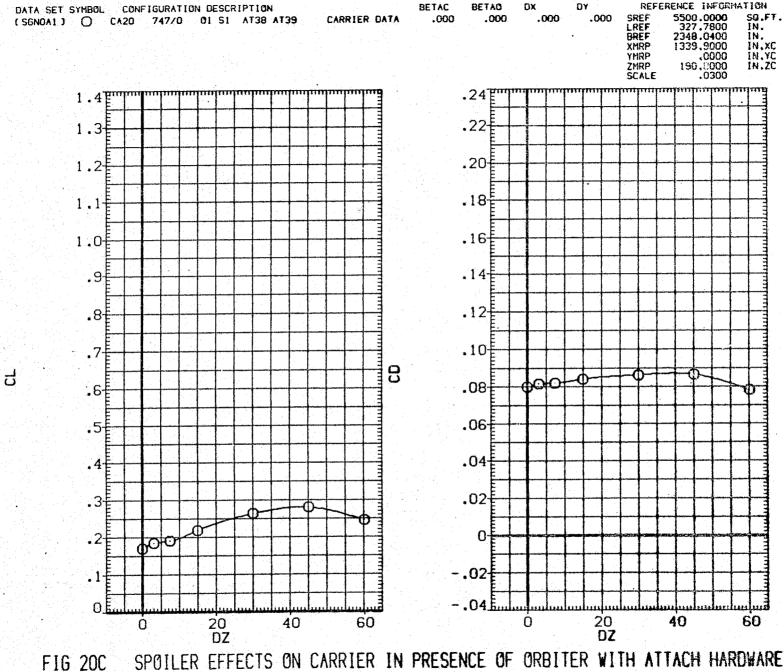
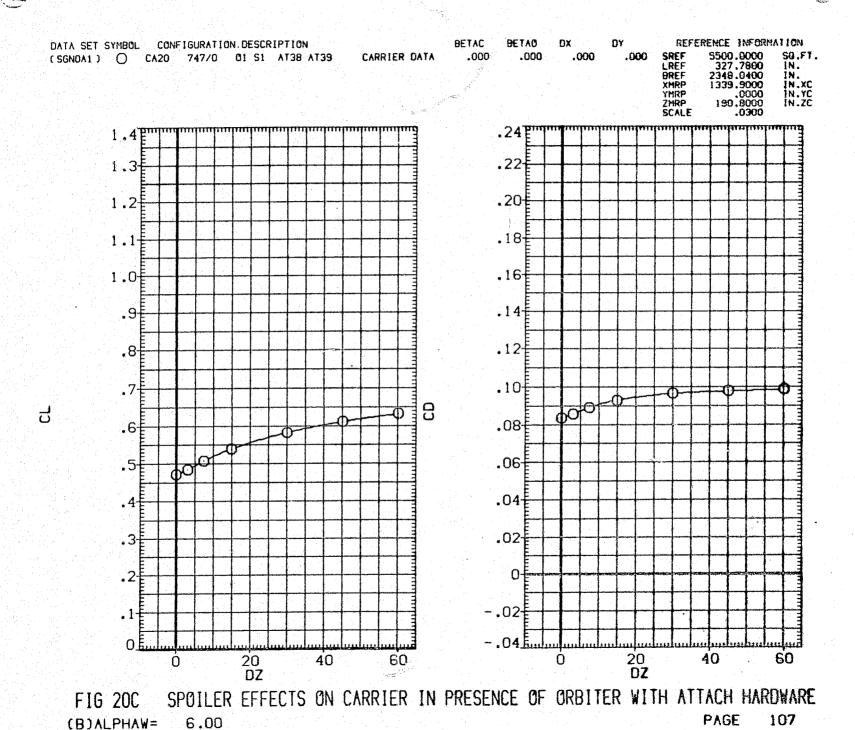


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWAR

-



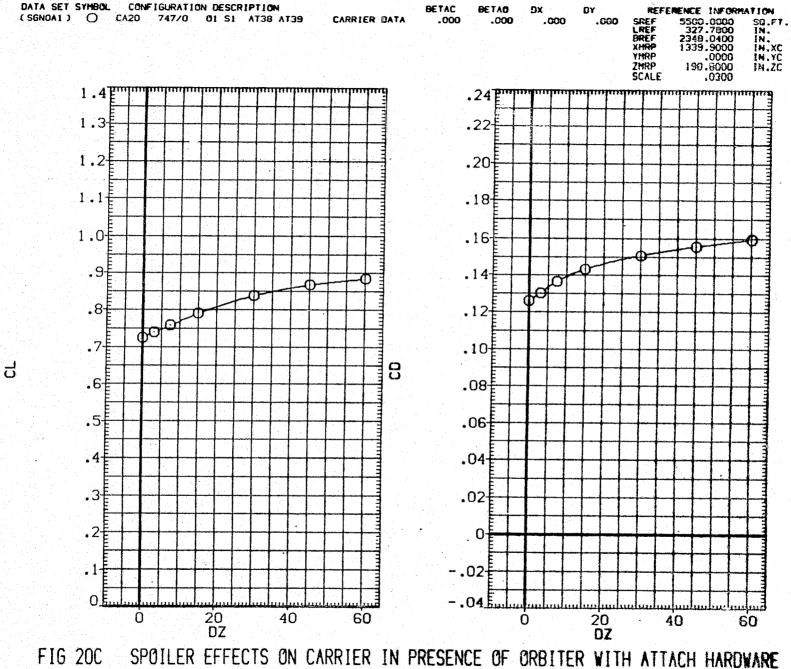
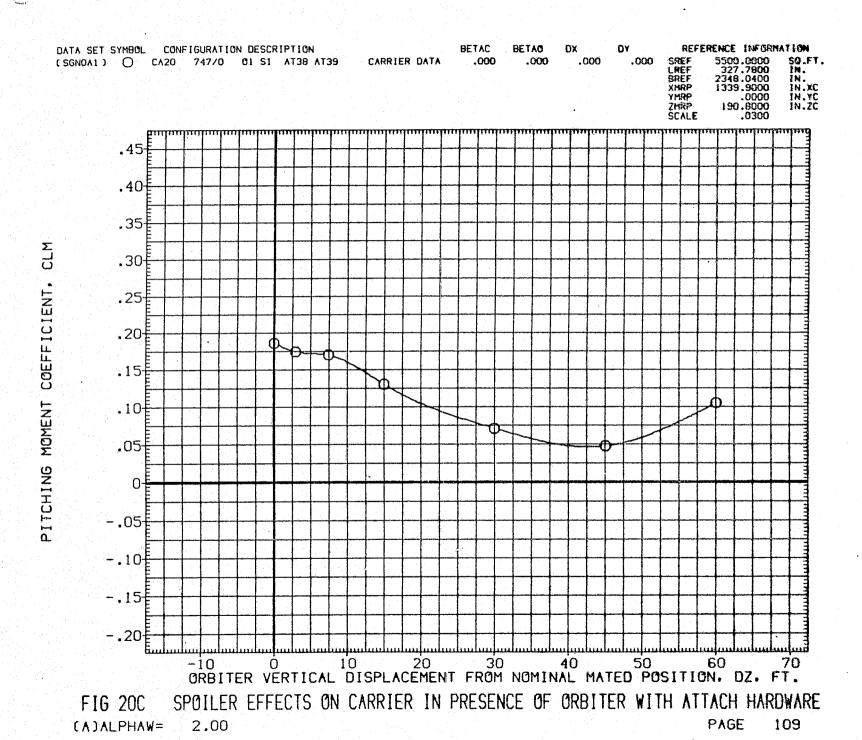
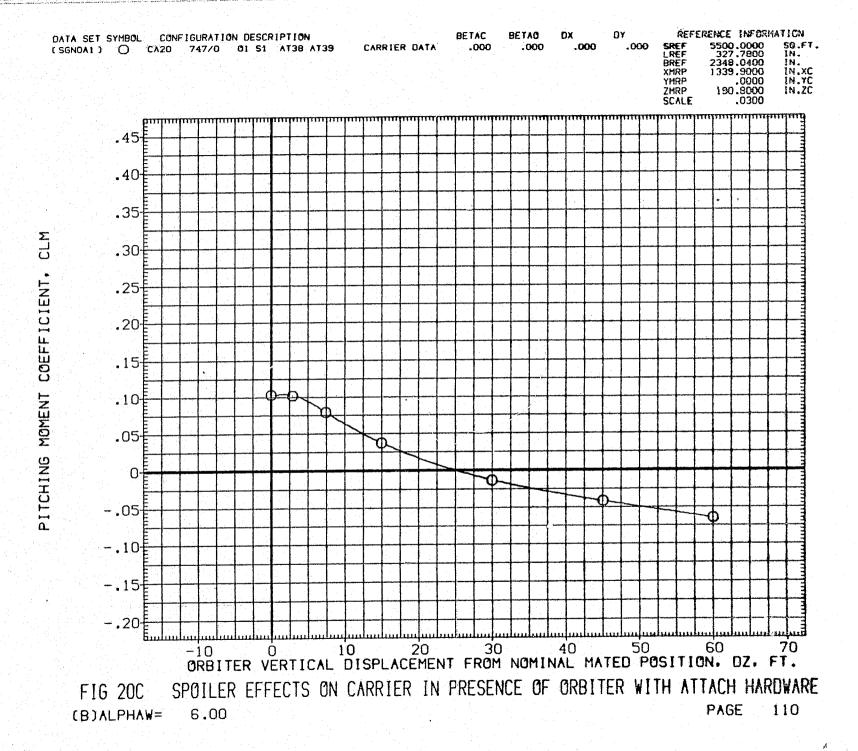
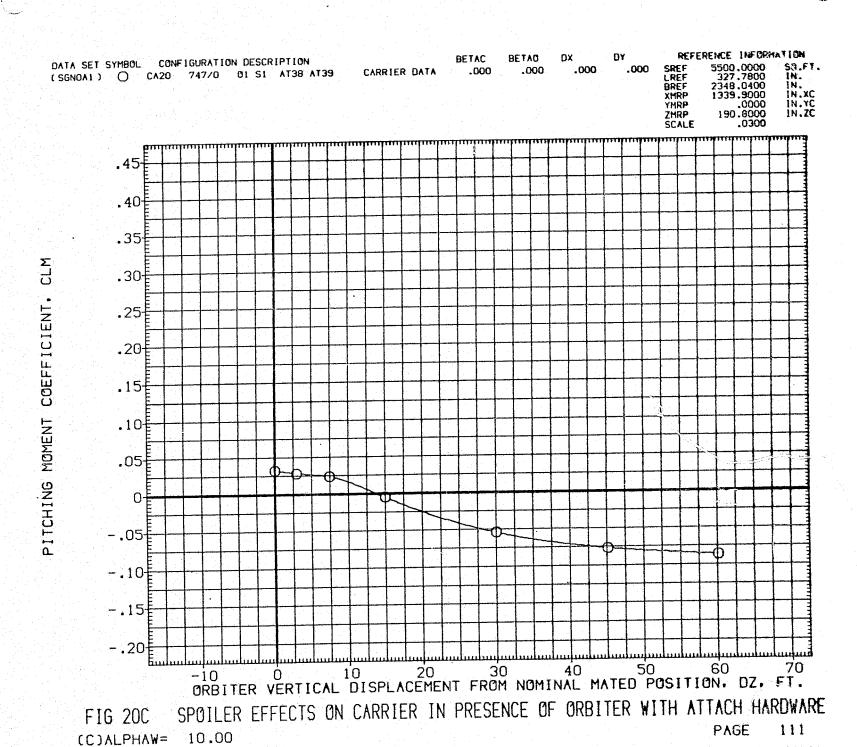


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARD (C)ALPHAW= 10.00







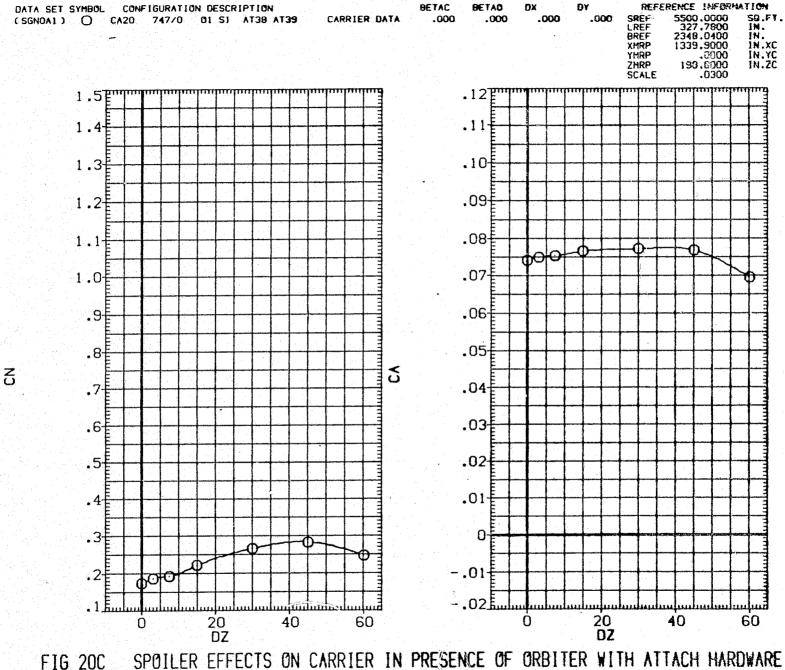


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARD (A) ALPHAW= 2.00 PAGE 112

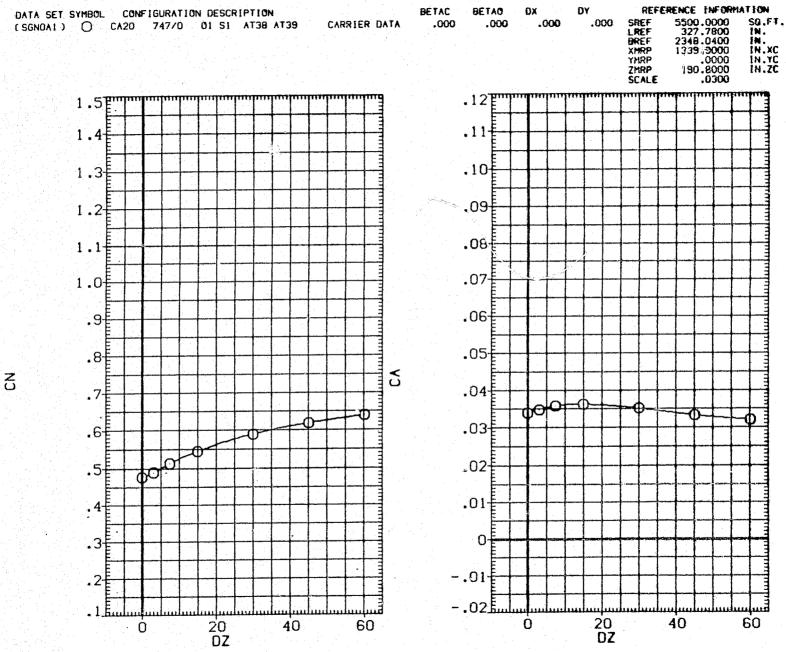


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)ALPHAW= 6.00 PAGE 113

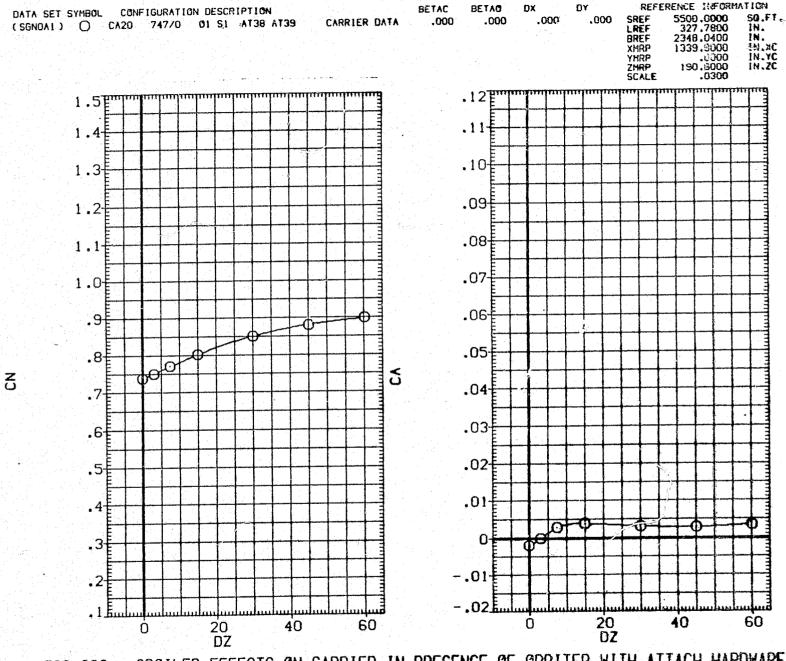


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(C)ALPHAW= 10.00

PAGE 114

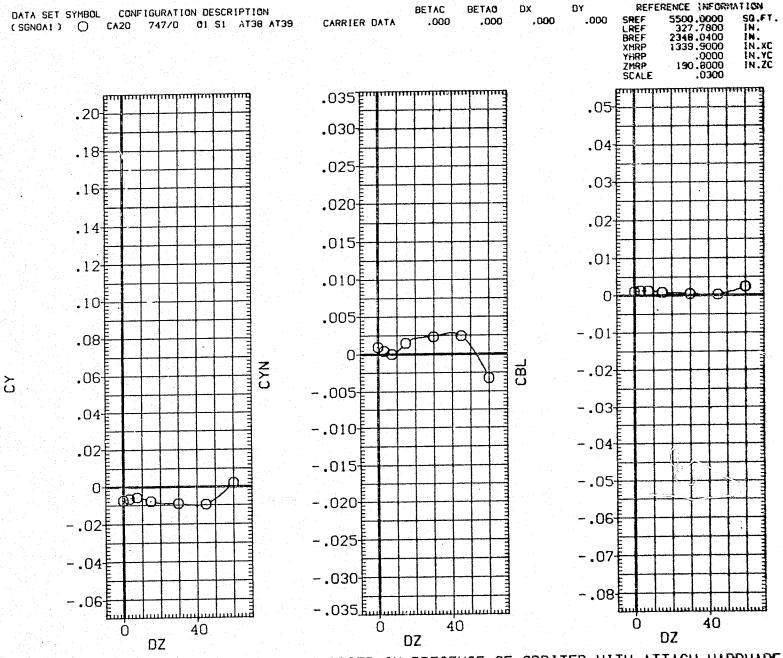
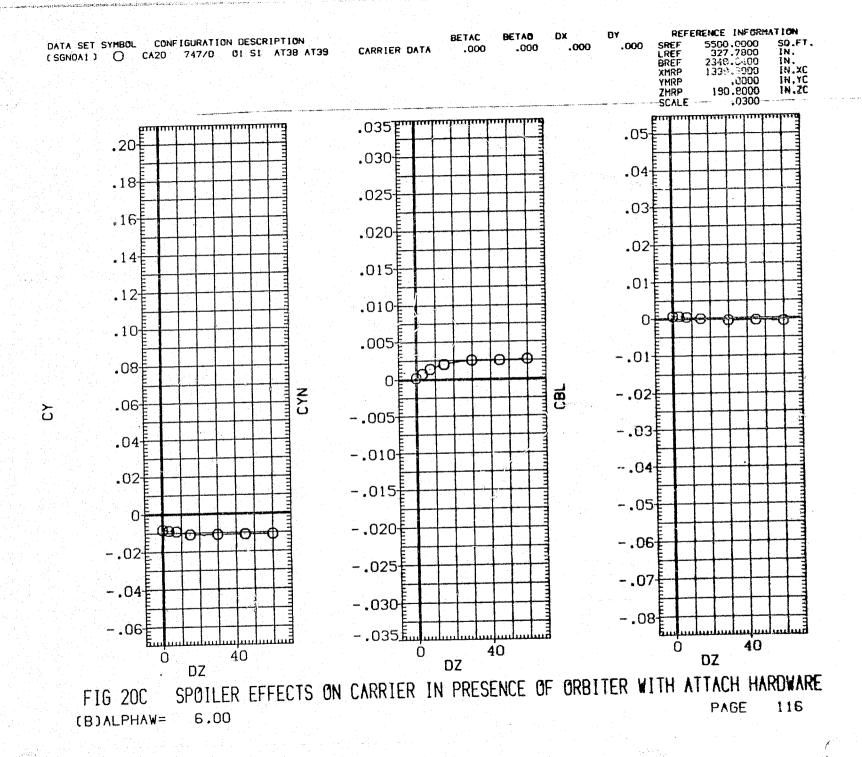


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE
(A)ALPHAW= 2.00
PAGE 115



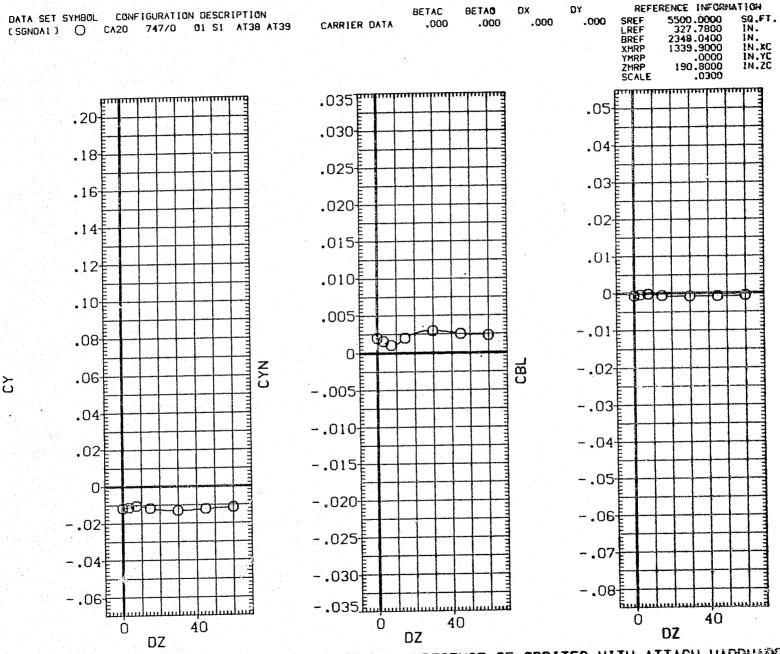
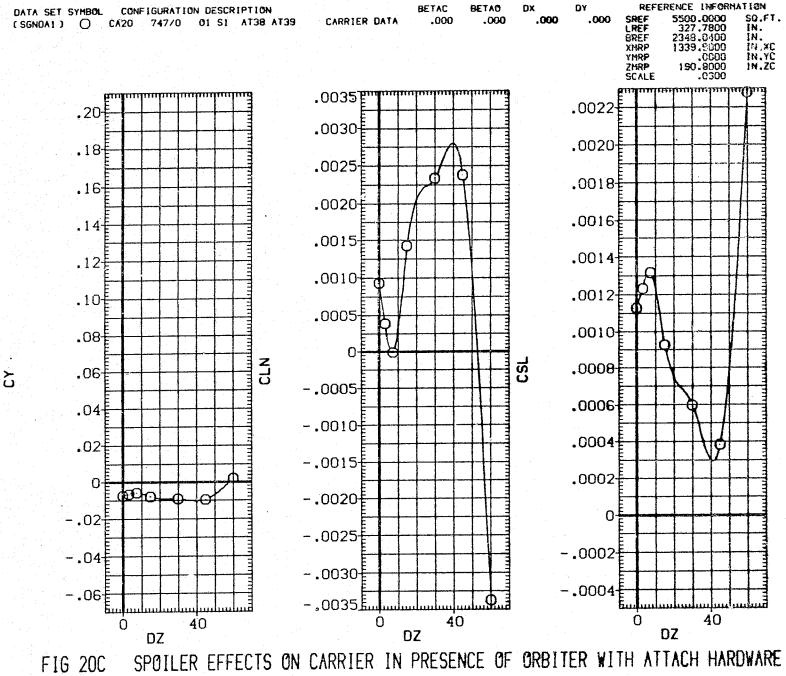


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF GRBITER WITH ATTACH HARDWARE (C)ALPHAW= 10.00



PAGE 118 (A)ALPHAW= 2.00

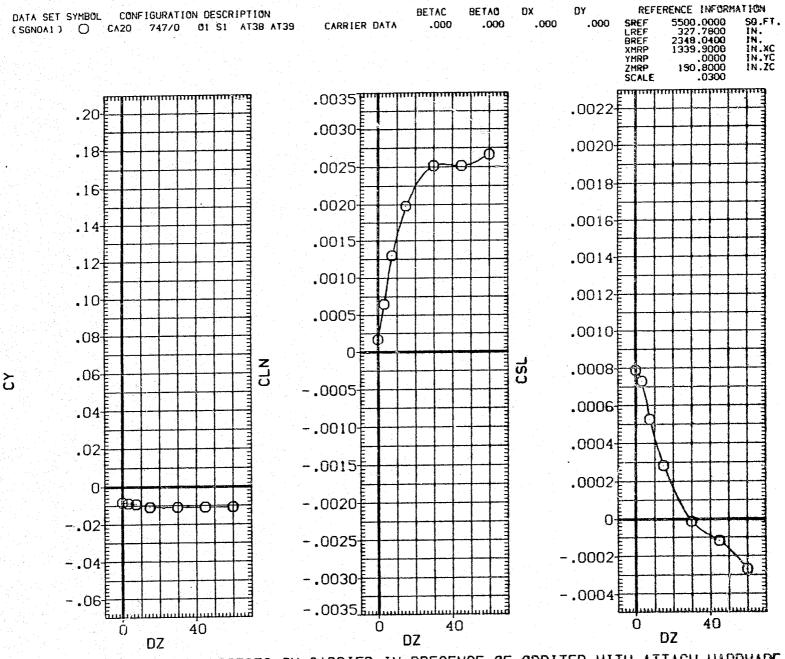
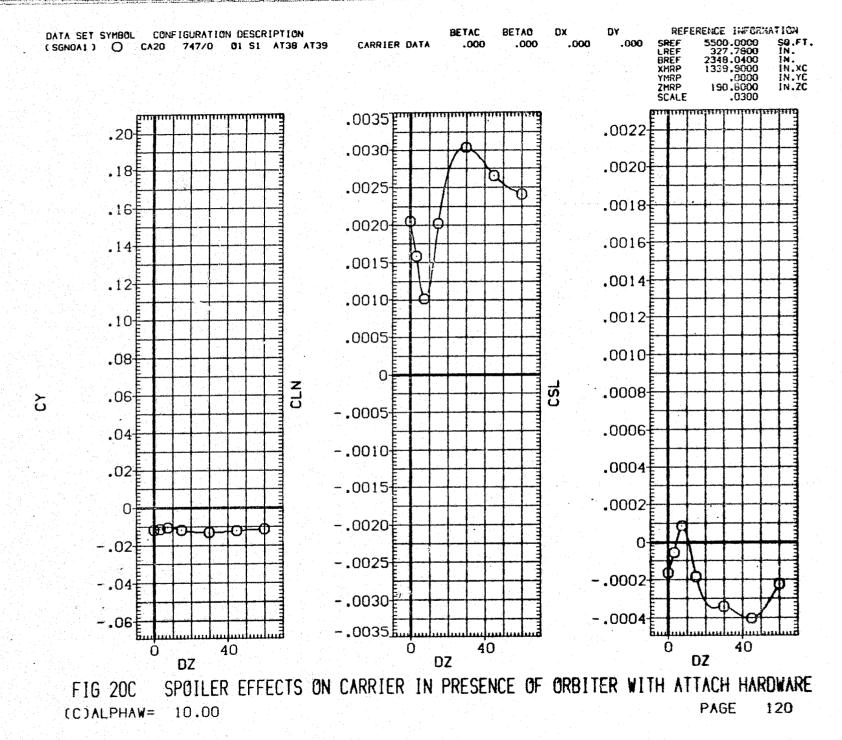


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE
(B)ALPHAW= 6.00
PAGE 119



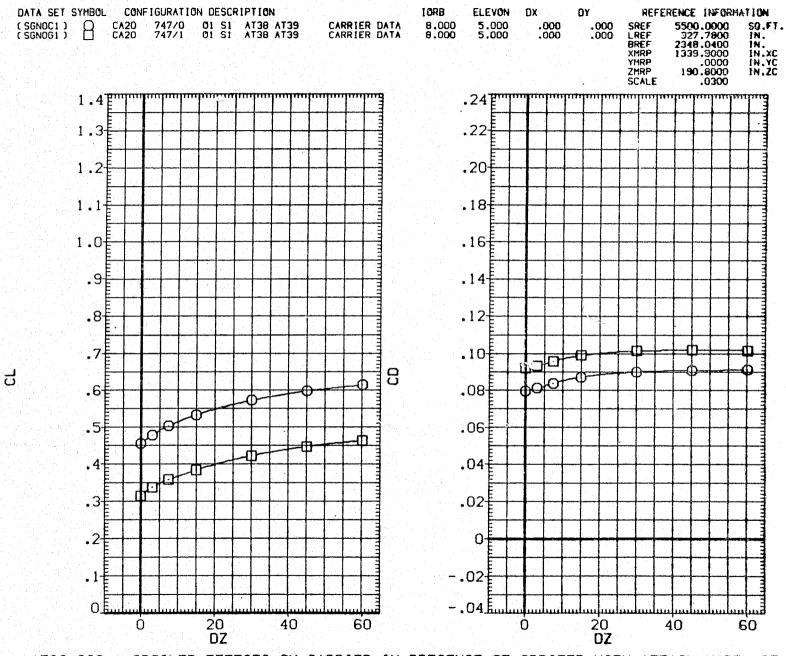


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)BETA = -5.00

PAGE 121

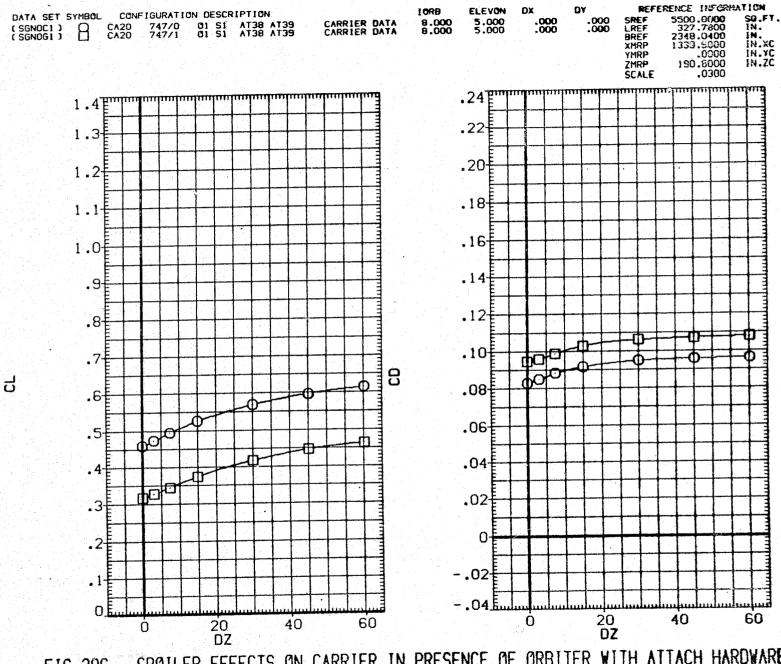
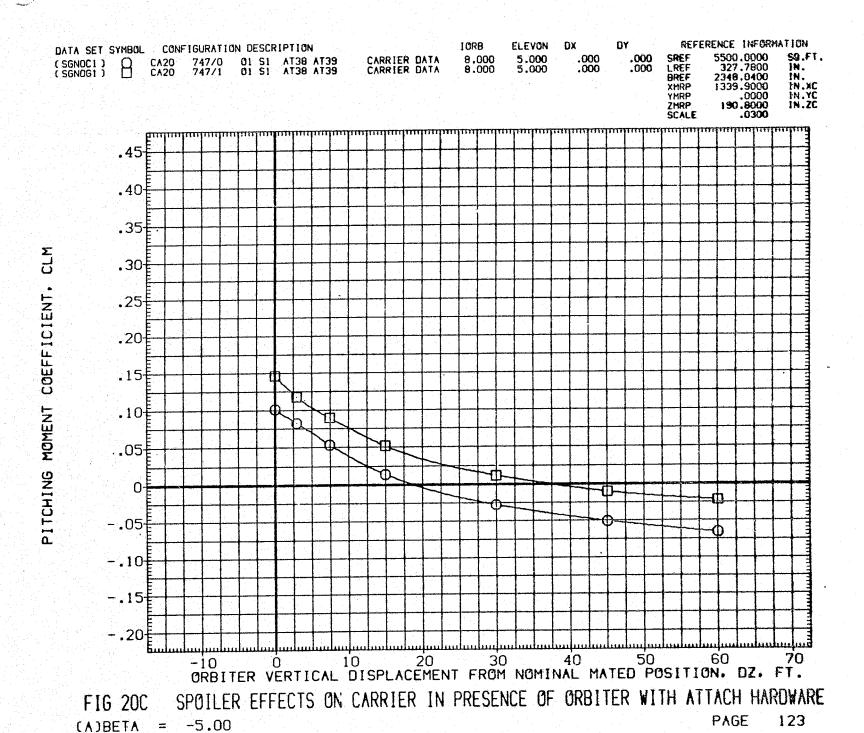
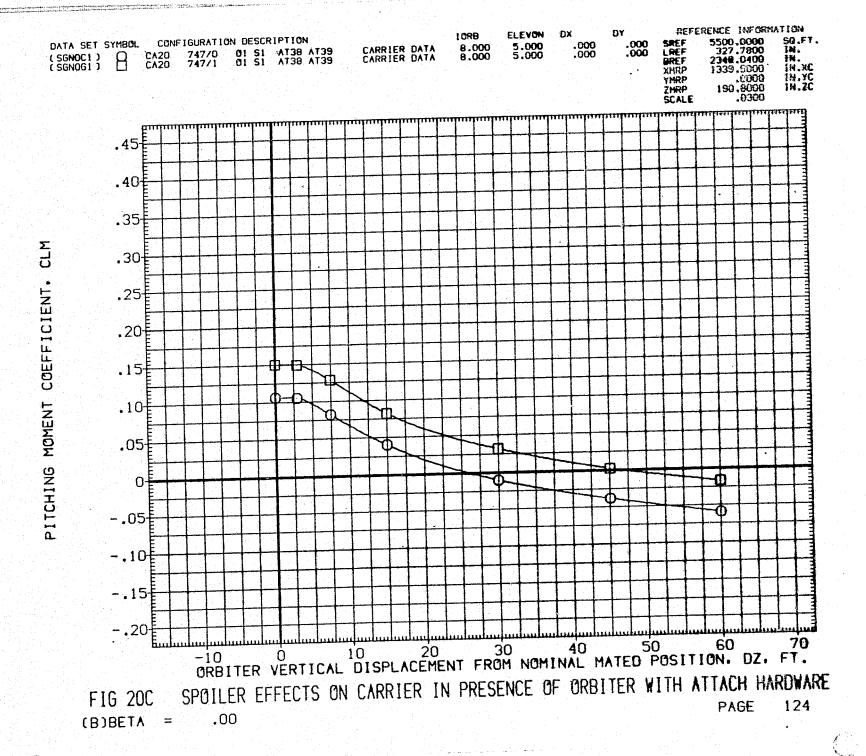


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)BETA = .00

PAGE 122





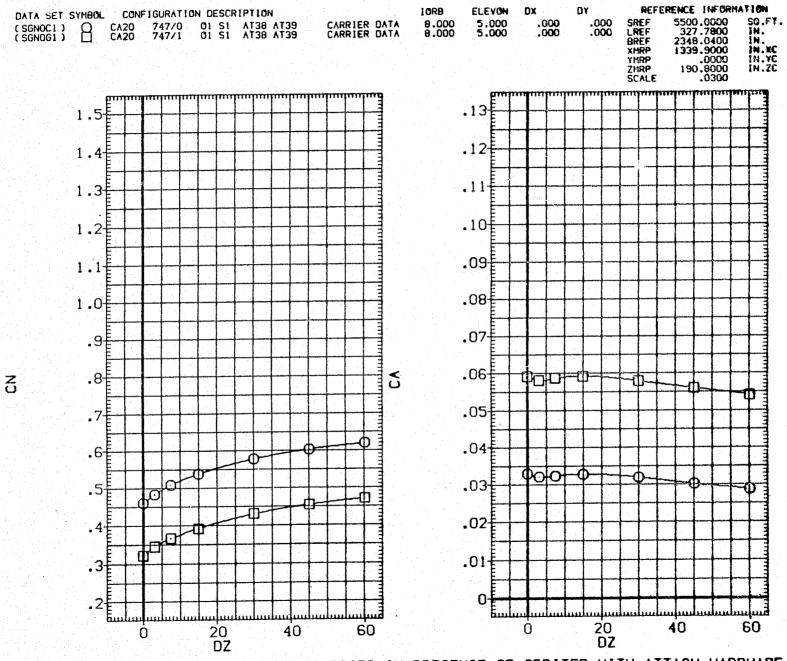


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)BETA = -5.00

PAGE 125

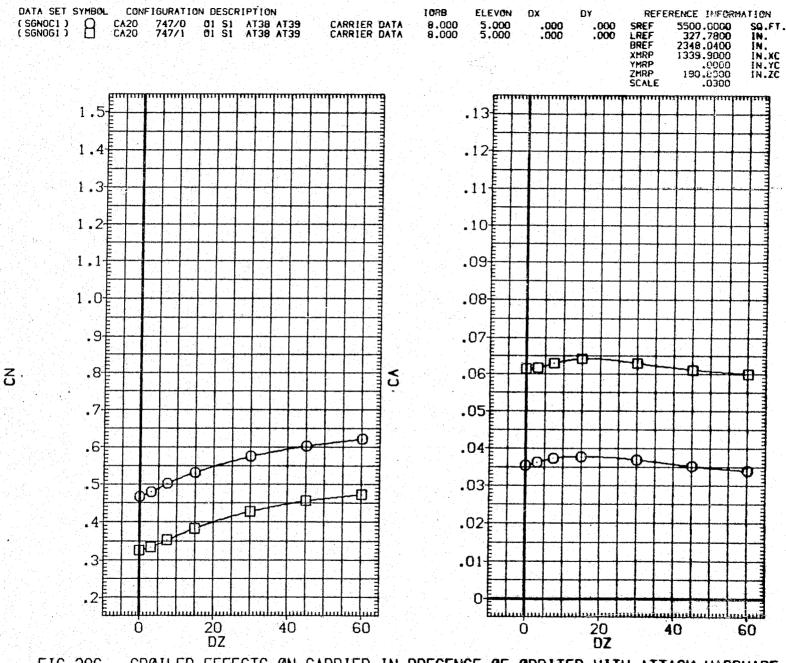


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)BETA = .00

PAGE 126

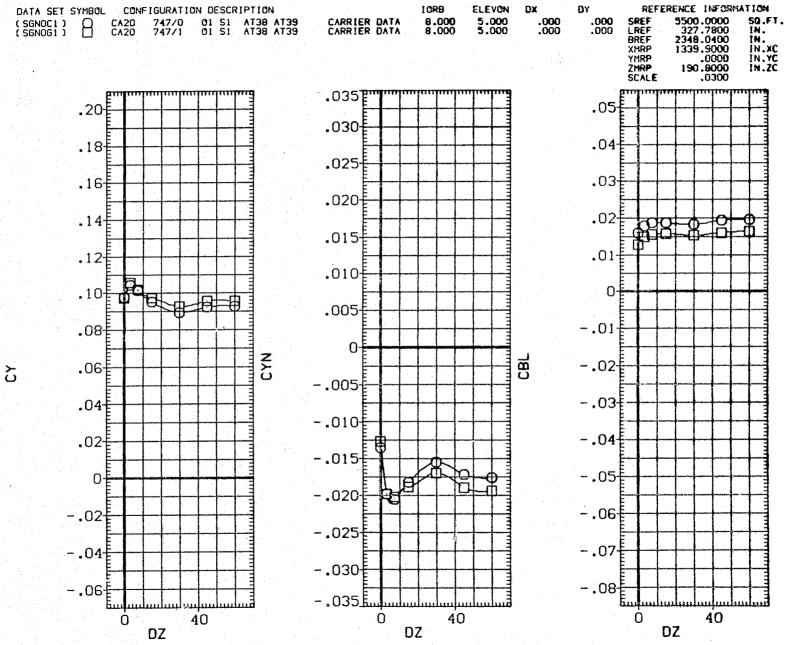
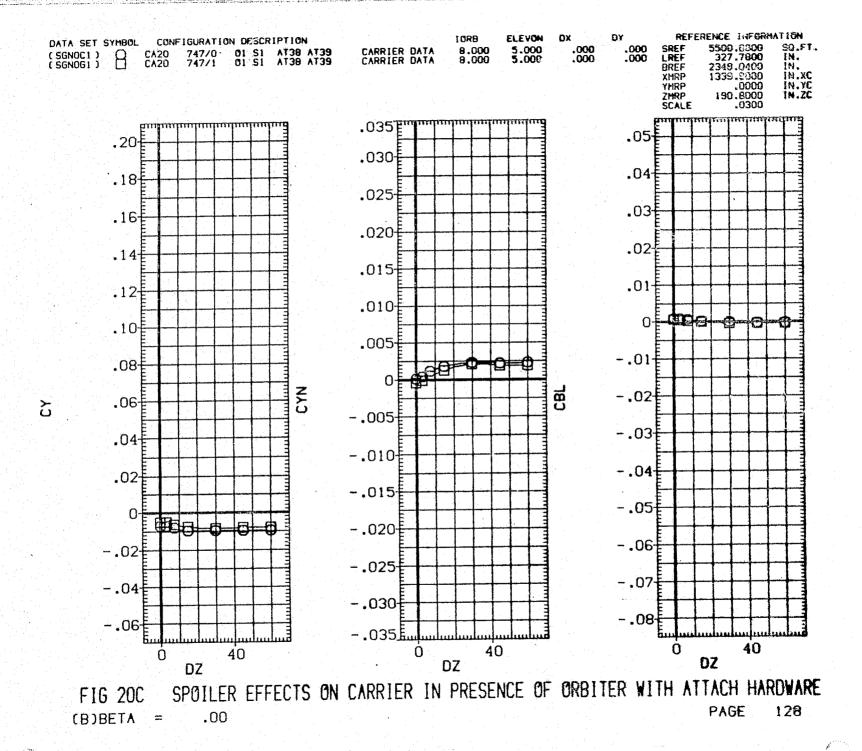


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)BETA = -5.00

PAGE 127



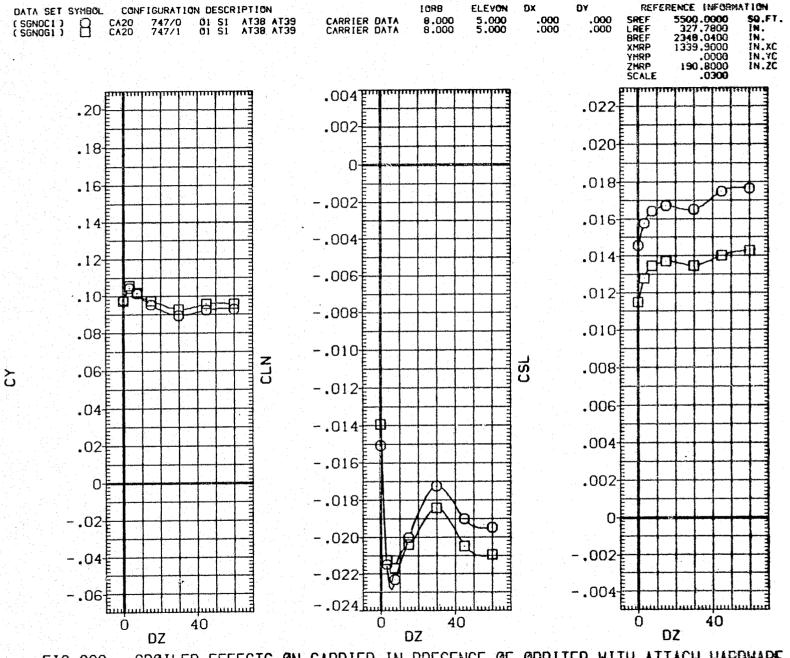


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)BETA = -5.00

PAGE 129

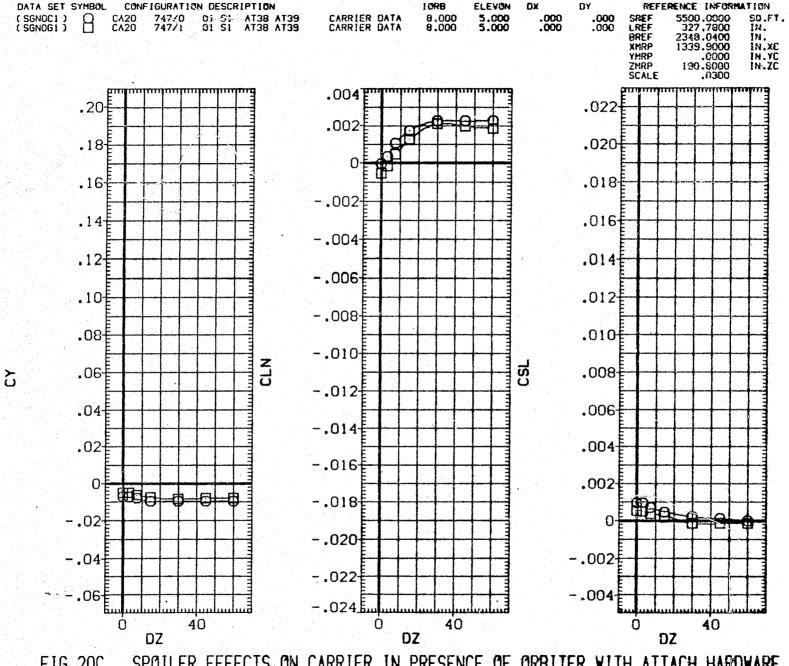


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)BETA = .00

PAGE 130

(=)

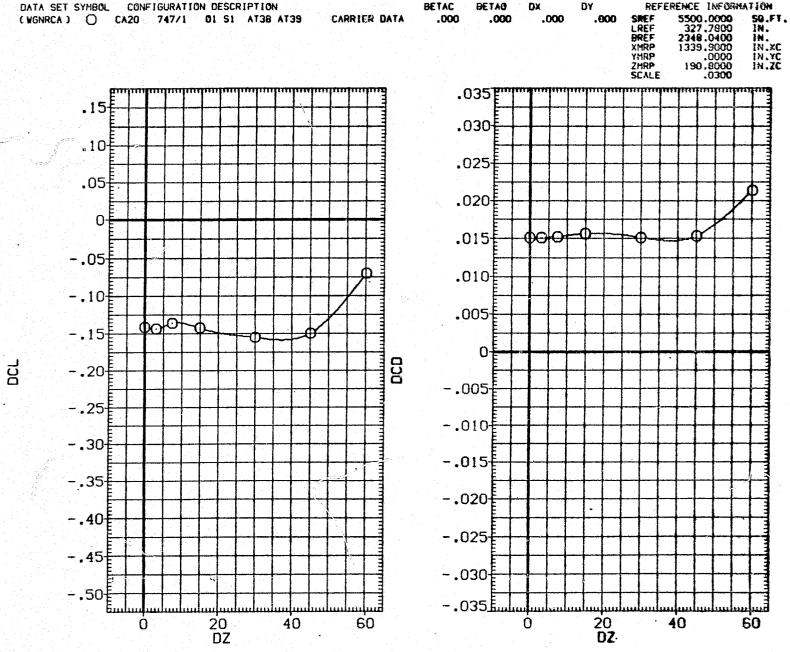


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A) ALPHAW = 2.00

PAGE 131

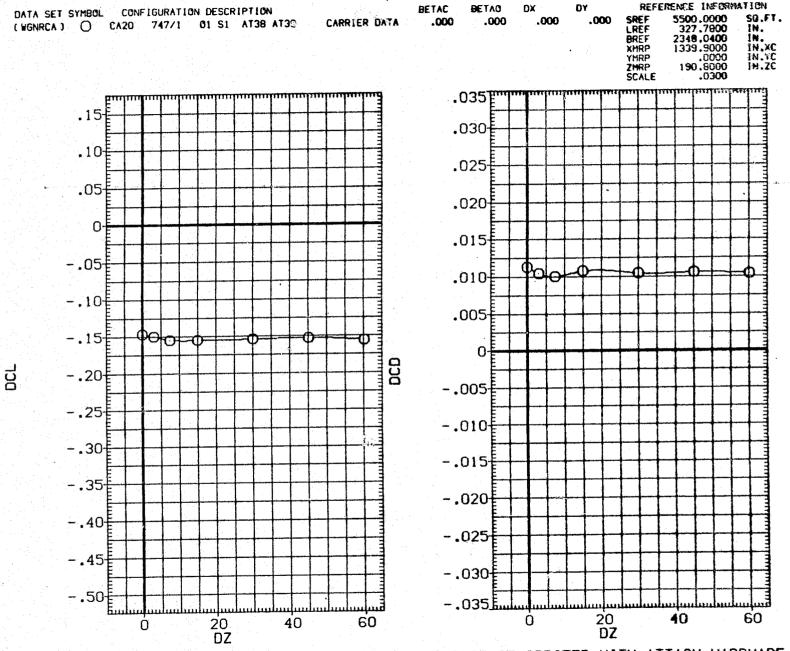


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B) ALPHAW = 6.00

PAGE 132

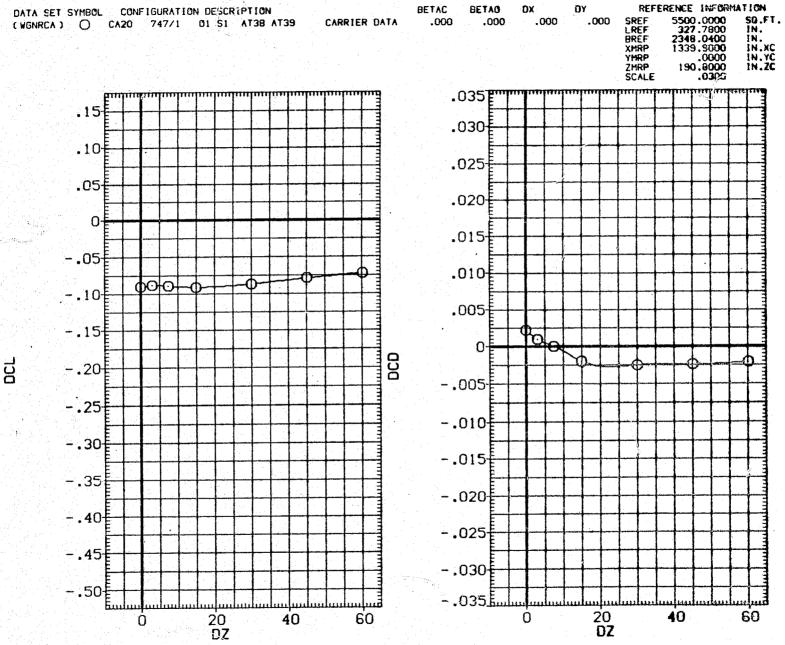
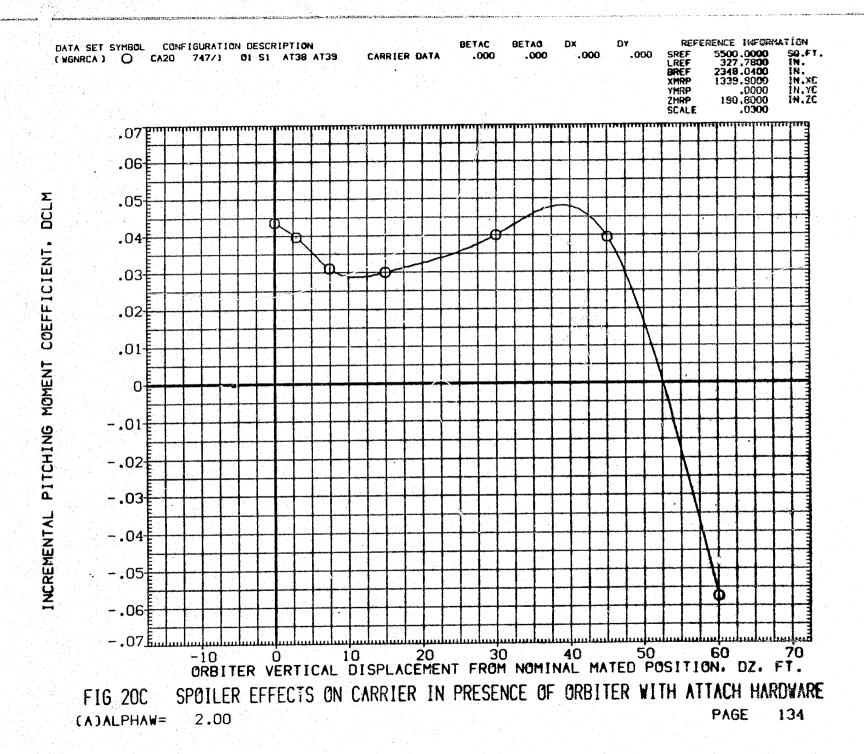
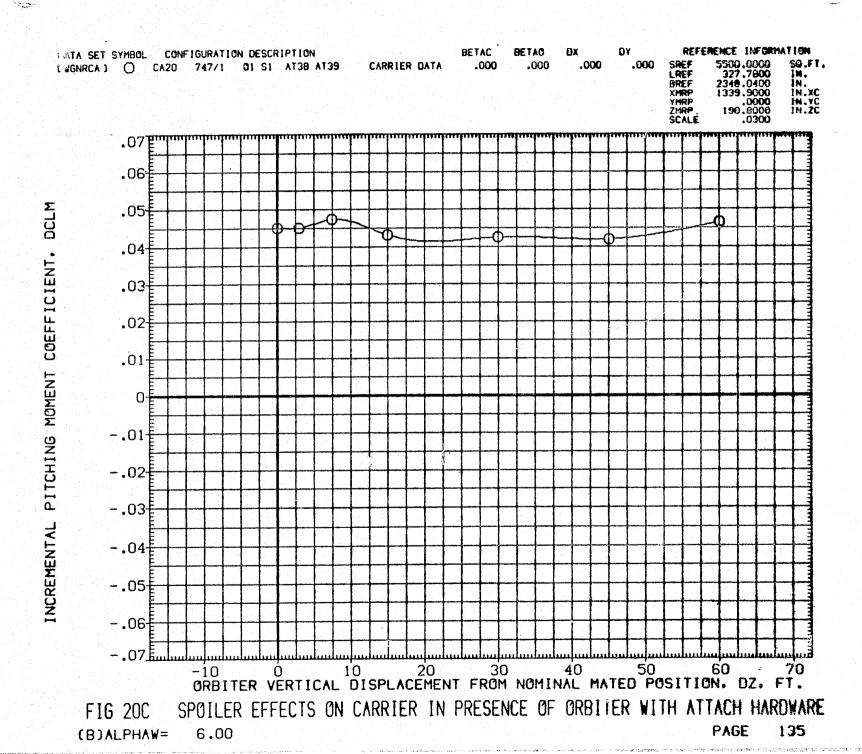
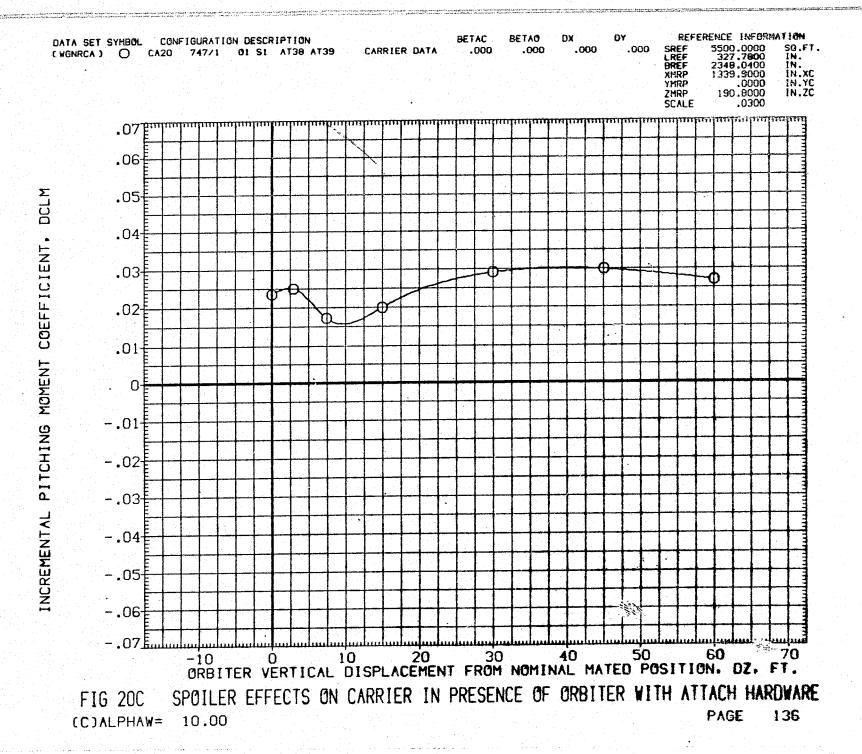


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE (C)ALPHAW= 10.00 PAGE 133







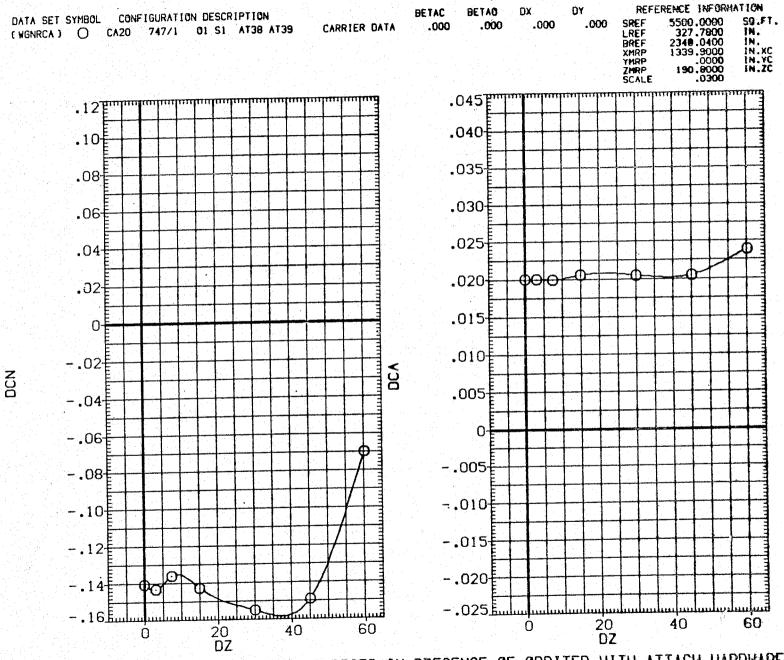
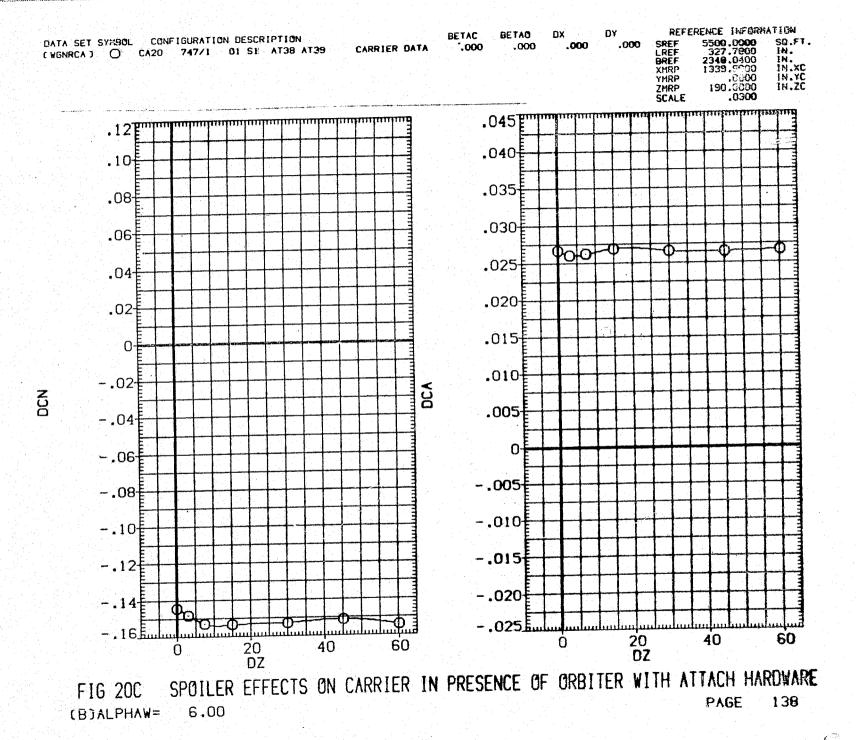
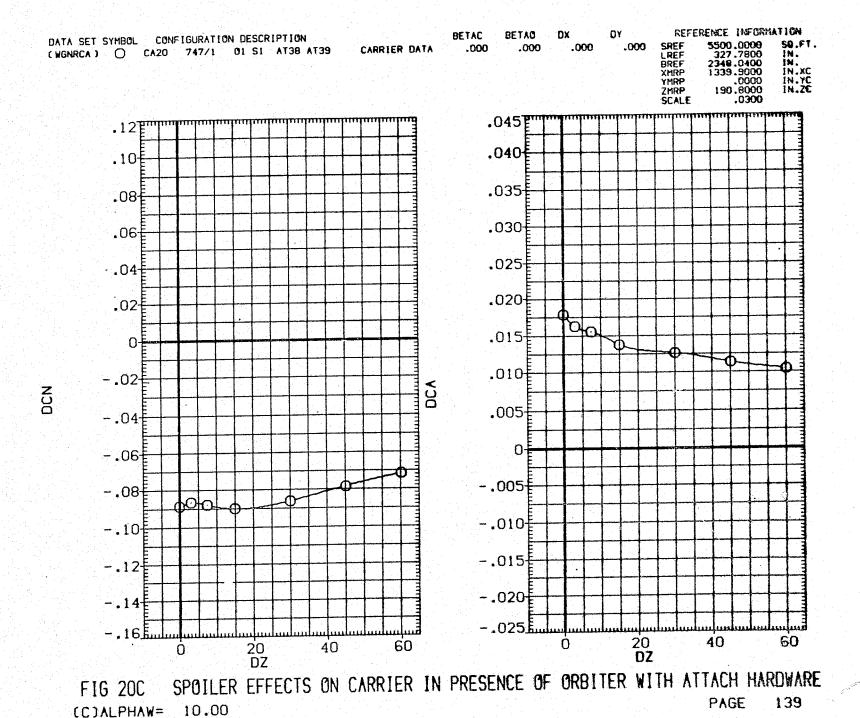


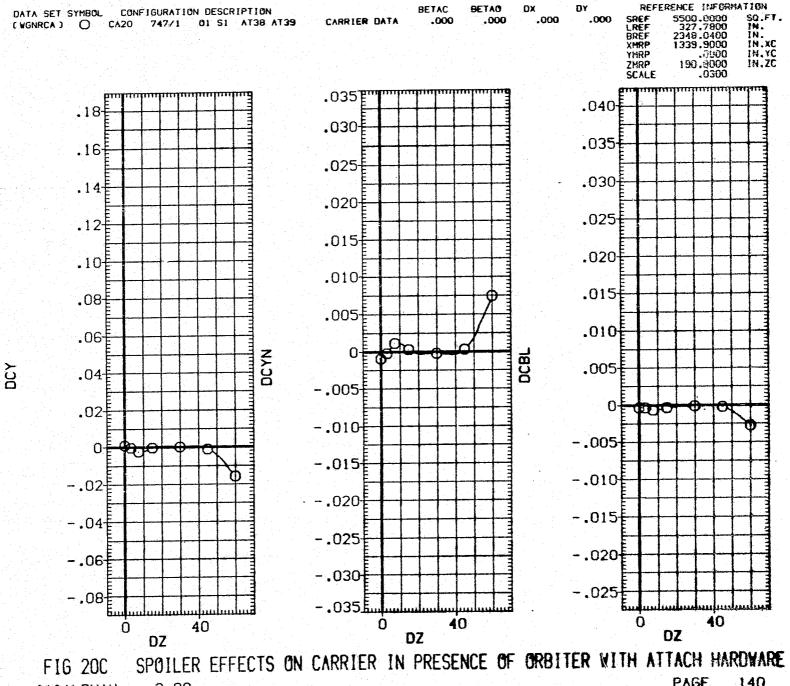
FIG 20C SPUILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)ALPHAW= 2.00

PAGE 137

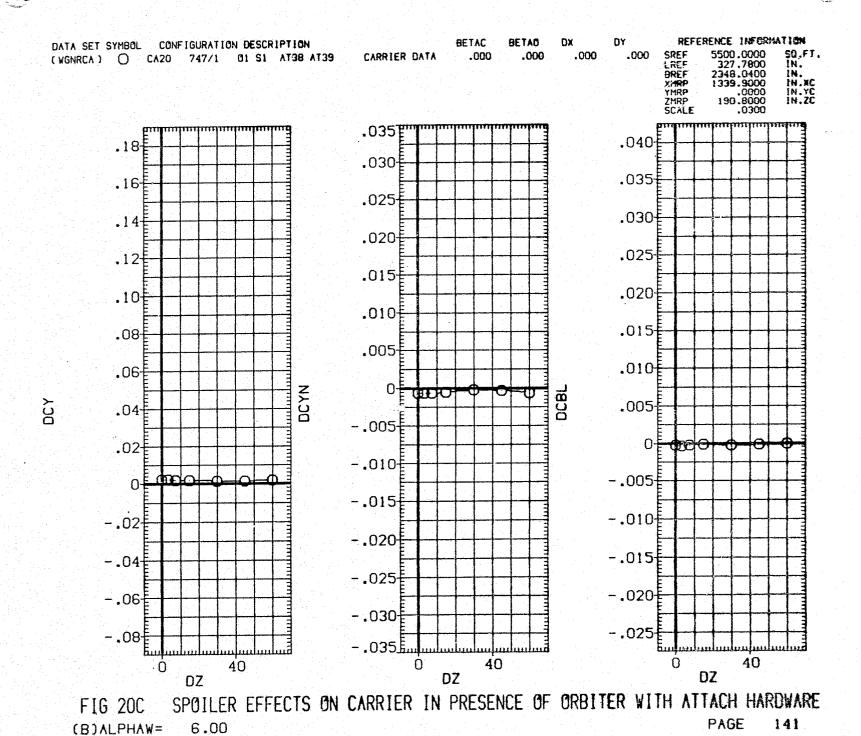


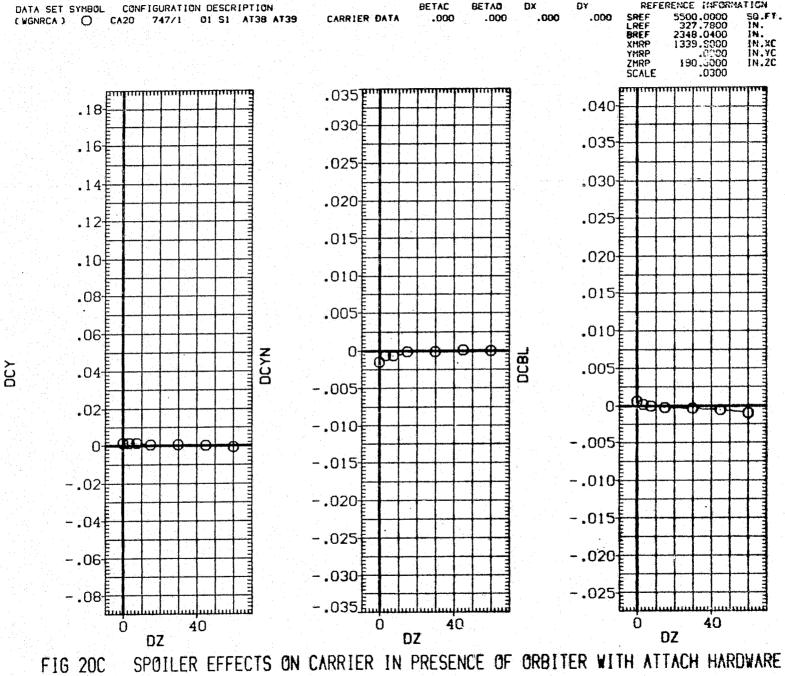




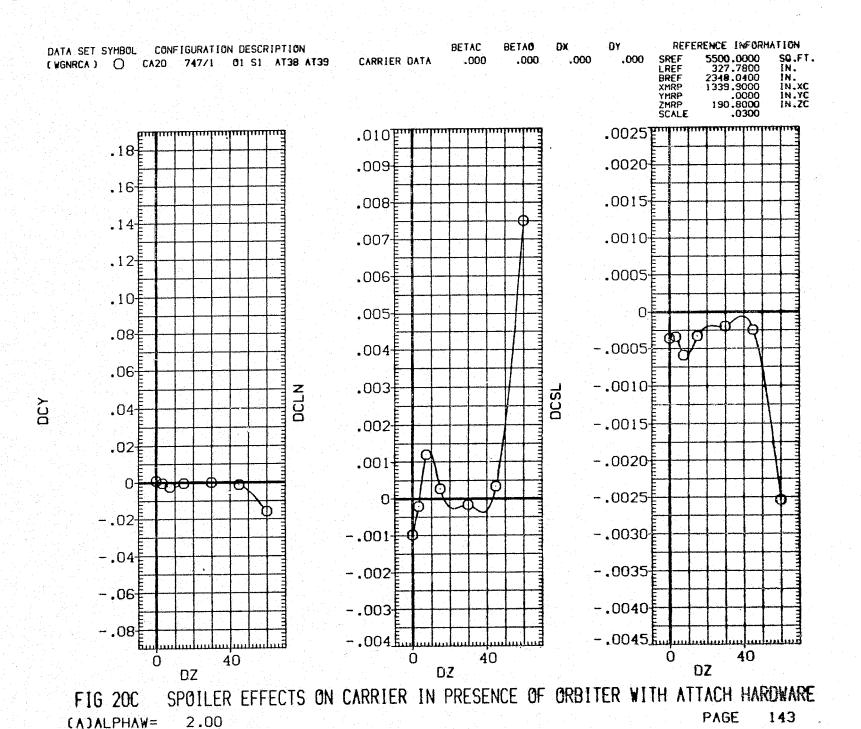
(A) ALPHAW = 2.00

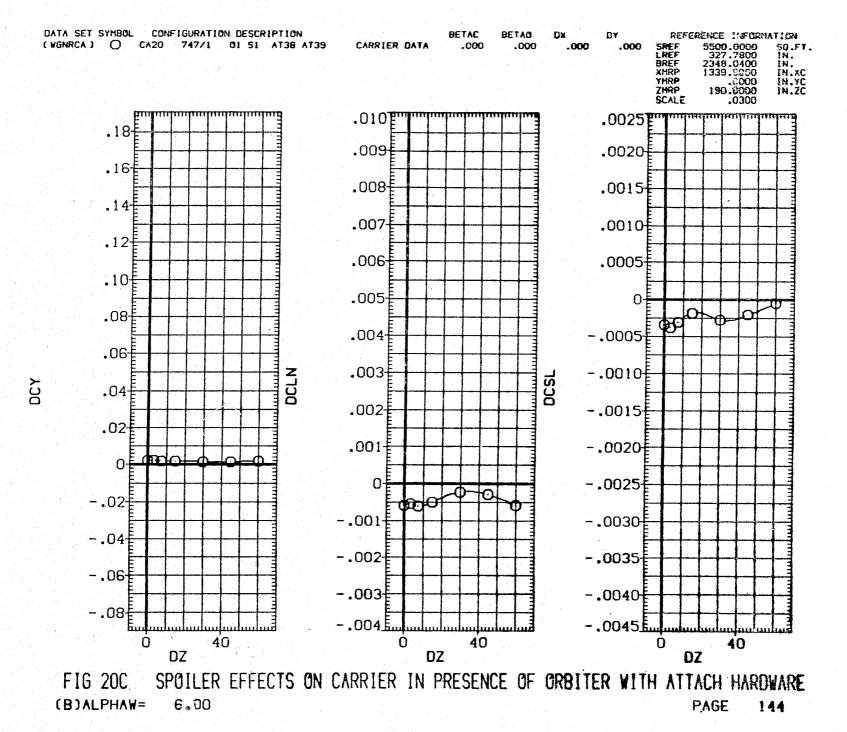
PAGE 140



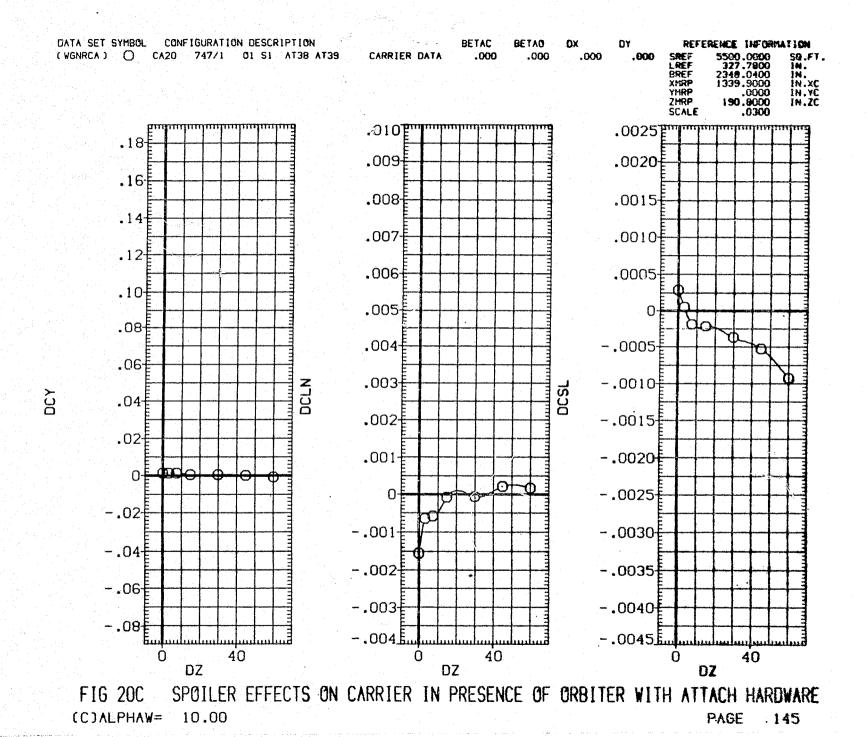


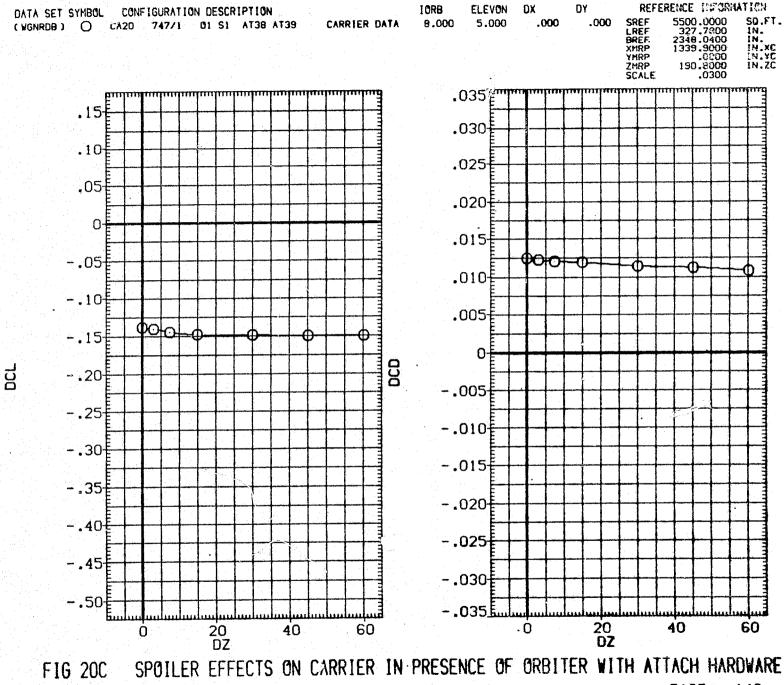
PAGE 142 (C)ALPHAW= 10.00





Anna Anna





PAGE 146 = -5.00 (A)BETA

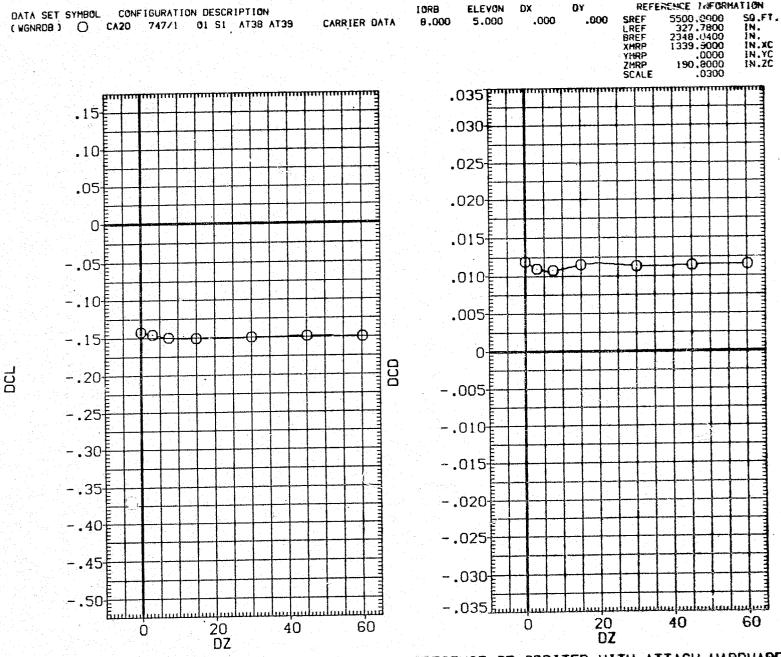
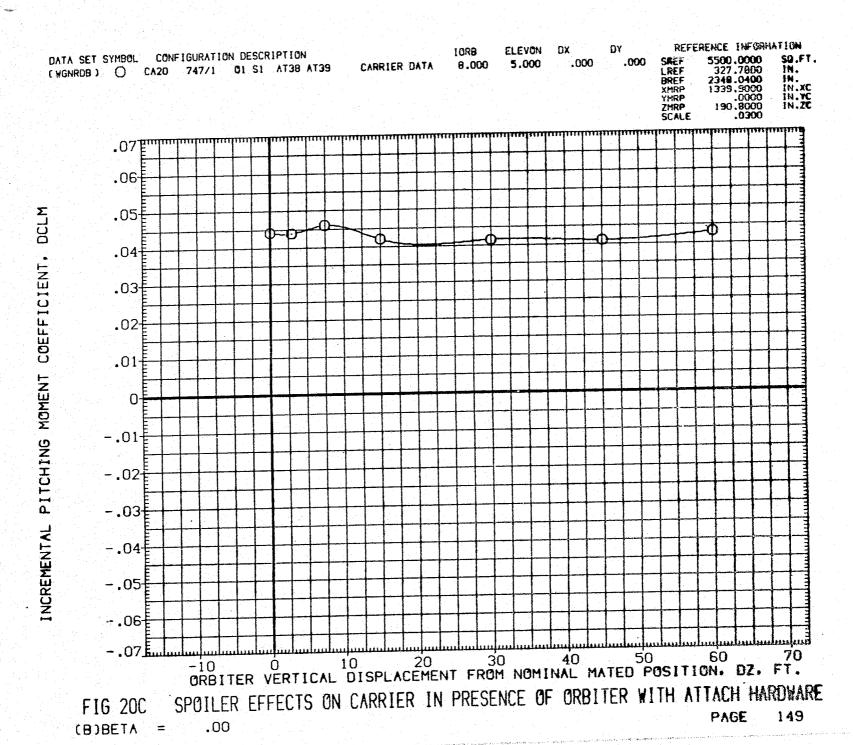


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)BETA = .00

PAGE 147



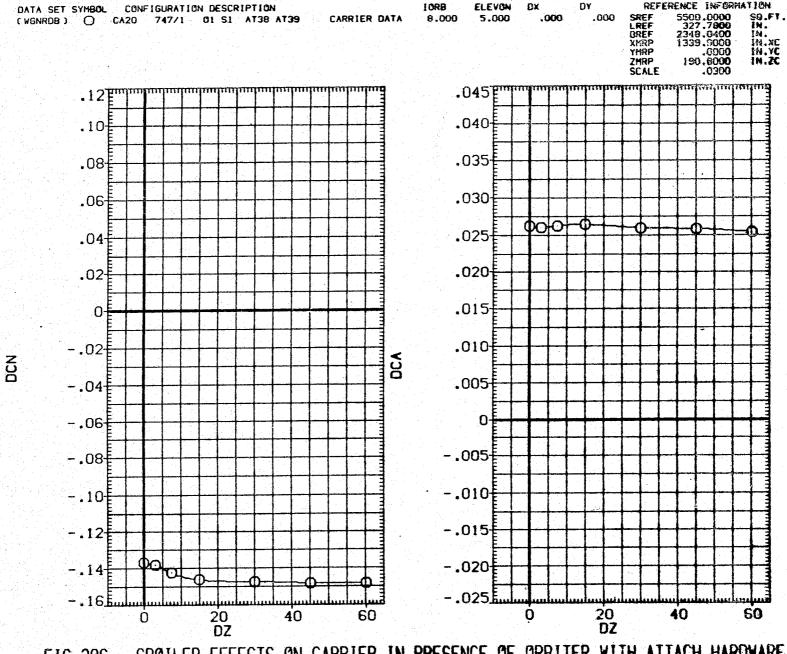


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(A)BETA = -5.00

PAGE 150



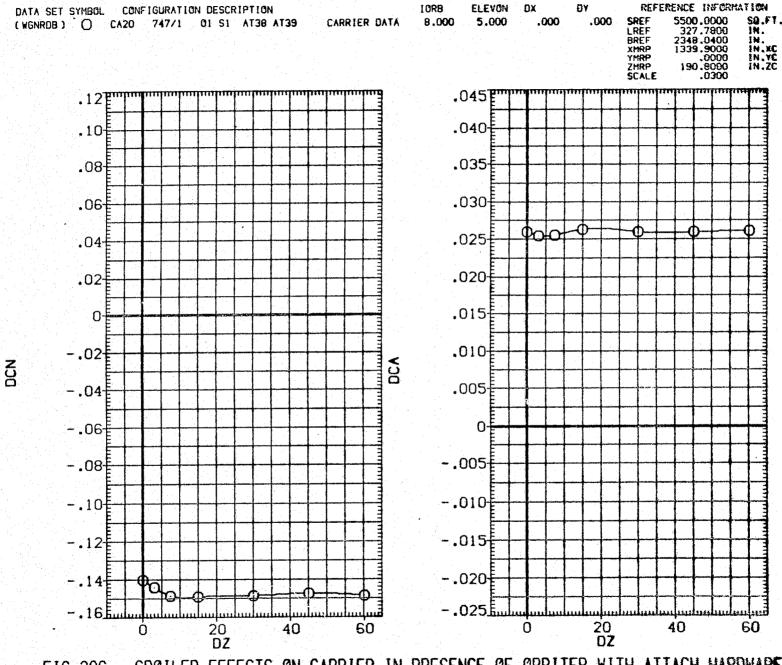


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)BETA = .00

PAGE 151

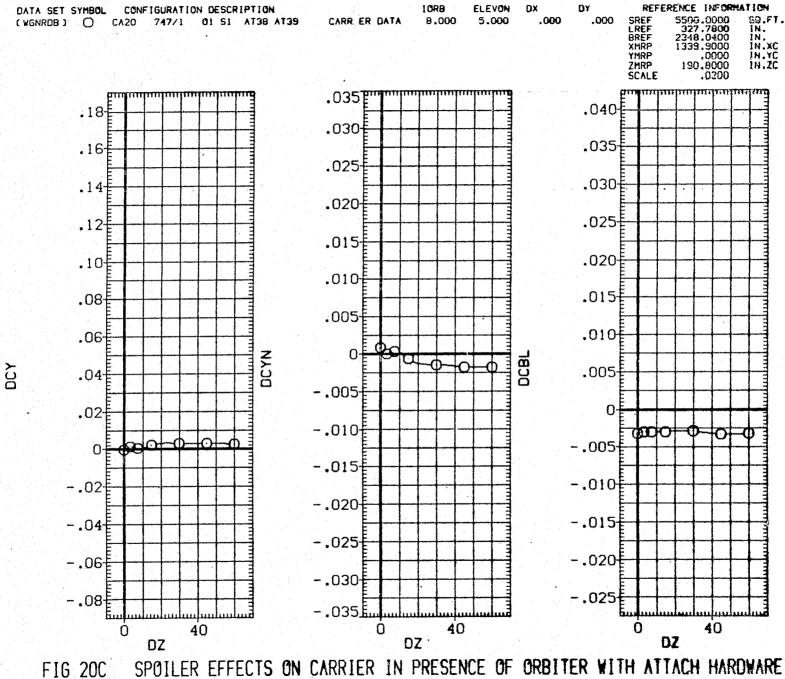


FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARD (A)BETA = -5.00

PAGE 152

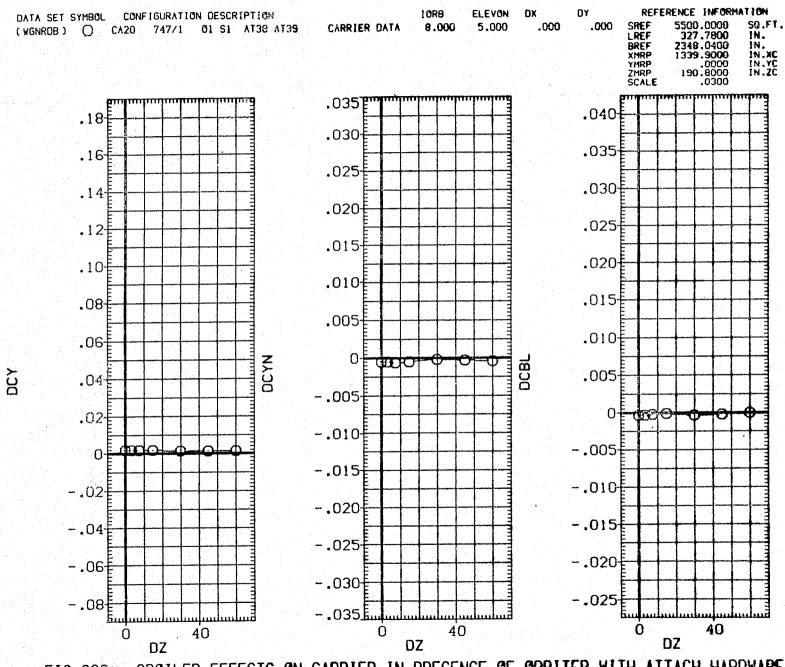
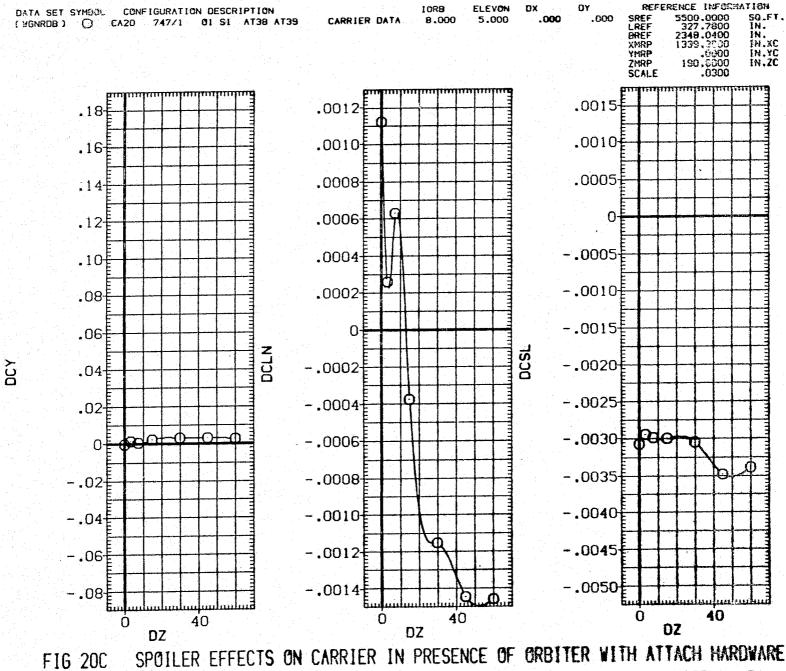


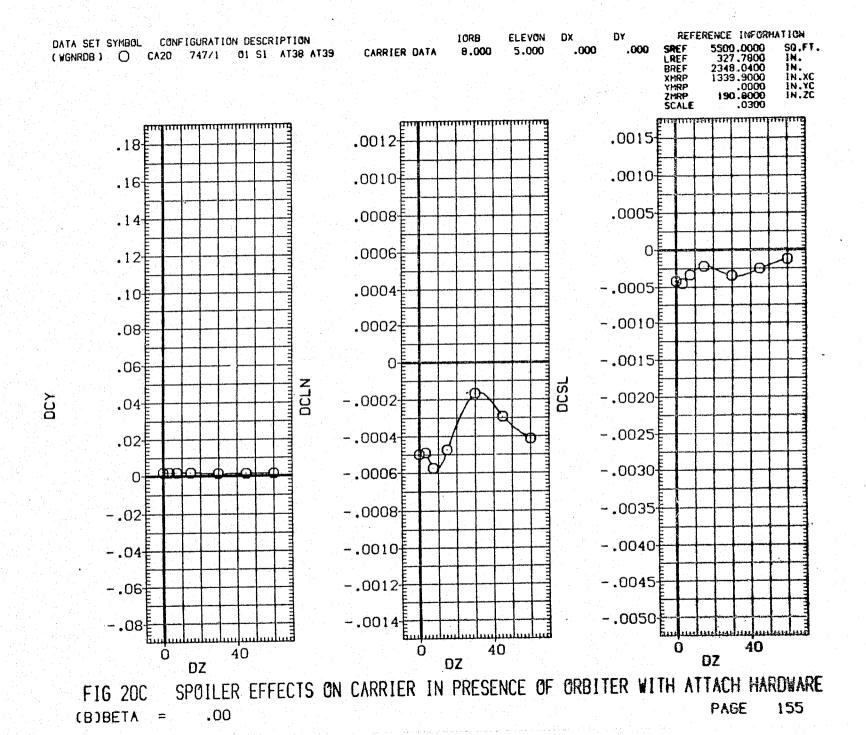
FIG 20C SPOILER EFFECTS ON CARRIER IN PRESENCE OF ORBITER WITH ATTACH HARDWARE

(B)BETA = .00

PAGE 153



154 PAGE = -5.00 (A)BETA



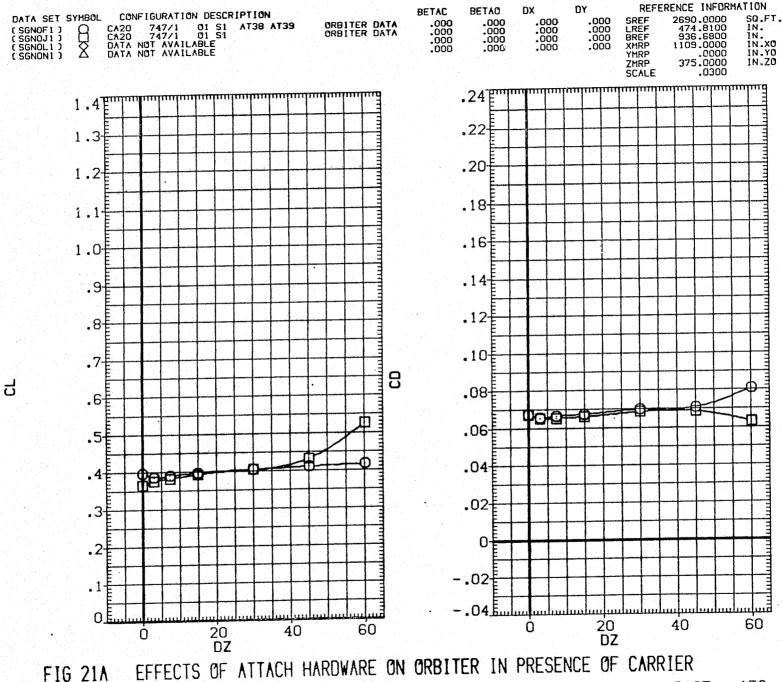


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

PAGE 156

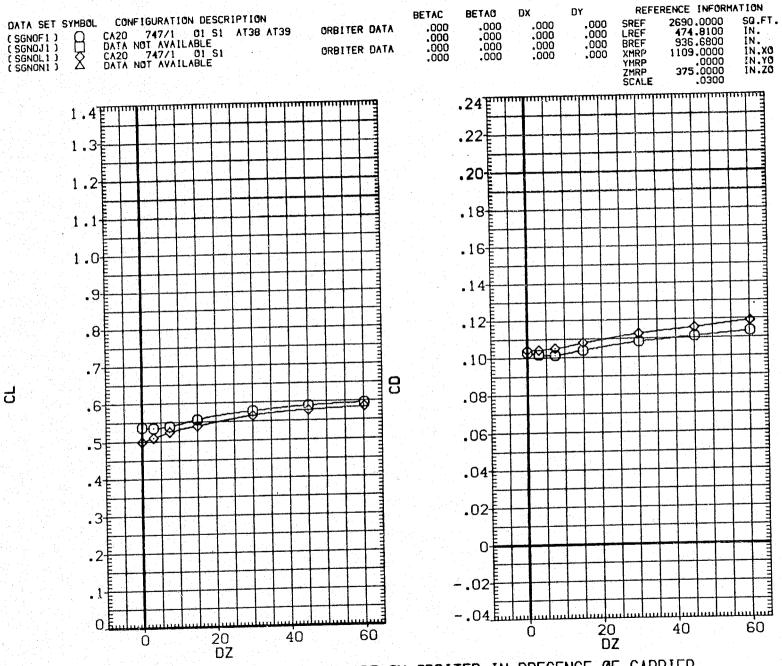


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

(B) ALPHAO = 12.00

157

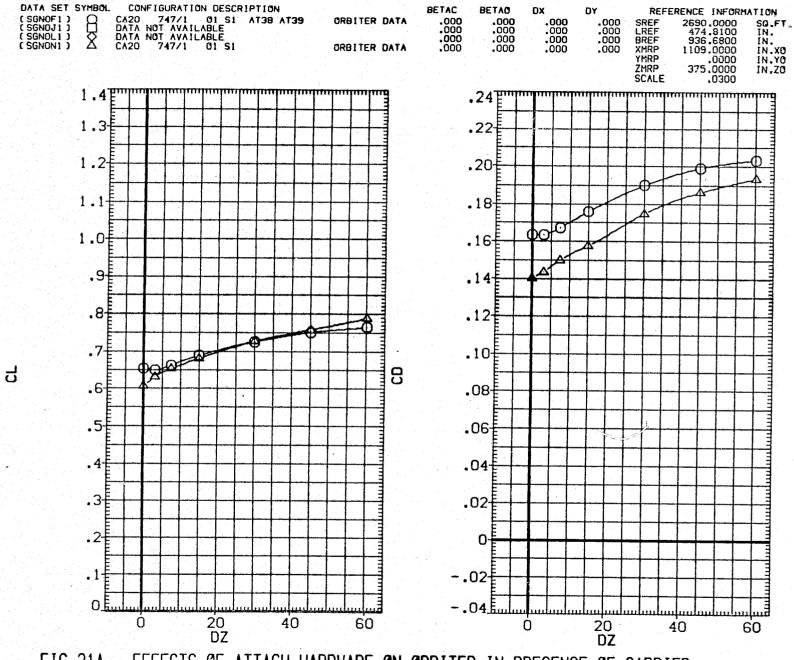
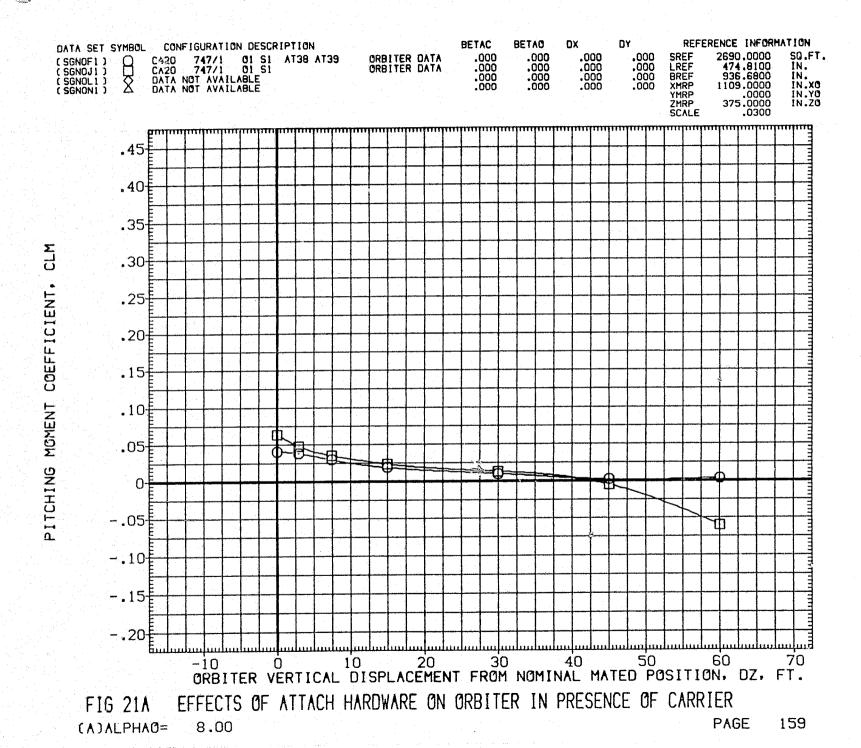
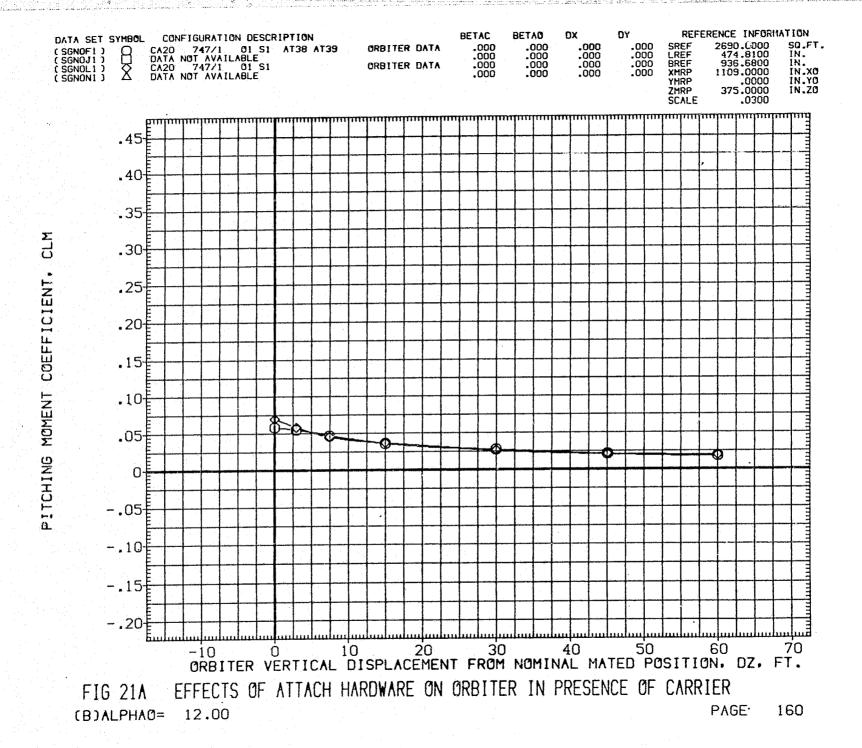
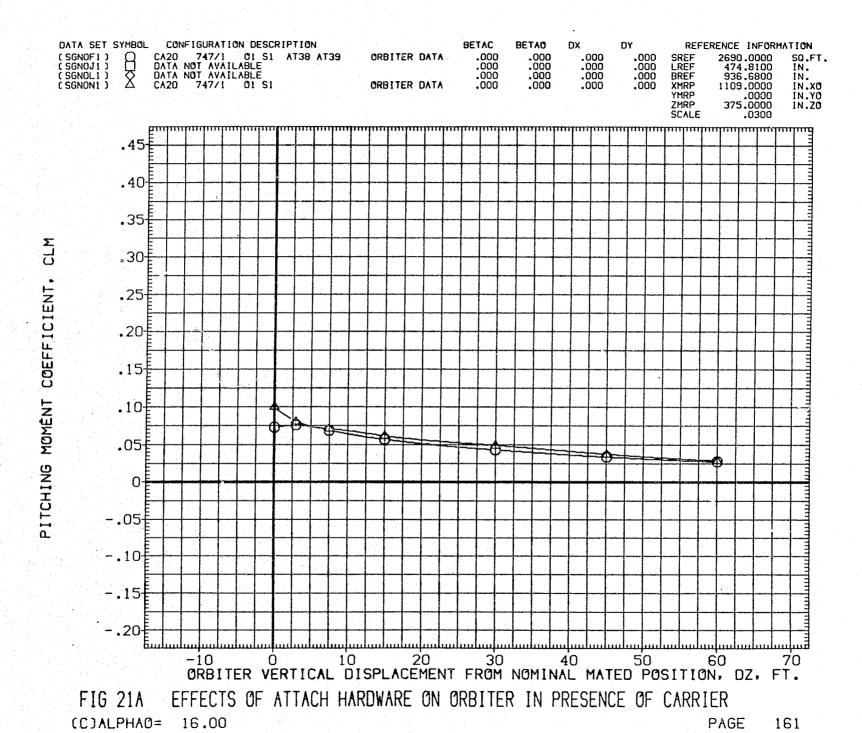


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER (C)ALPHAO= 16.00

158







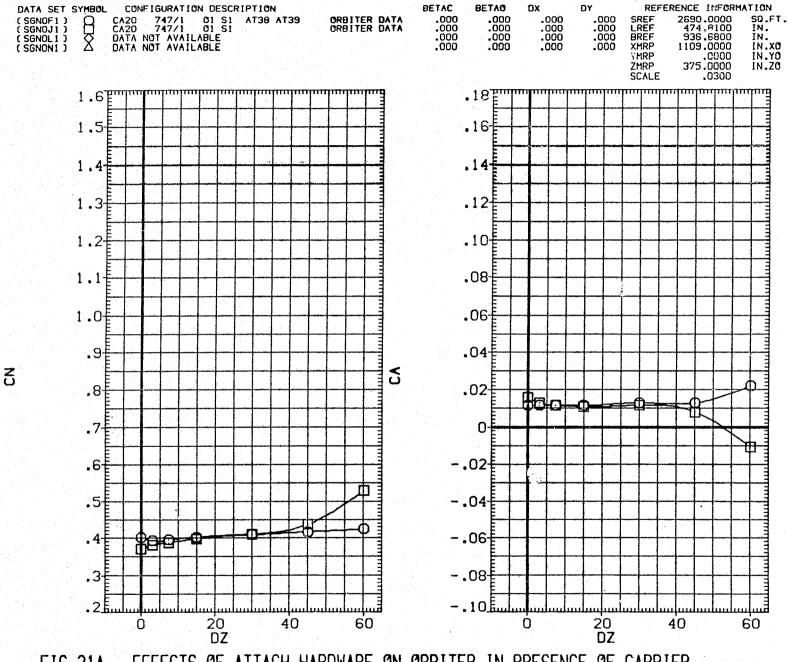


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

(A)ALPHAO= 8.00

PAGE 162

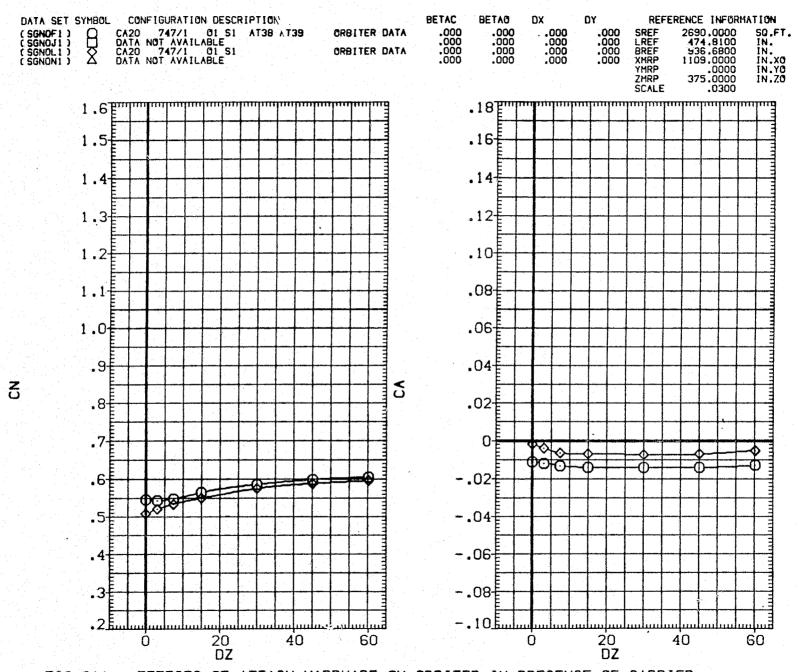


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

(B) ALPHAO = 12.00

PAGE

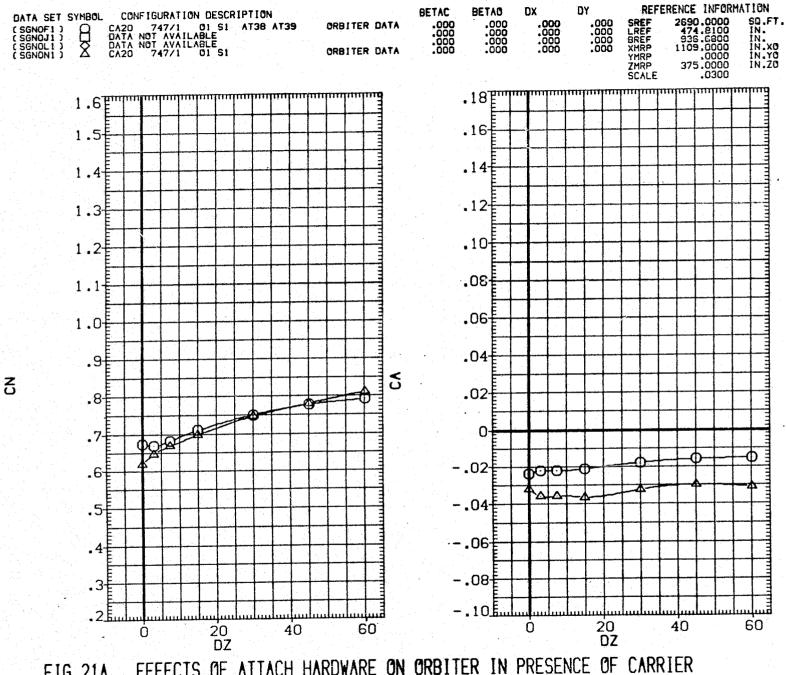
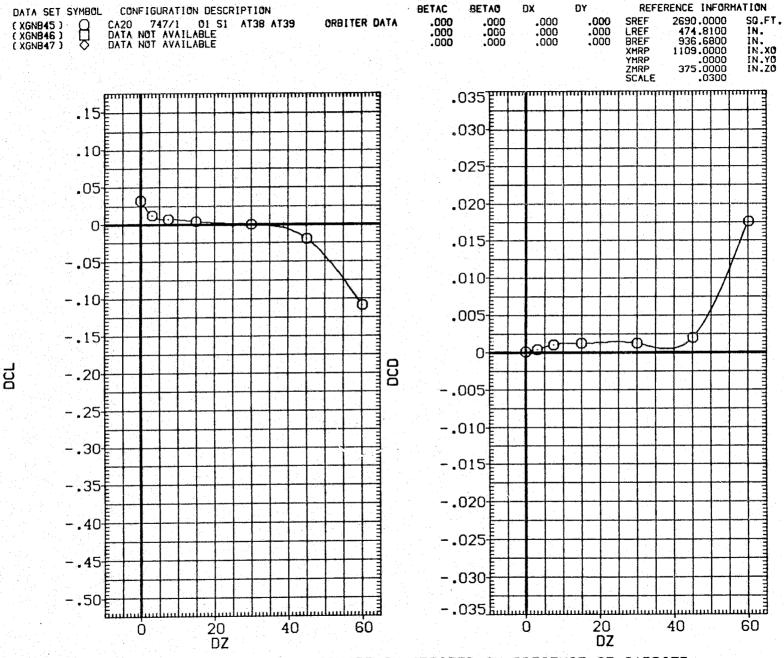


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

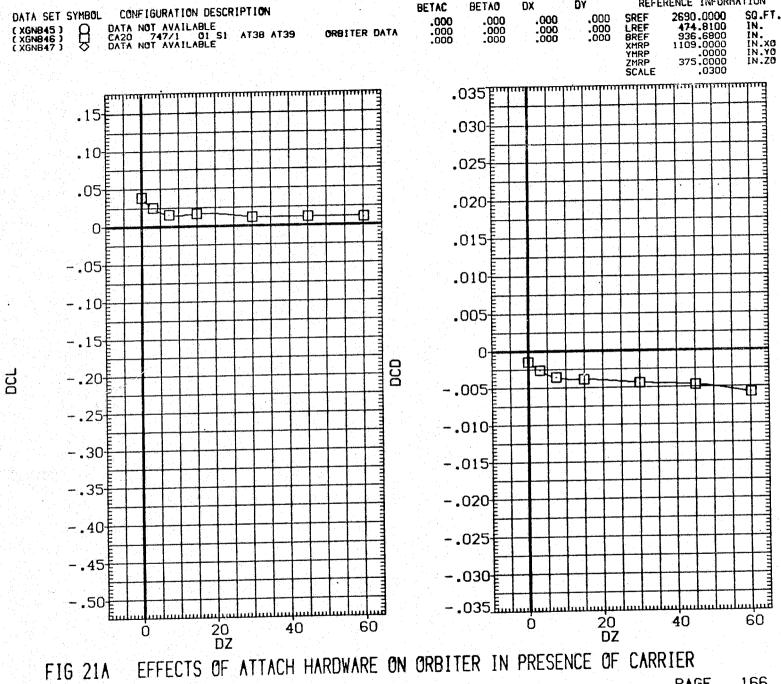
COALPHAGE 16.00

PAGE



EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER FIG 21A (A)ALPHAO= 8.00

PAGE



REFERENCE INFORMATION

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166 PAGE (B)ALPHA0= 12.00



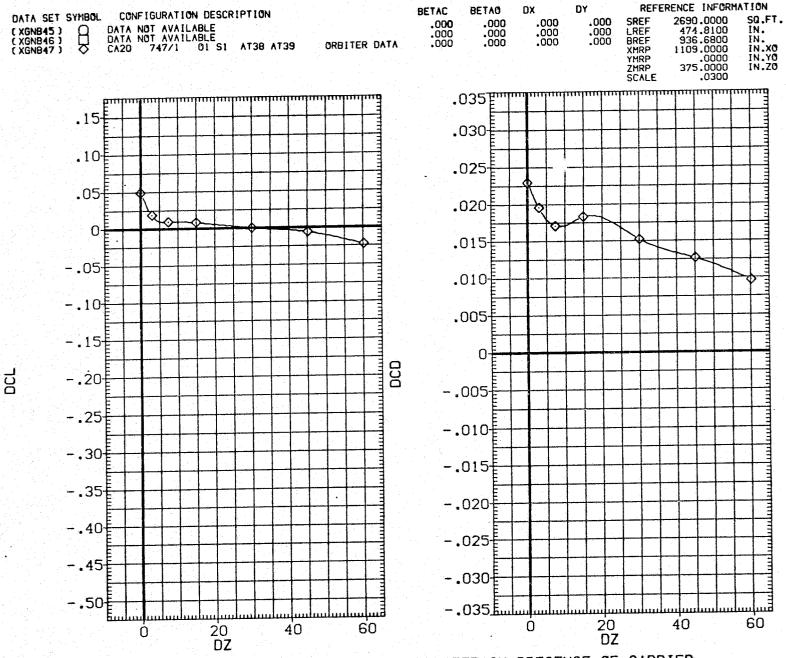
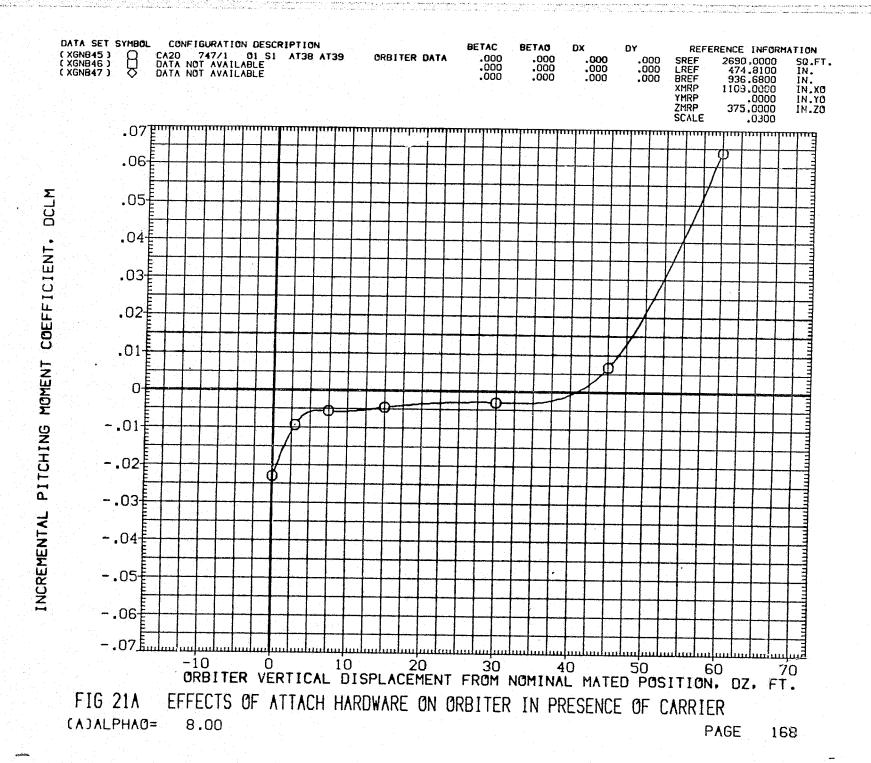
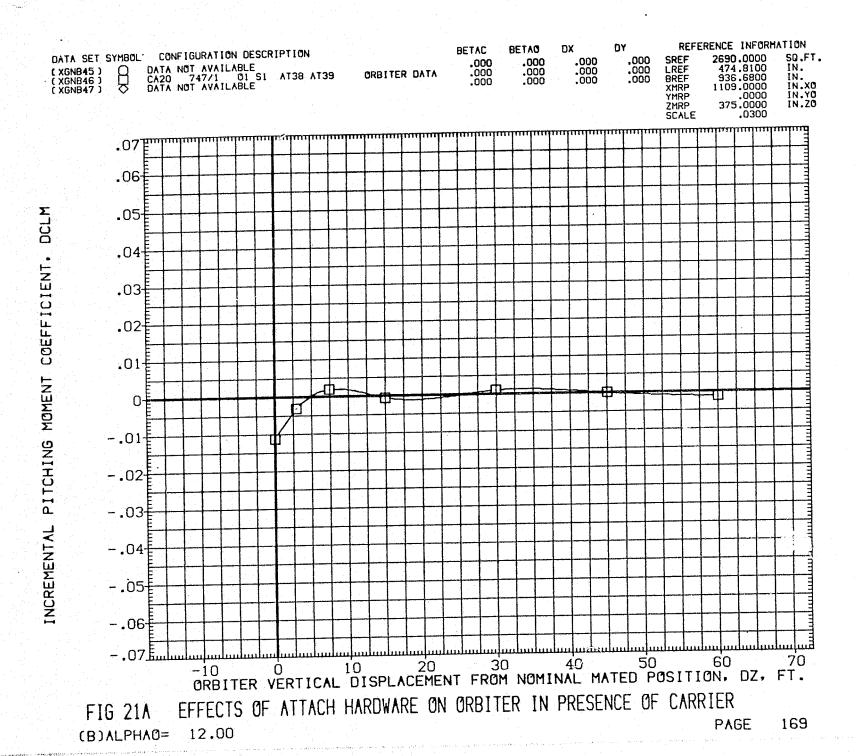


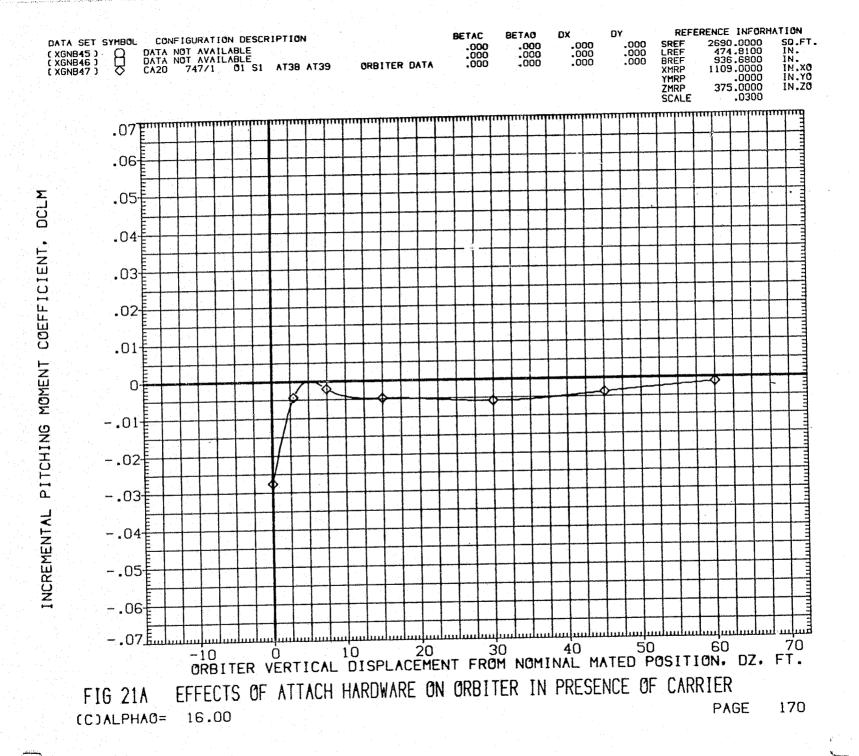
FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

(C) ALPHAO = 16.00

PAGE







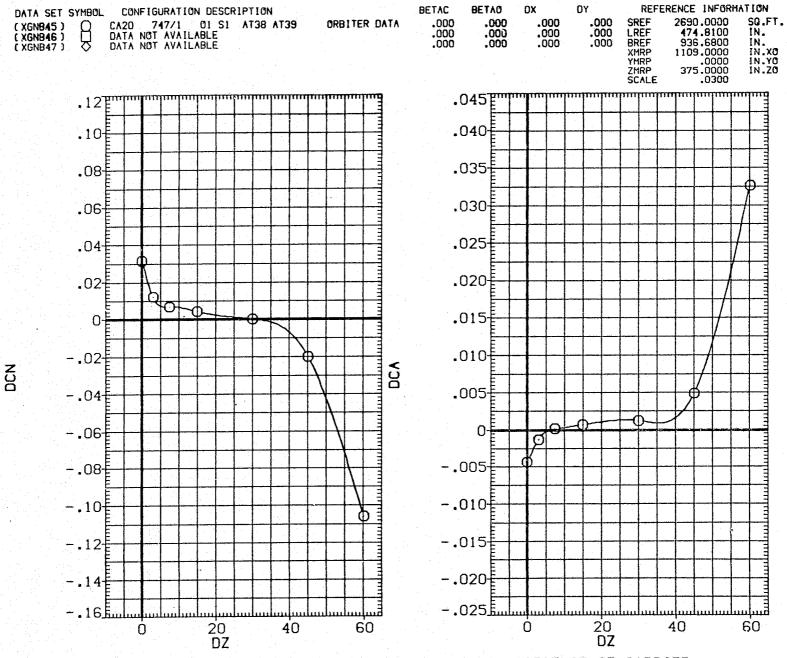


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

(A)ALPHAO= 8.00

PAGE

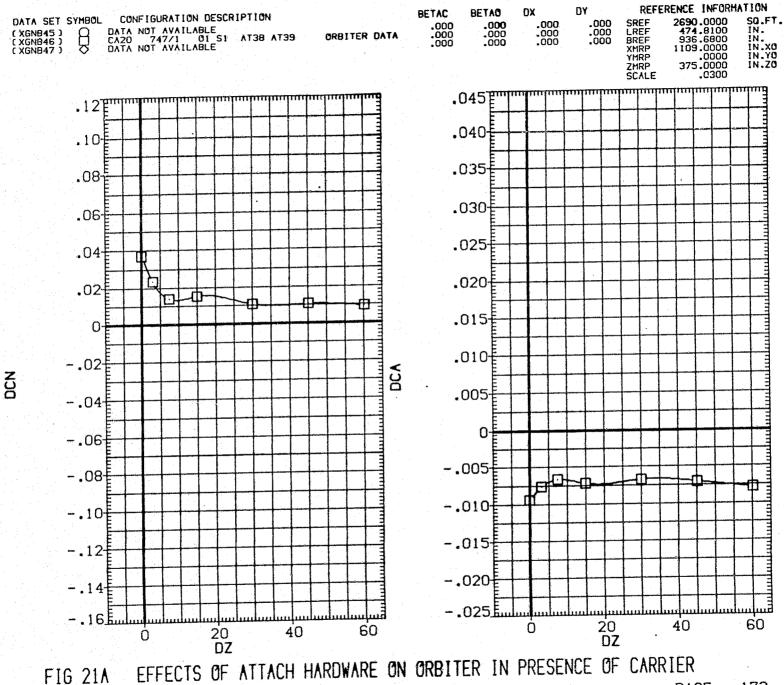


FIG 21A 172 PAGE (B)ALPHA0= 12.00

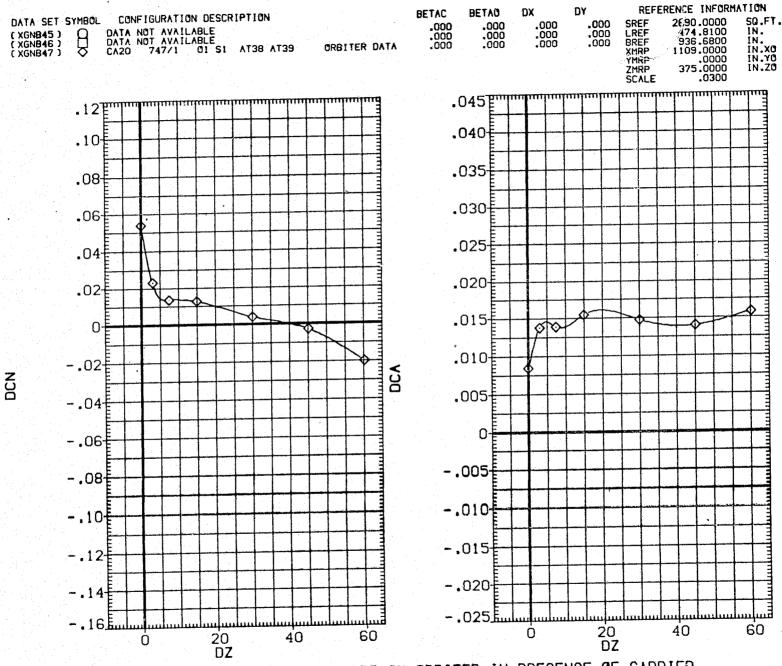
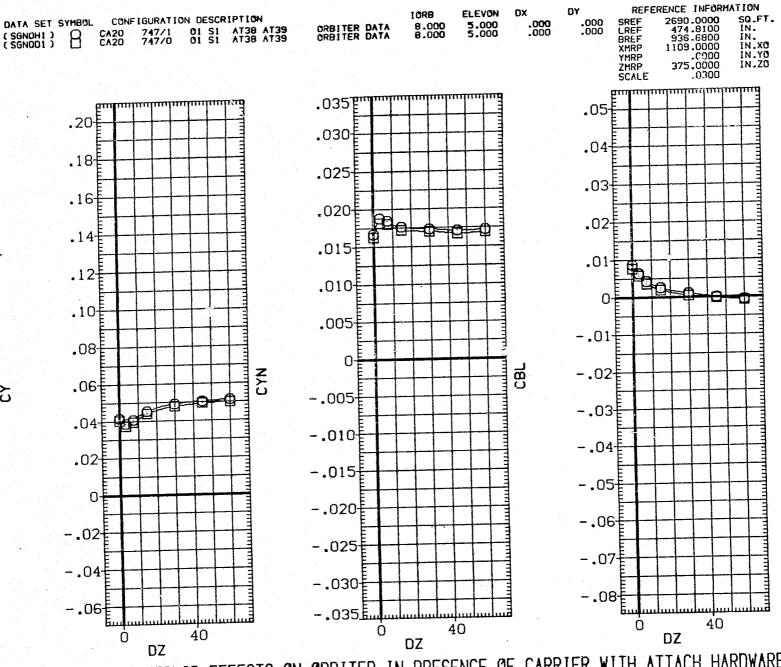


FIG 21A EFFECTS OF ATTACH HARDWARE ON ORBITER IN PRESENCE OF CARRIER

COALPHAGE 16.00



SIDESLIP EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE PAGE 174

(A)BETA0 = -5.00

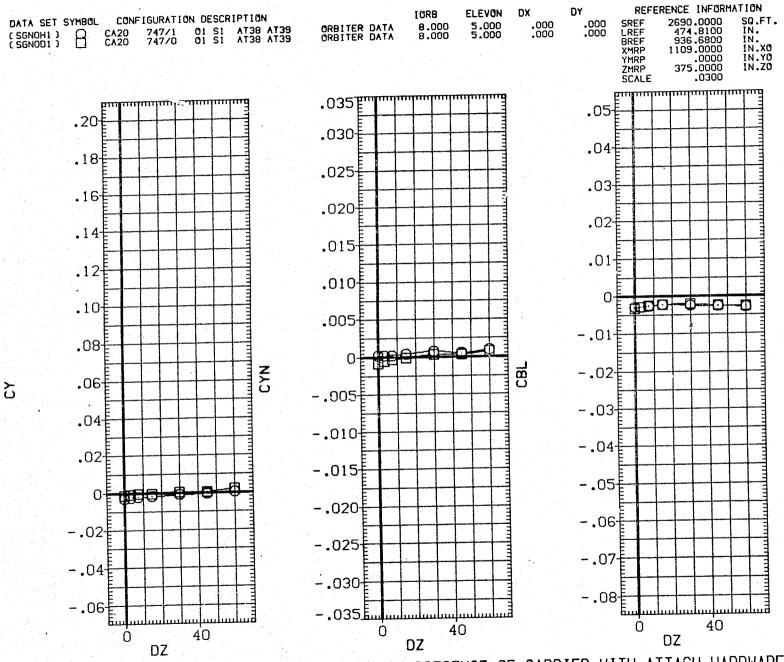


FIG 21B SIDESLIP EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B)BETAO = .00

PAGE 175

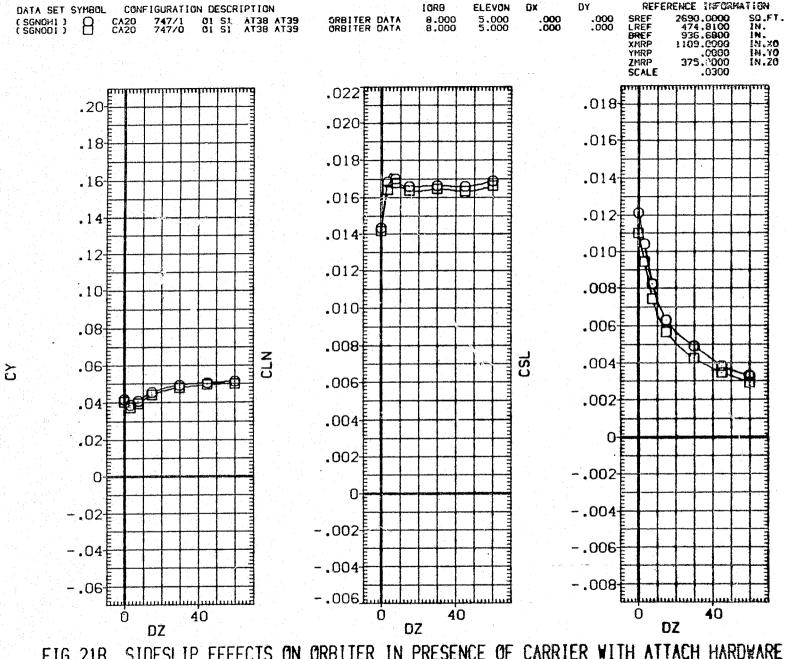


FIG 21B SIDESLIP EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)BETAO = -5.00

PAGE 176

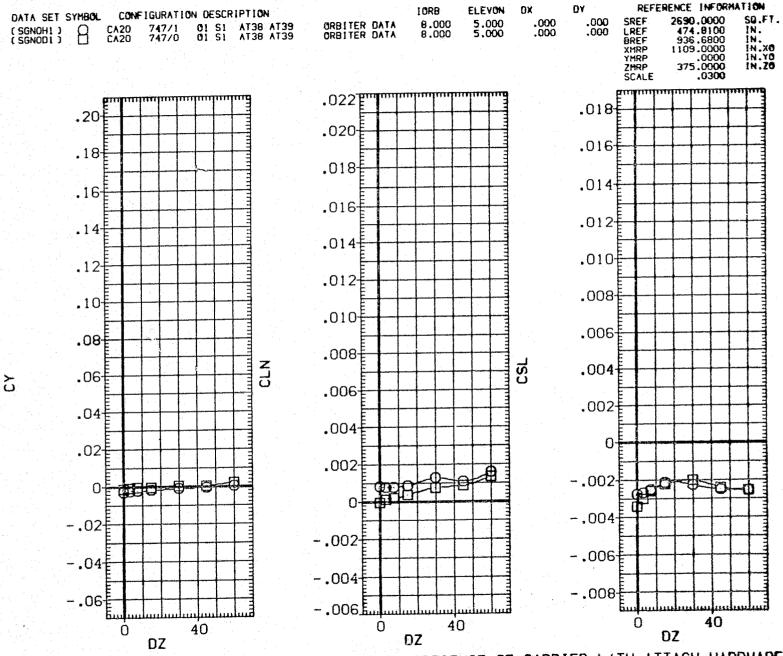


FIG 21B SIDESLIP EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B)BETAG = .00

PAGE 177

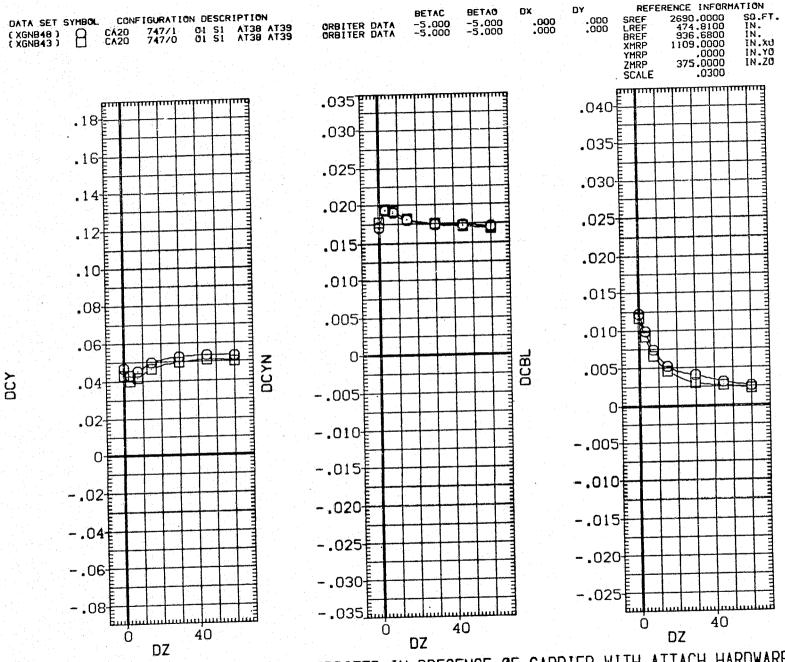


FIG 21B SIDESLIP EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

PAGE 178

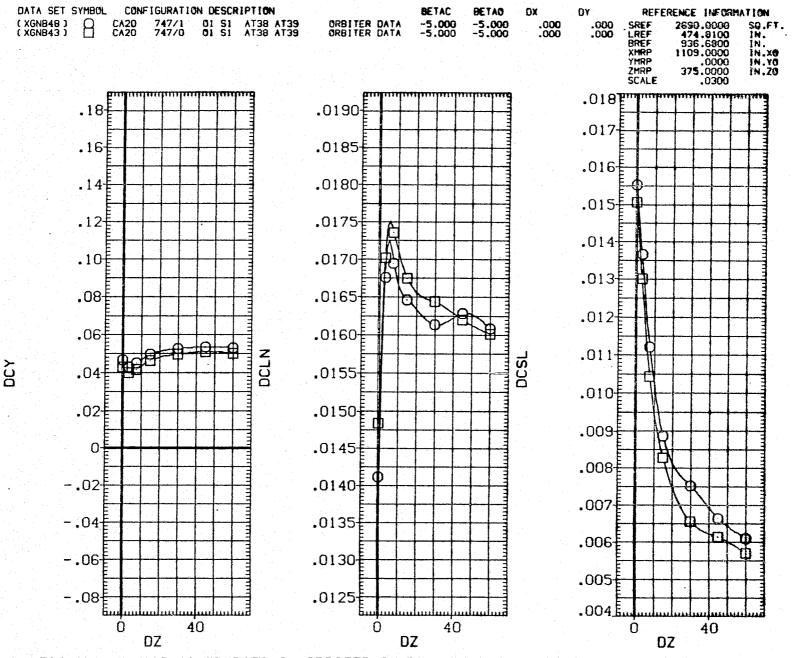


FIG 21B SIDESLIP EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE
(A)ALPHAO= 12.00
PAGE 179

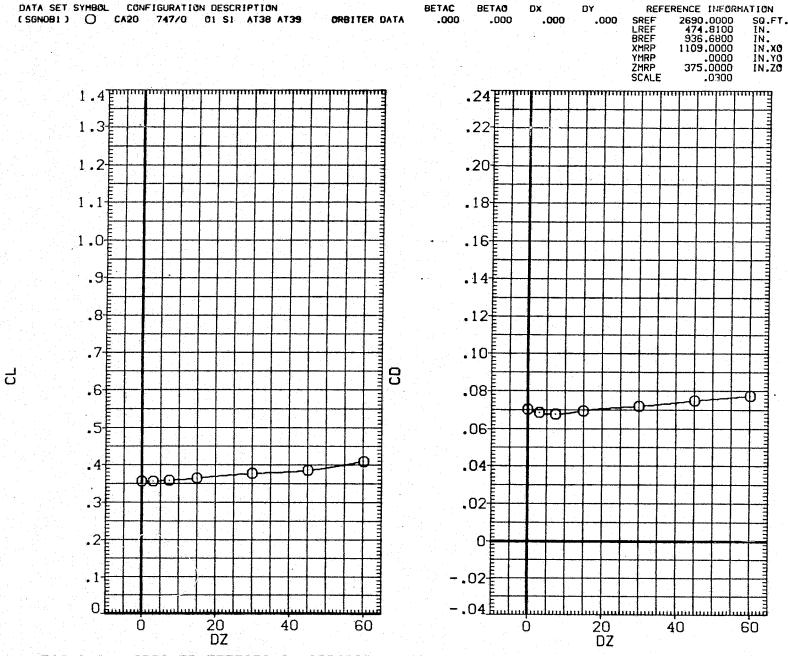


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)ALPHAO= 8.00

PAGE 180

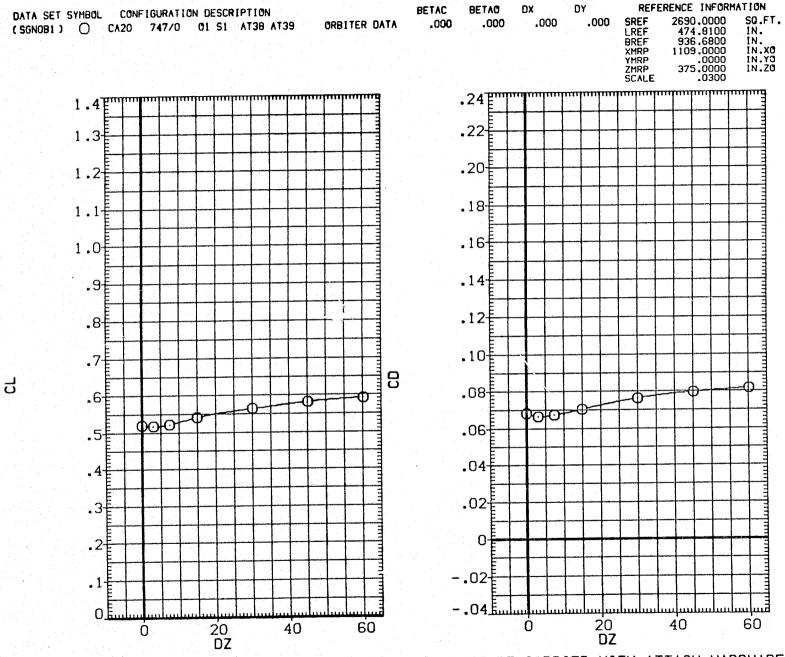


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B) ALPHAO = 12.00

PAGE 181

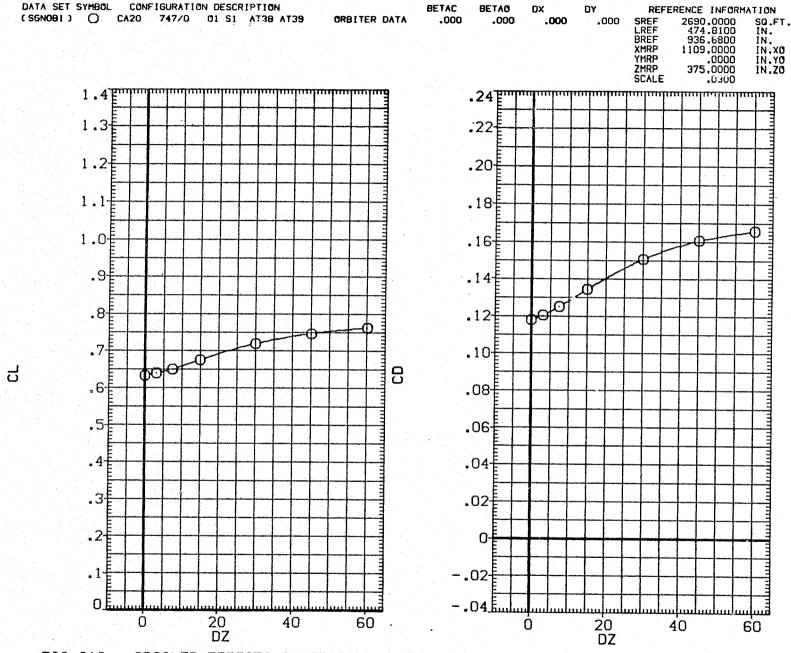
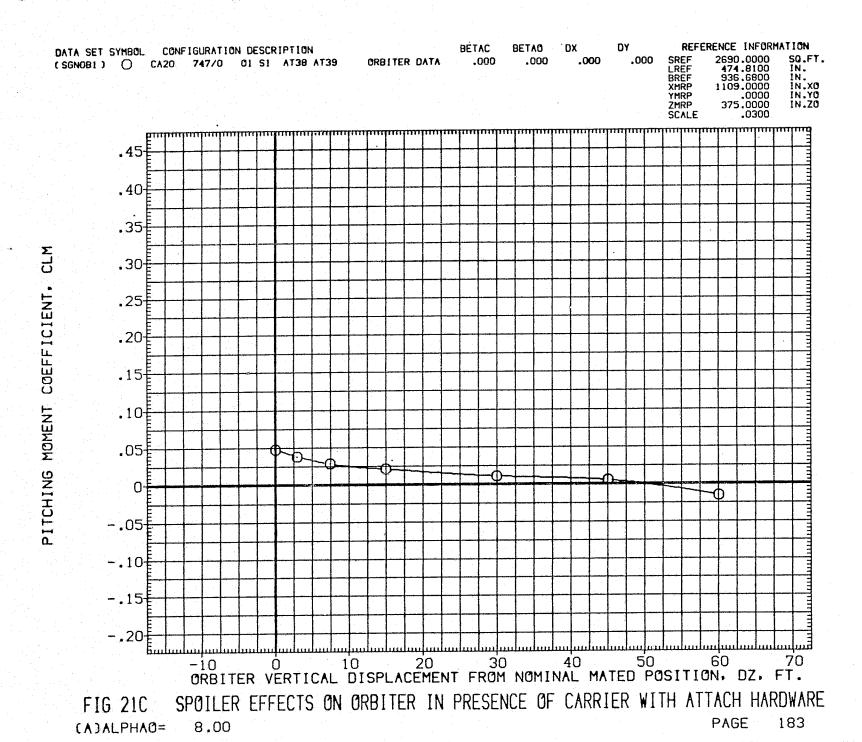
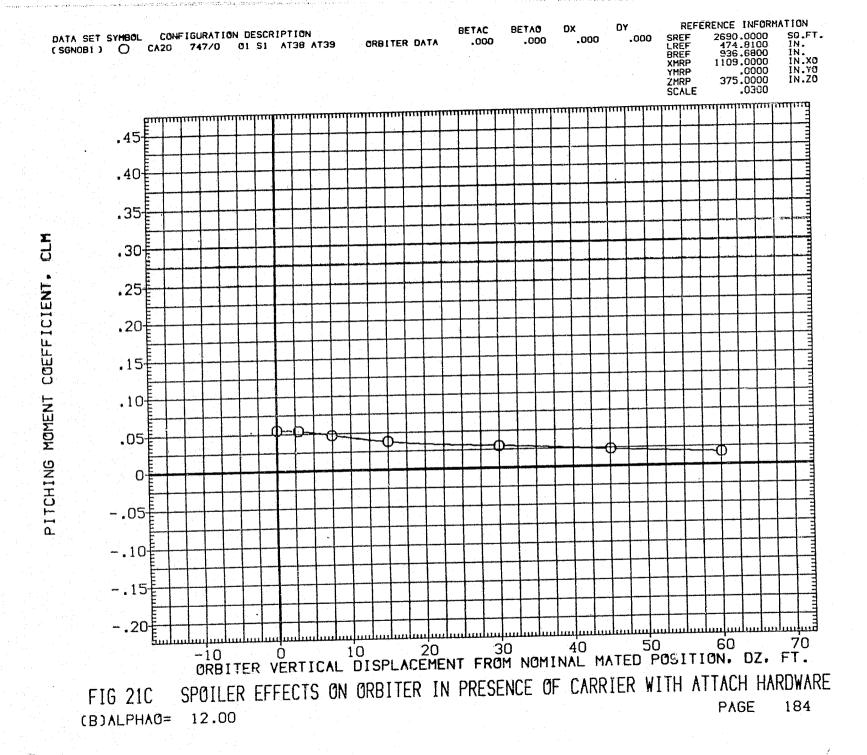


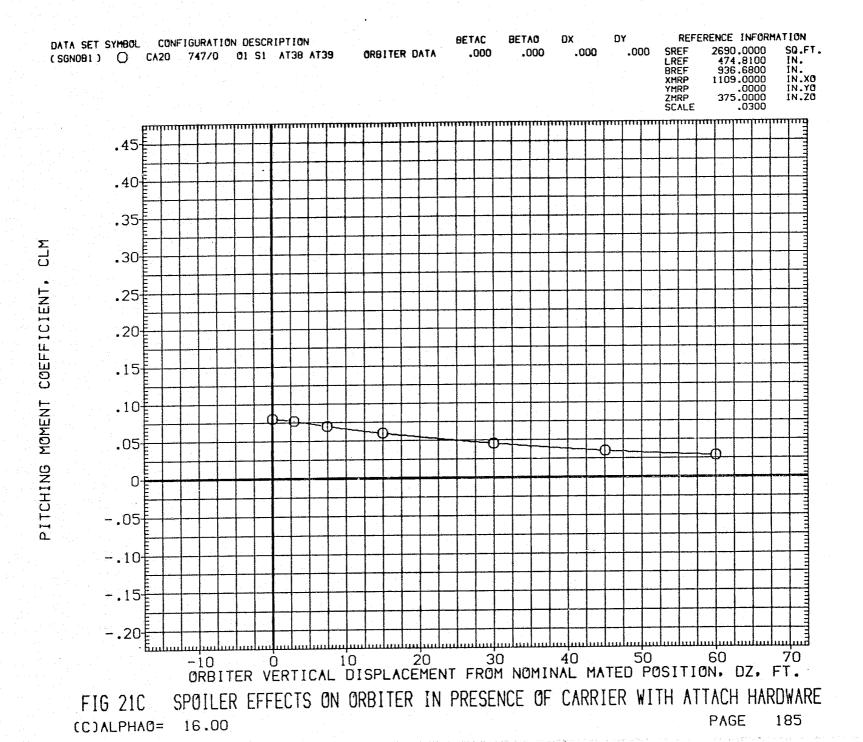
FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(C)ALPHAO= 16.00

PAGE 182







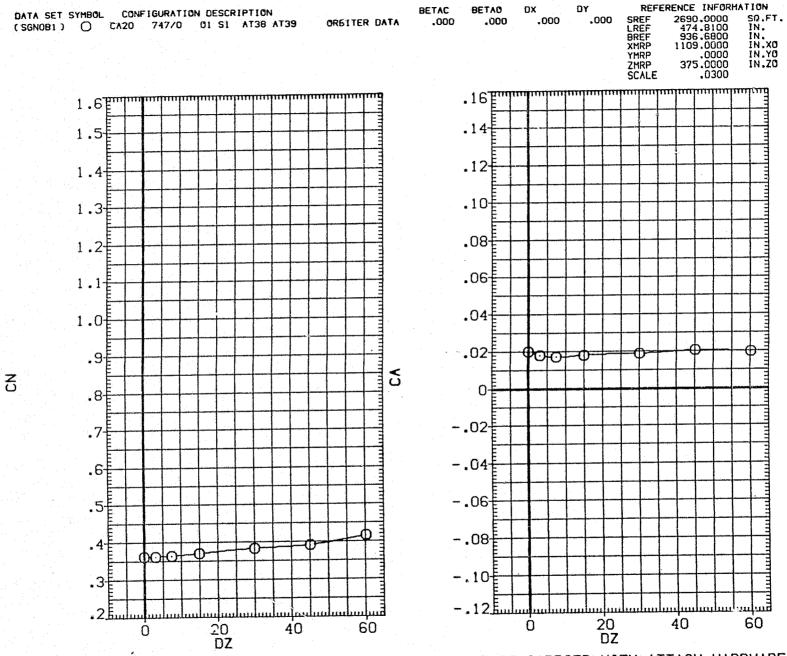


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)ALPHAO= 8.00

PAGE 186

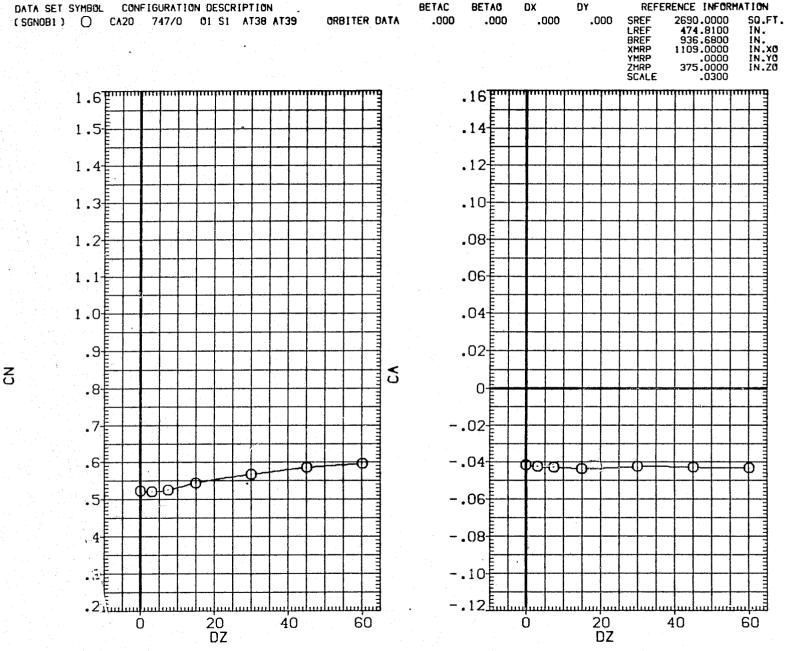


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B) ALPHAO: 12.00 PAGE 187

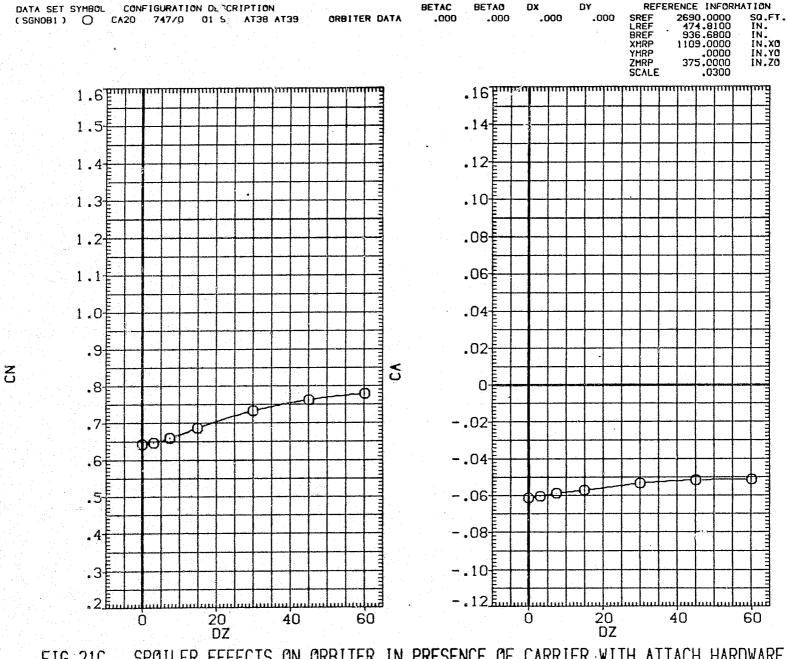
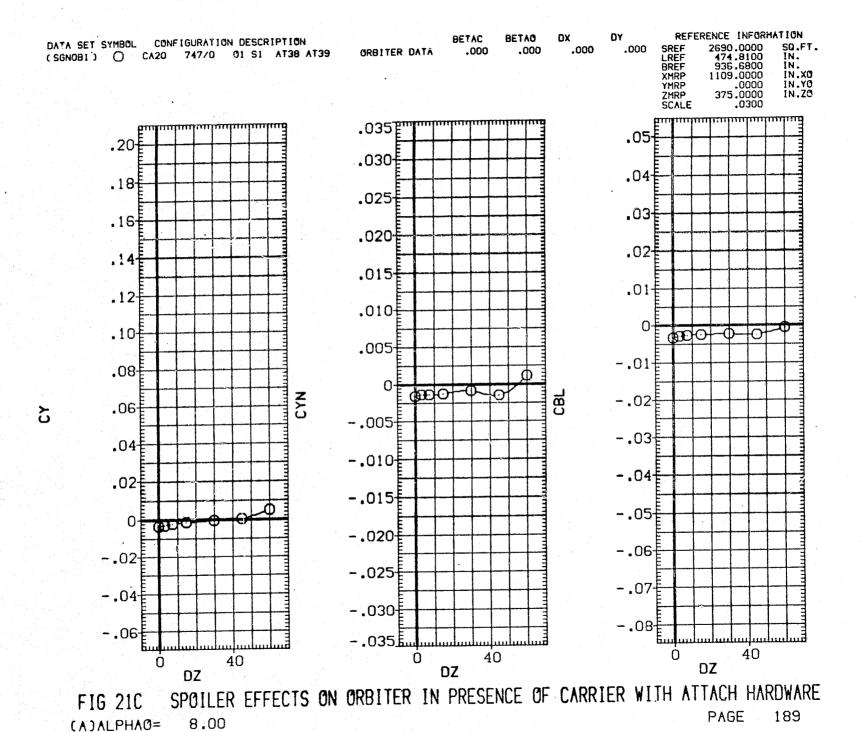


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(C)ALPHAO= 16.00 PAGE 188

The same



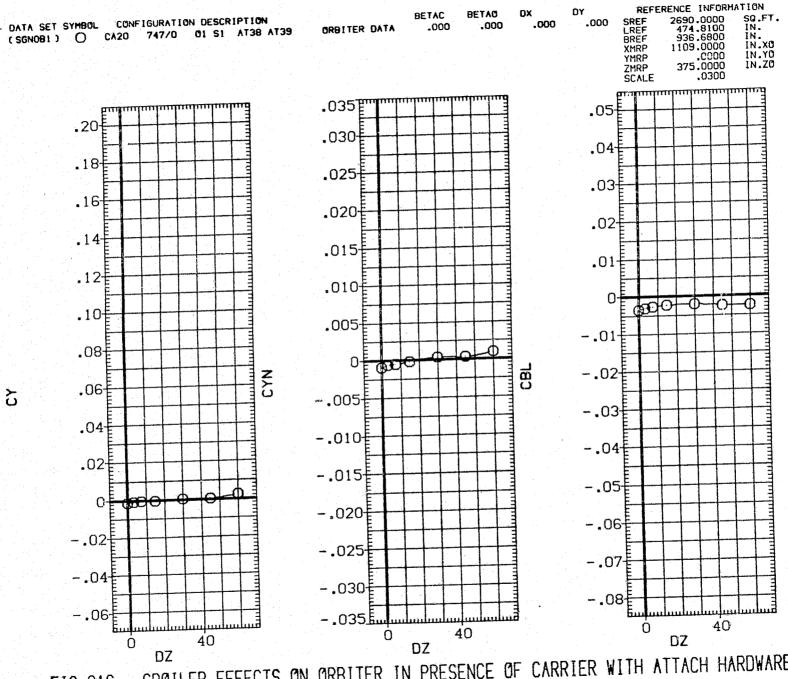


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B) ALPHAO = 12.00

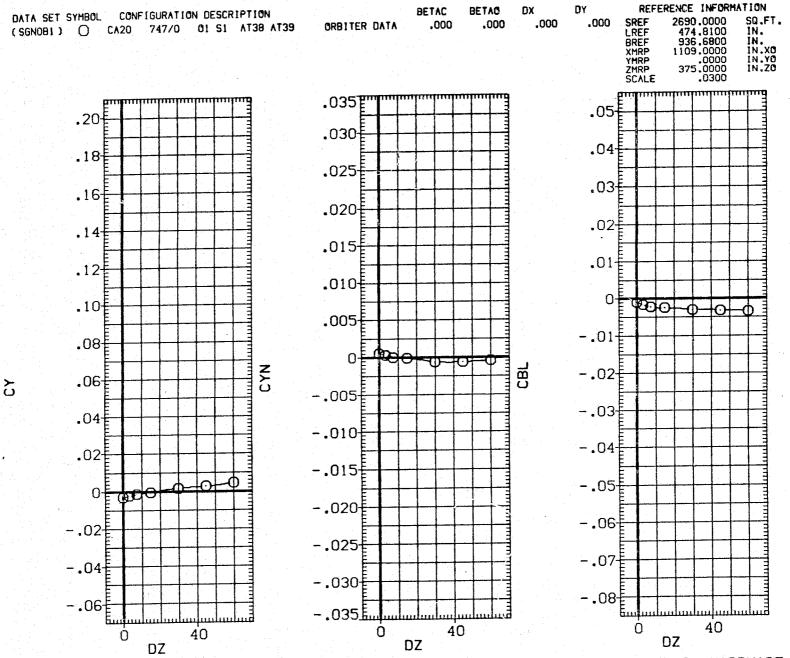


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(C)ALPHAO= 16.00

PAGE 191

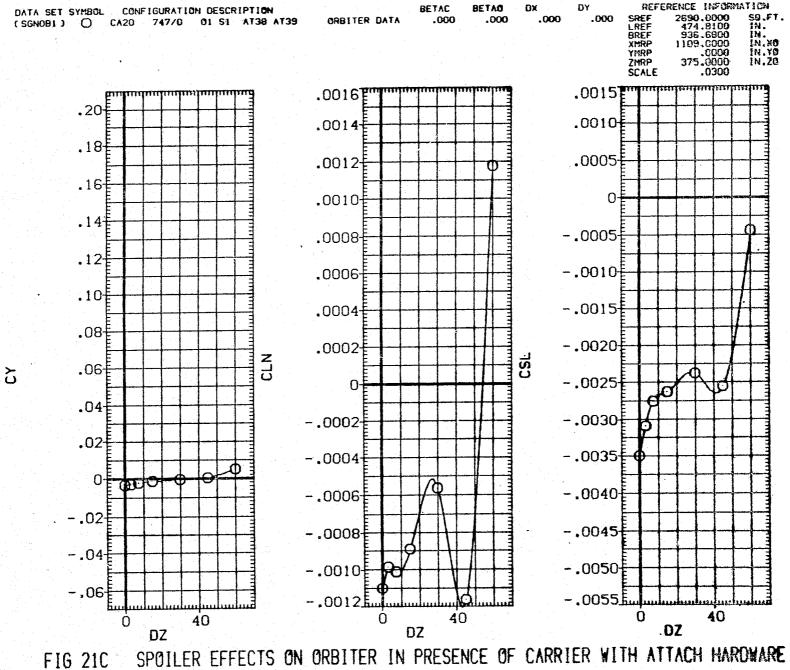


FIG 21C 192 PAGE (A)ALPHAO= 8.00

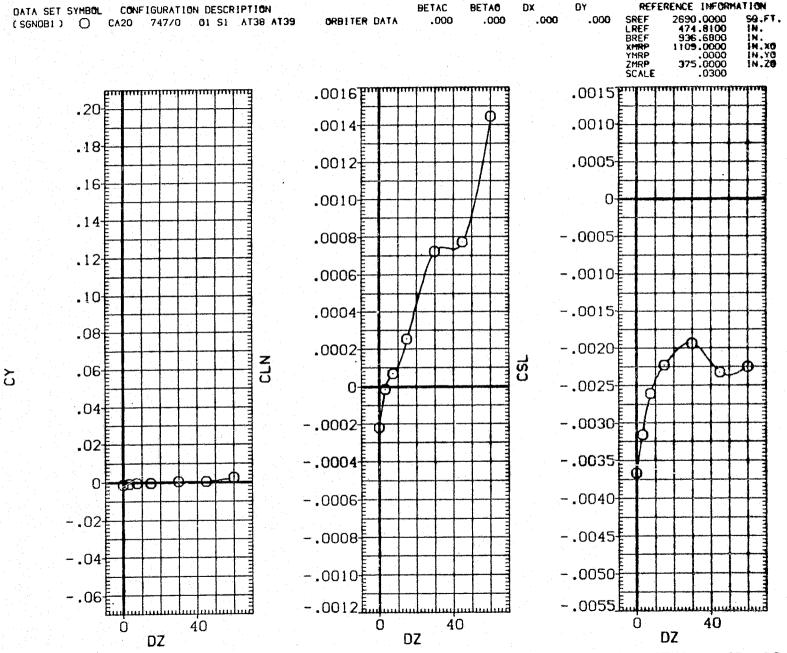


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B)ALPHAO= 12.00 PAGE 193

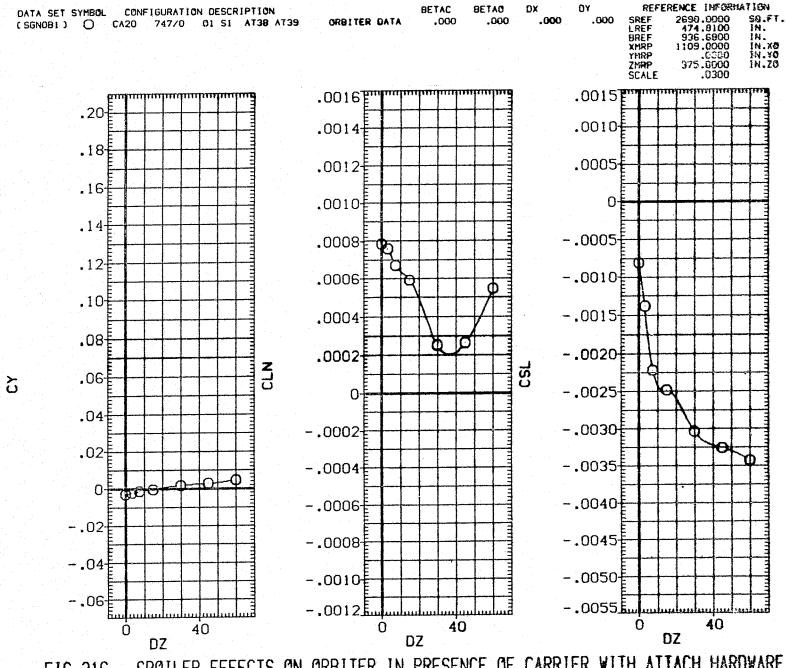


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE (C)ALPHAO= 16.00 PAGE 194

-

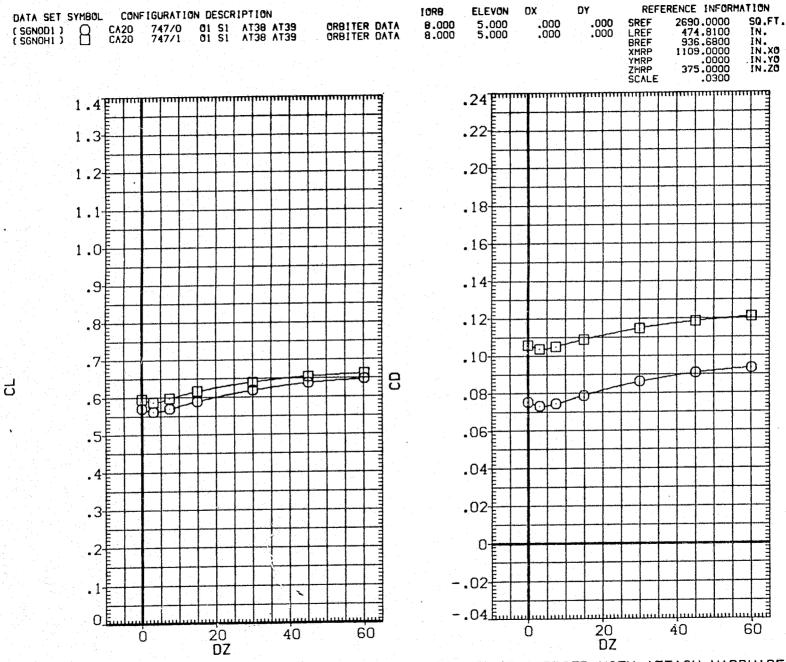
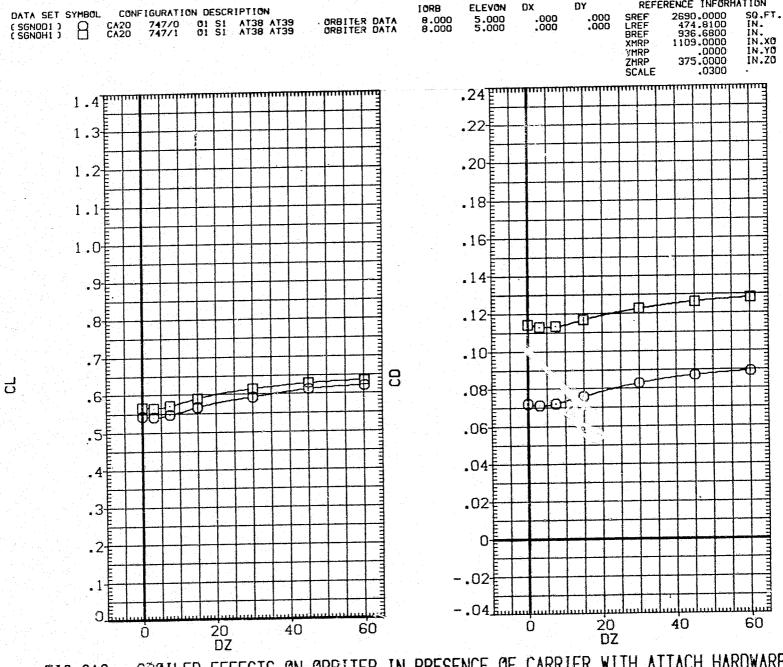


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

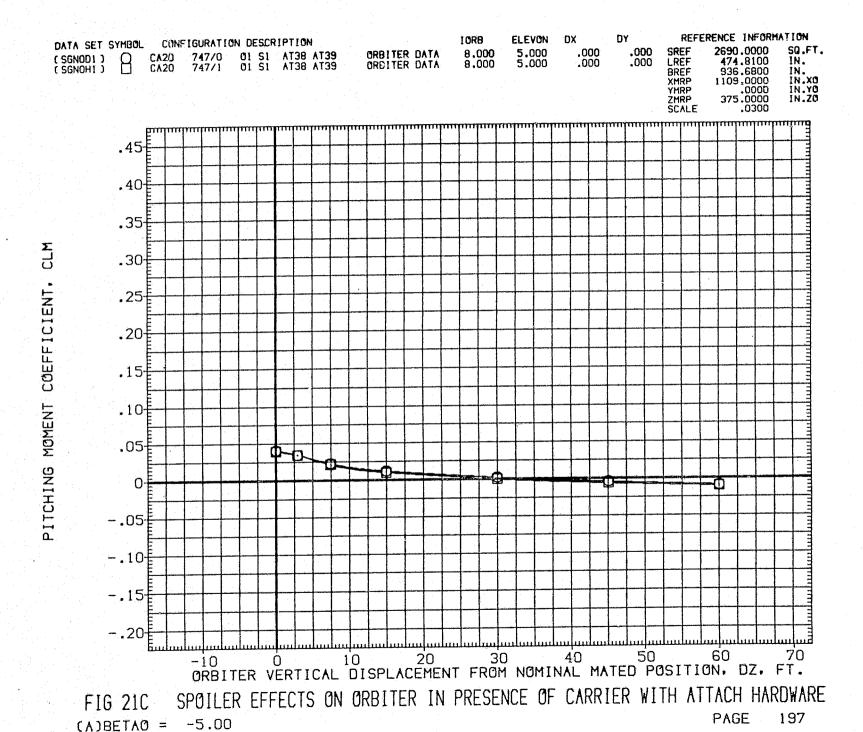
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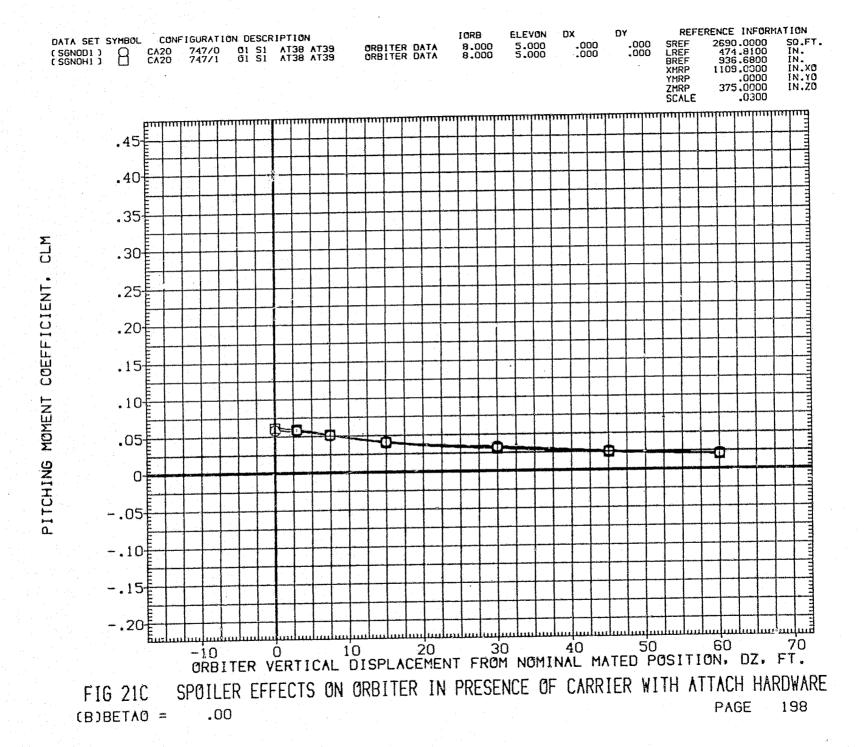
PAGE 195



REFERENCE INFORMATION

SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE FIG 21C PAGE 196 .00 (B)BETAO =





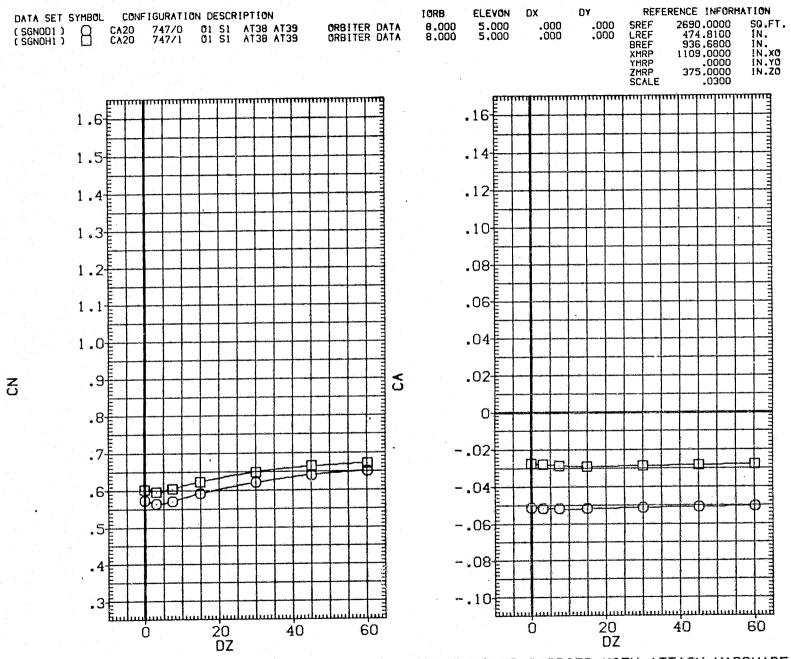
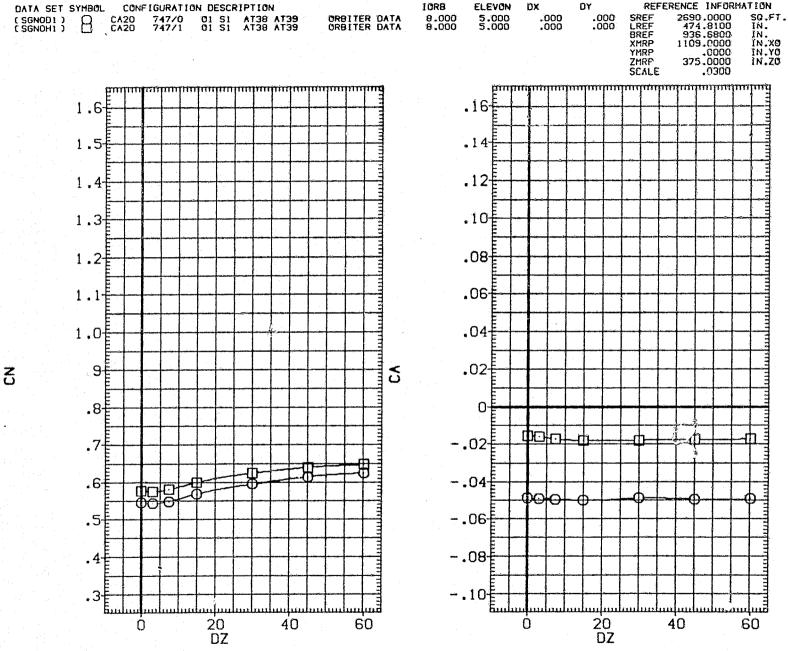


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)BETAO = -5.00

PAGE 199



dasai arrawa 🖟 ra deno deno de medico de recurso do miderio de la desenva da recurso da la desenva de racionar de la composició del composició de la composició del composició de la composició de la composició del composició de

SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE FIG 21C PAGE 200 .00 (B)BETAO =

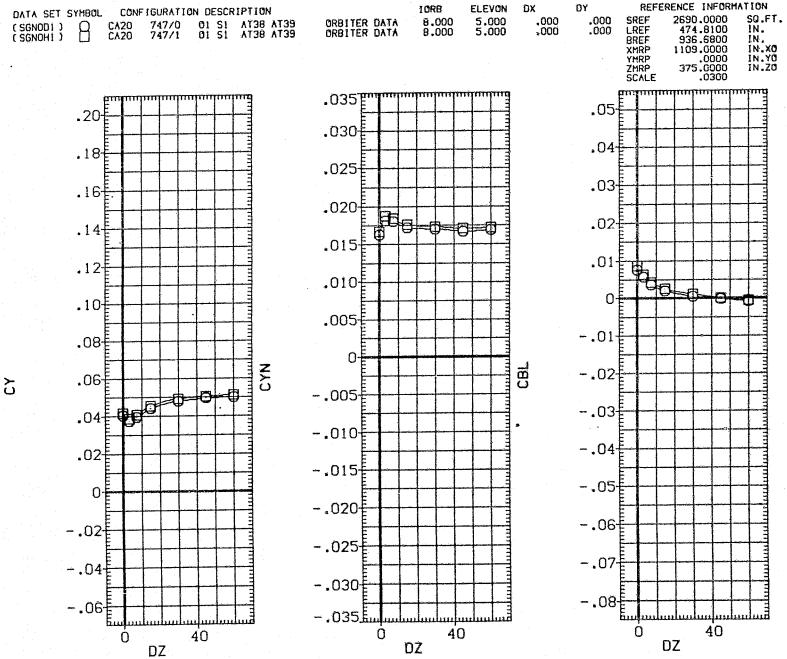


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)BETAO = -5.00

PAGE 201

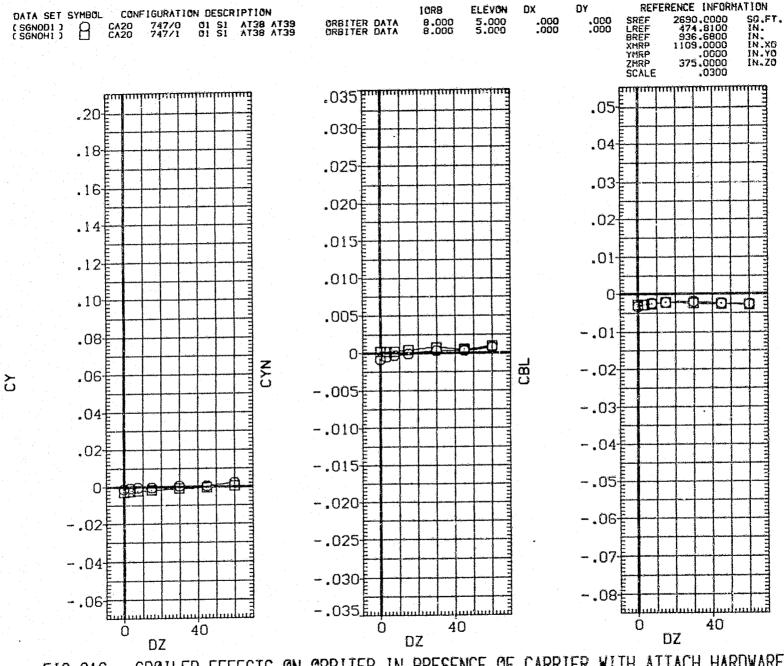
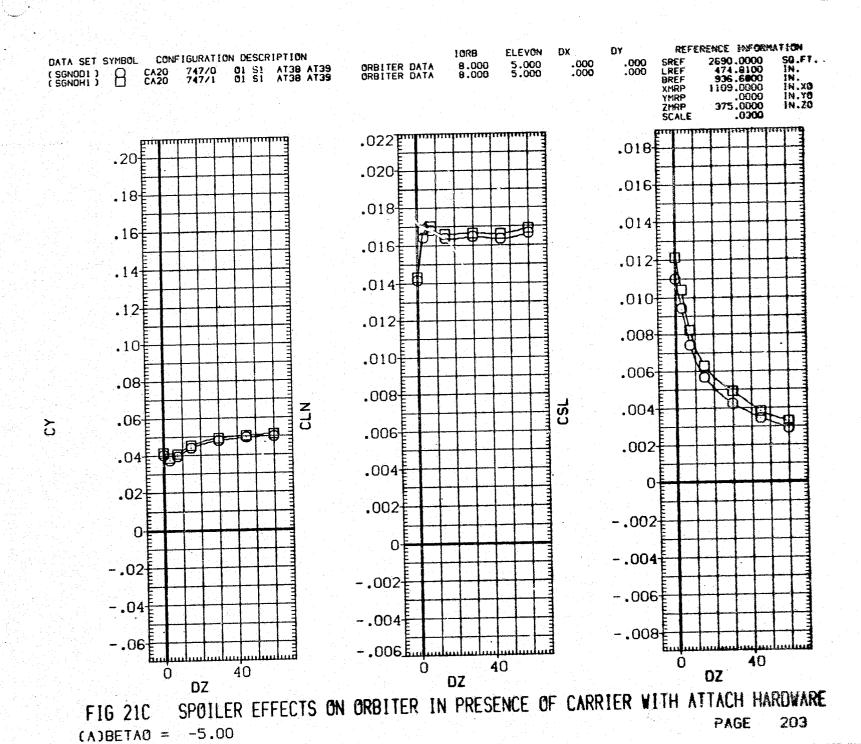
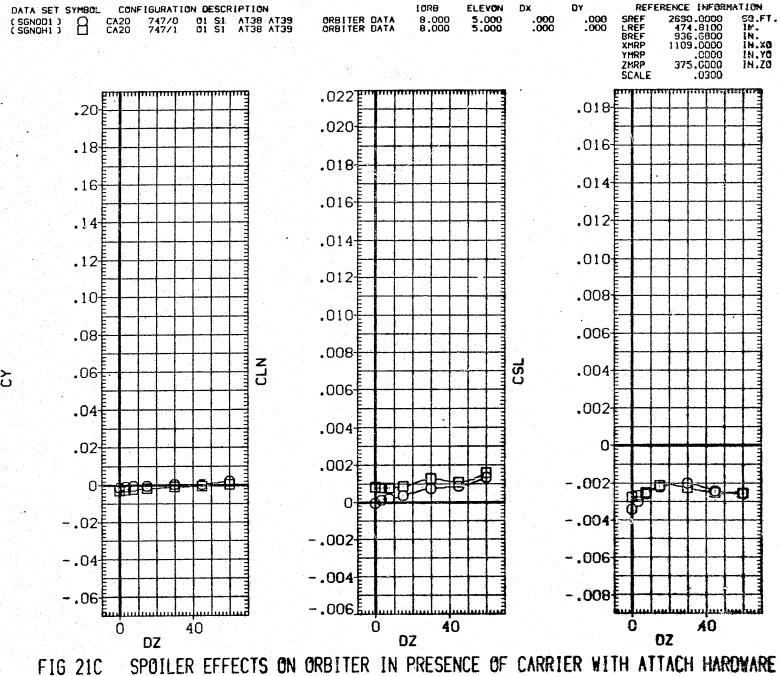


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B)BETAO = .00

PAGE 202





PAGE 204 (B)BETAO = .00

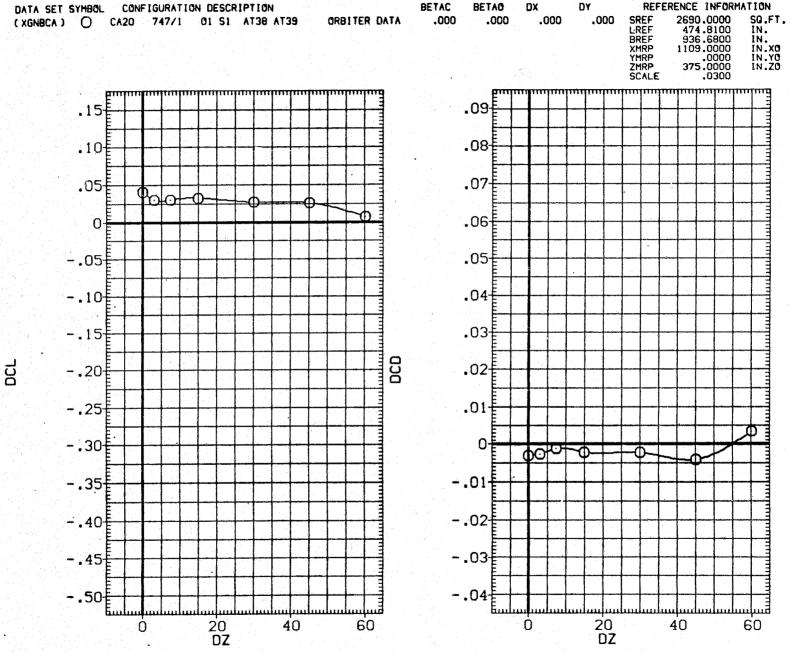


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)ALPHAO= 8.00

PAGE 205

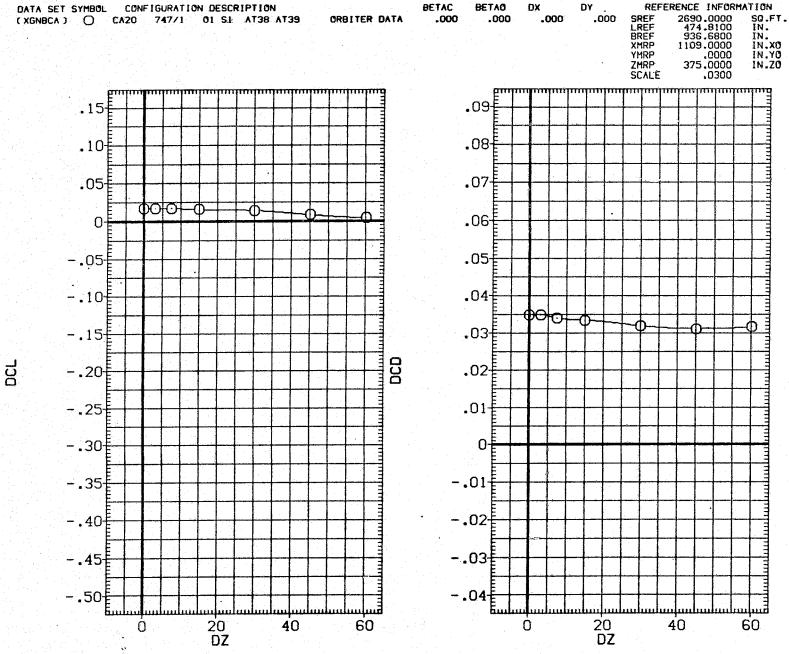


FIG 21C .SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE
(B)ALPHAO= 12.00 PAGE 206

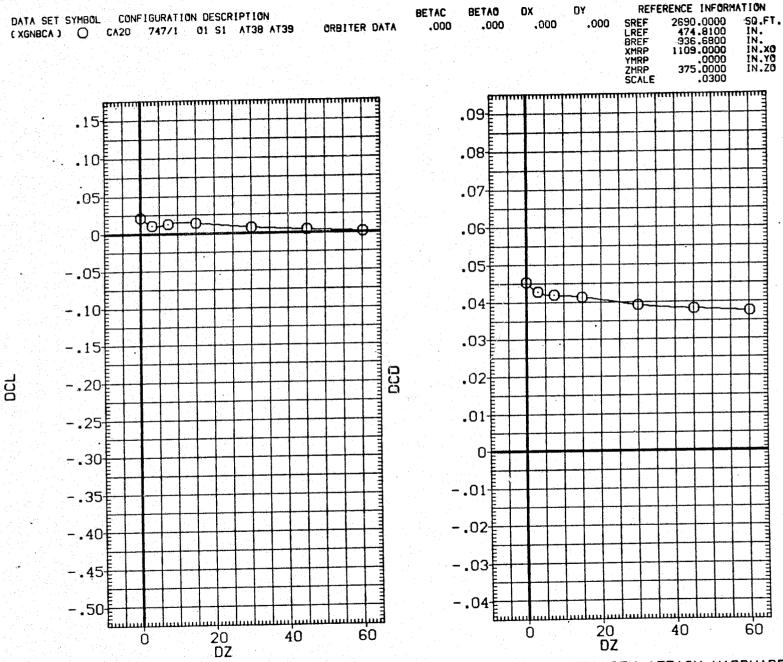
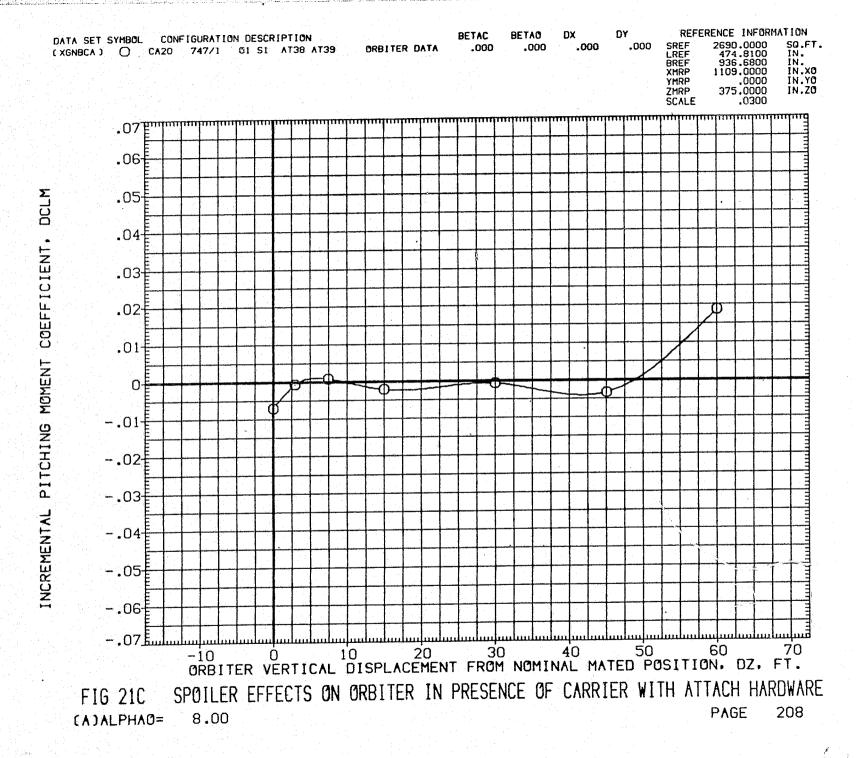
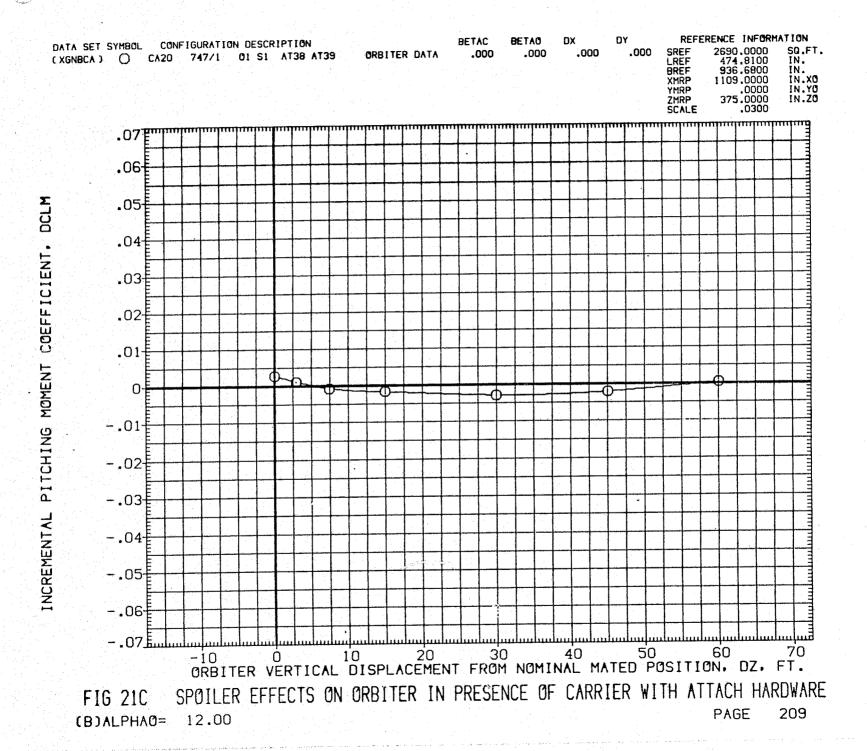


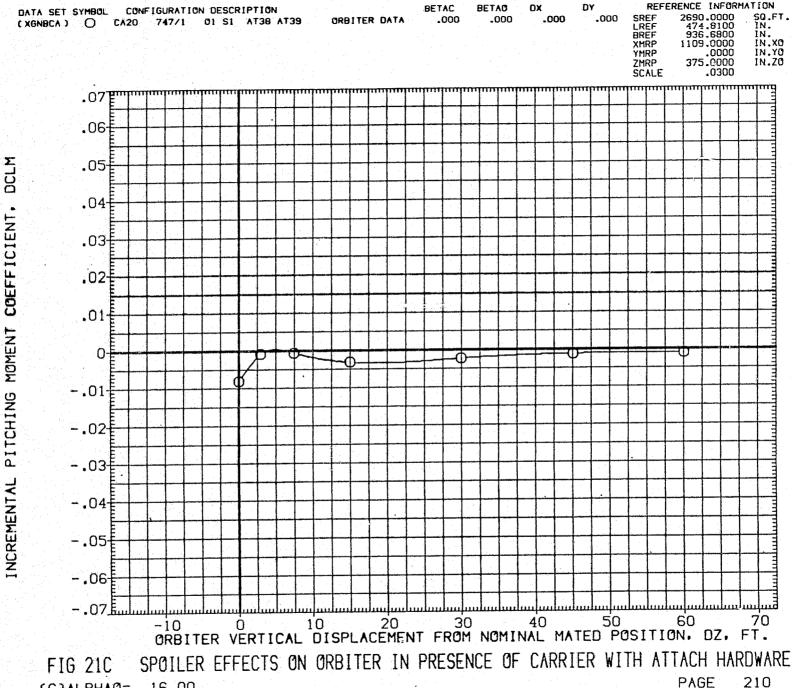
FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(C)ALPHAO= 16.00

PAGE 207







PAGE 210 (C)ALPHA0= 16.00

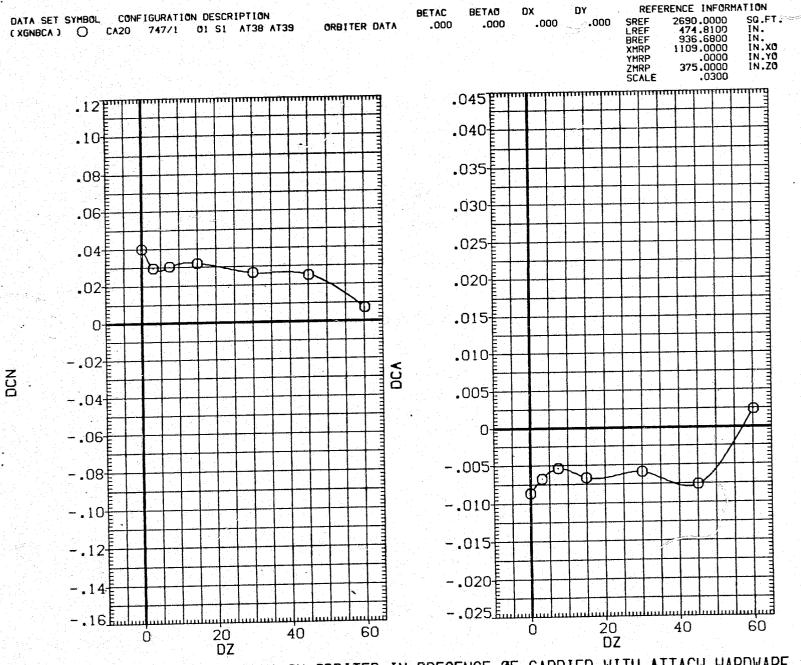
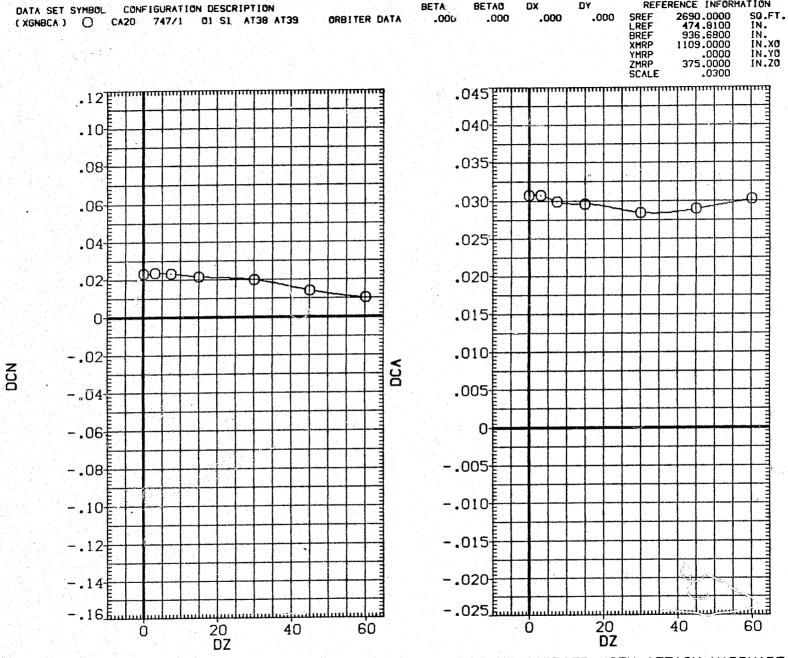


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)ALPHAO= 8.00

PAGE 211



REFERENCE INFORMATION

DY

SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE FIG 21C PAGE 212 (B)ALPHA0= 12.00

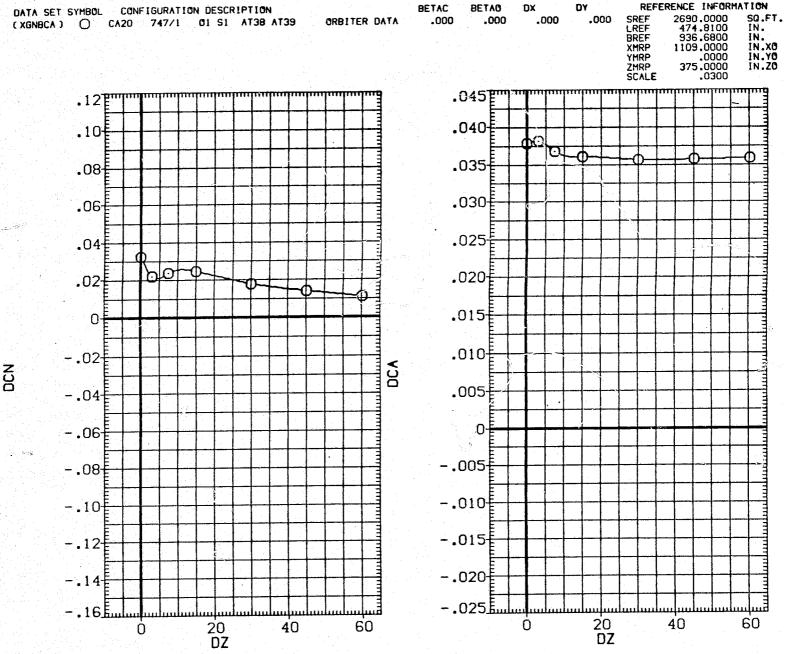


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(C)ALPHAO= 16.00

PAGE 213

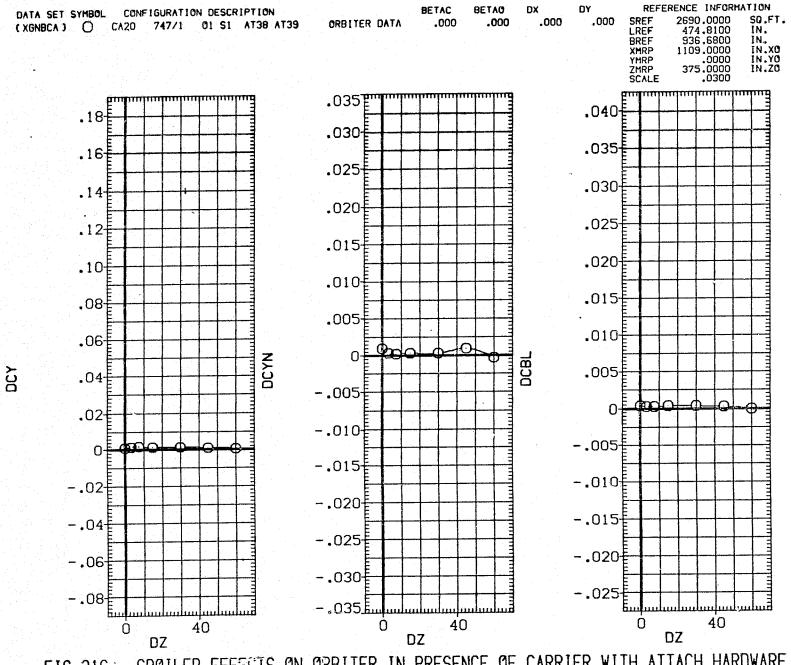


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)ALPHAO= 8.00

PAGE 214

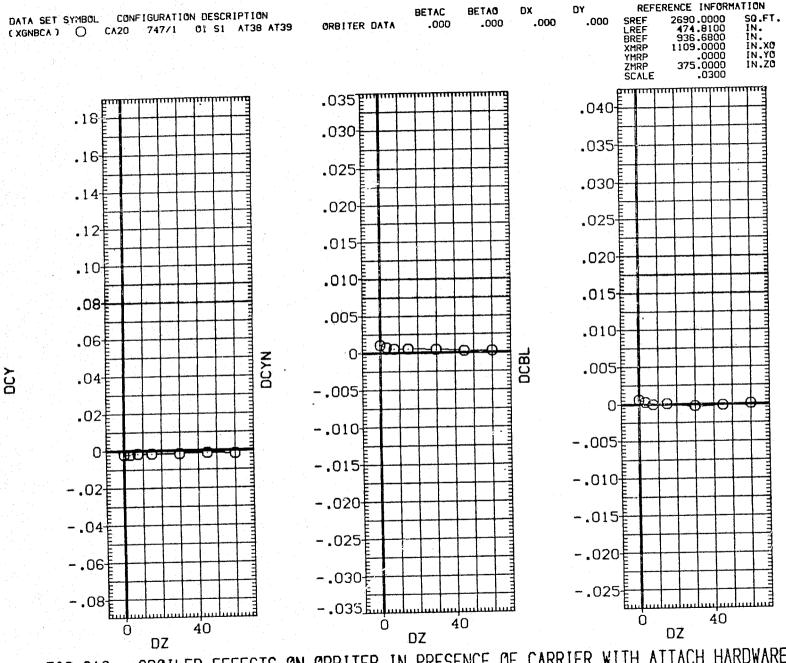


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B) ALPHAO = 12.00

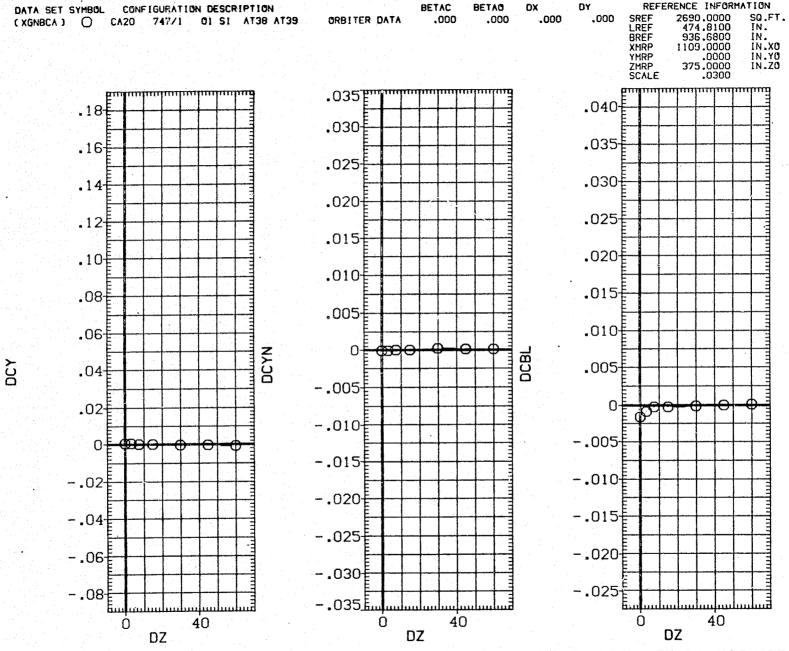


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE (C)ALPHAO= 16.00

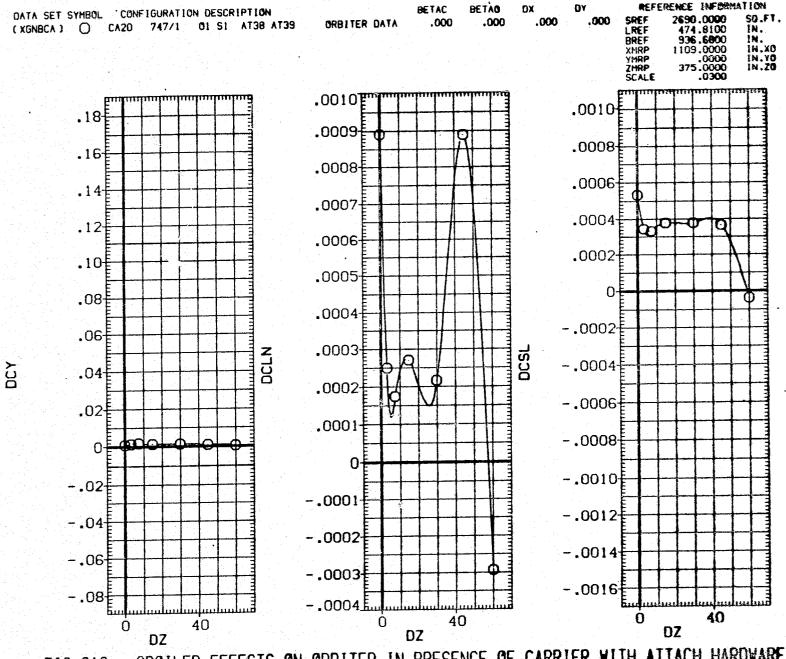


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A) ALPHAO = 8.00

PAGE 217

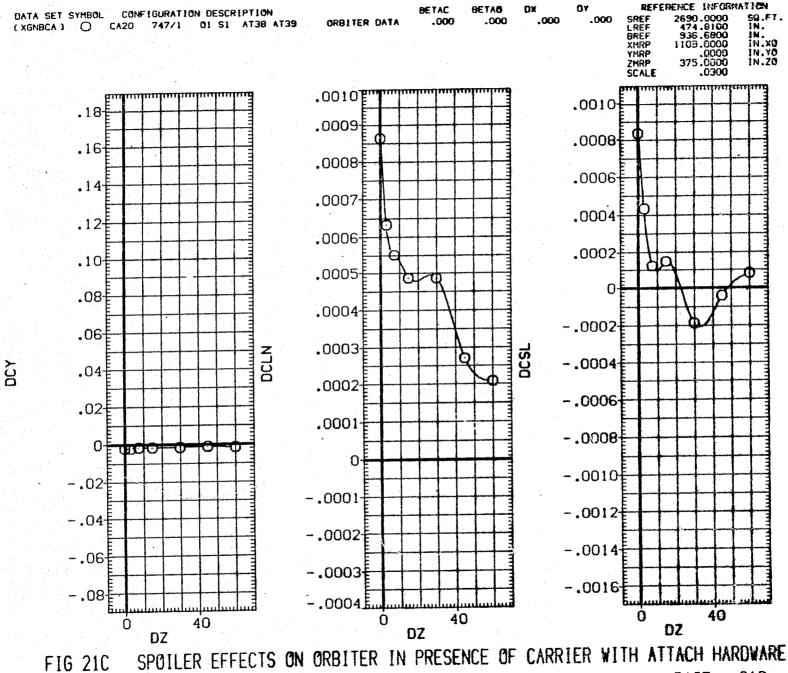


FIG 21C PAGE 218 12.00 (B) ALPHAO=

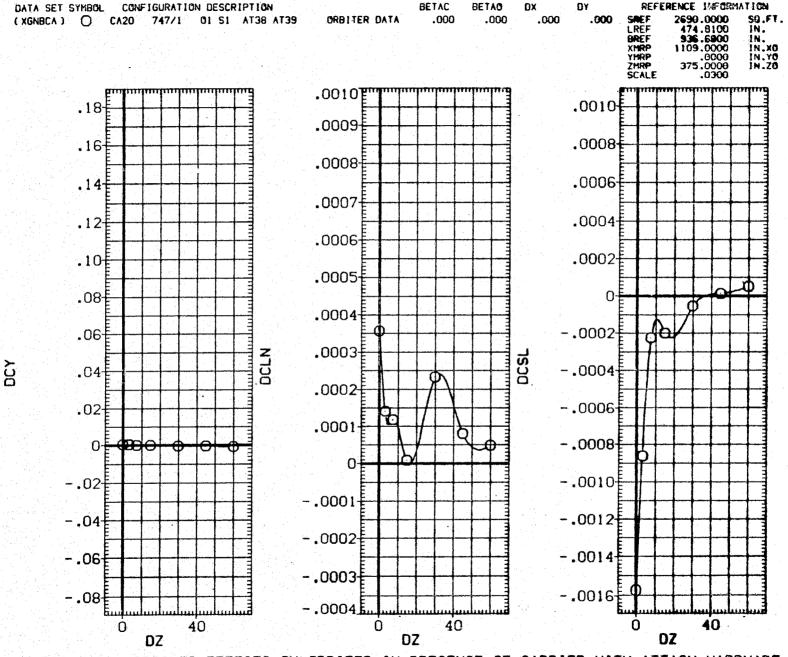


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(C)ALPHAO= 16.00 PAGE 219

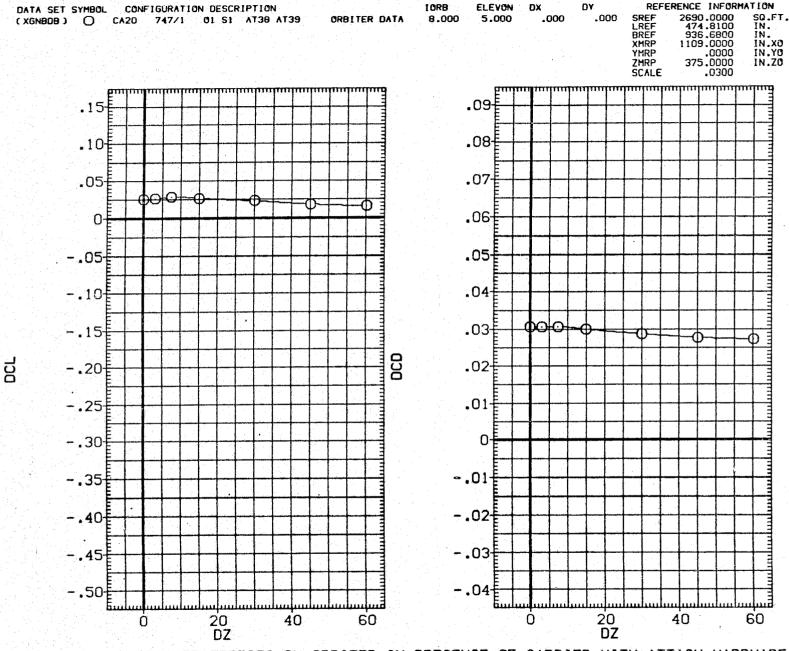
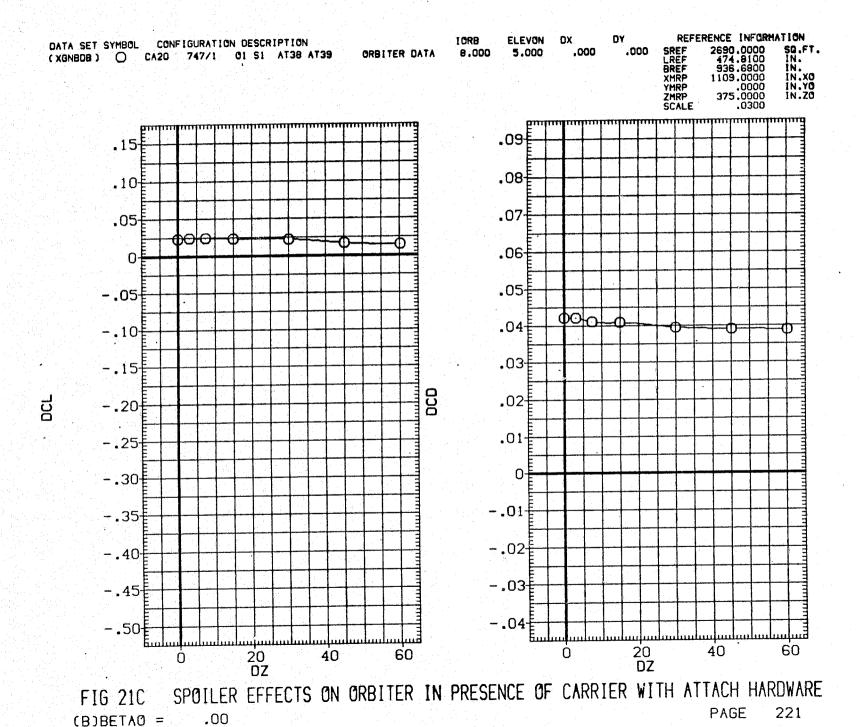
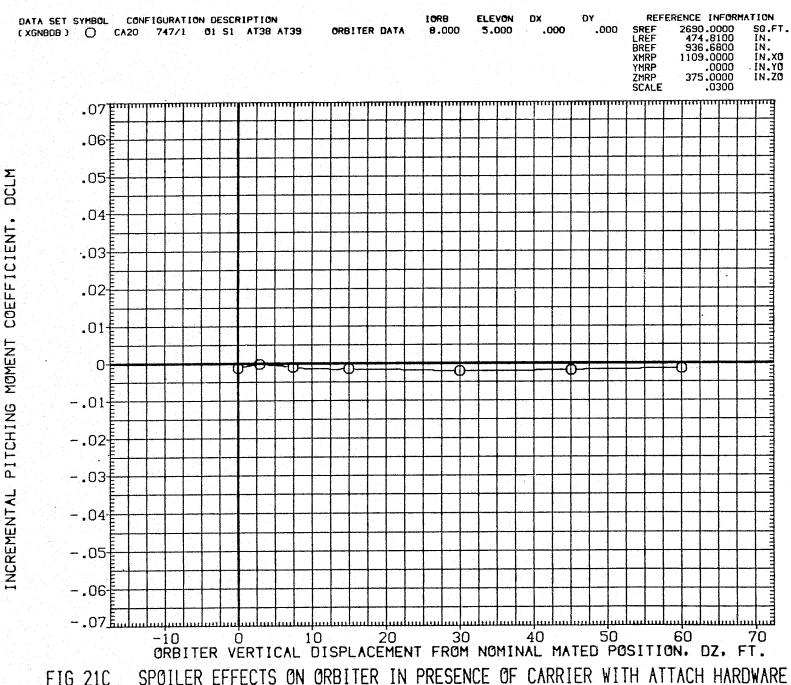


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

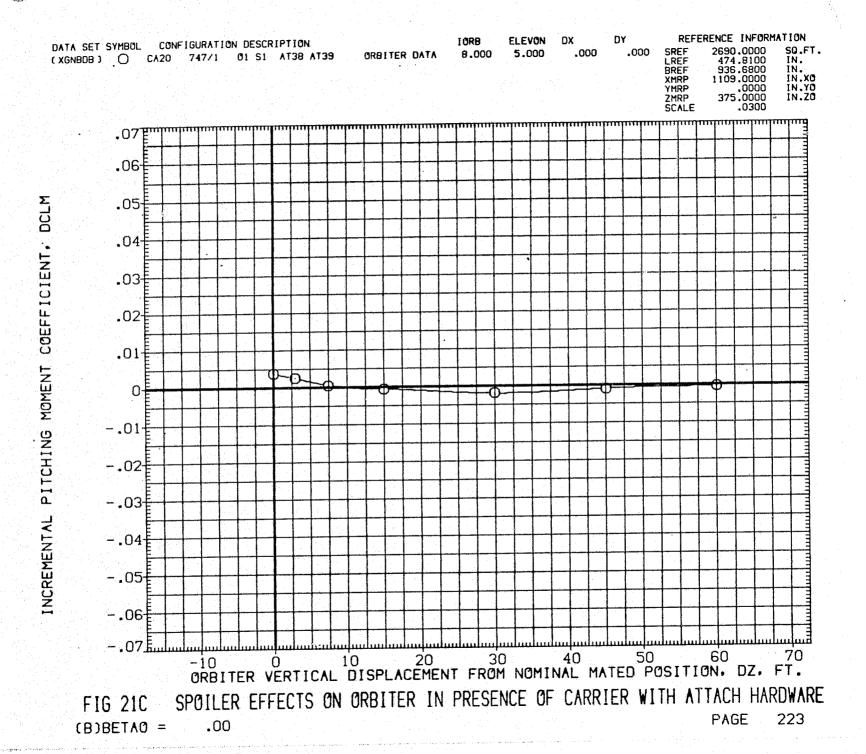
(A)BETAO = -5.00

PAGE 220





SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE FIG 21C 222 PAGE (A)BETAO = -5.00



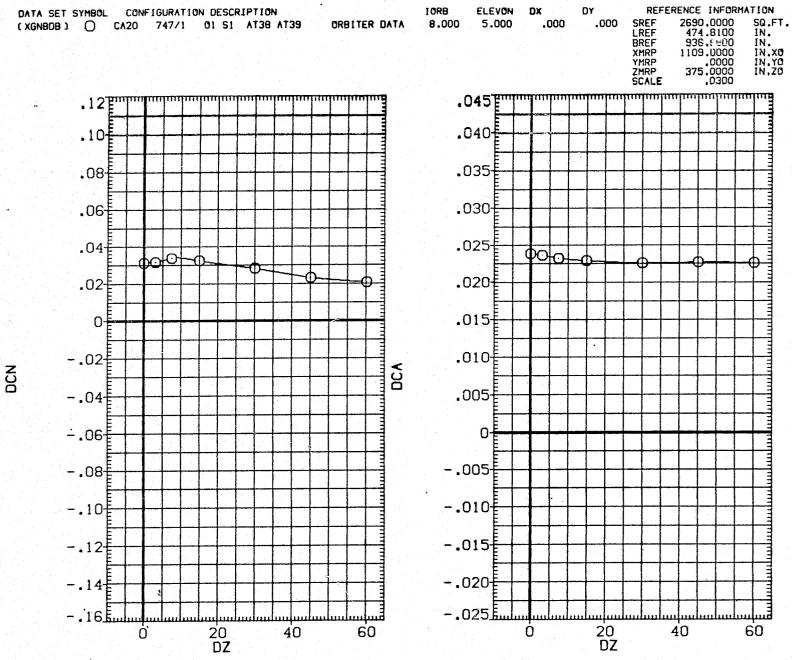


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)BETAO = -5.00

PAGE 224

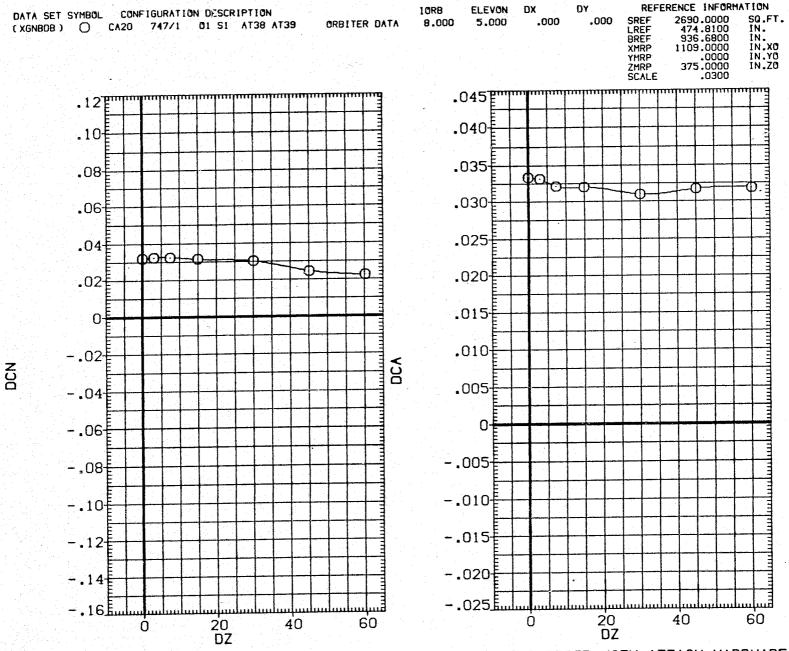


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B)BETAO = .00

PAGE 225

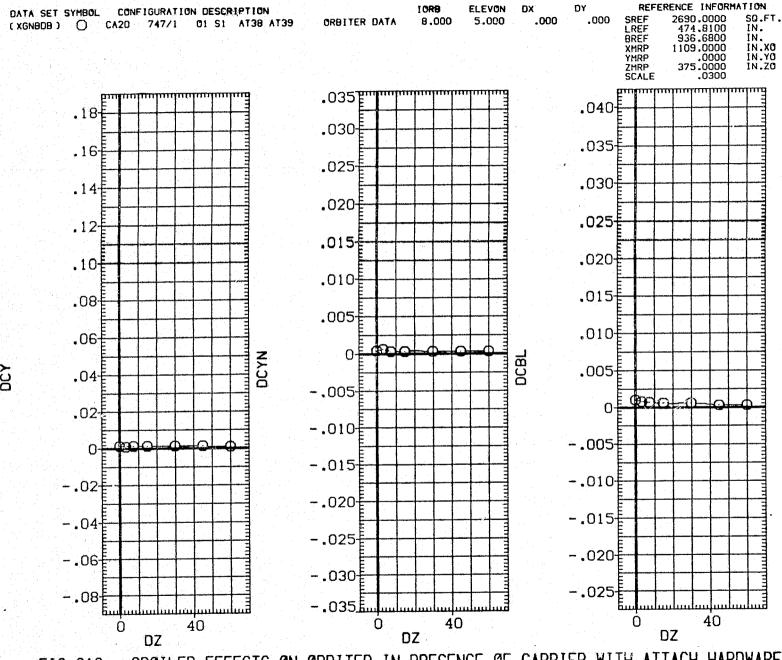


FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(A)BETAO = -5.00

PAGE 226

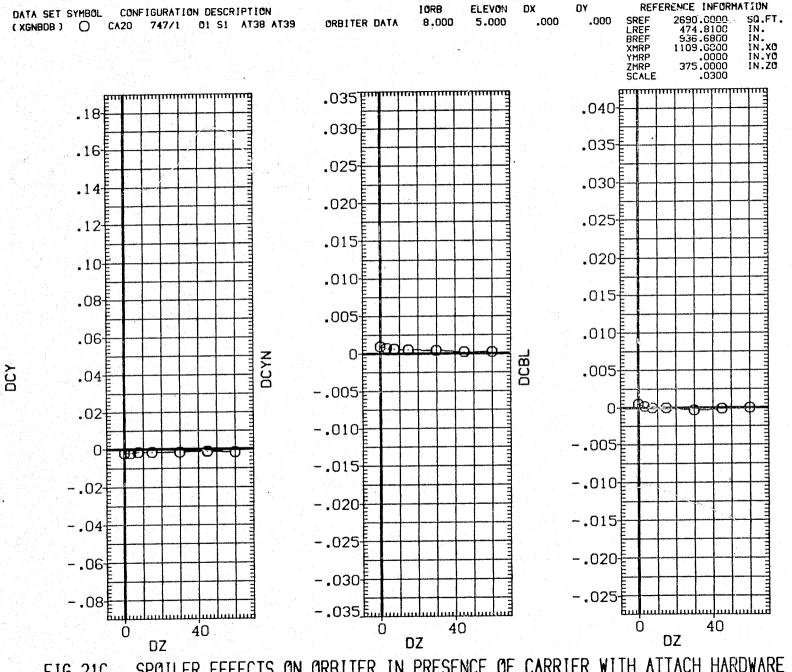
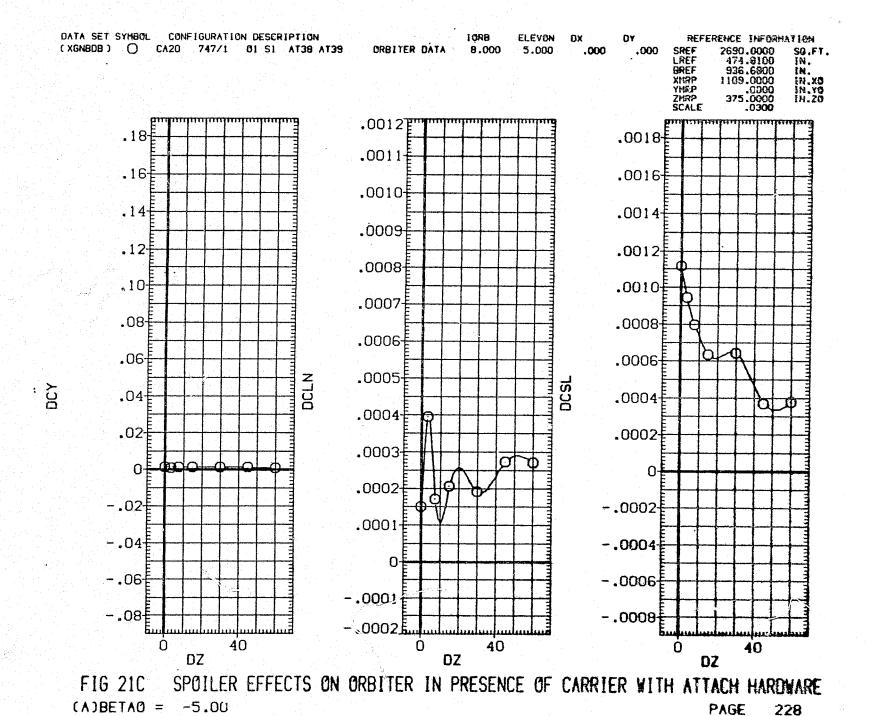
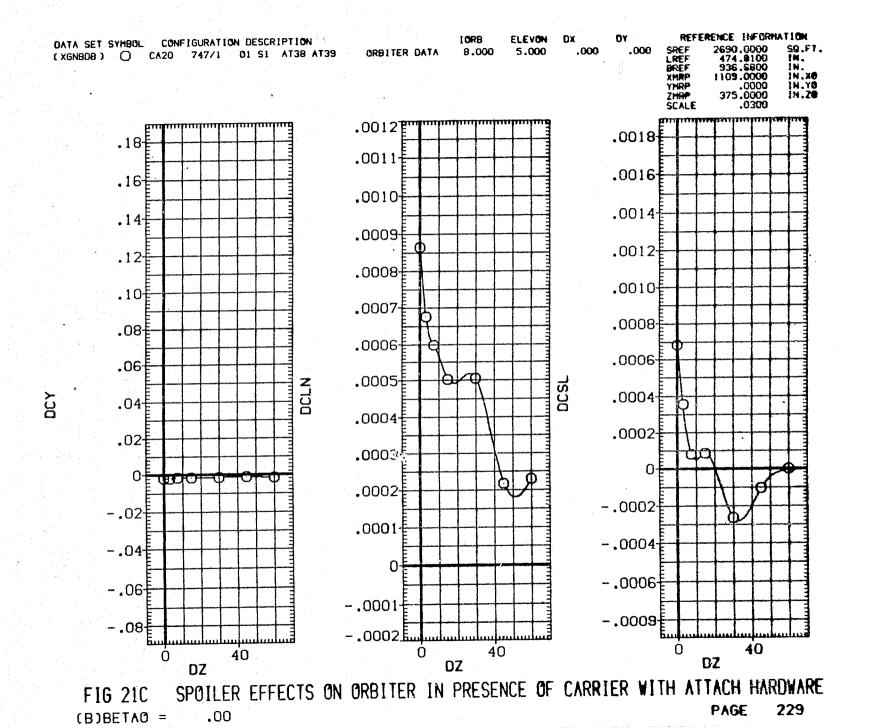


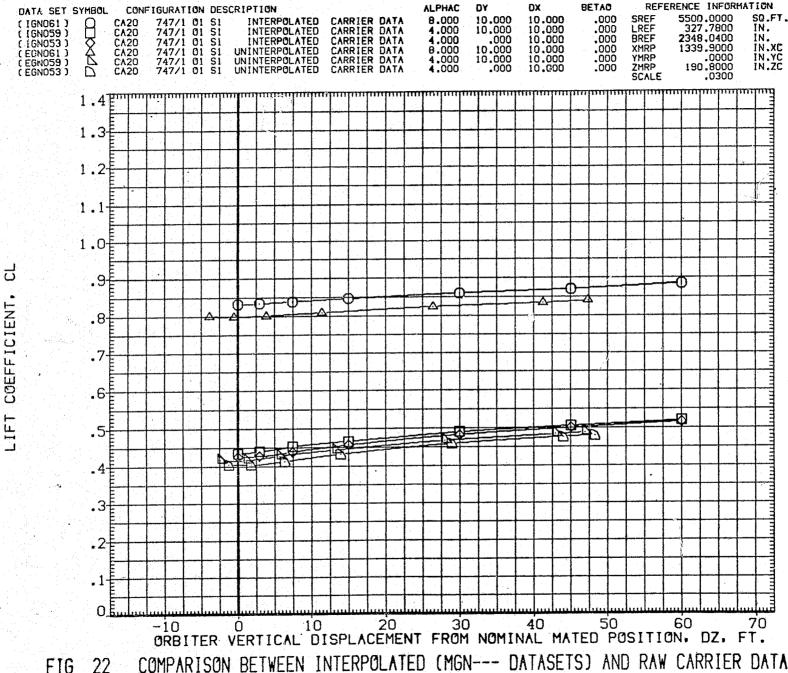
FIG 21C SPOILER EFFECTS ON ORBITER IN PRESENCE OF CARRIER WITH ATTACH HARDWARE

(B)BETAO = .00

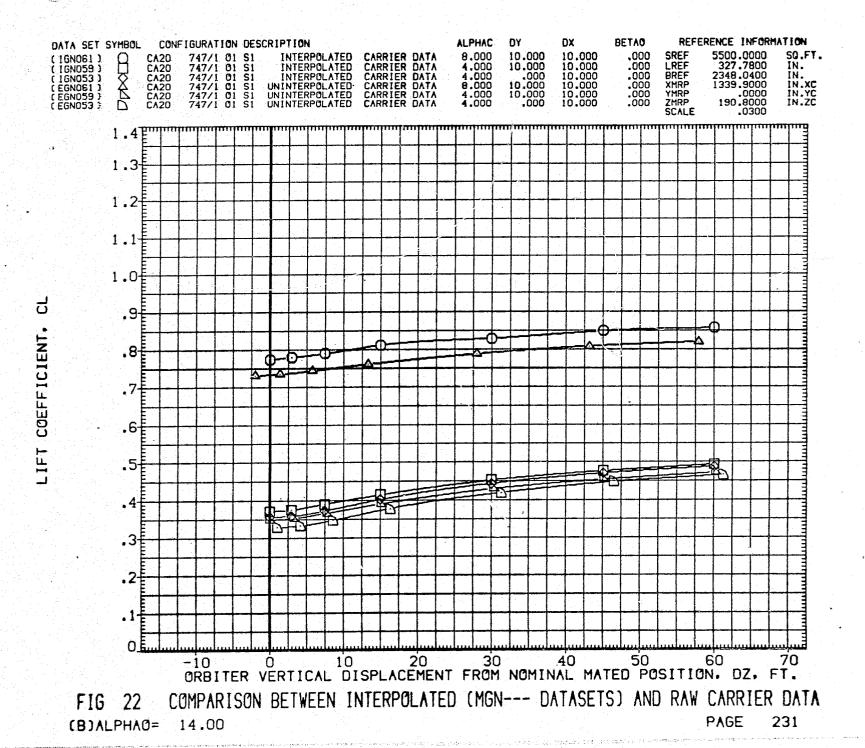
PAGE 227

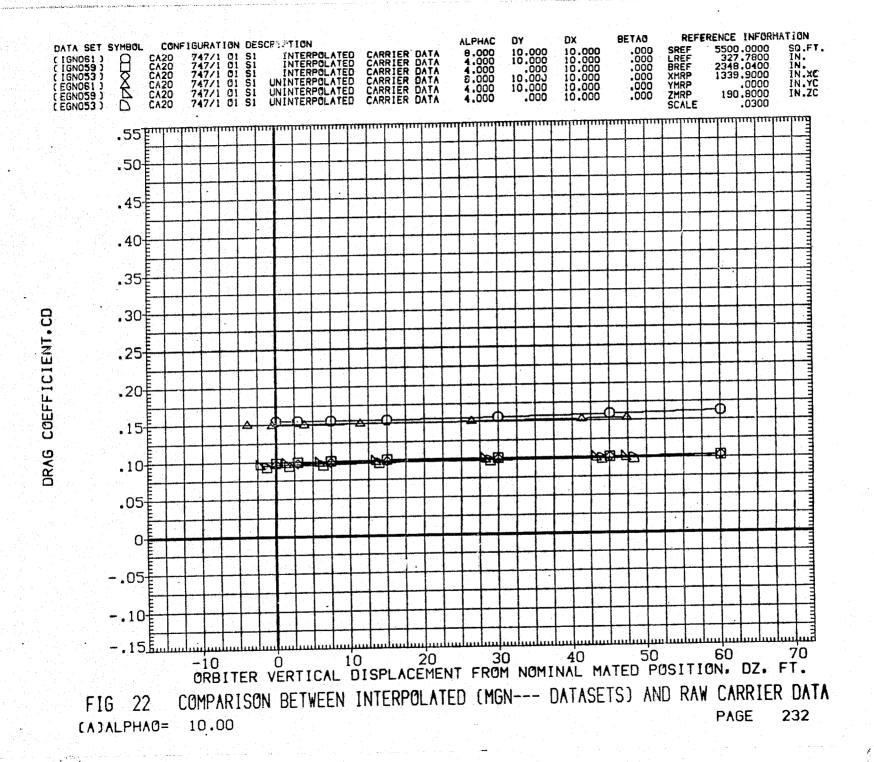


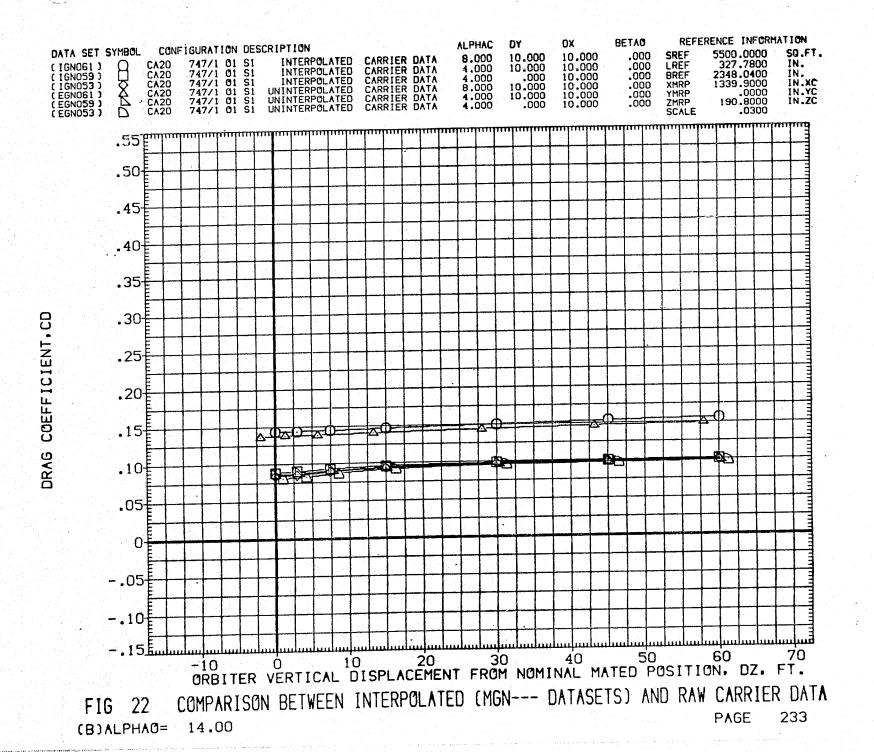


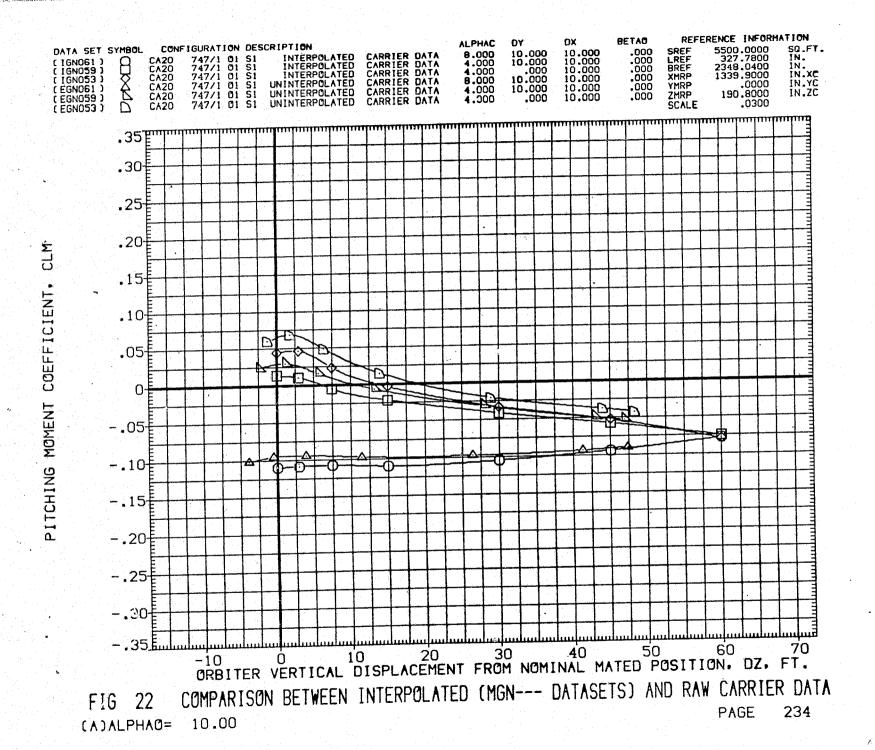


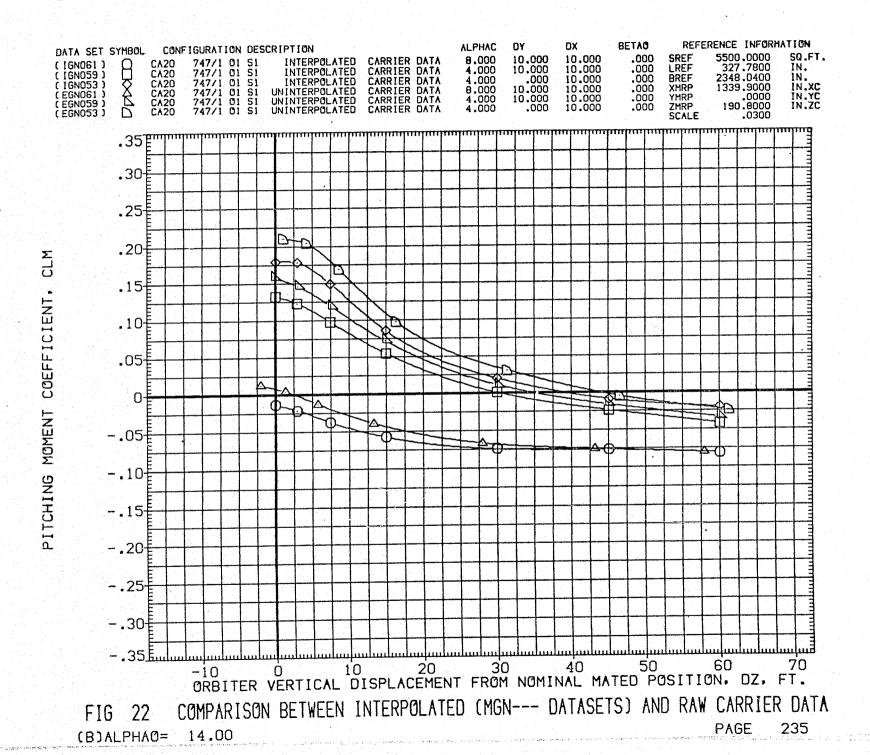
COMPARISON BETWEEN INTERPOLATED (MGN--- DATASETS) AND RAW CARRIER DATA PAGE 230 (A)ALPHAO = 10.00

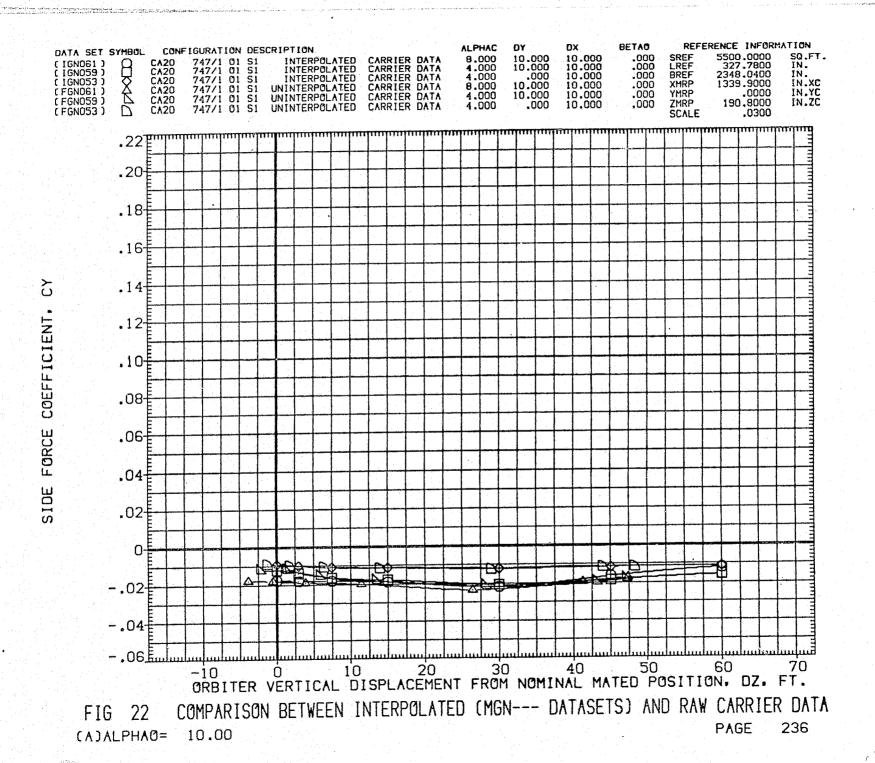


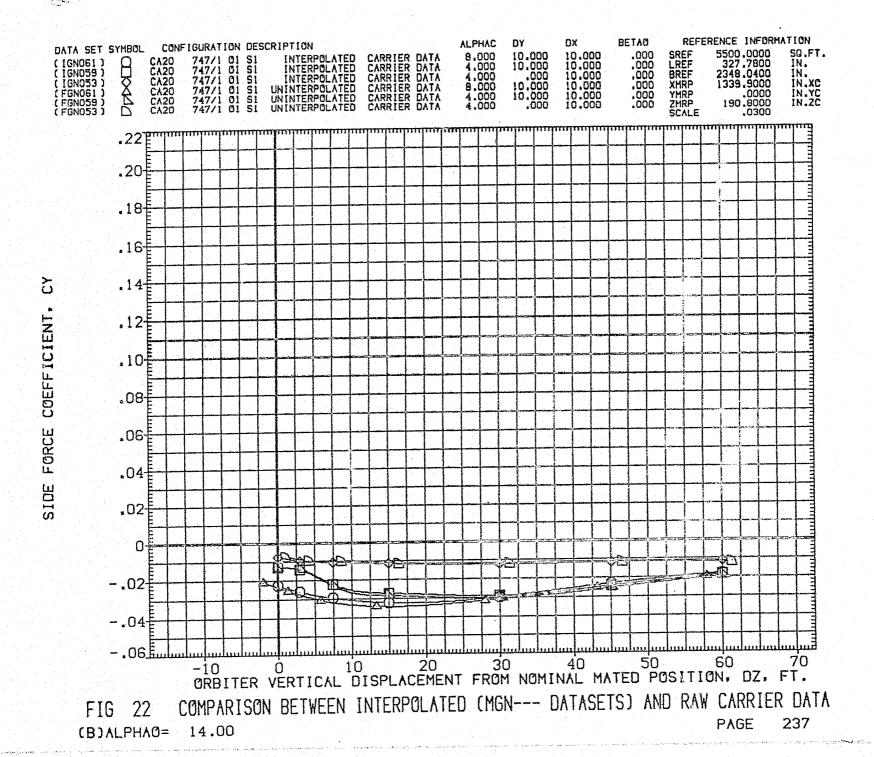


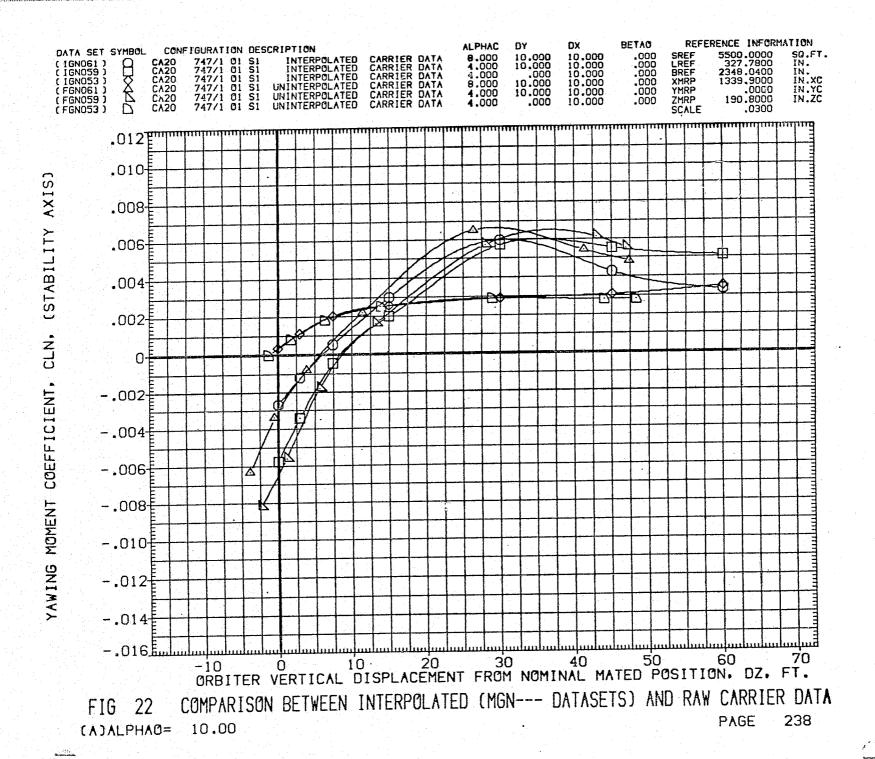


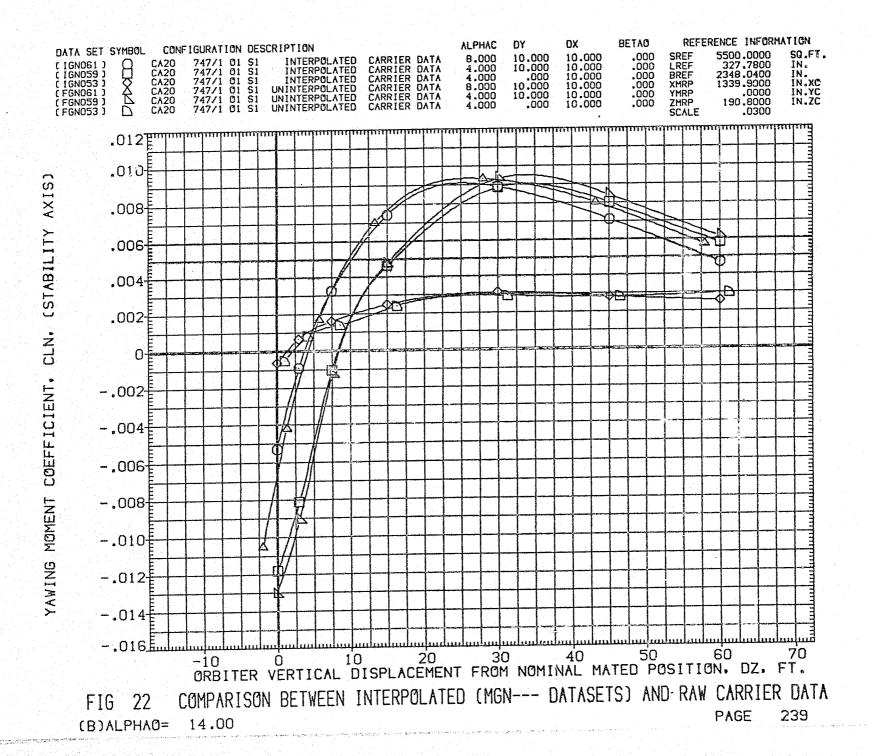


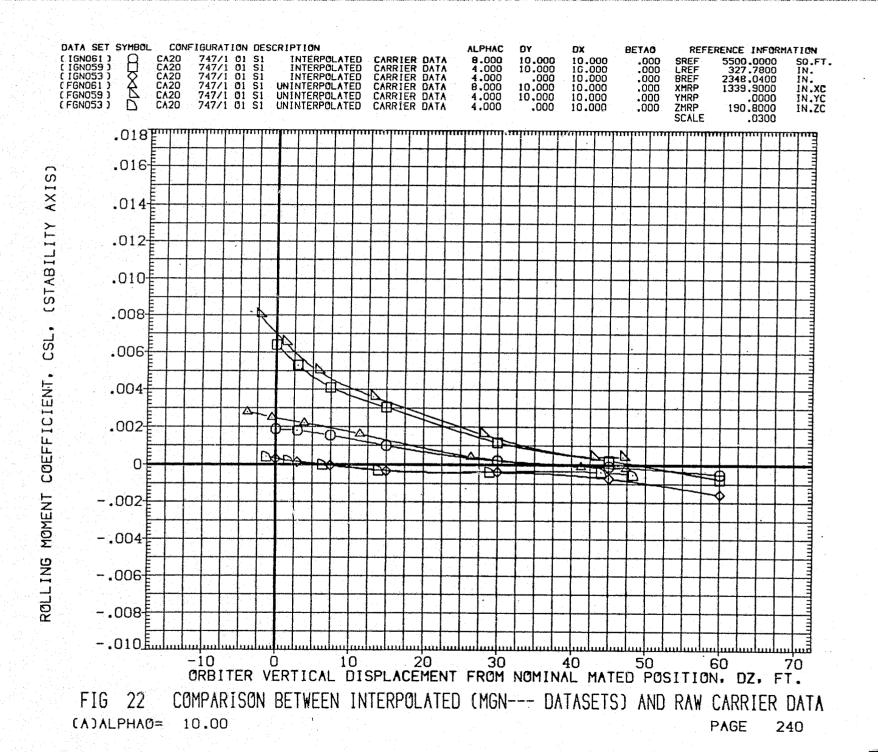


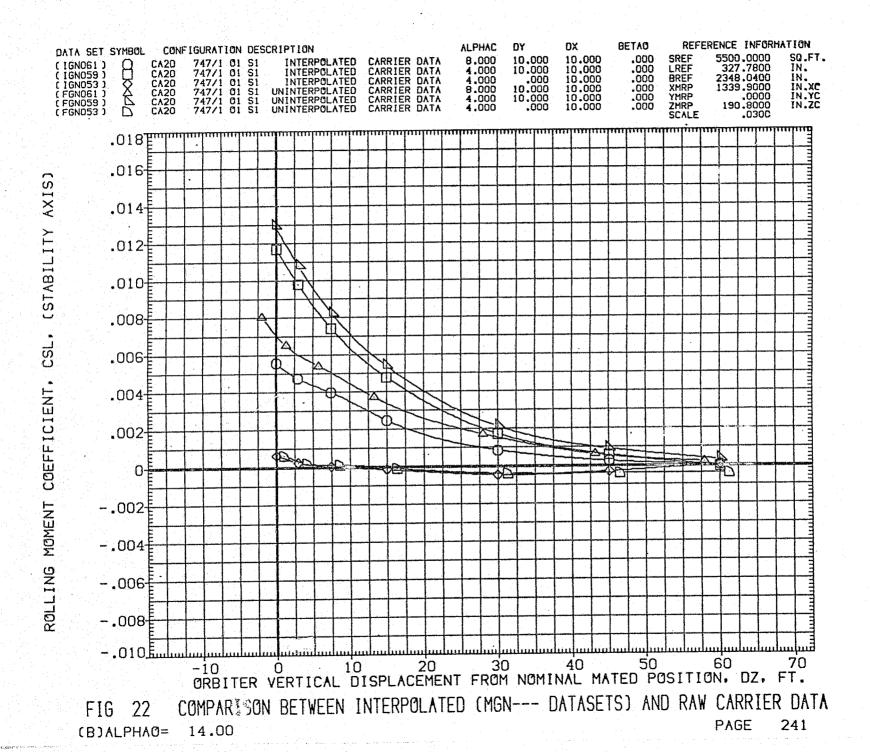


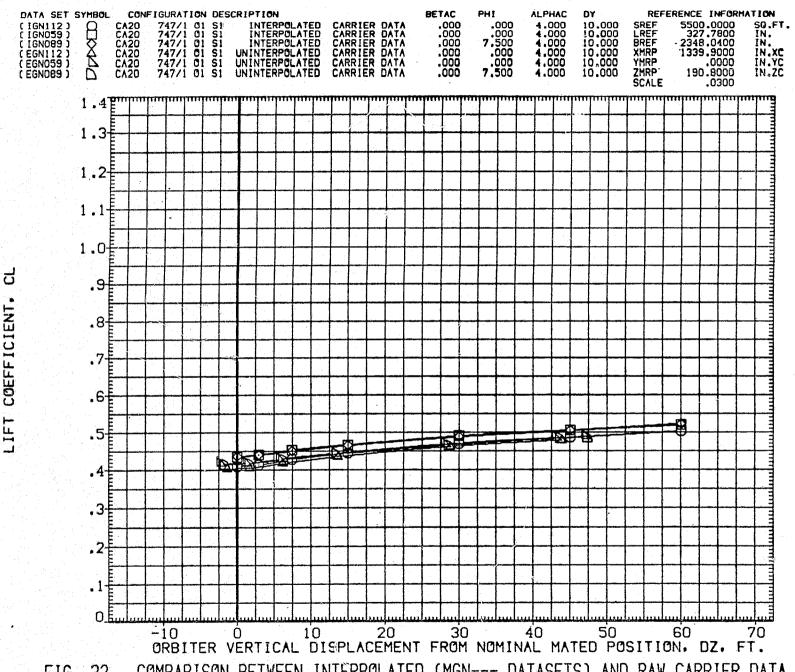








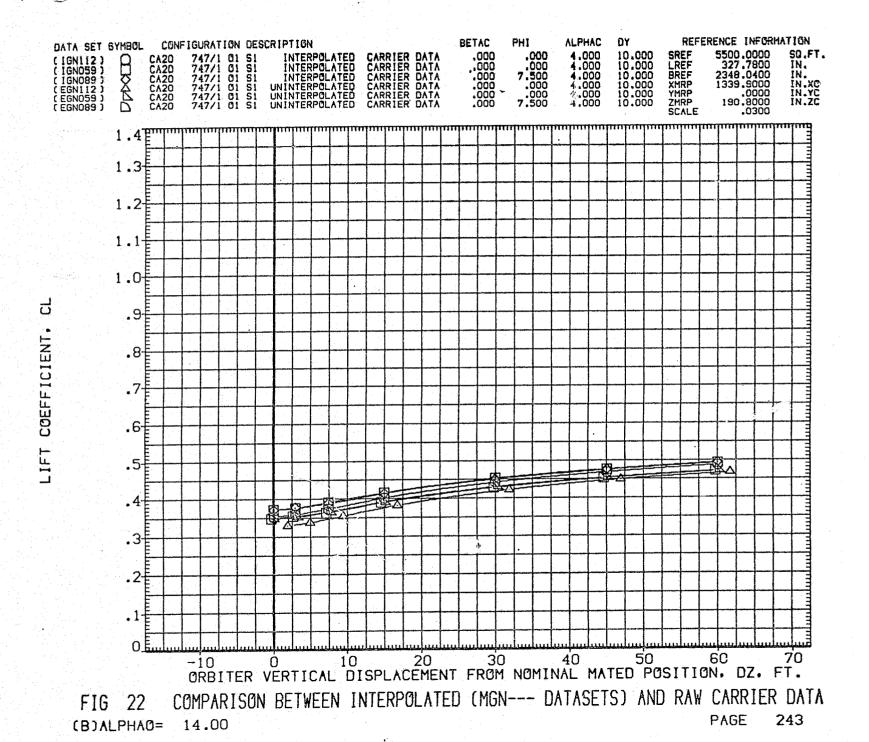


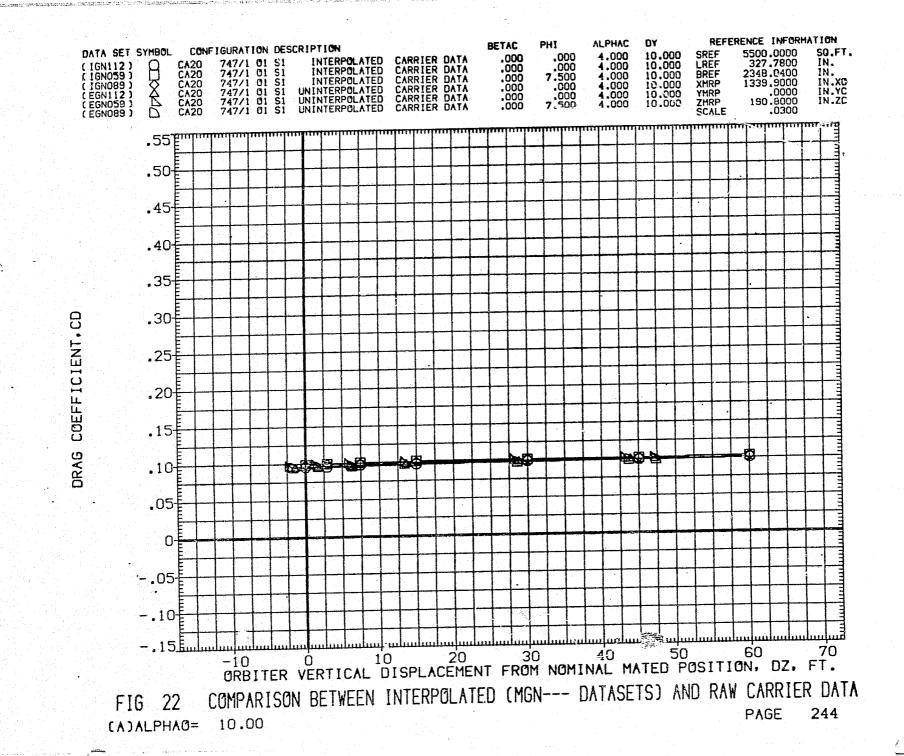


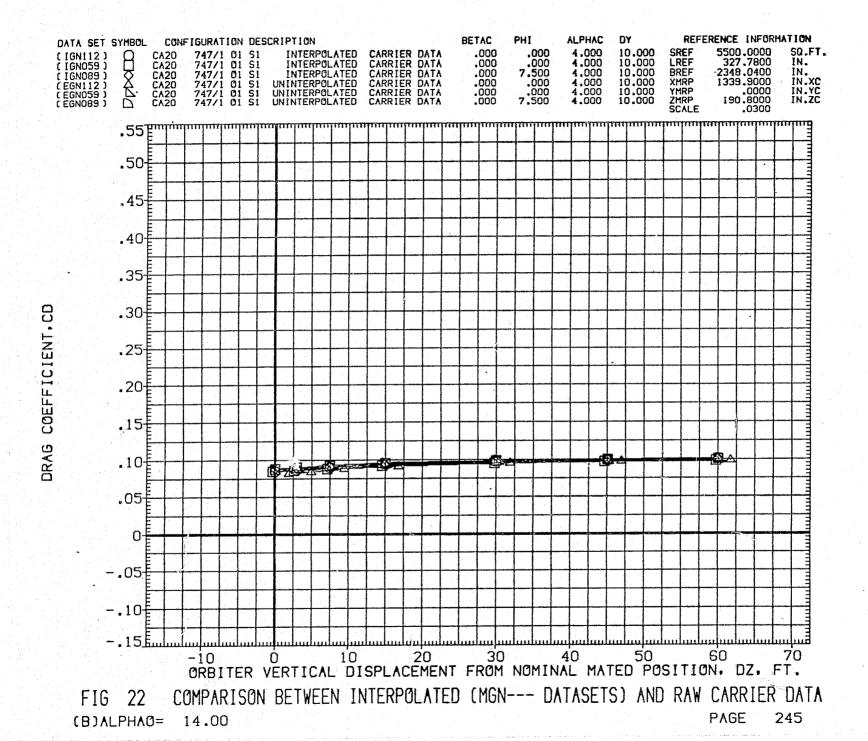
COMPARISON BETWEEN INTERPOLATED (MGN--- DATASETS) AND RAW CARRIER DATA PAGE 242 (A)ALPHAO= 10.00

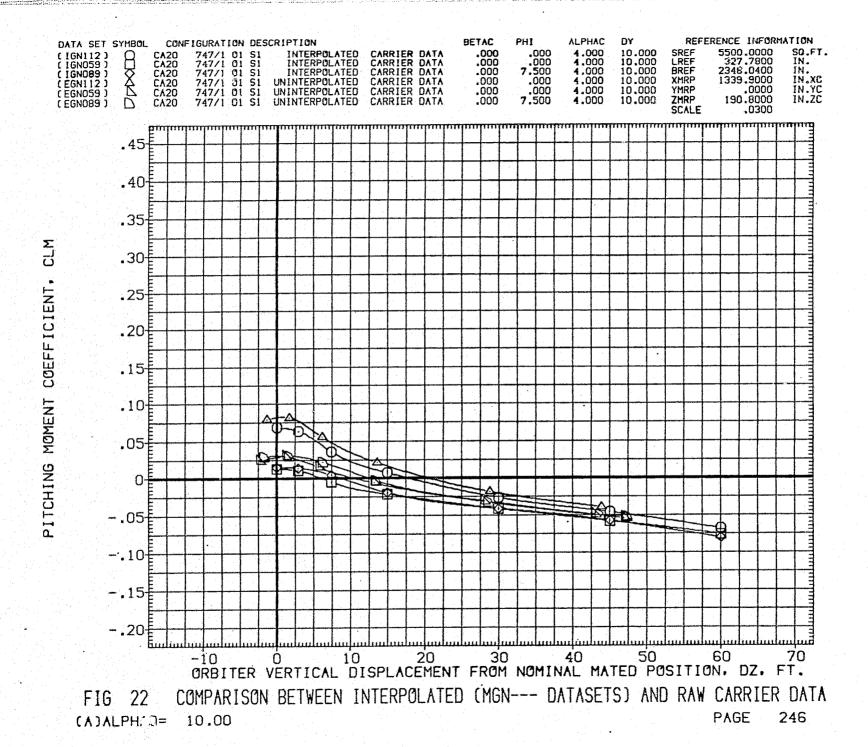
COEFFICIENT

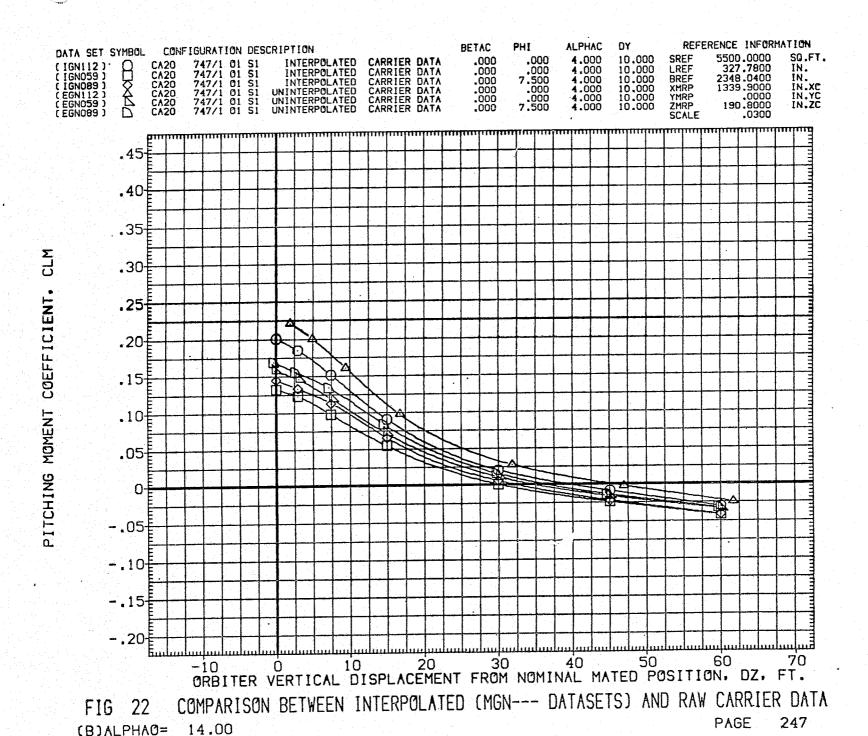
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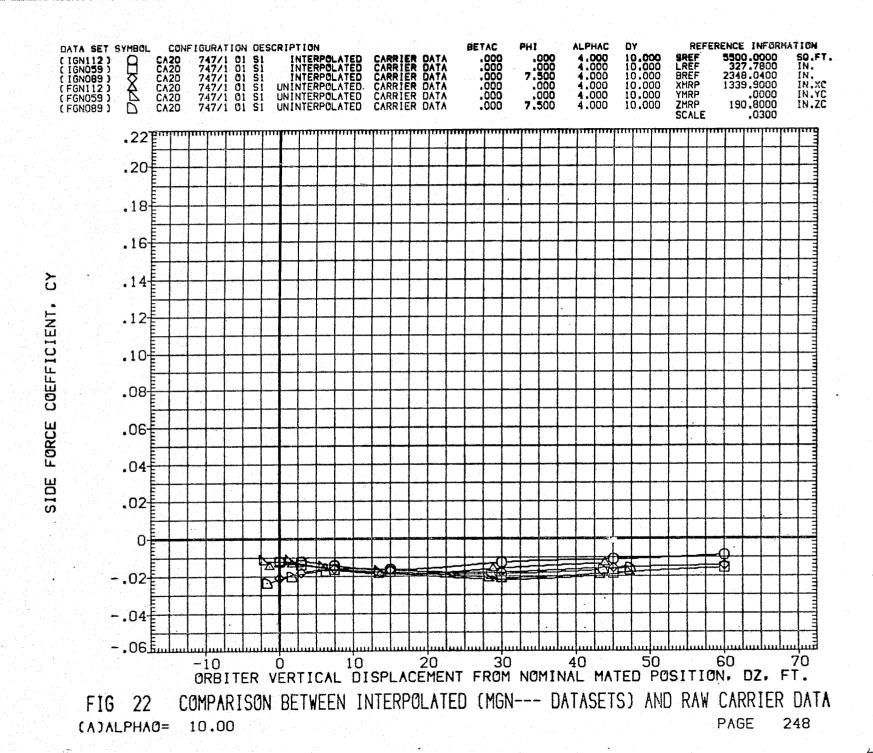












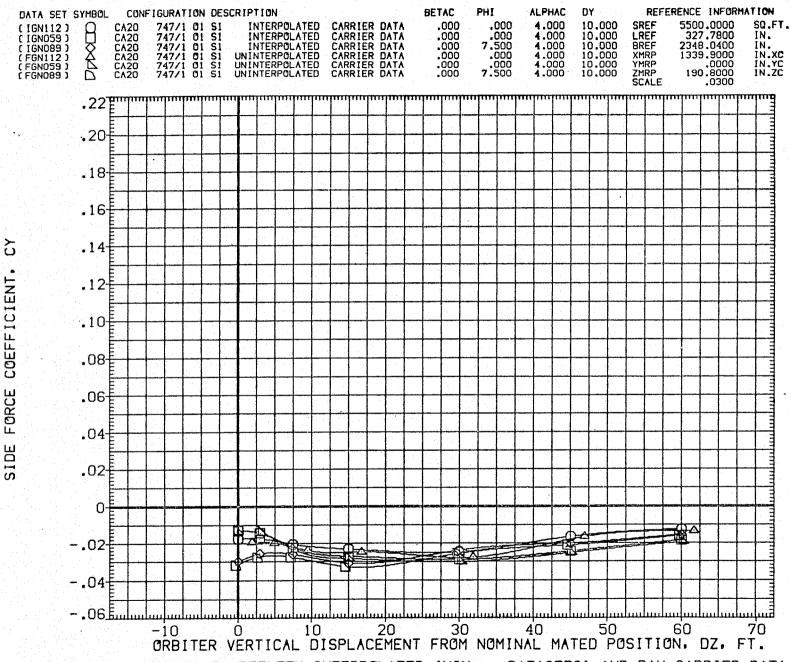
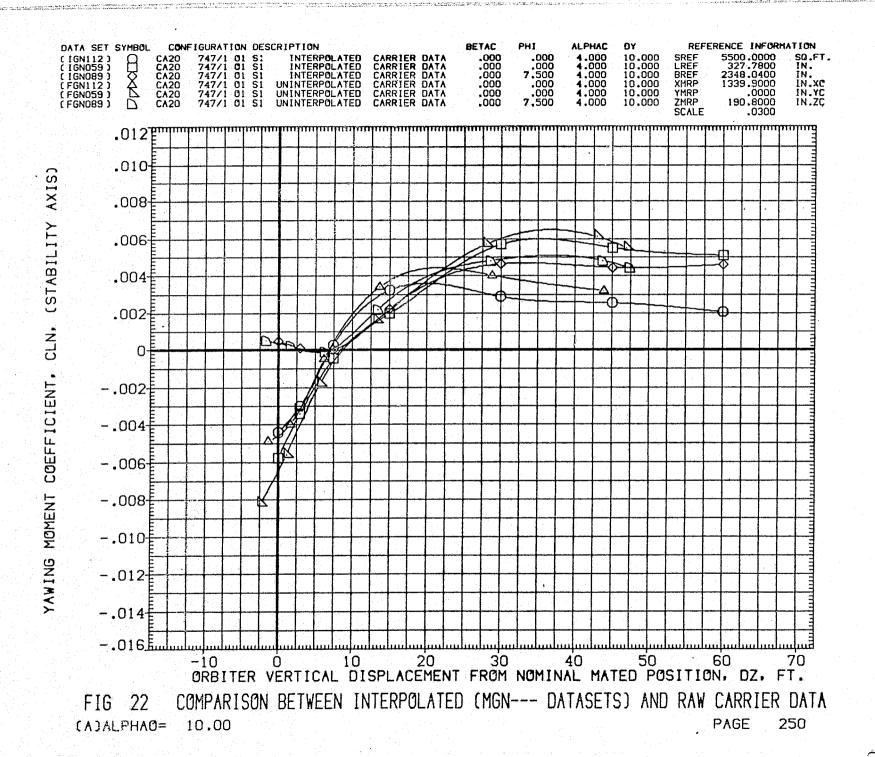
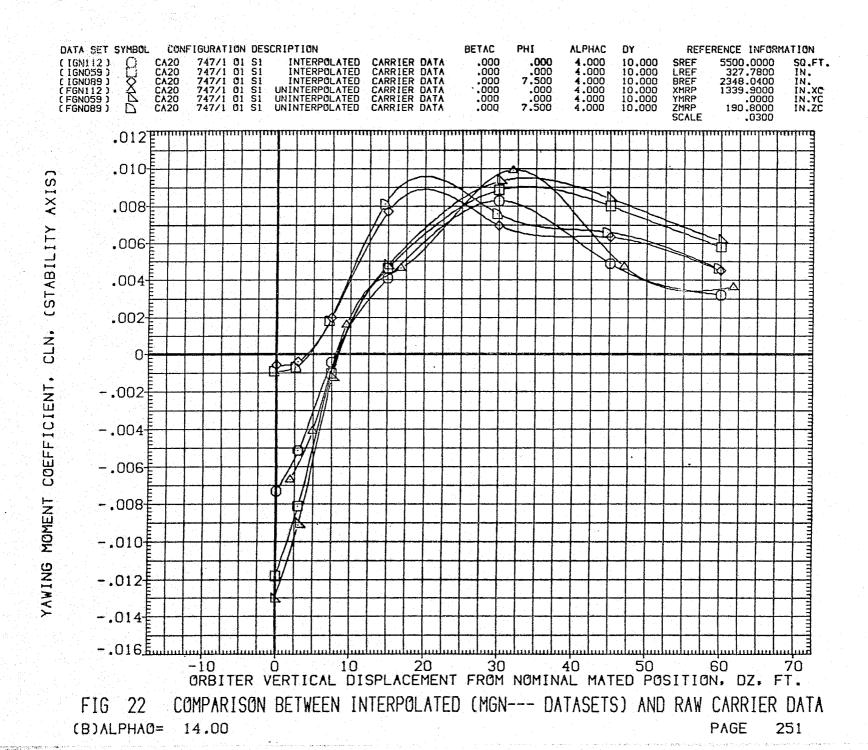
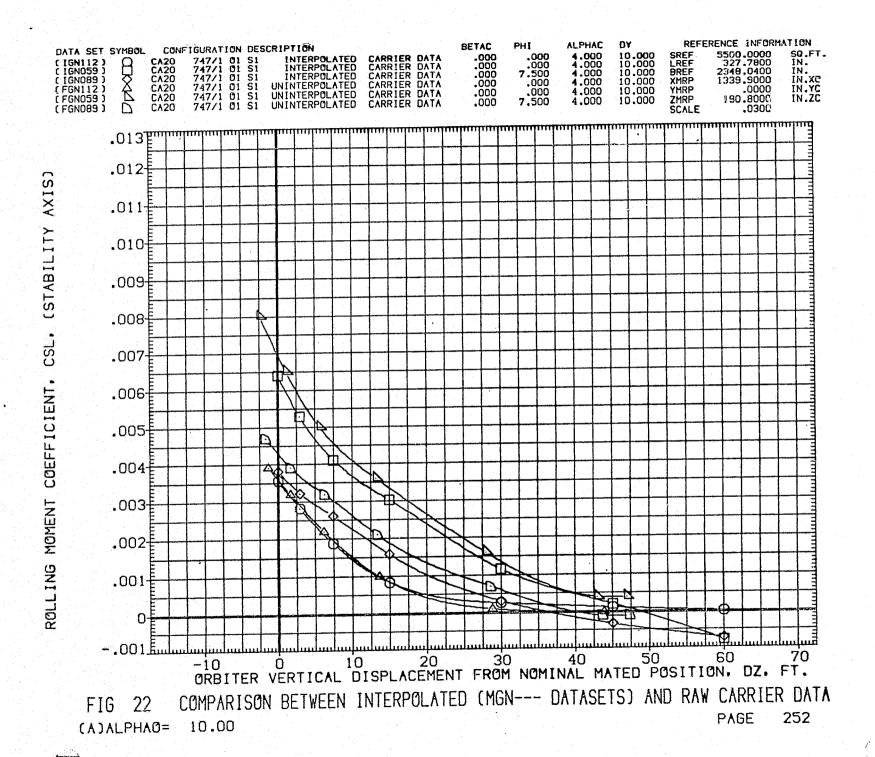
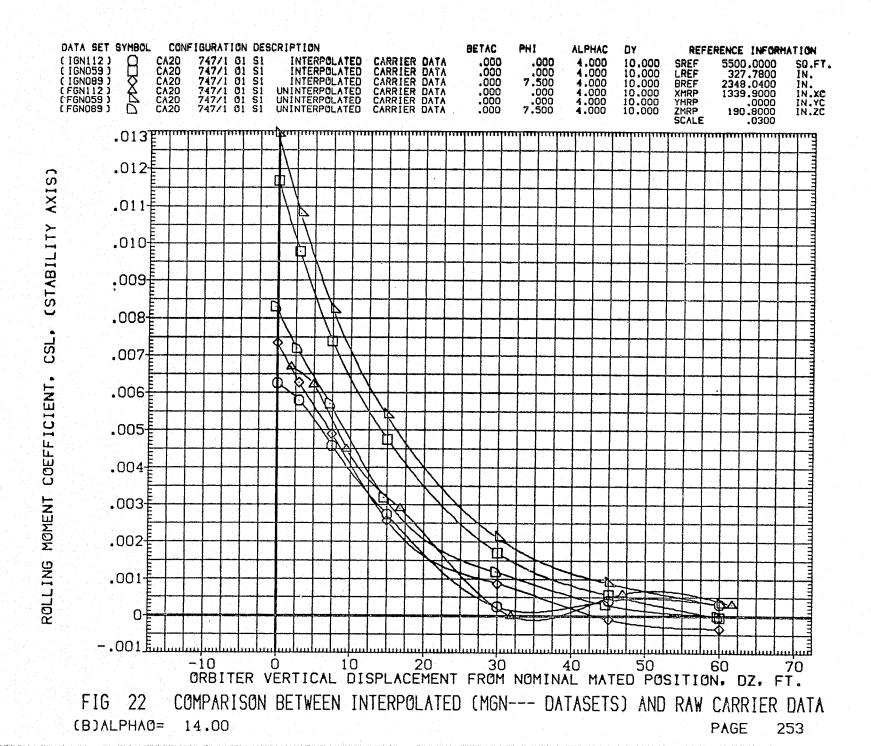


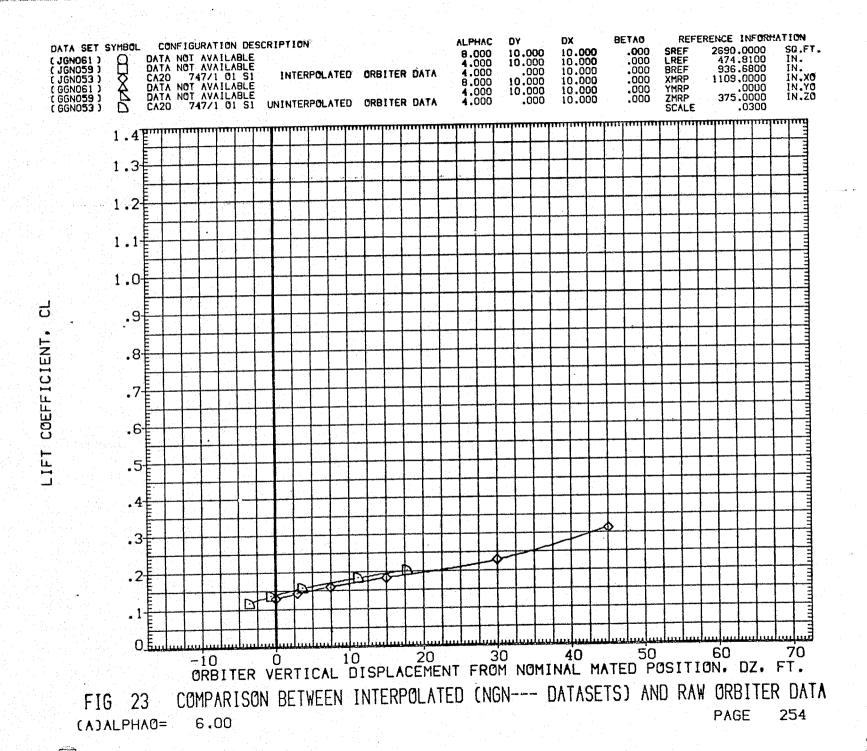
FIG 22 COMPARISON BETWEEN INTERPOLATED (MGN--- DATASETS) AND RAW CARRIER DATA
(B)ALPHAO= 14.00 PAGE 249











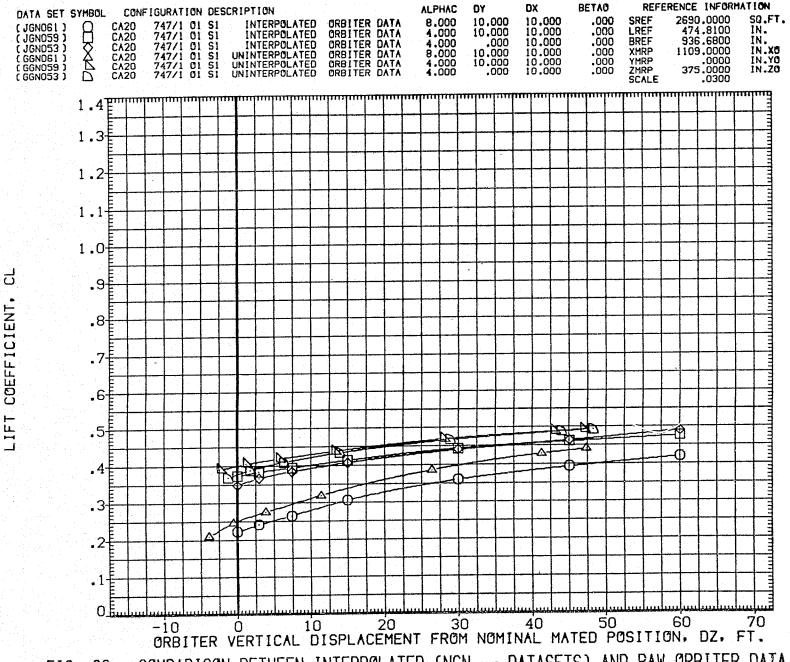


FIG 23 COMPARISON BETWEEN INTERPOLATED (NGN--- DATASETS) AND RAW ORBITER DATA

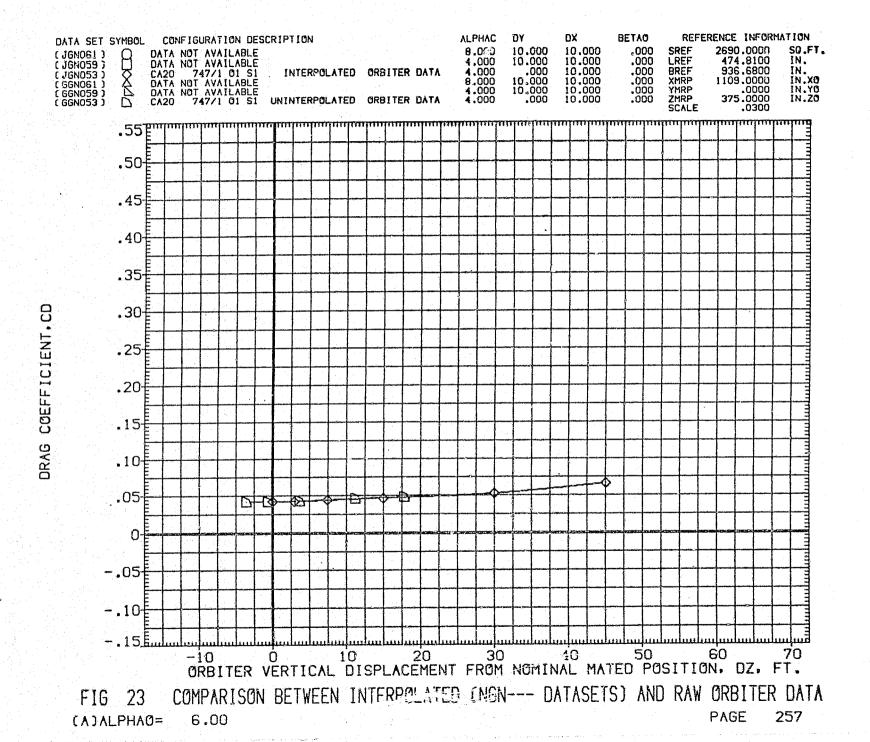
(B) ALPHAG= 10.00 PAGE 255

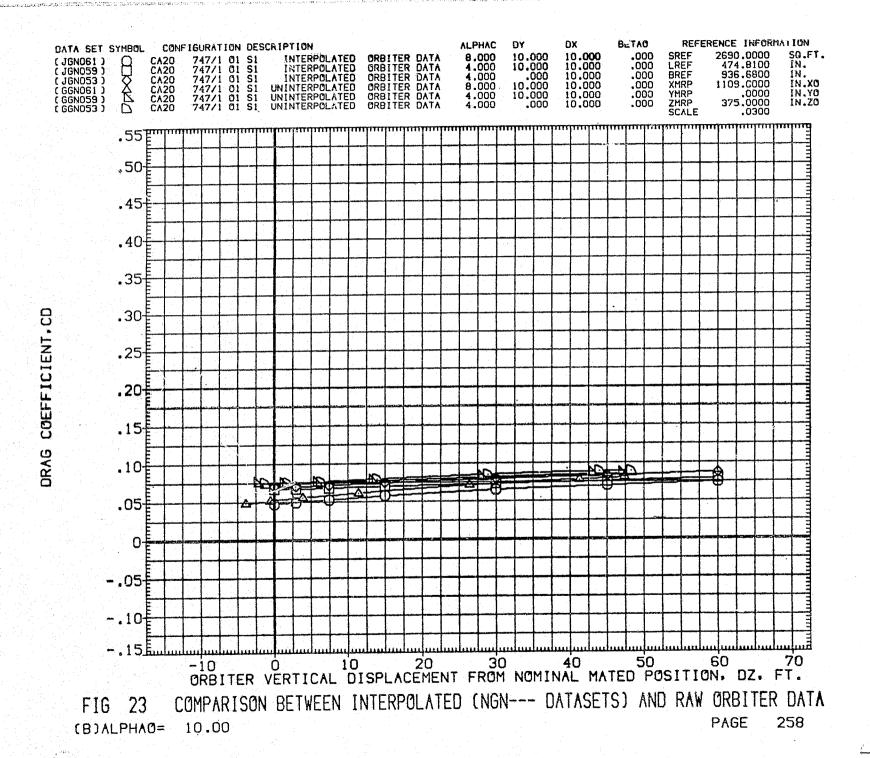
(C)ALPHAO=

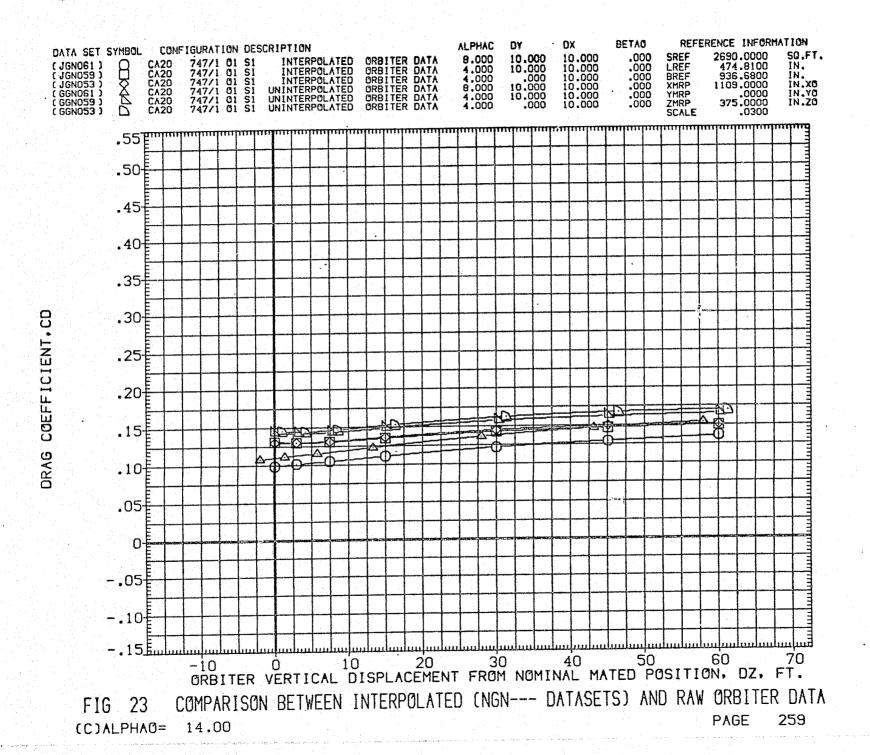
14.00

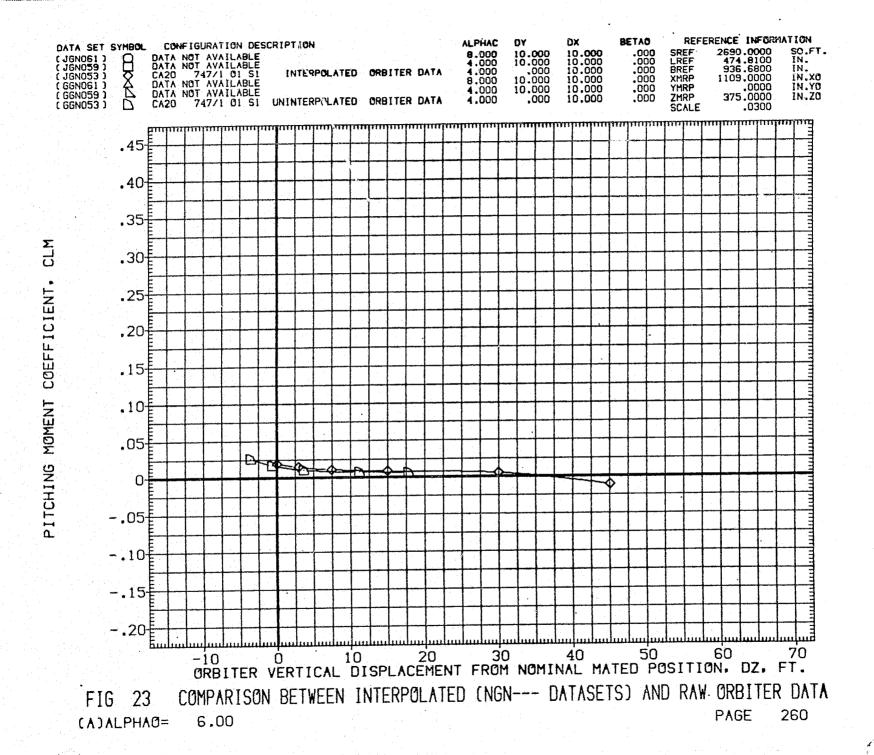
PAGE

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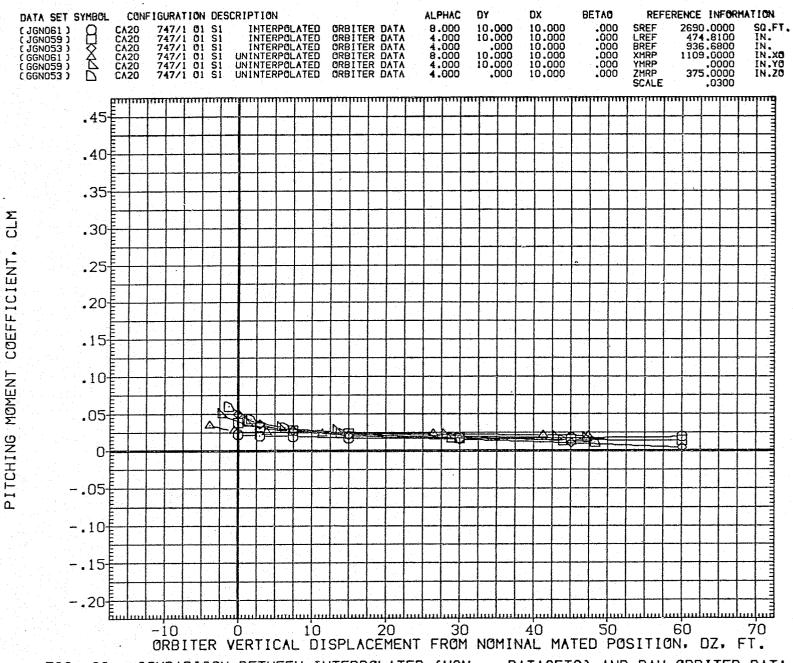


FIG 23 COMPARISON BETWEEN INTERPOLATED (NGN--- DATASETS) AND RAW ORBITER DATA

(B) ALPHAO = 10.00

PAGE 261

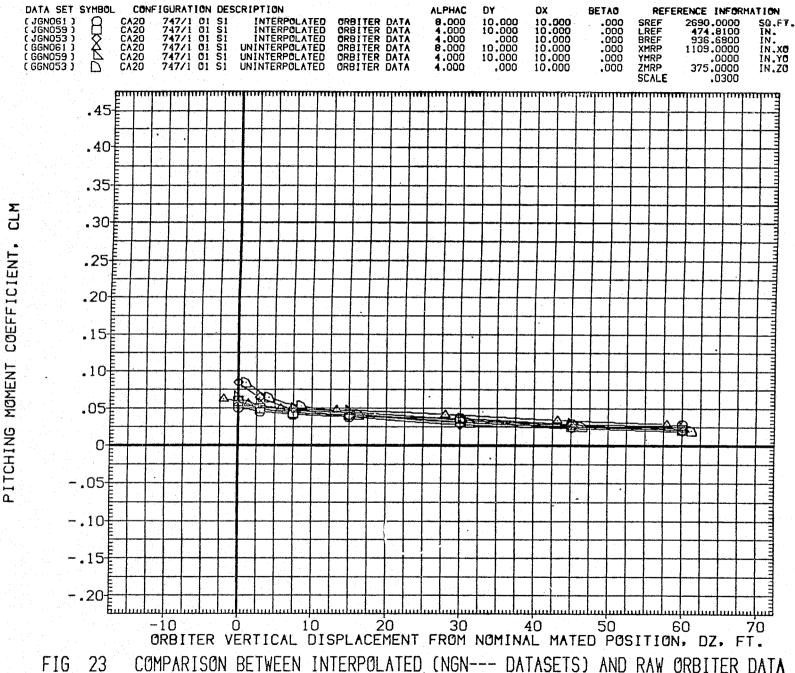
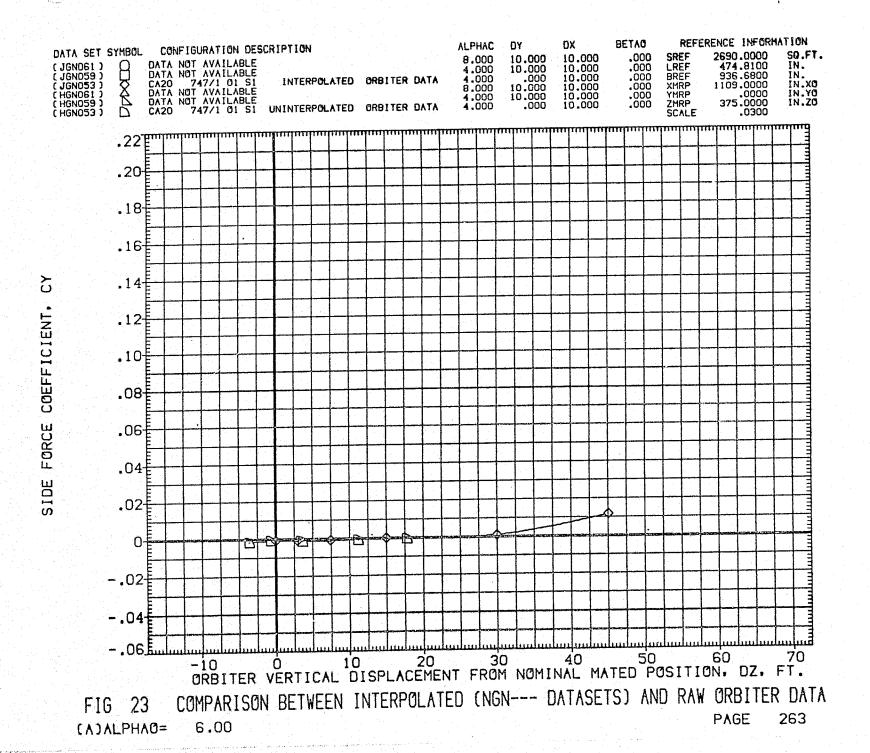
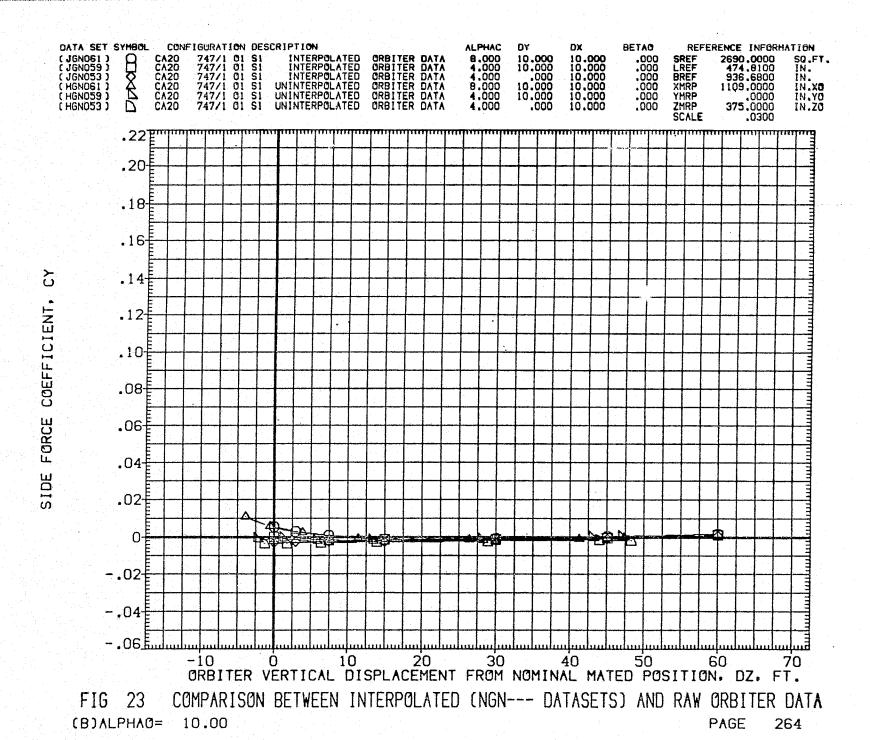


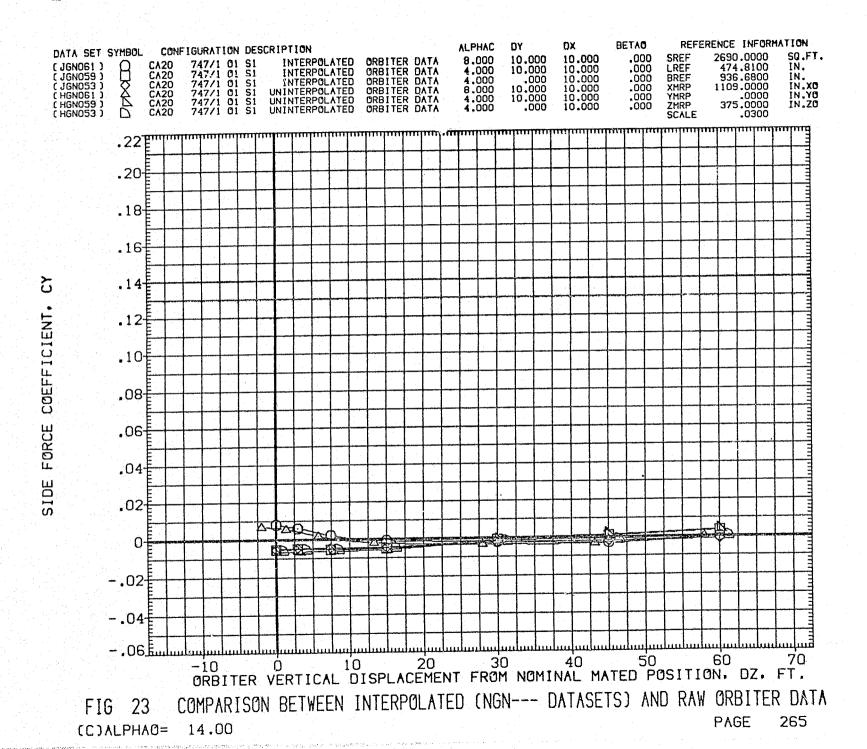
FIG 23 COMPARISON BETWEEN INTERPOLATED (NGN--- DATASETS) AND RAW ORBITER DATA

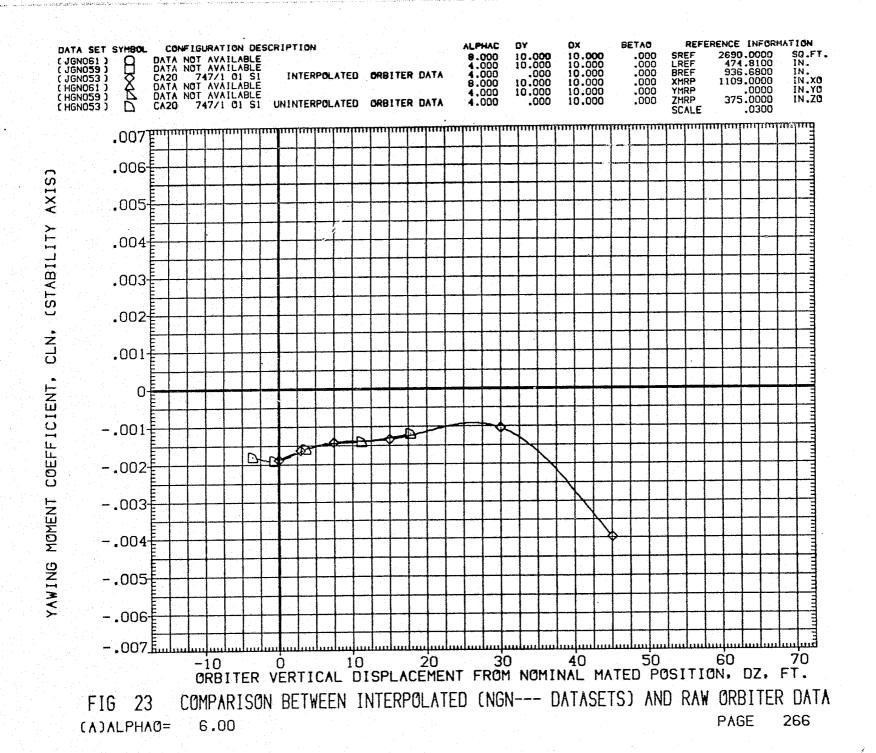
(C)ALPHAO= 14.00

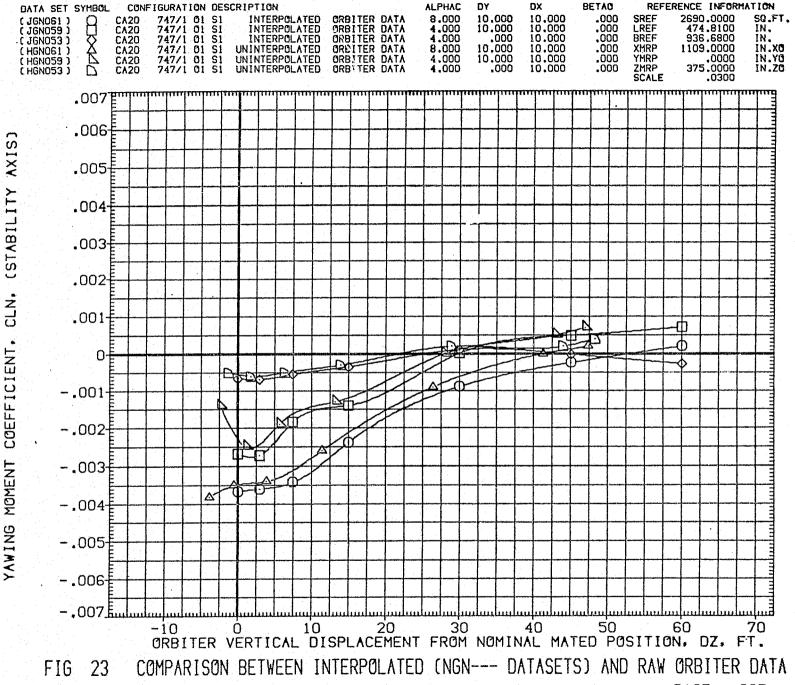
PAGE 262



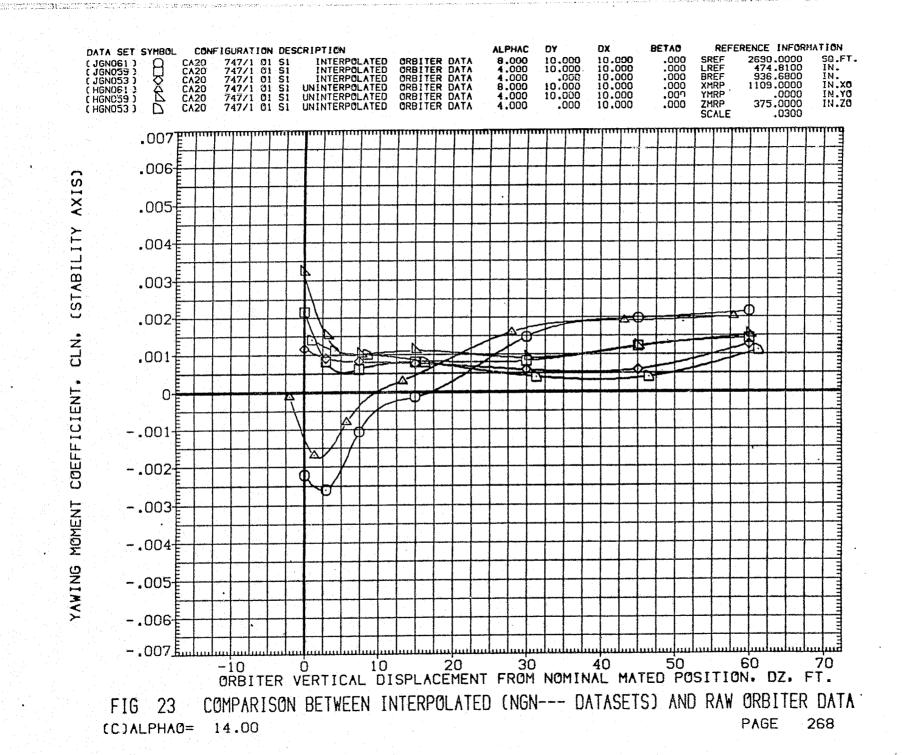


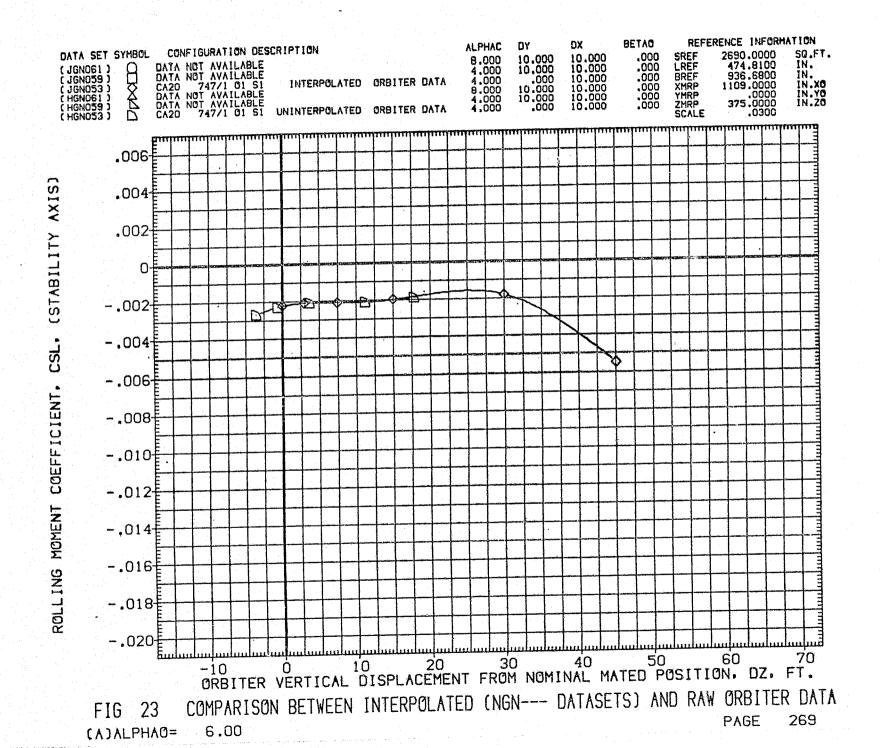






PAGE 267 (B)ALPHAO= 10.00





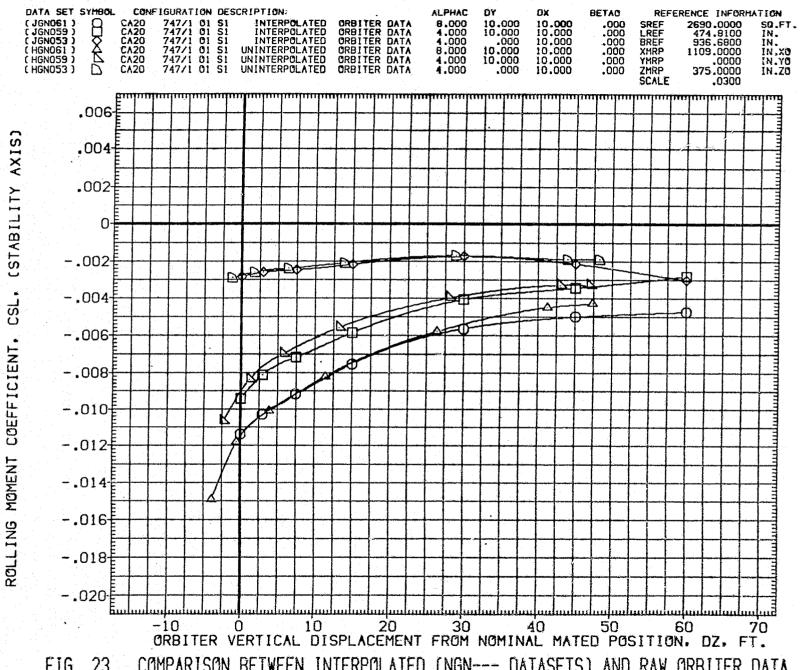
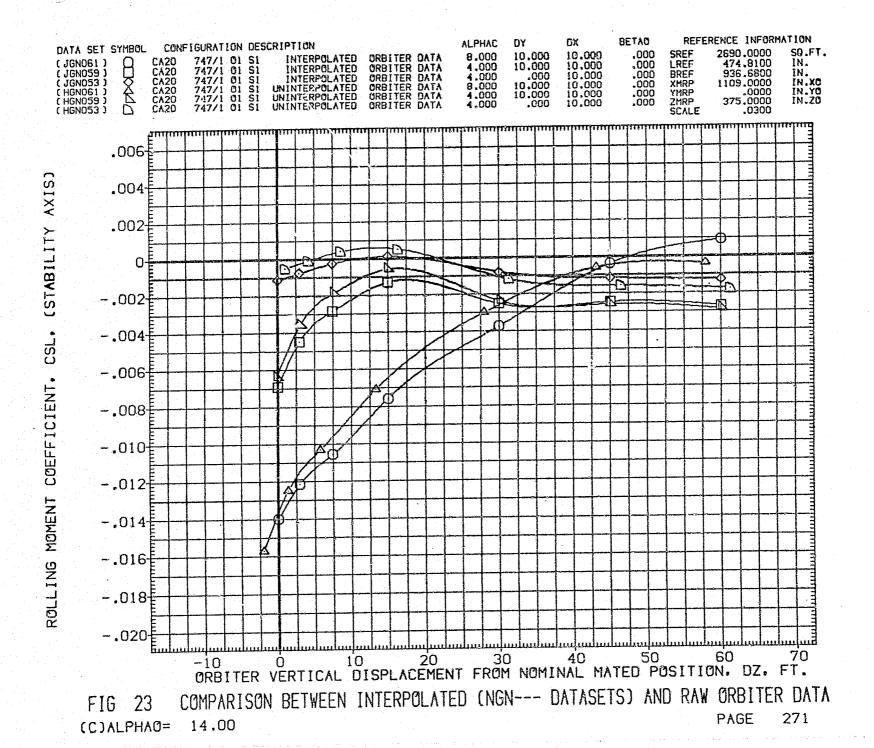
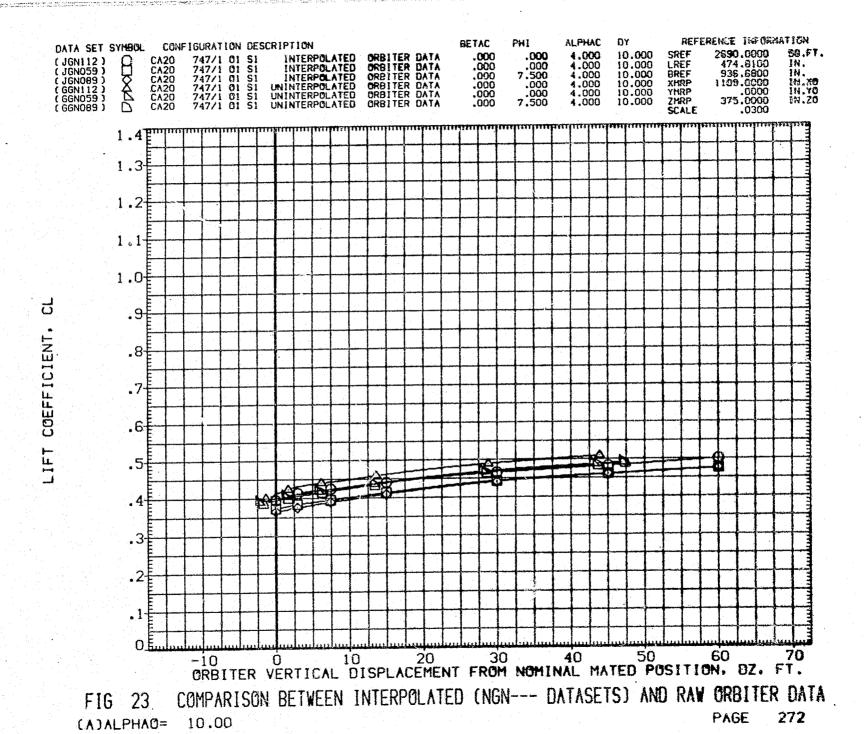
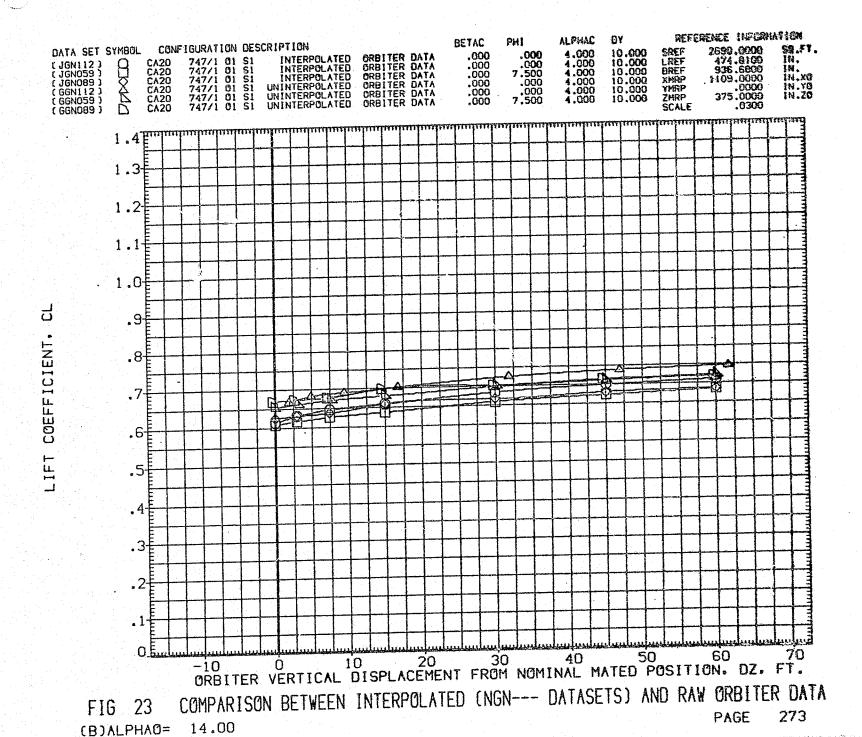


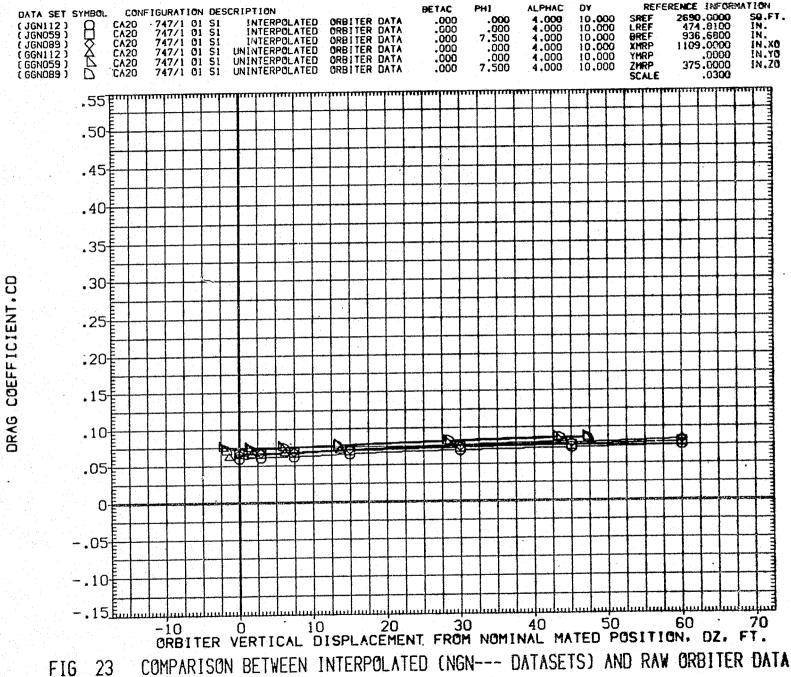
FIG 23 COMPARISON BETWEEN INTERPOLATED (NGN--- DATASETS) AND RAW ORBITER DATA

(B)ALPHAO= 10.00 PAGE 270

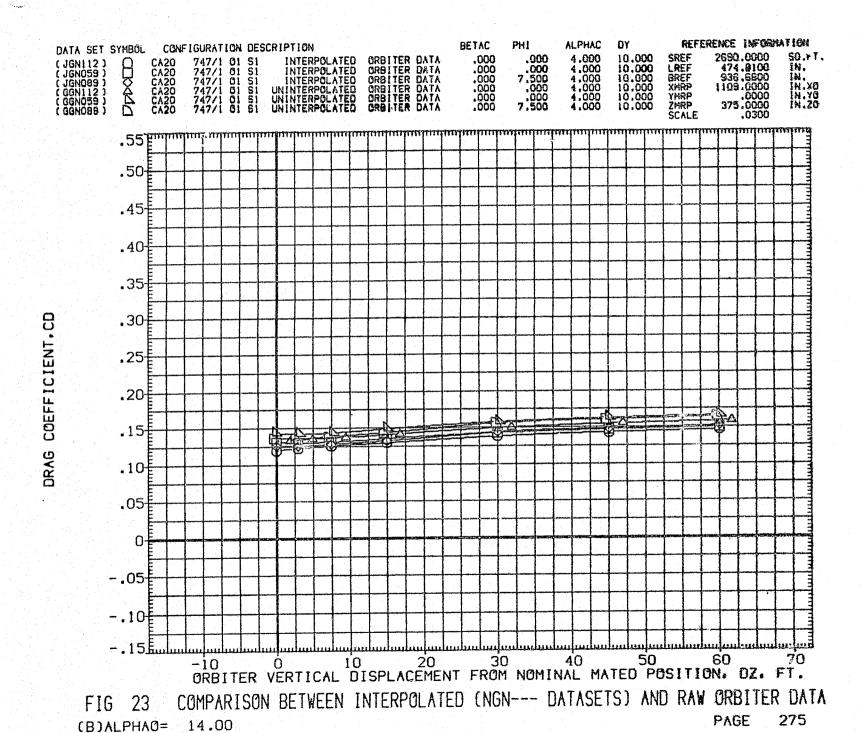


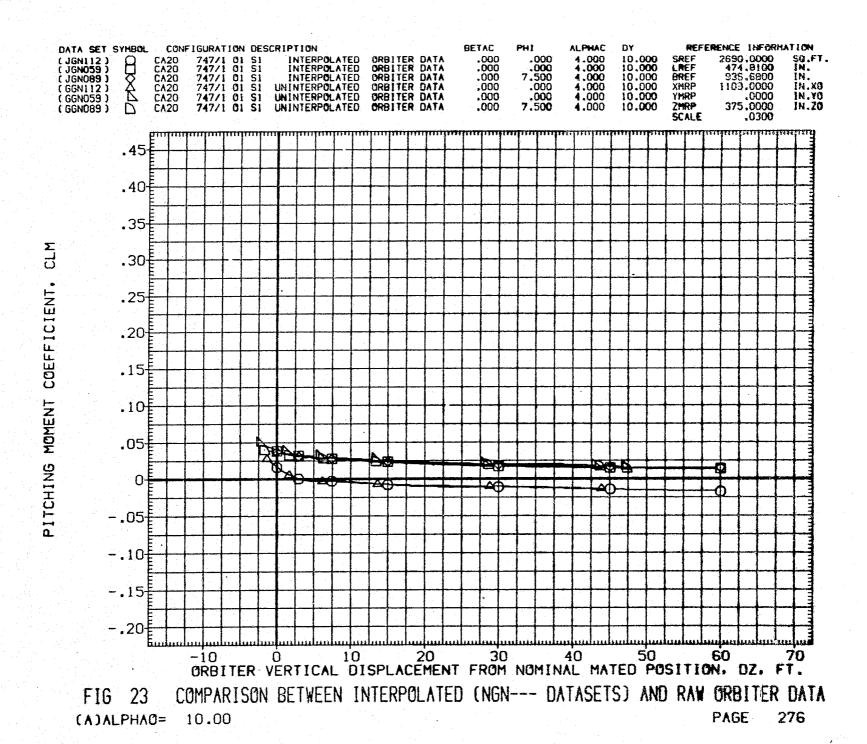


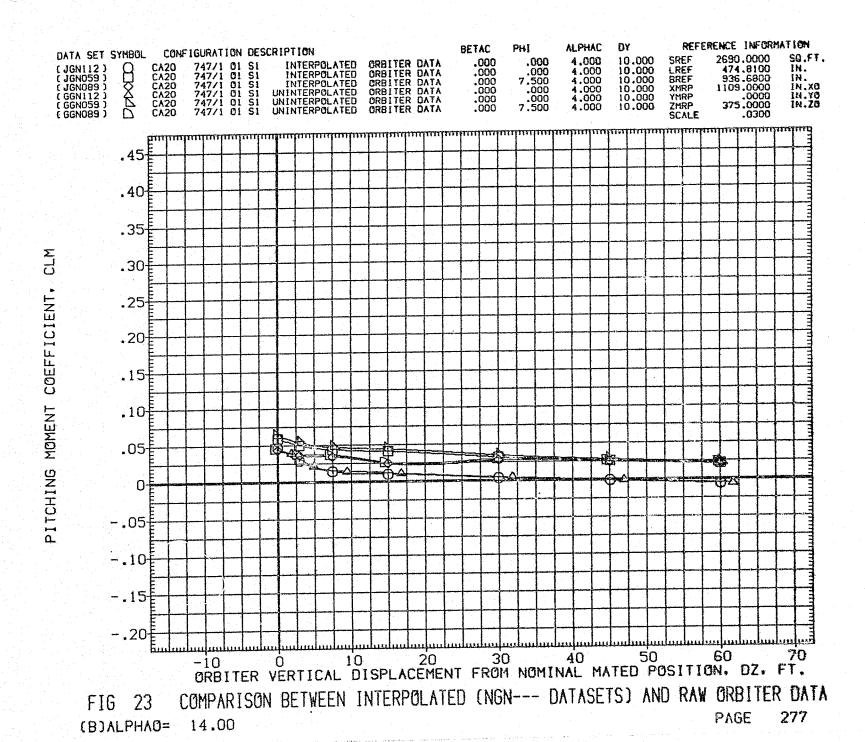




274 PAGE 10.00 (A)ALPHAO=







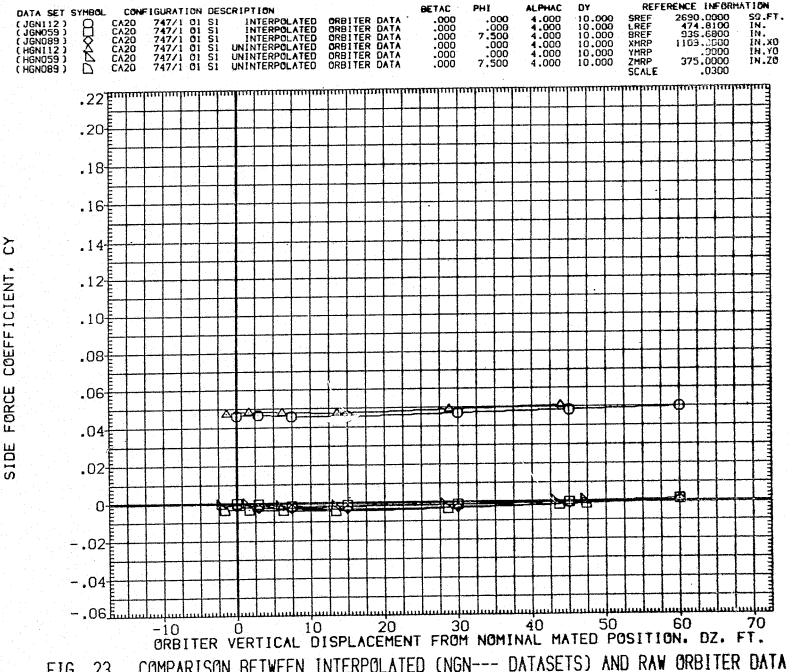
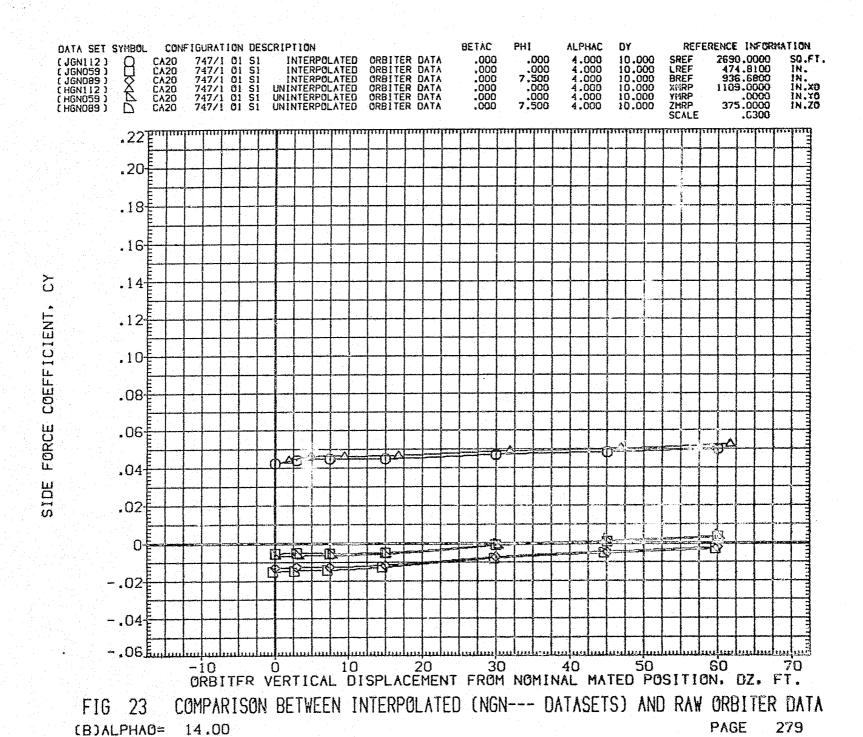
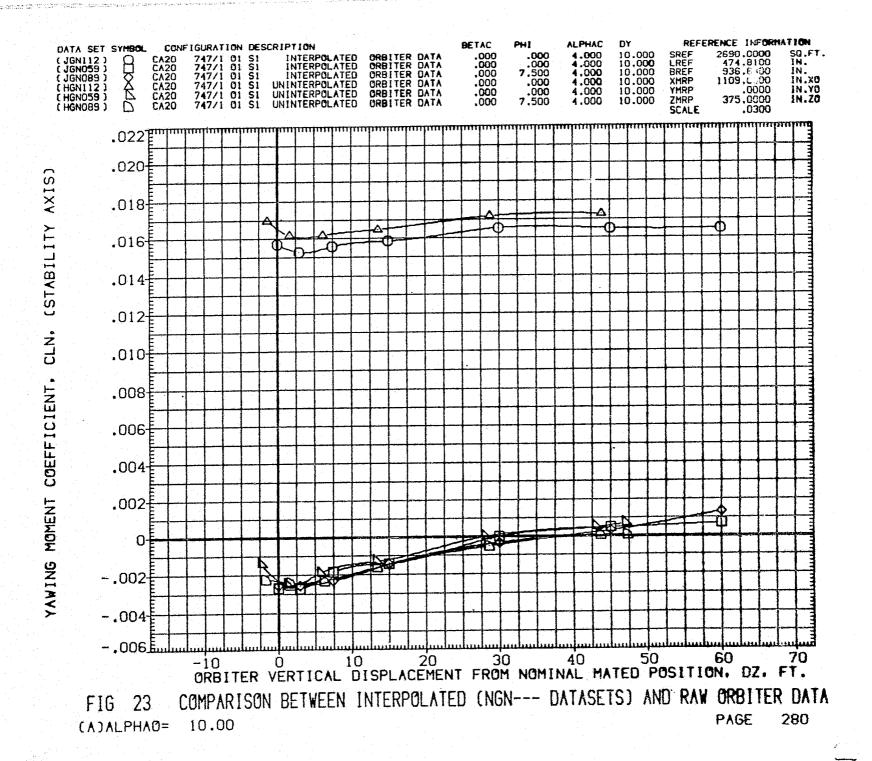


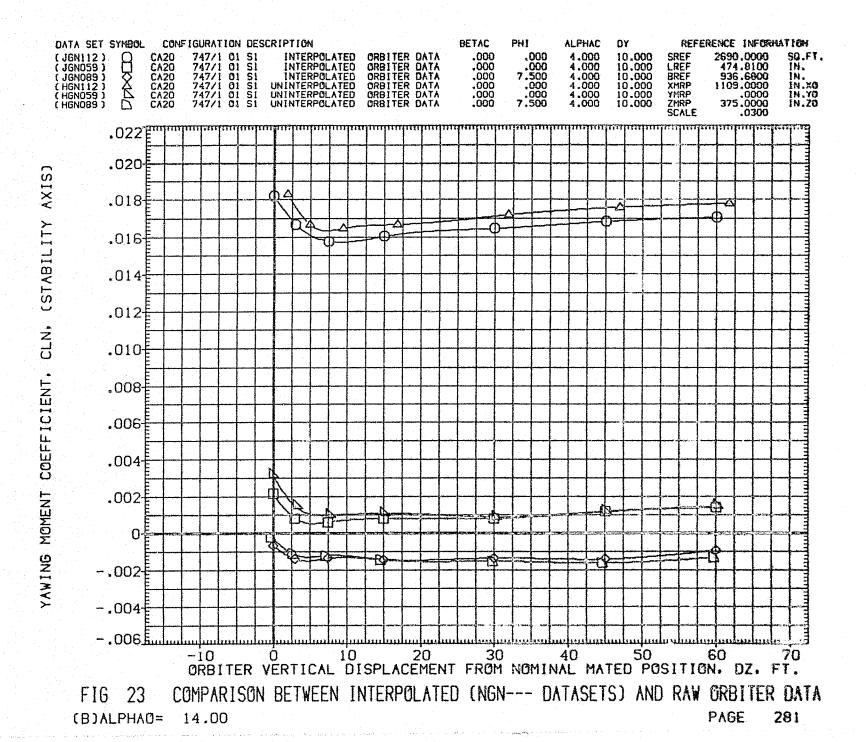
FIG 23 COMPARISON BETWEEN INTERPOLATED (NGN--- DATASETS) AND RAW ORBITER DATA

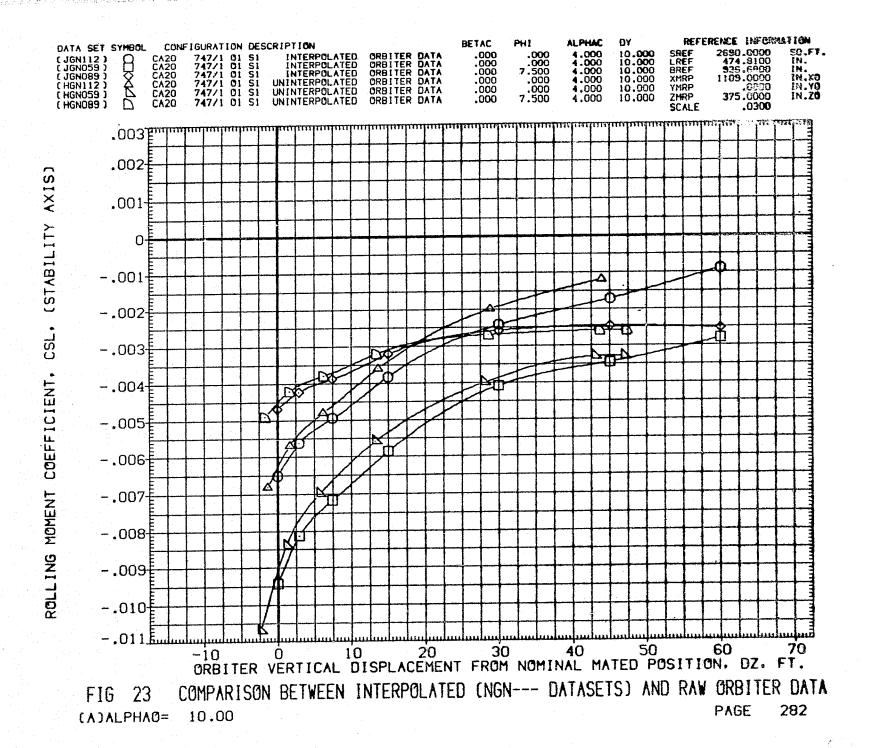
(A)ALPHAO= 10.00

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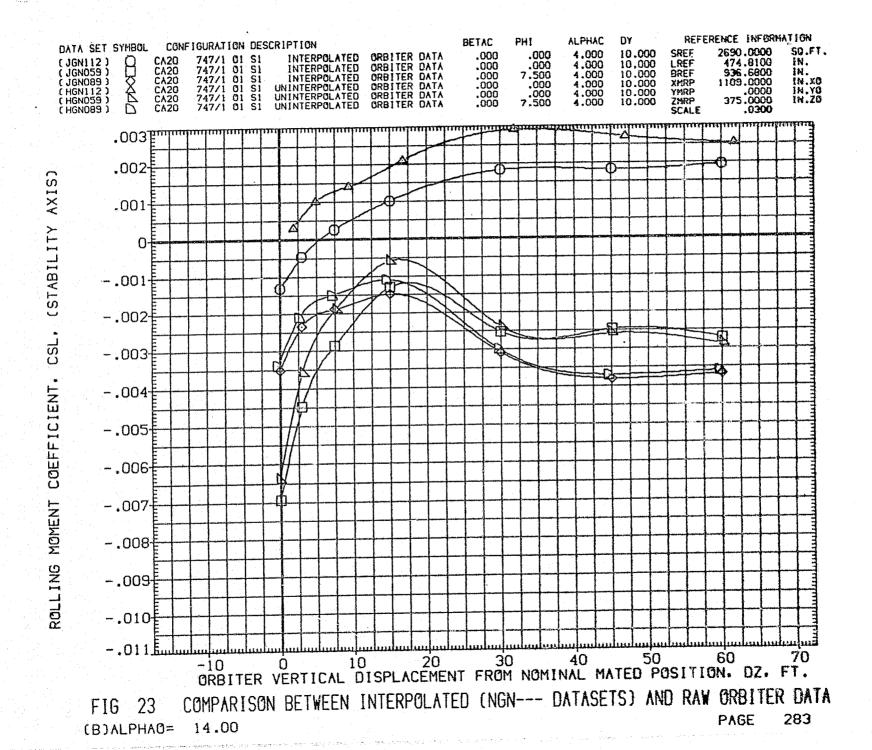


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 284

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

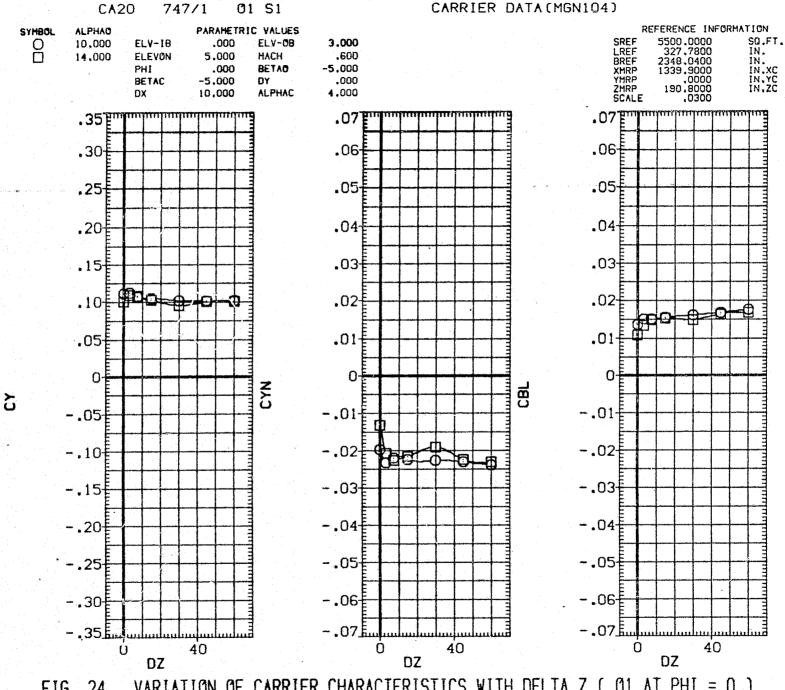


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 286

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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 287

DZ

DZ

DZ

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 288

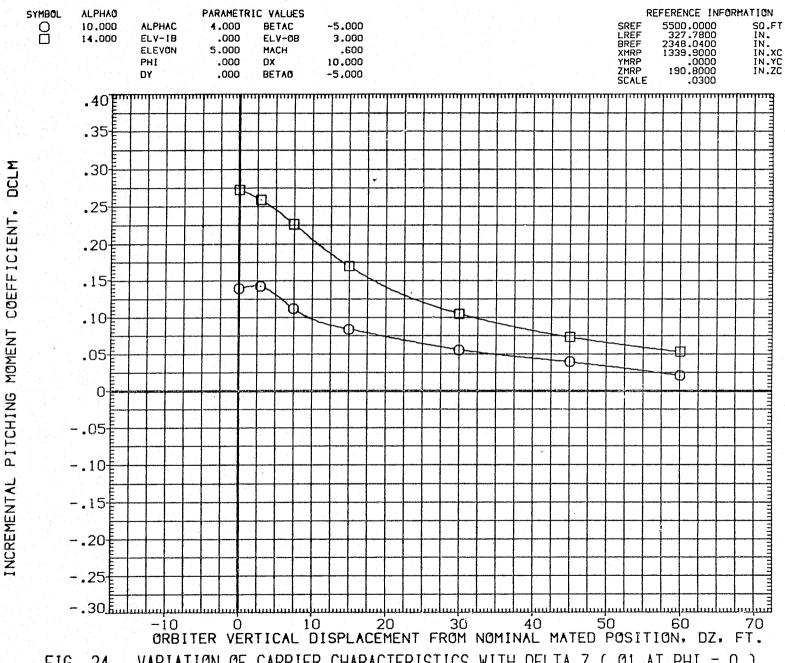


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

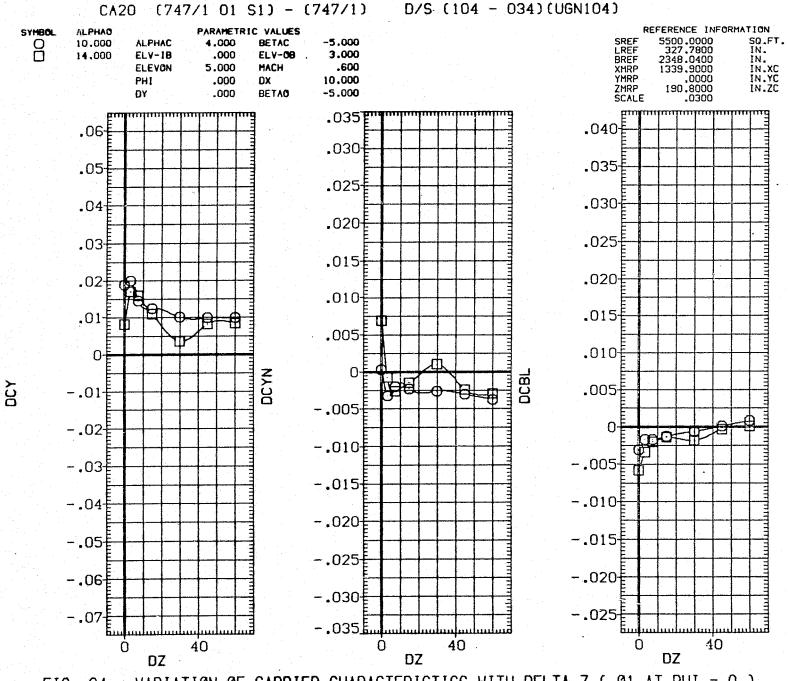


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 290

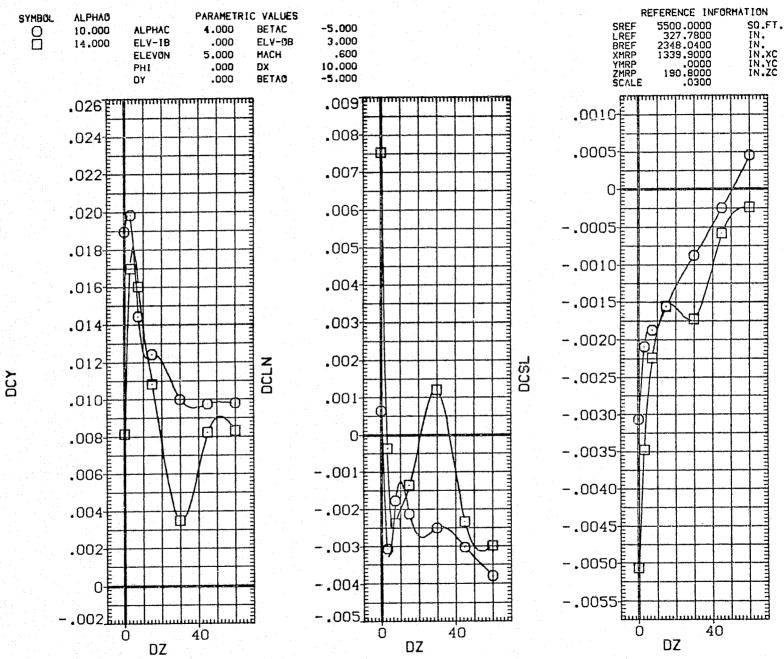


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 291

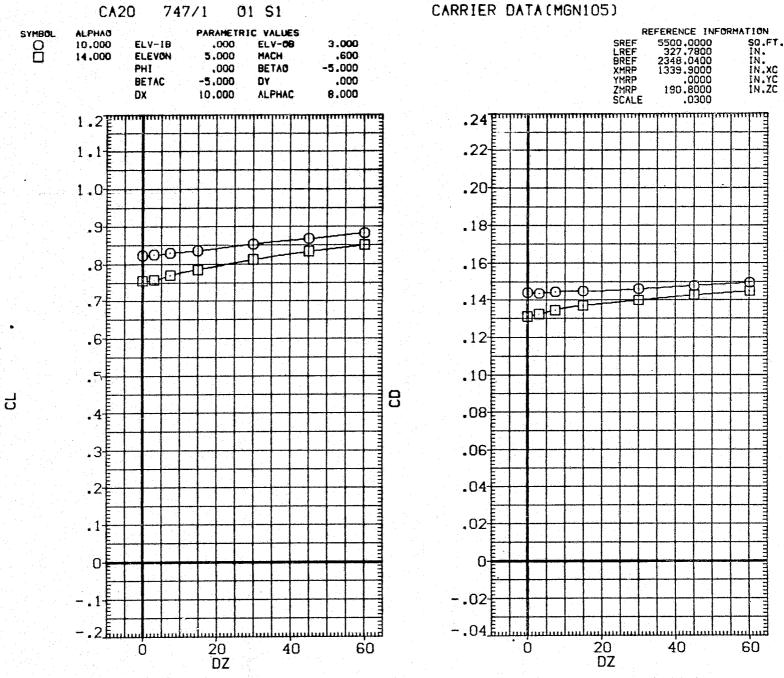


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 292

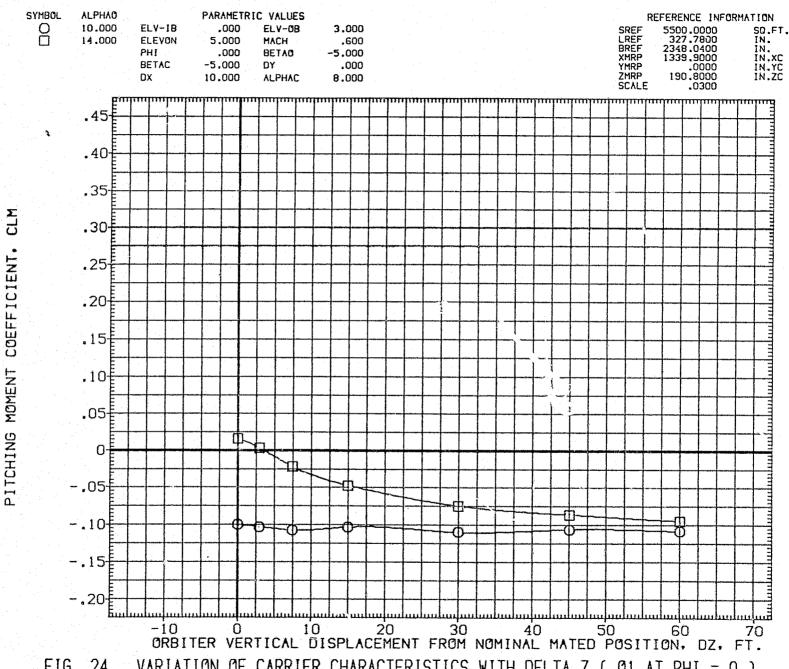


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 293

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 294

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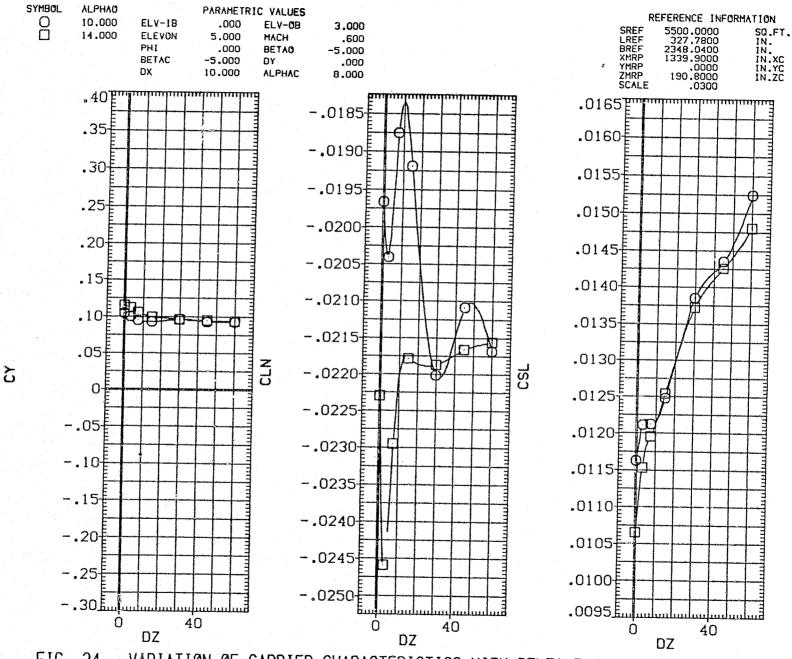


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

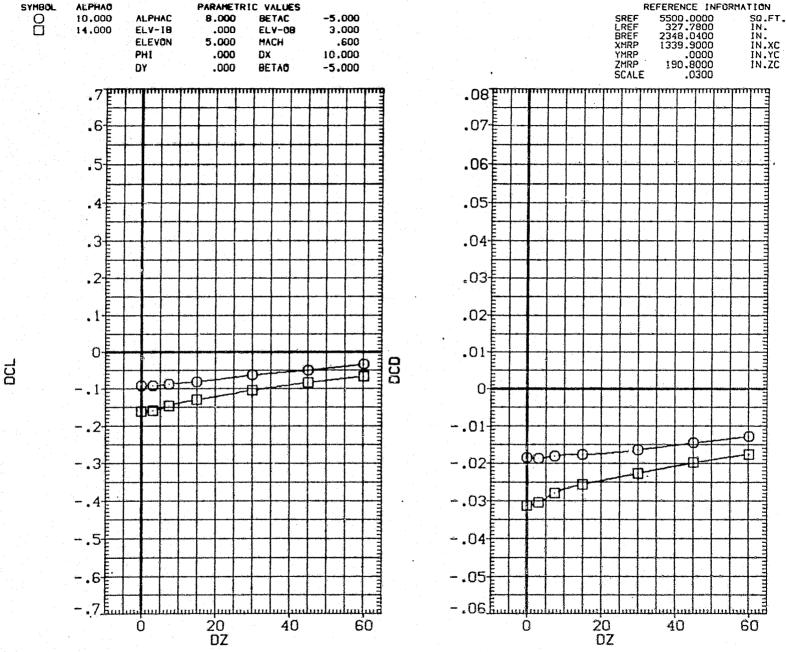
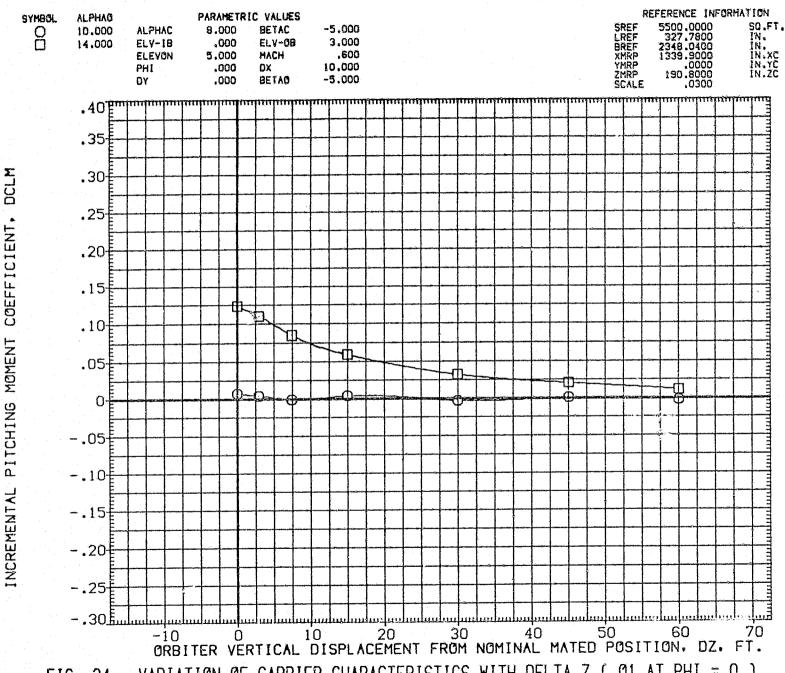


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 296



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24

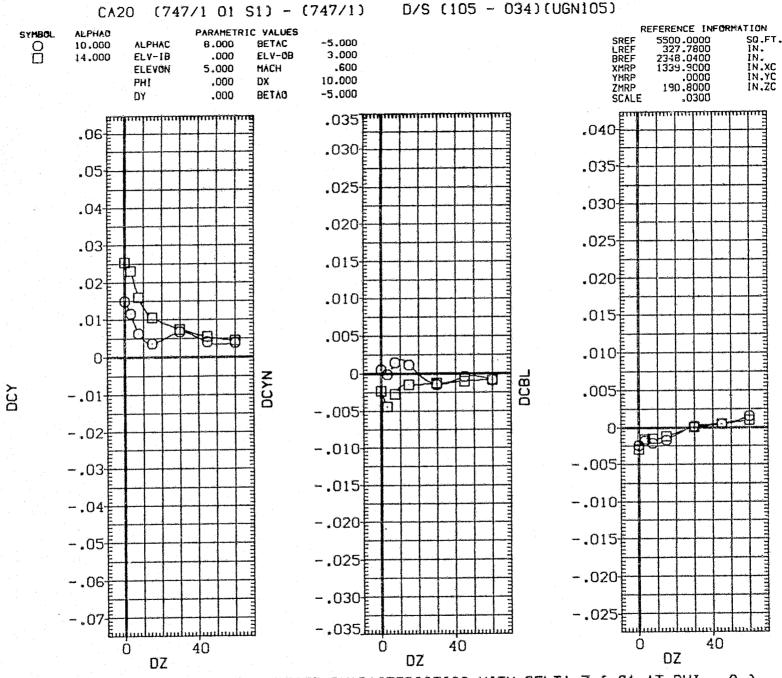


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 298

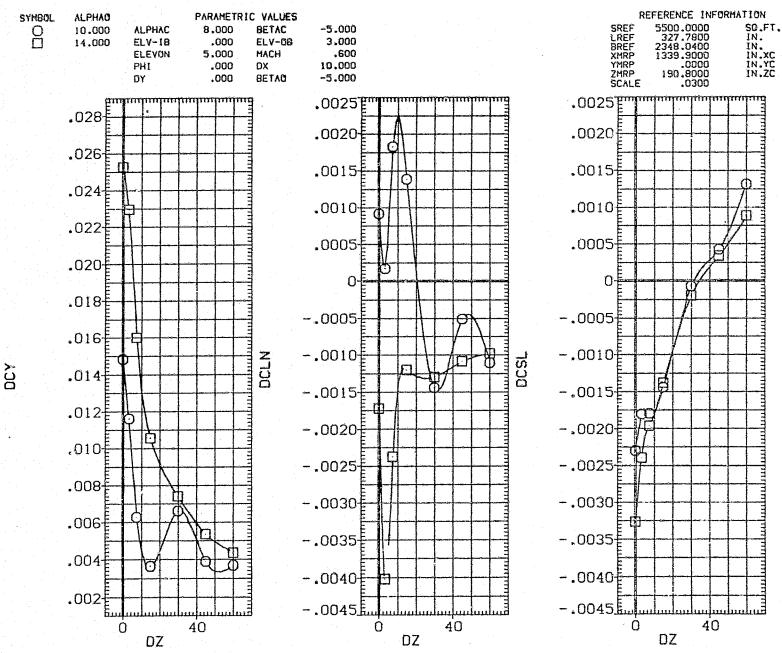


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 299

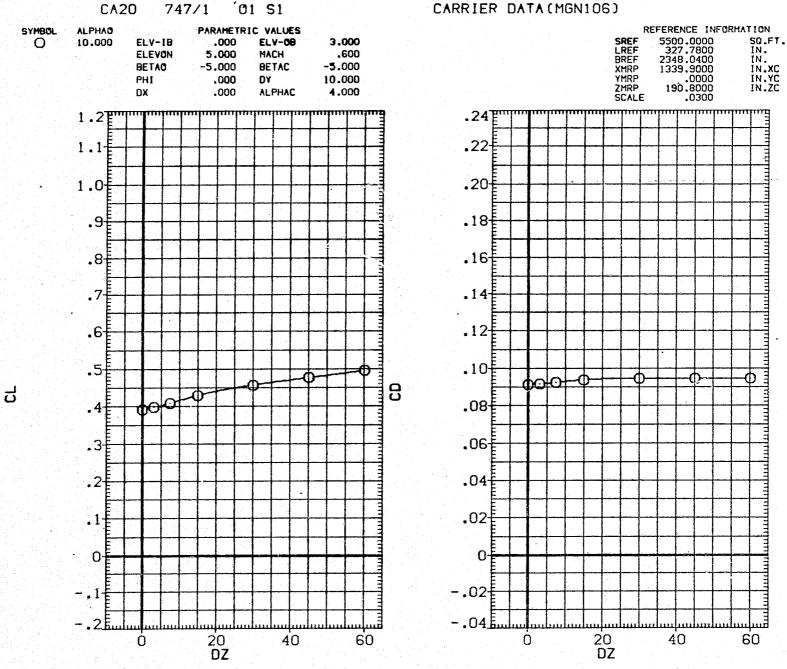


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 300

PAGE 301

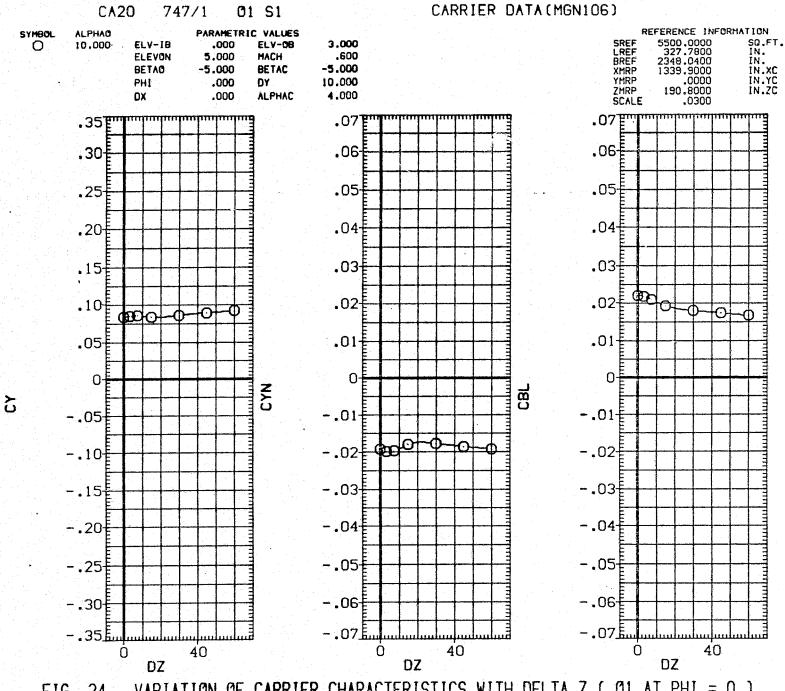
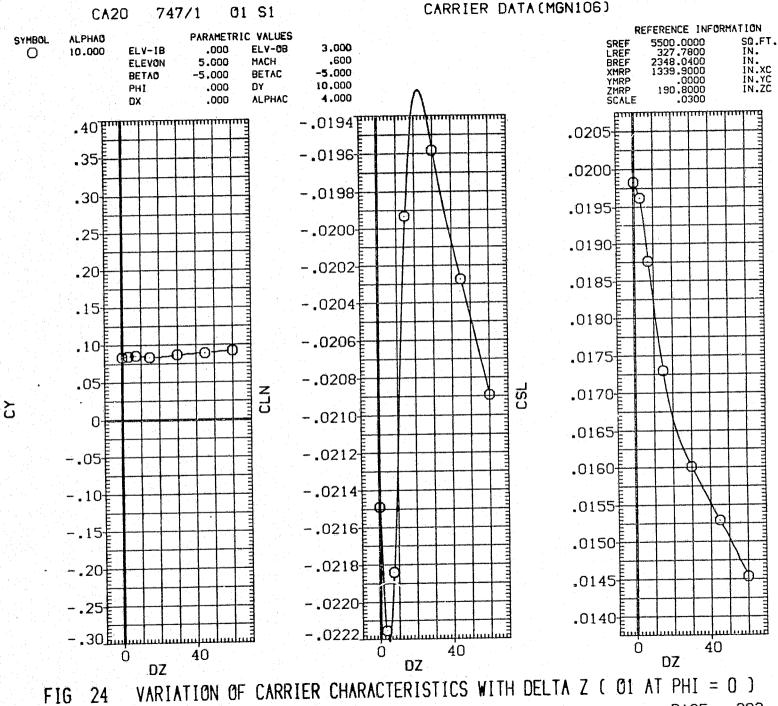


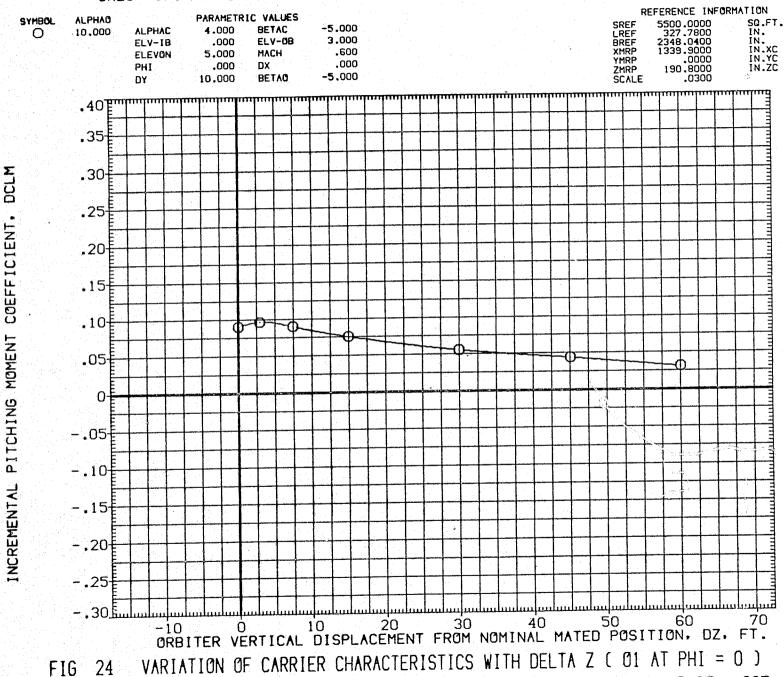
FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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303 PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 304



305

PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 306

DZ

0

40

0

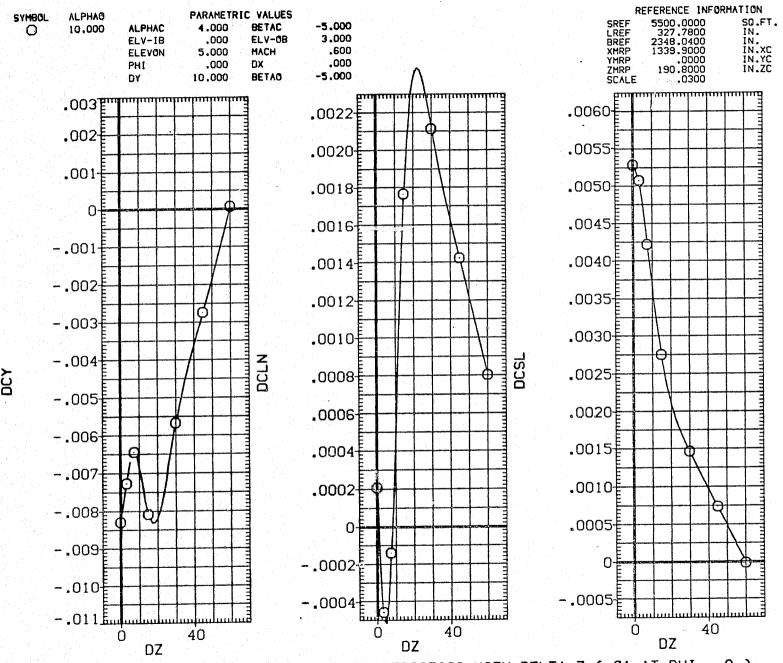
DZ

40

40

0

DZ



24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 307

FIG

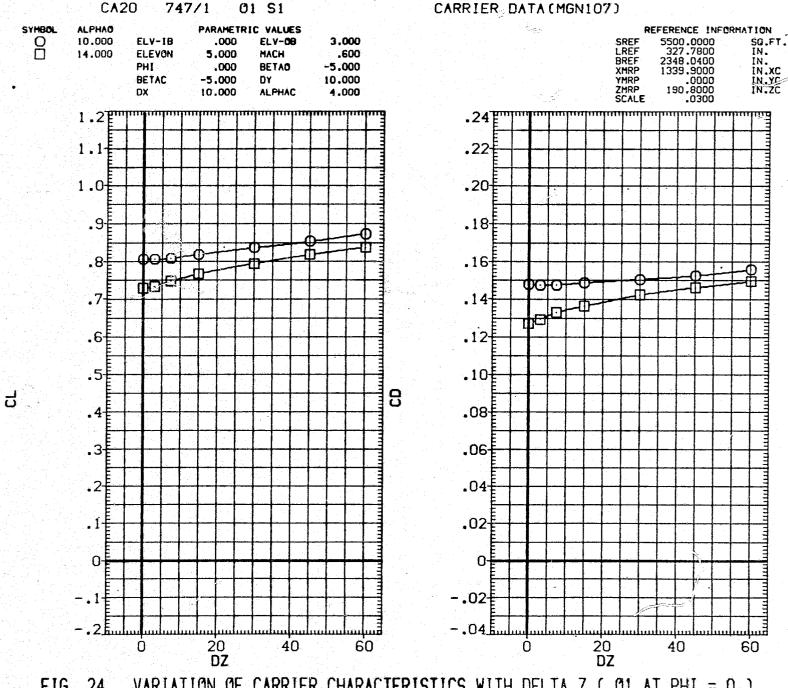


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 308

भागम क्रोक्रामानुबन्द के साथ के का अनुसन होती है। एक एक नामान्य है। एक नामाना क्राक्स क्राक्स अपने अन्य क्रा

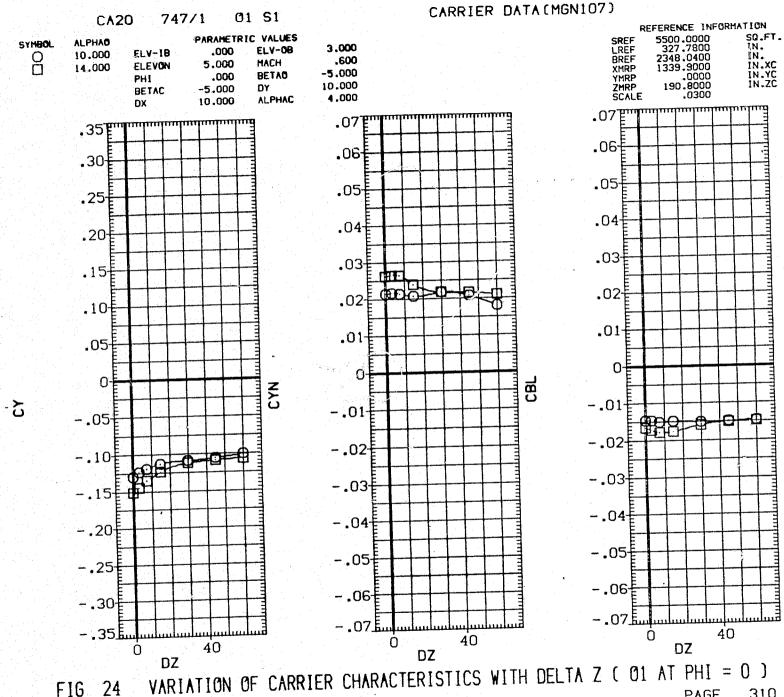


FIG 24 PAGE 310

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 311

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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60

20 **DZ**

0

40

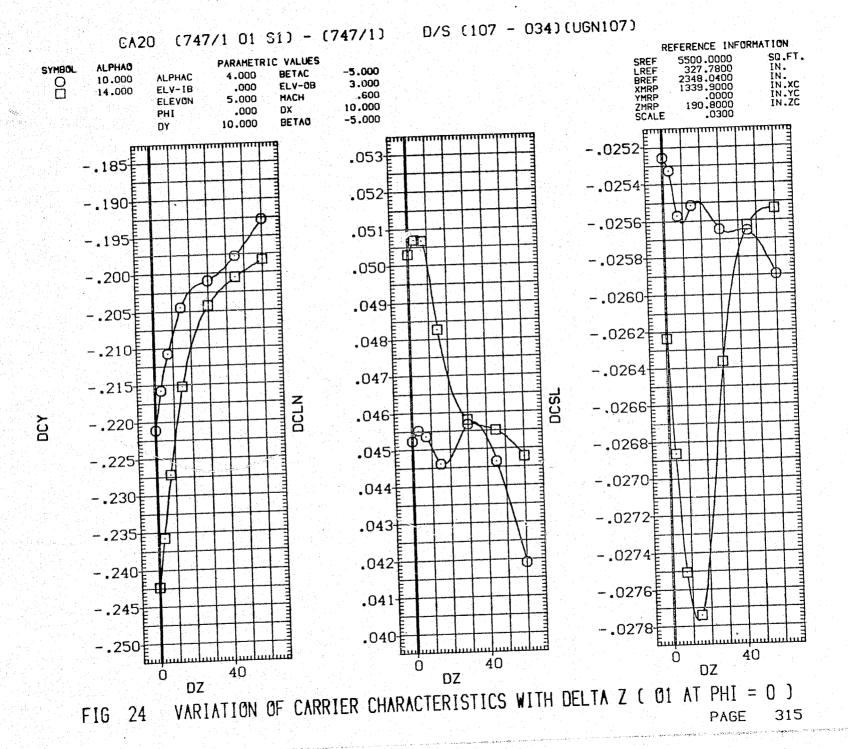
-.06<u>F</u>m

0

20 DZ 40

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 313

FIG 24 PAGE 314



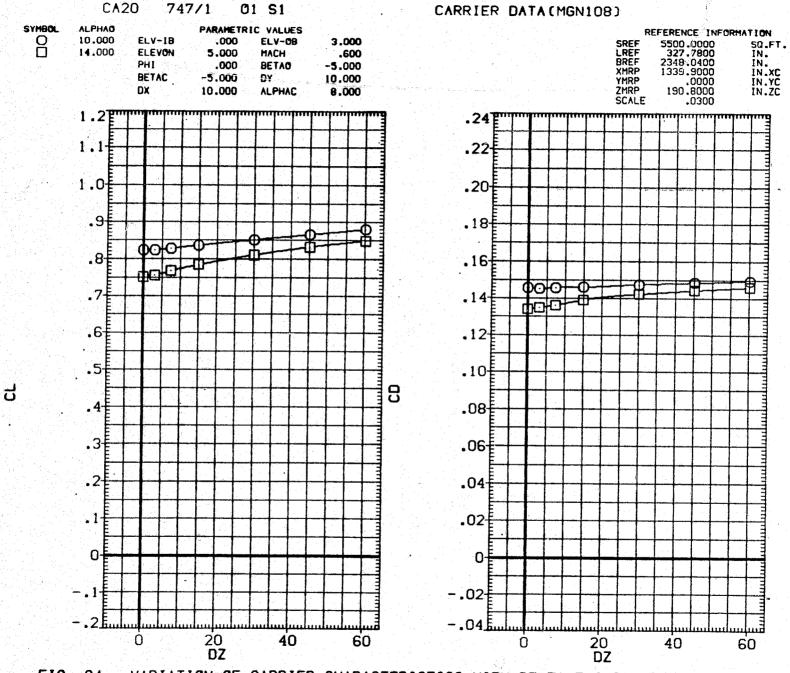
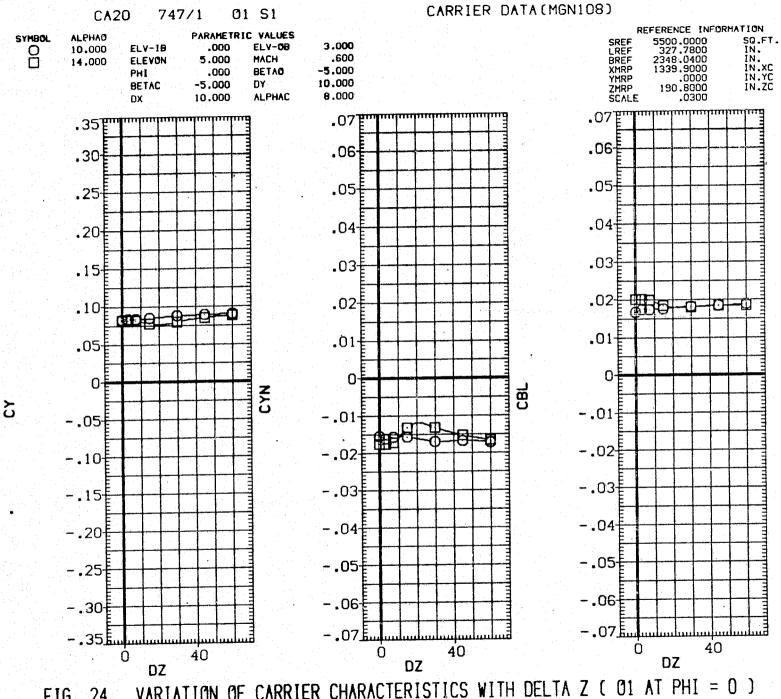


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 318

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 319

DZ

DZ

DZ

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 320

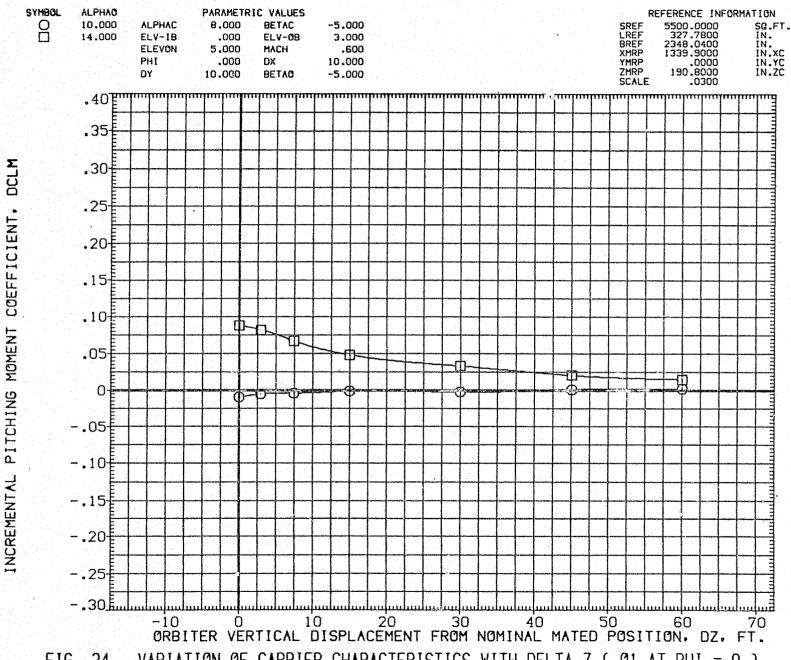


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

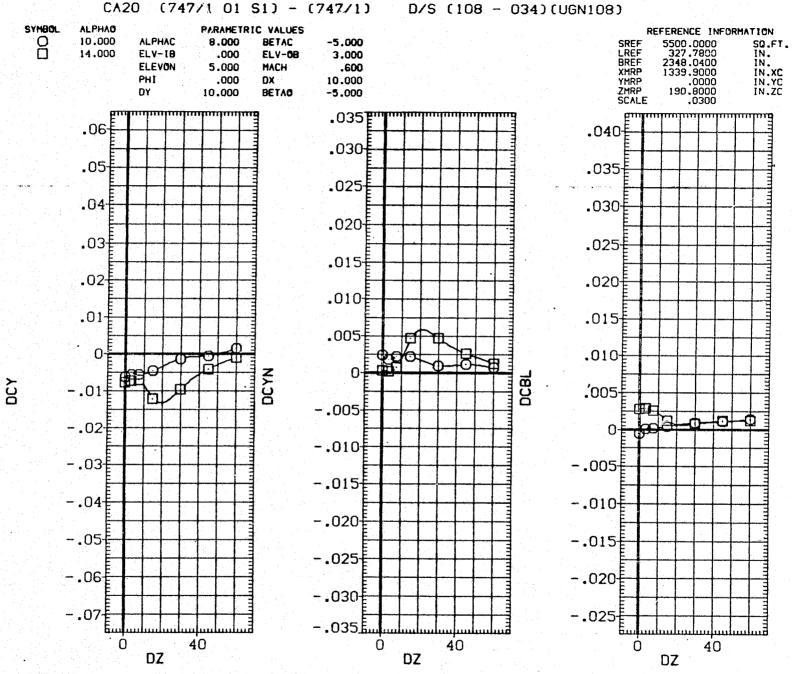


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 322

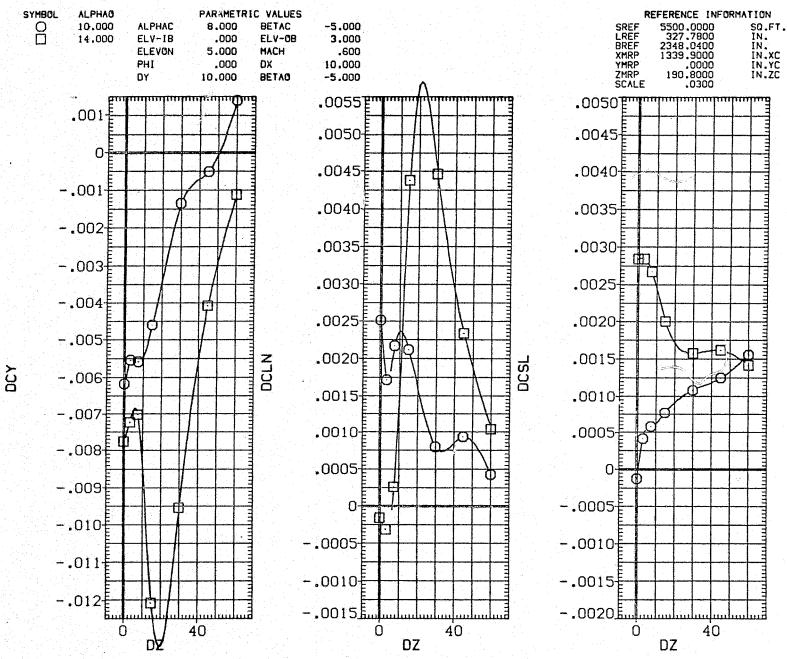
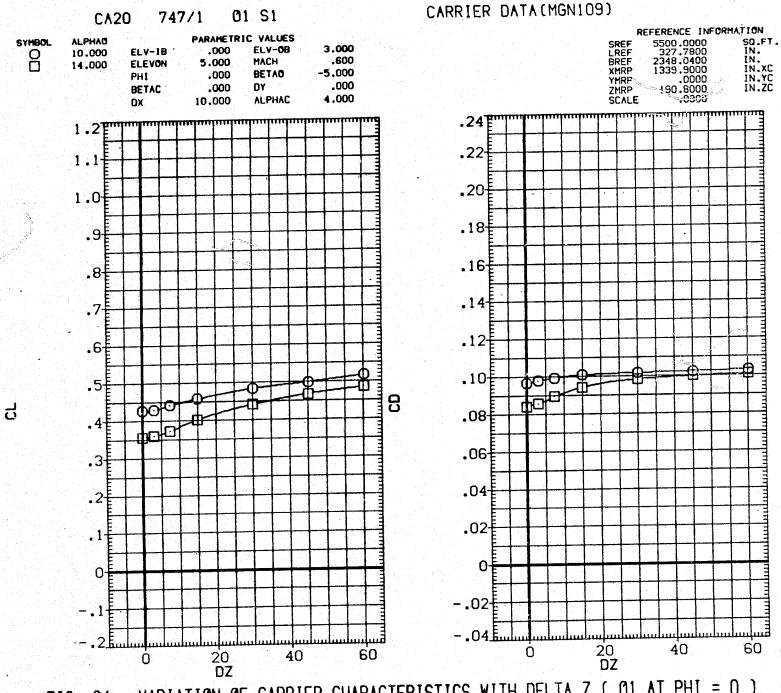


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 324

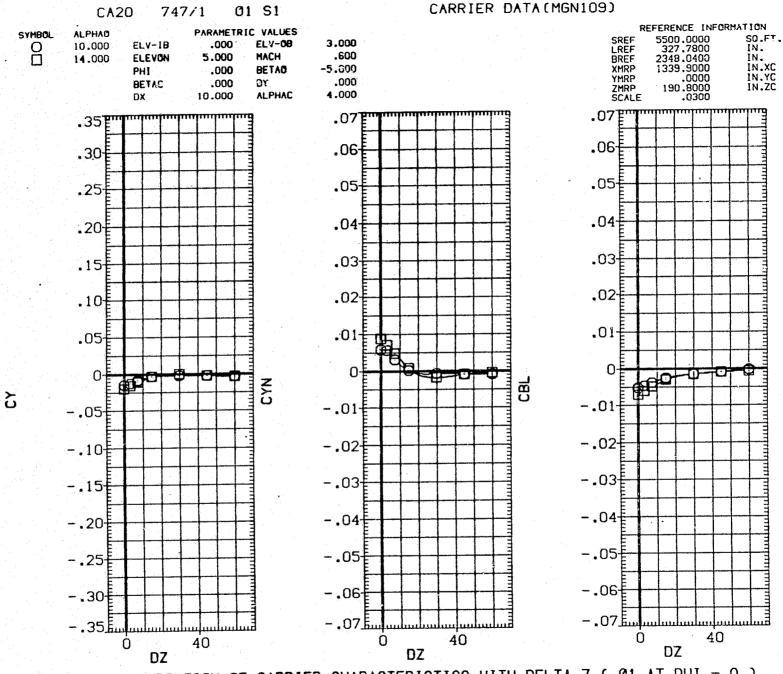
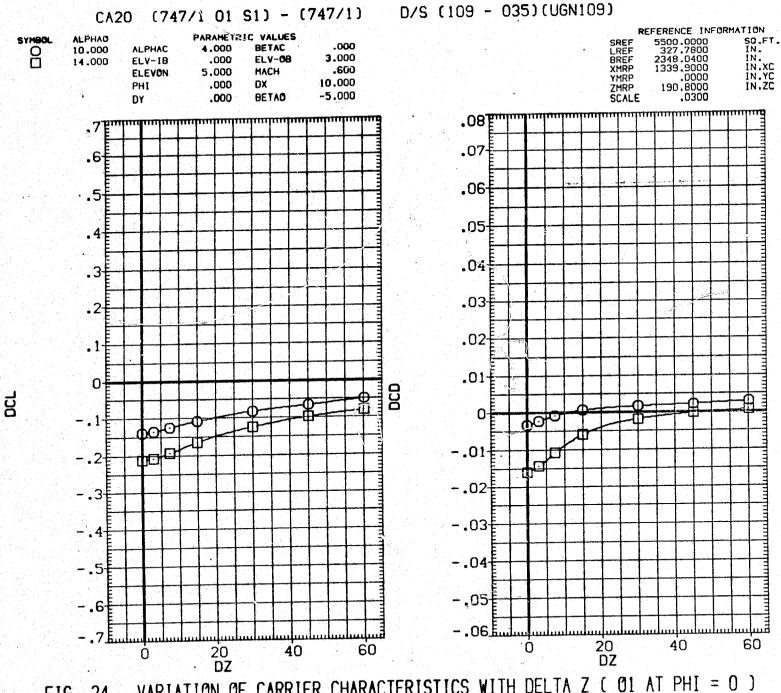


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG PAGE 328

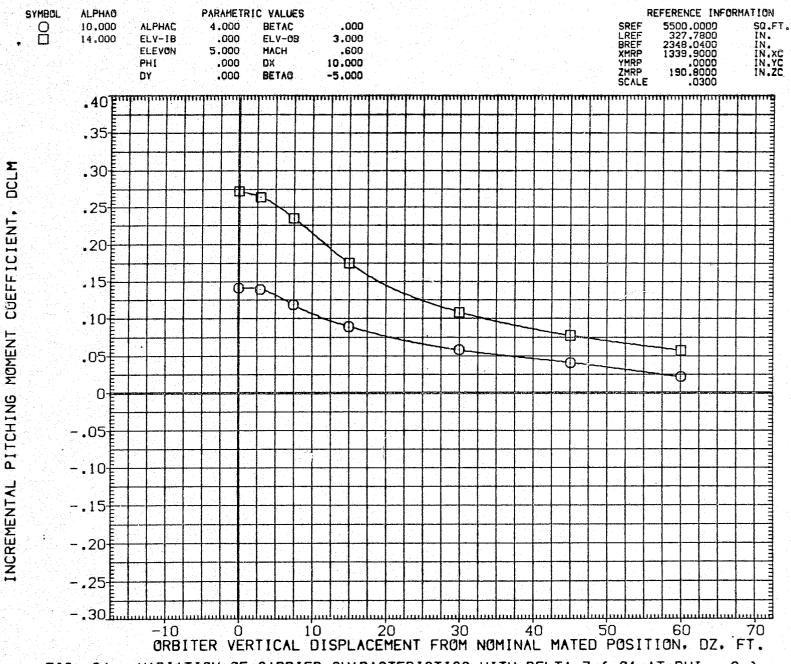
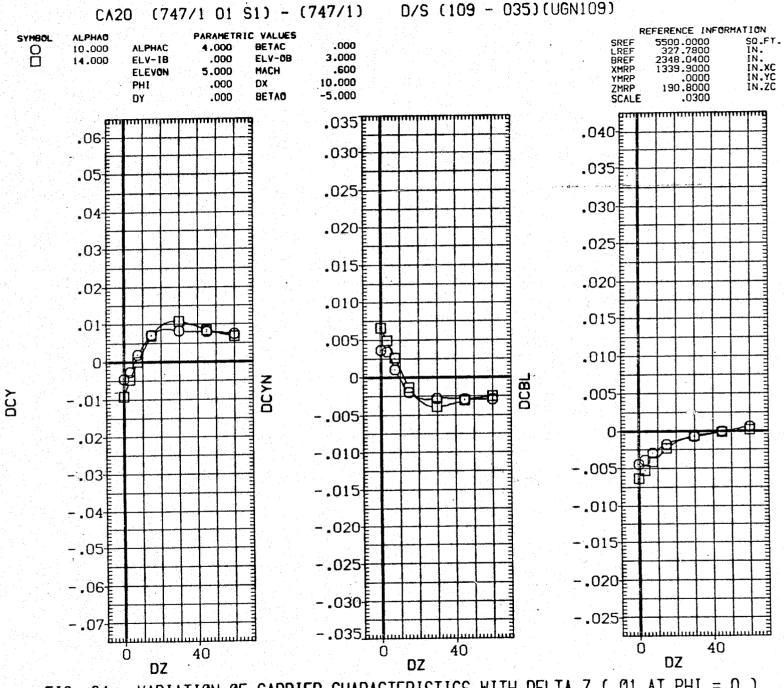


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 329



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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 330

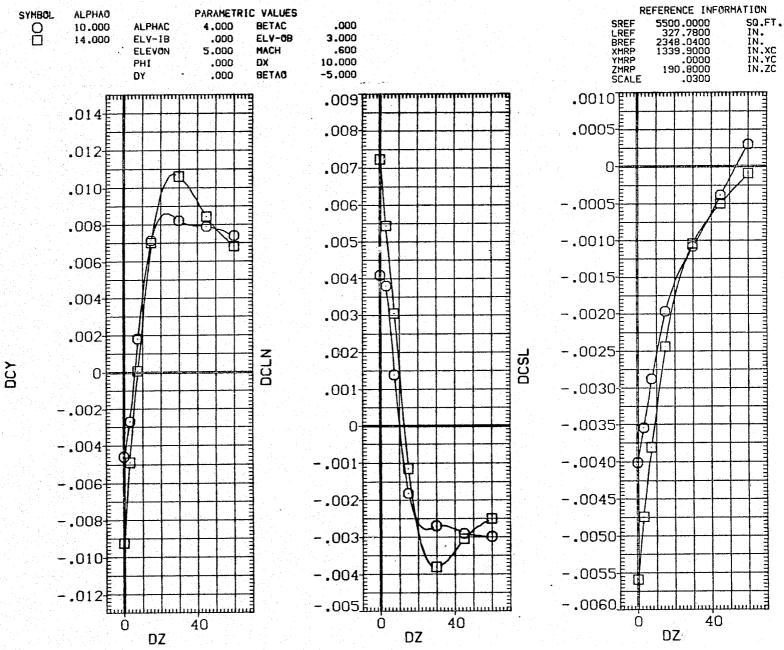


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 331

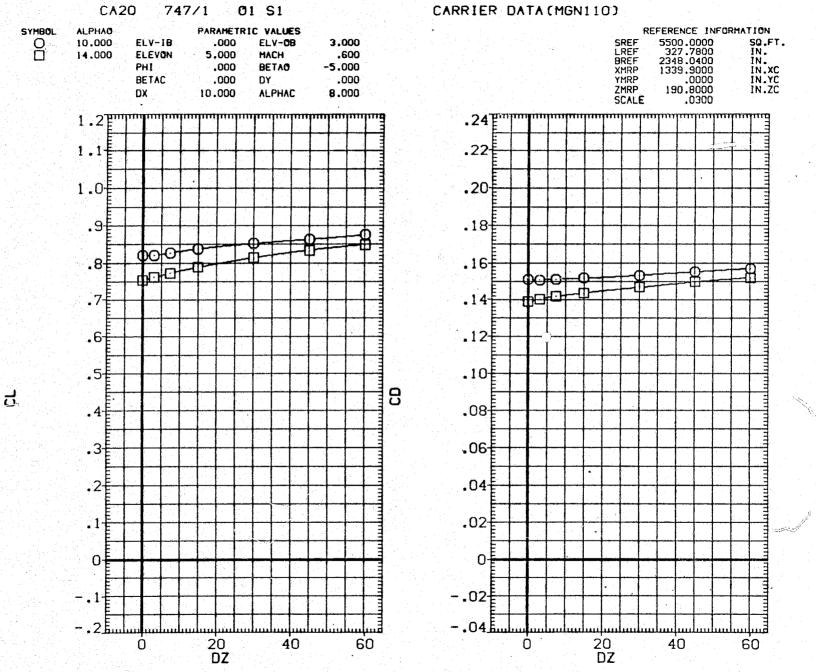
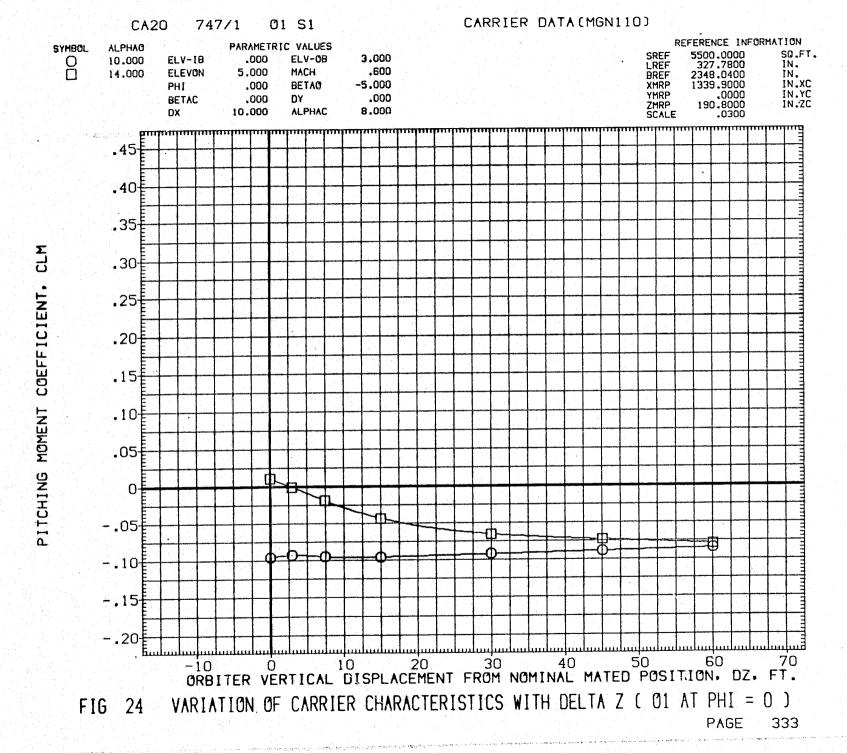


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



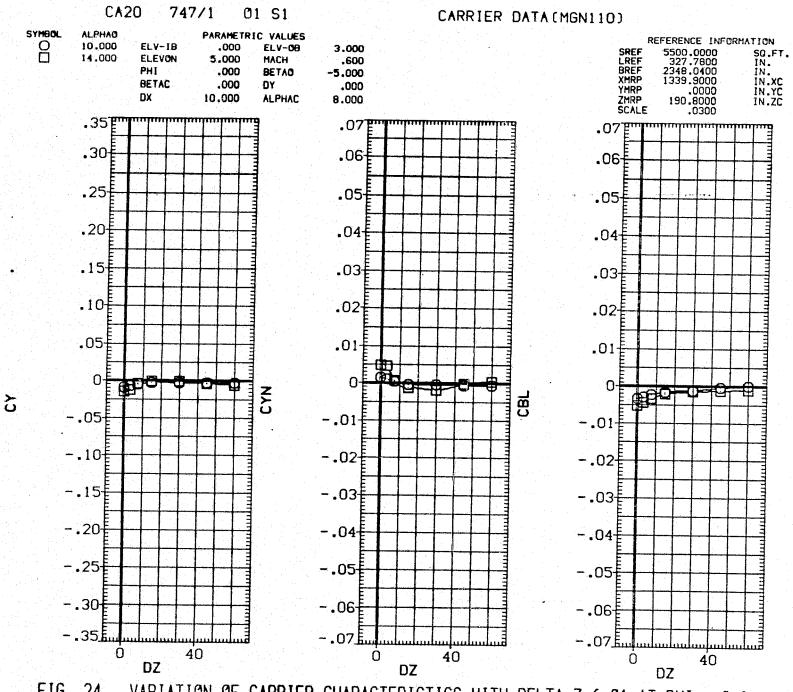


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 334

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 335

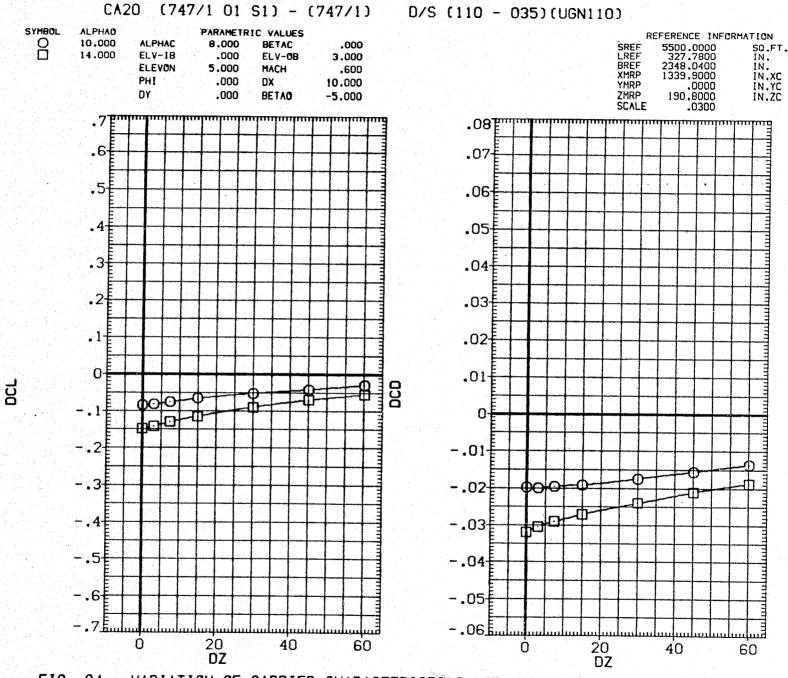


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 336

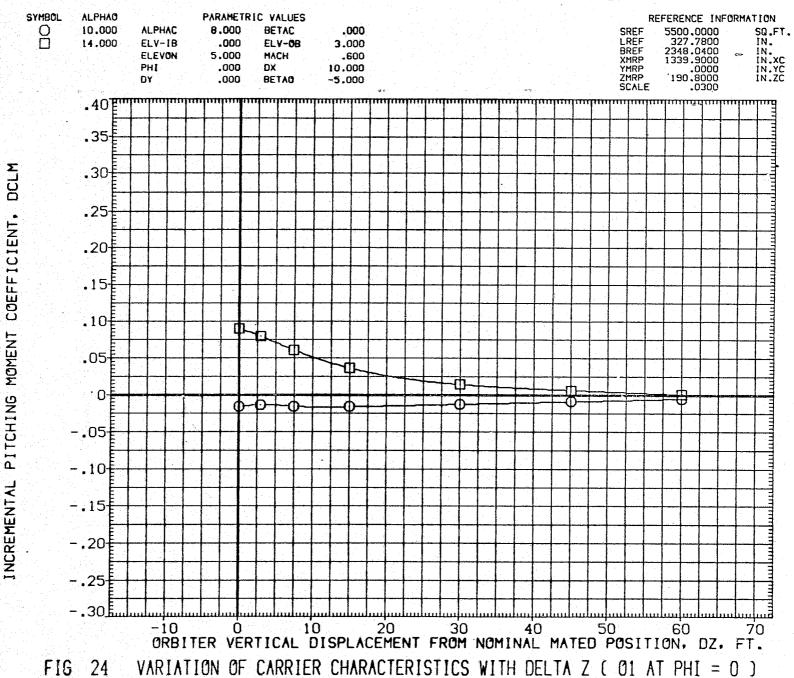


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 338

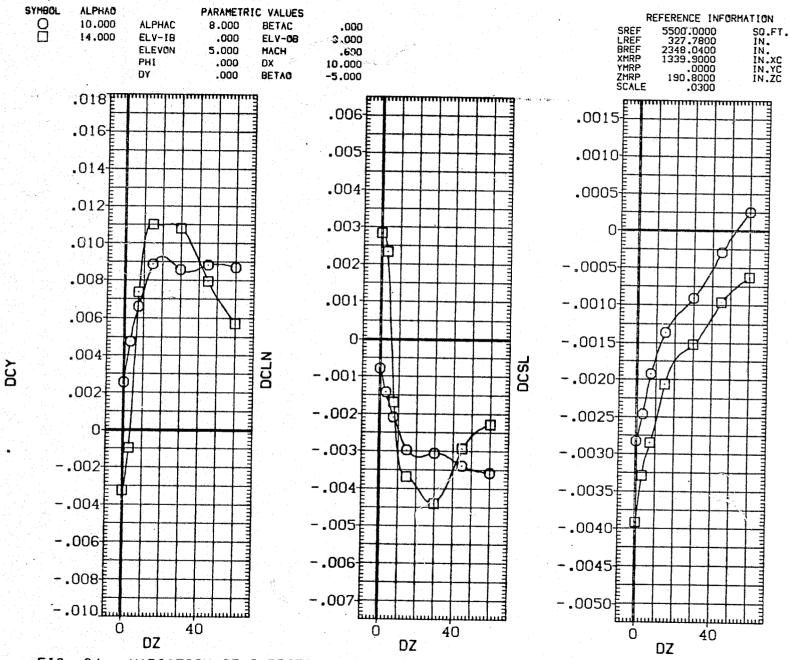


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 339

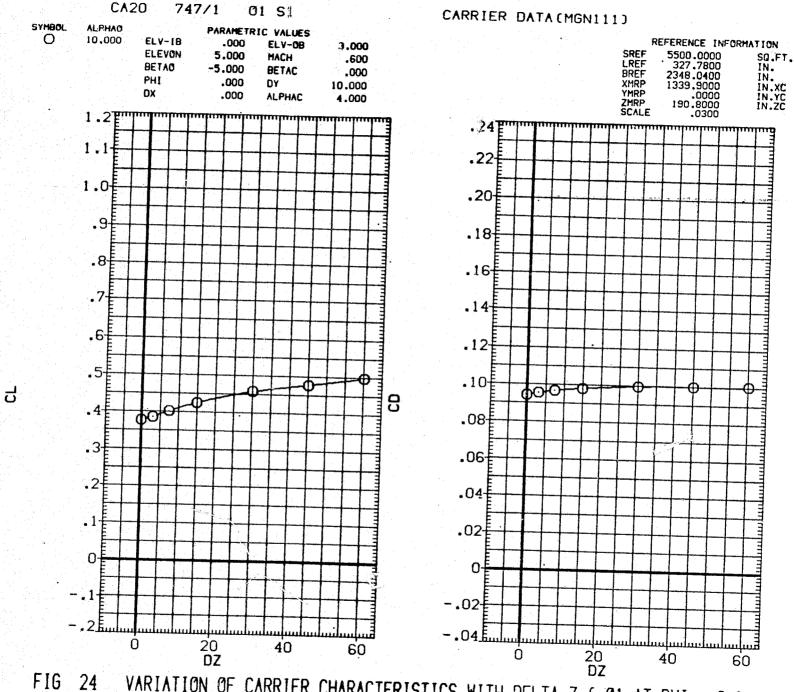


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

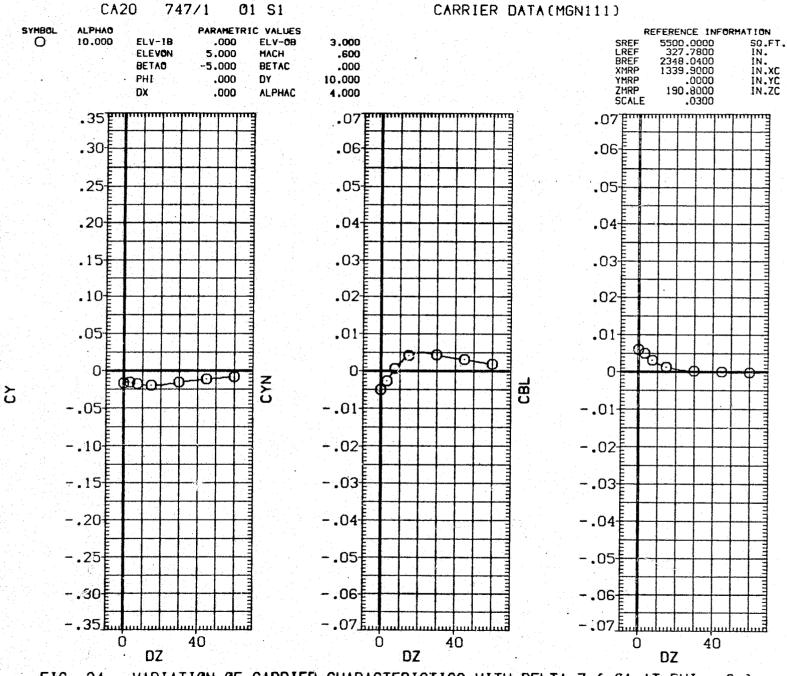


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 342

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 344

60

20 DZ 40

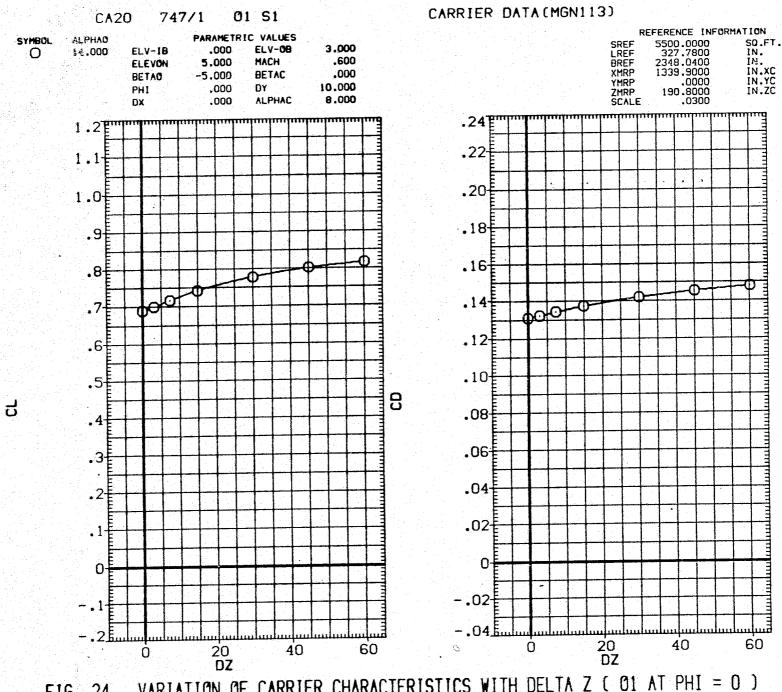
60

20 **DZ**

40

DZ VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0 24 FIG 347 PAGE

DZ



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 348

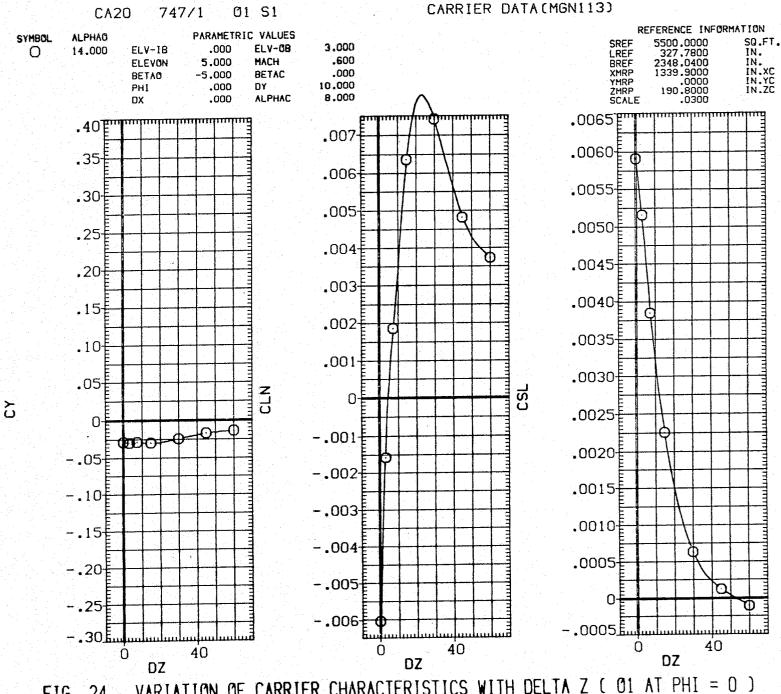
COEFFICIENT

MOMENT

PITCHING

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

24 FIG 350 PAGE



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 351

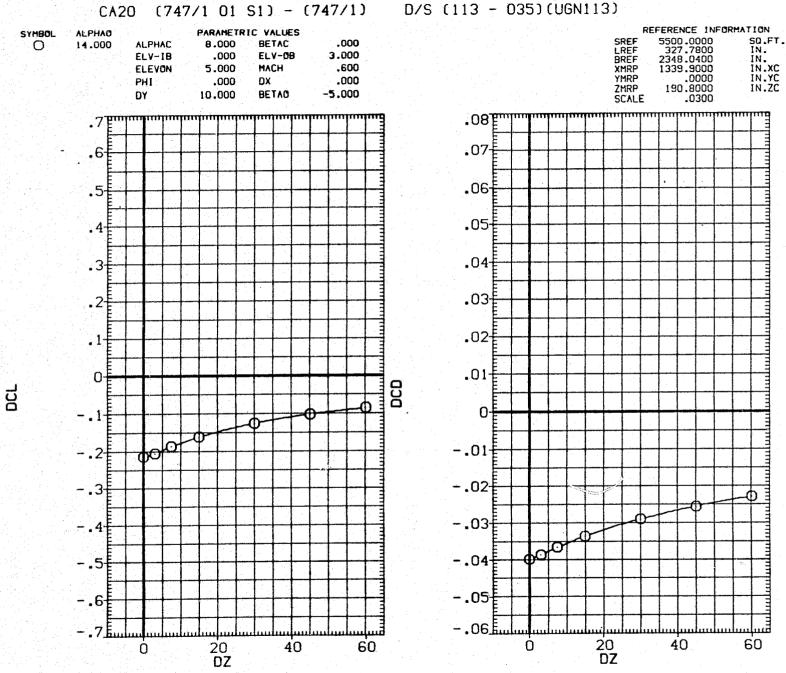
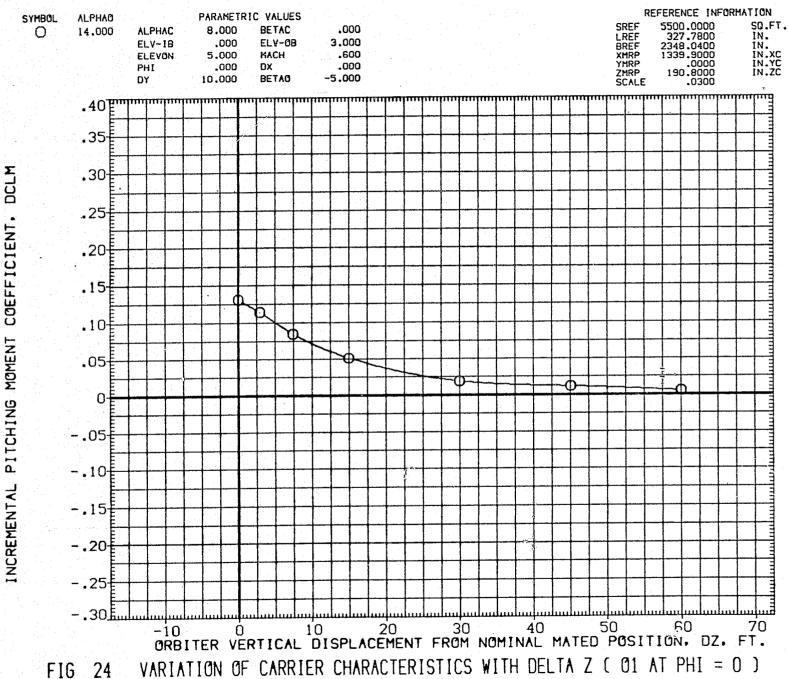


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 352

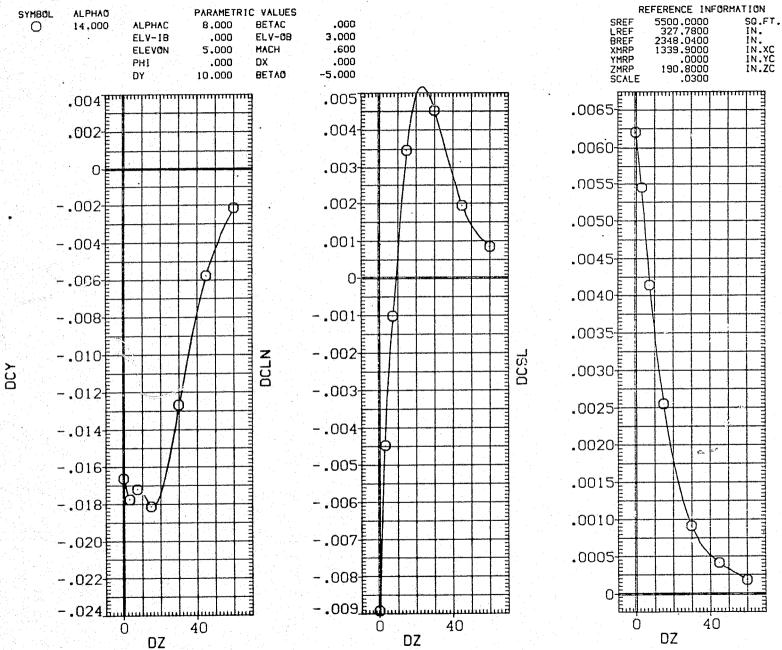
F 7



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) PAGE 353

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 354

DZ



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24

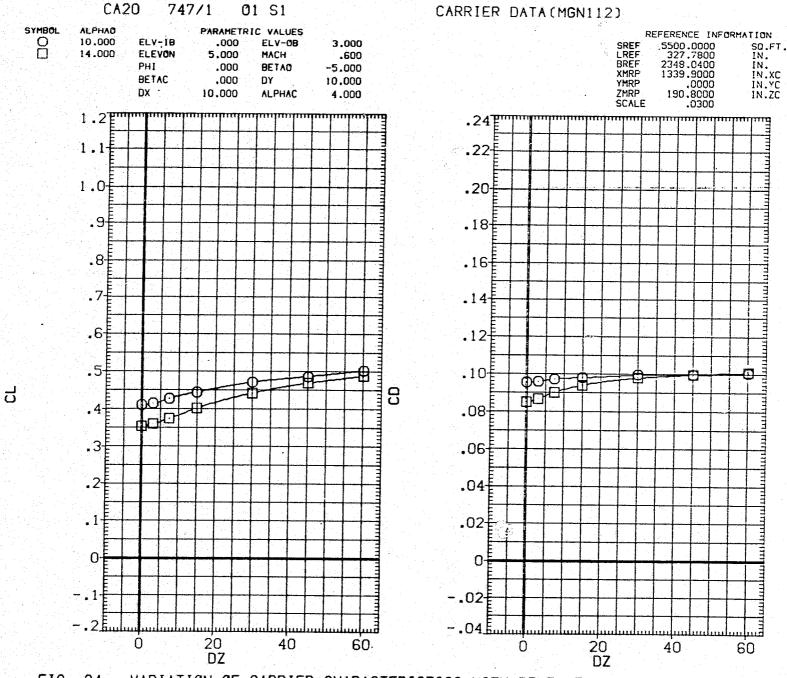


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 356



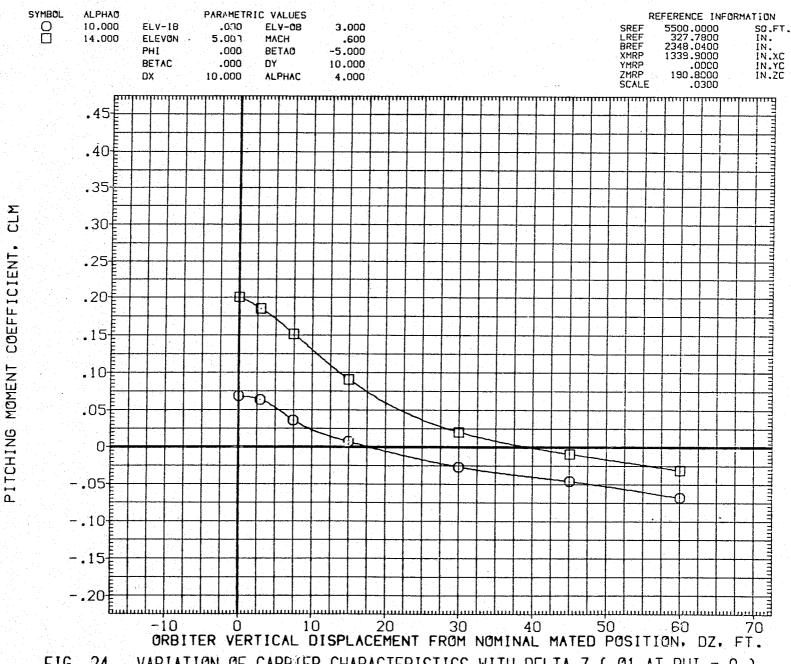


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 358

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 359

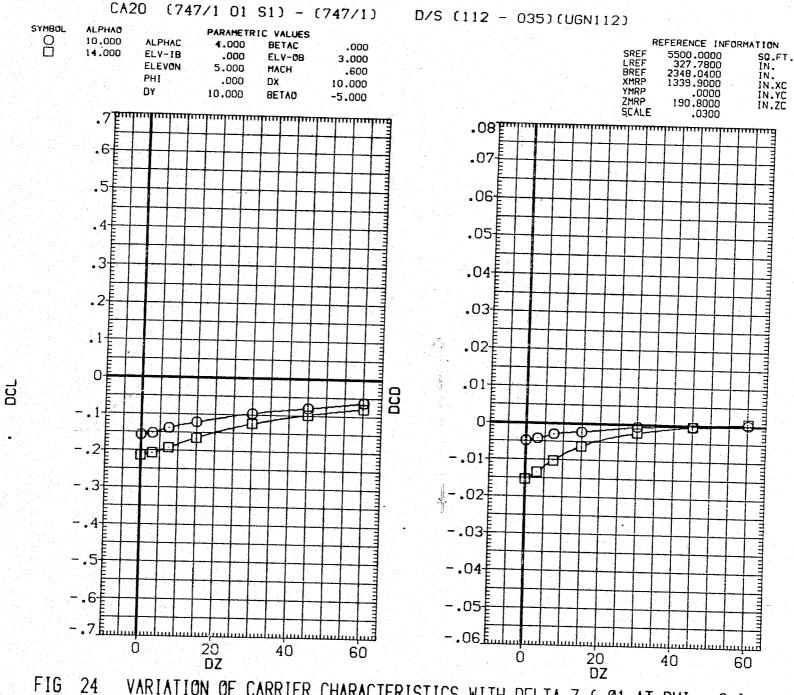
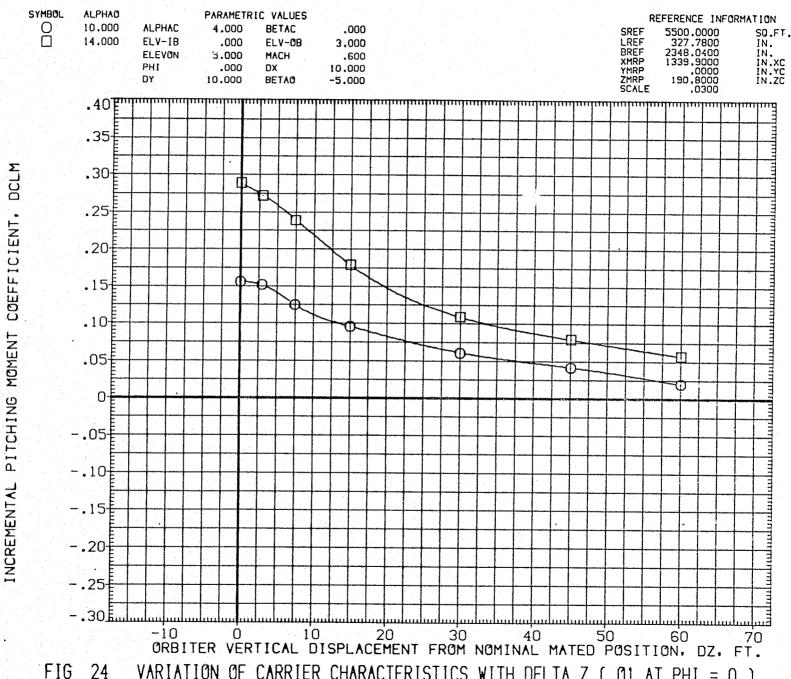


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 360



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 362

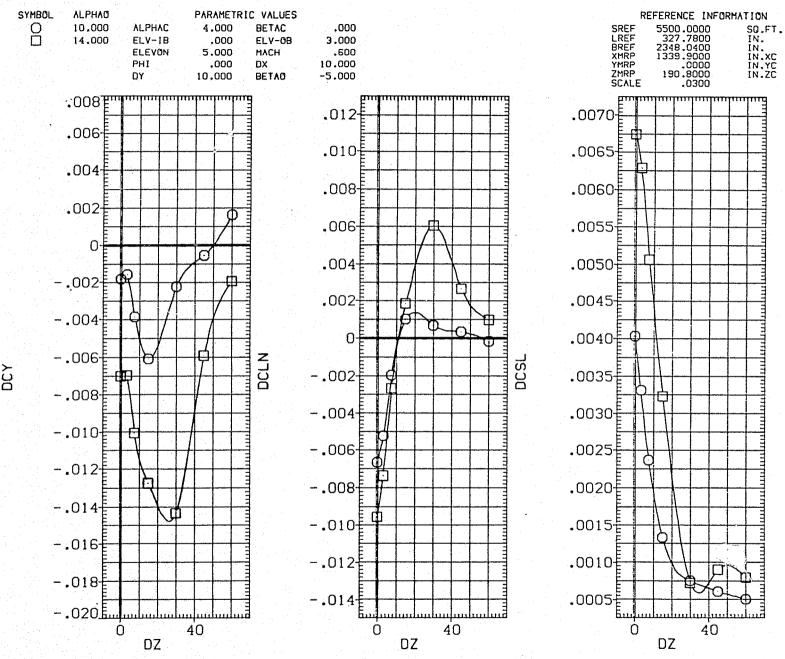


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 363

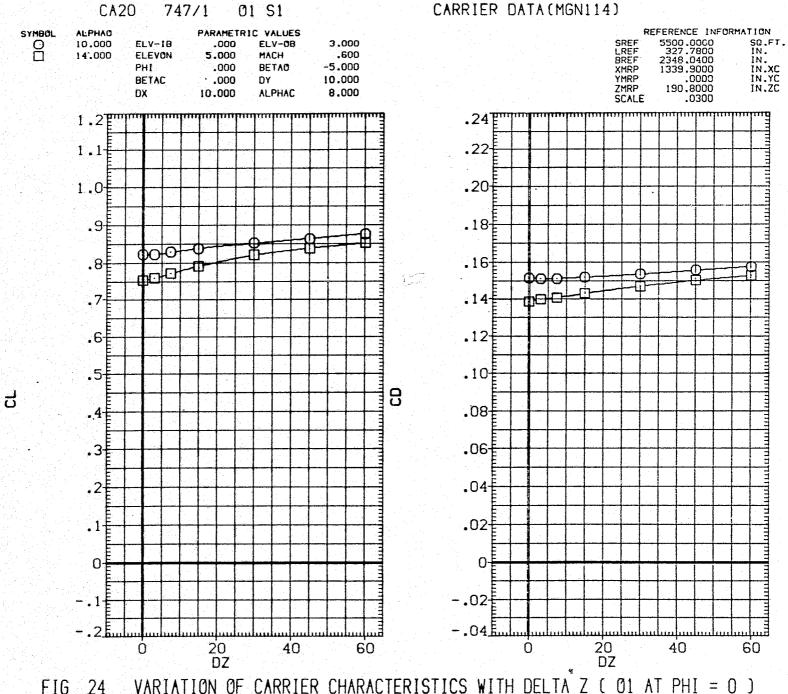
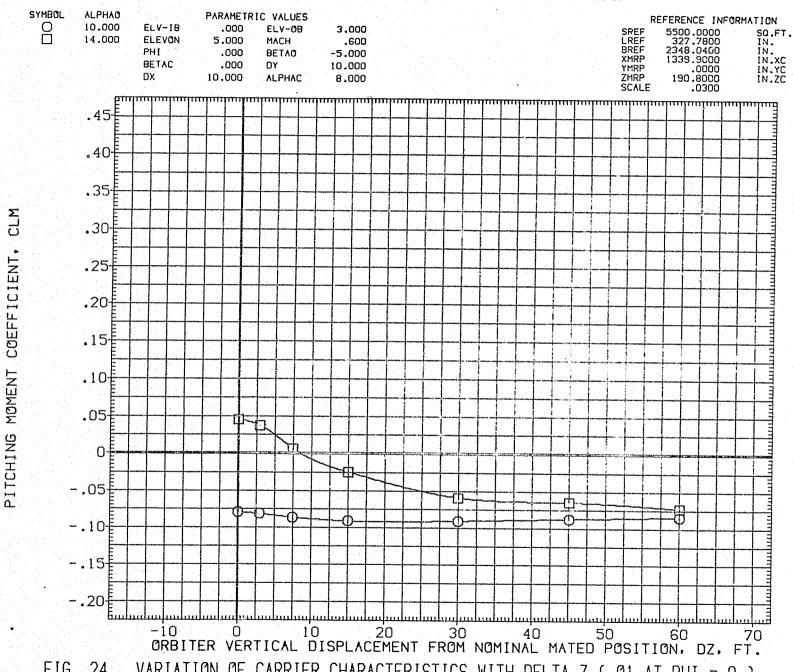


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) PAGE 364



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 367

DZ

DZ

DZ

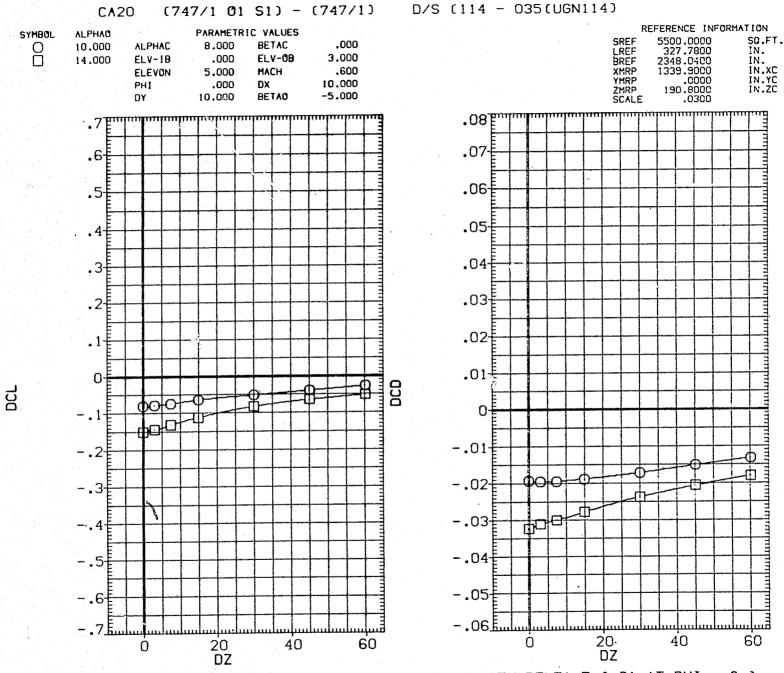


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 368

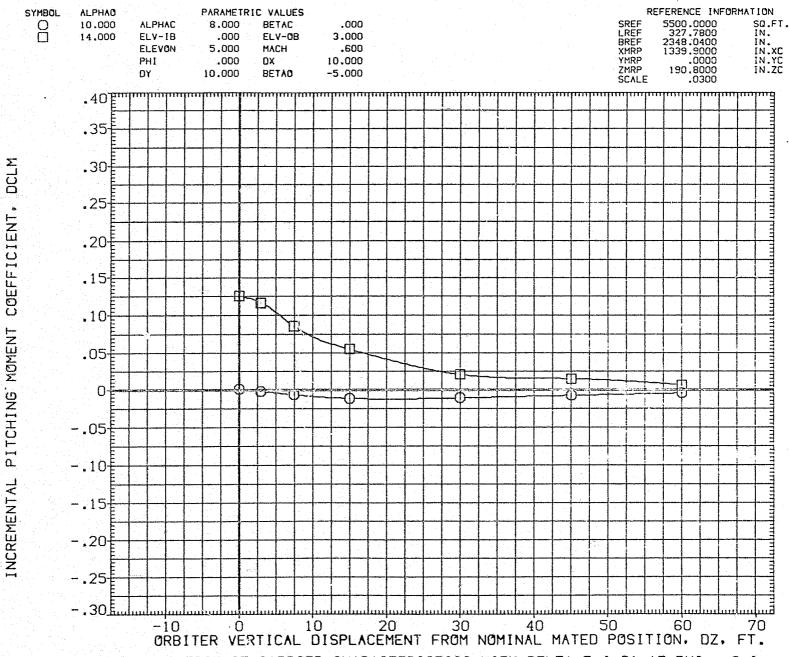


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 370 PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 372

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 373

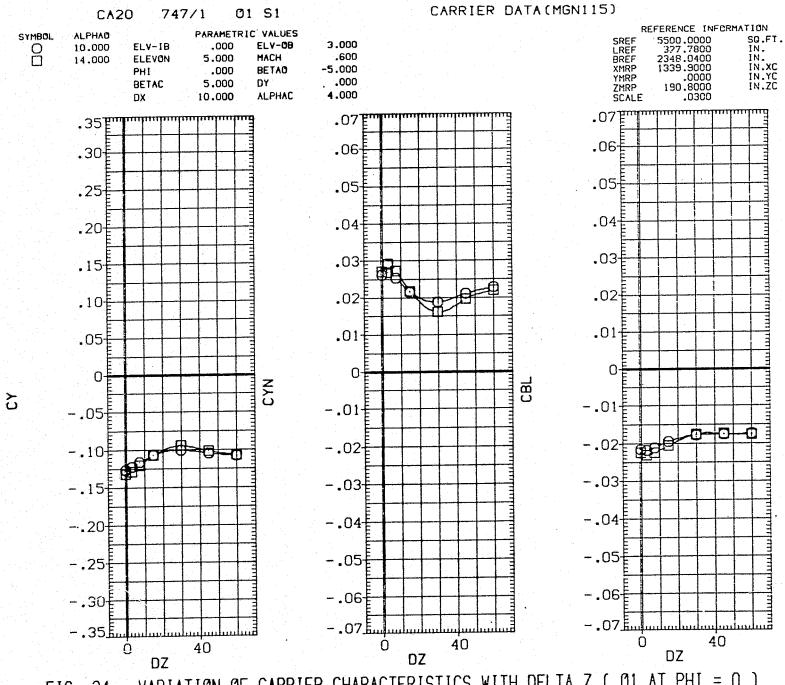


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 374

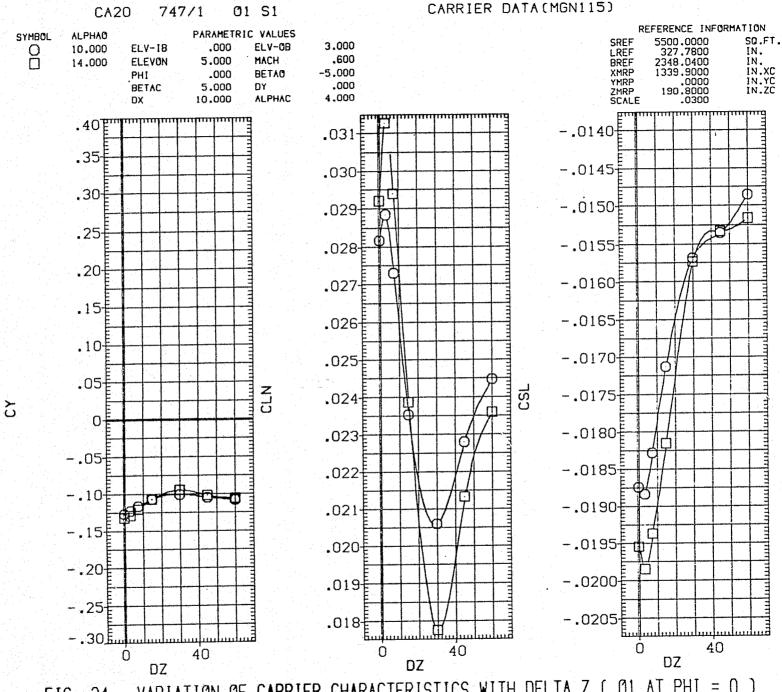


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 375

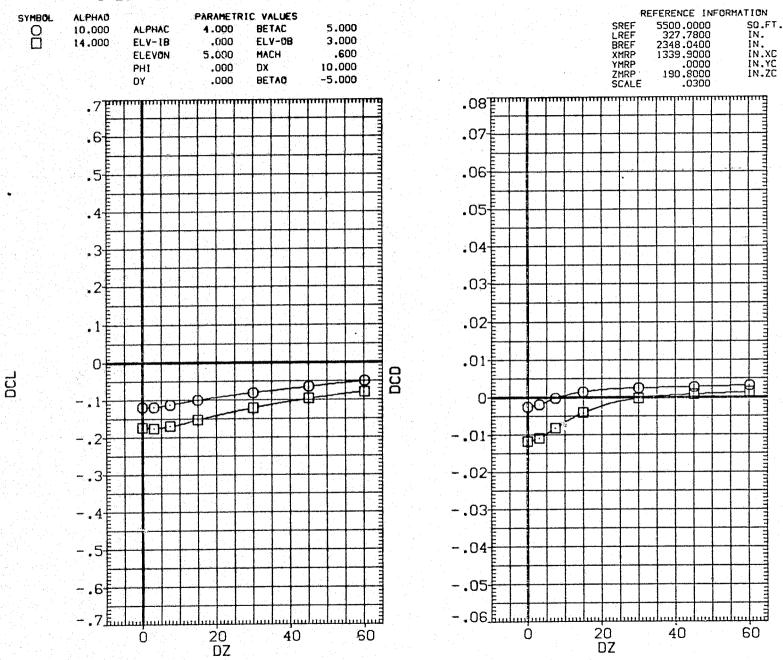


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 376

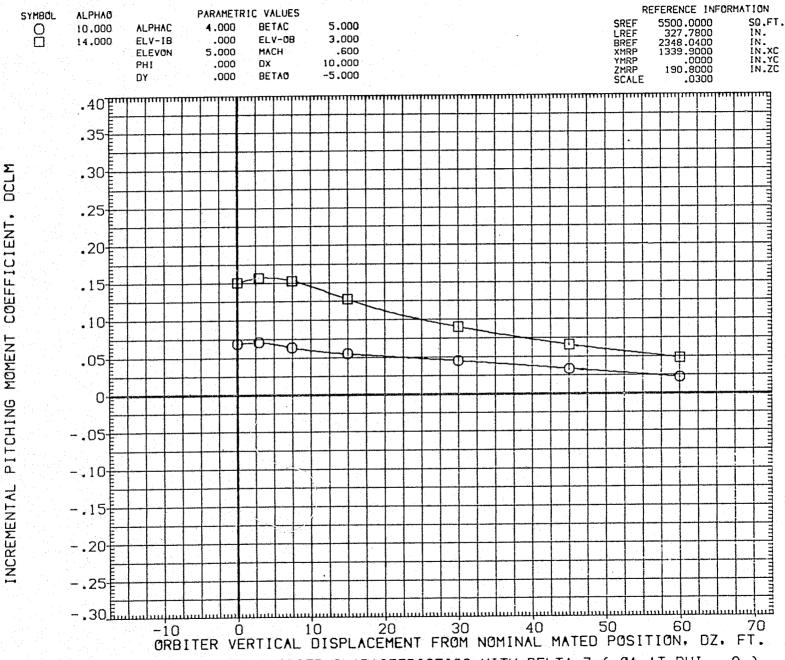


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 377

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 378

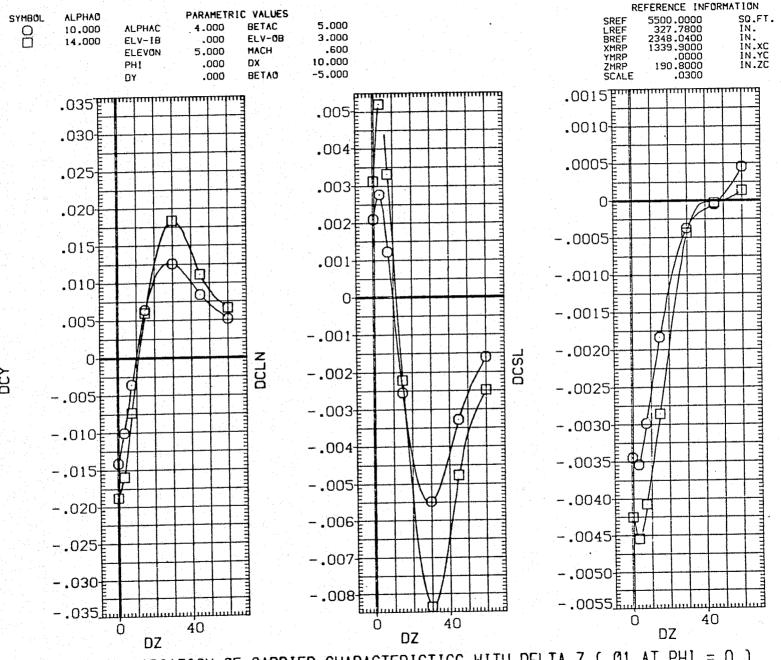


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 379

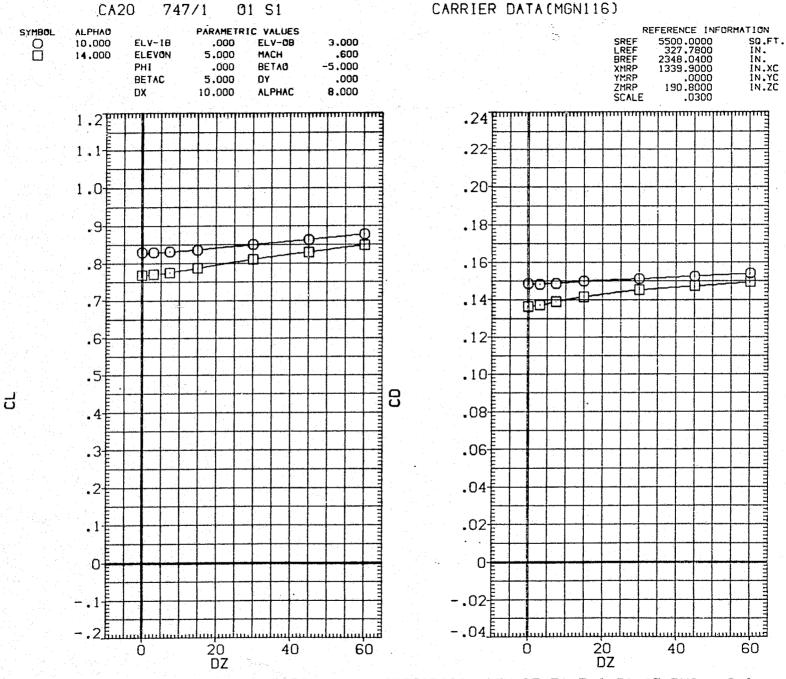


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 380

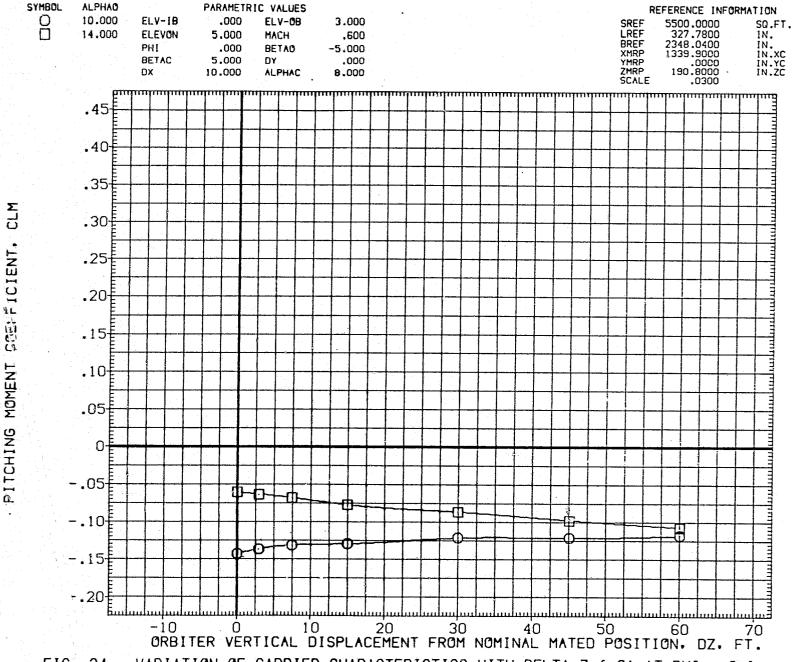


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 381

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 382

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 383

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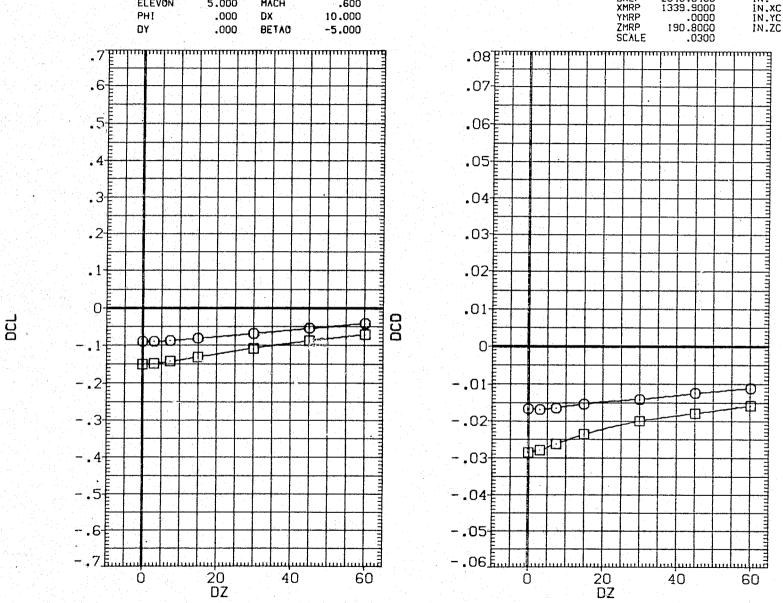


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) PAGE 384

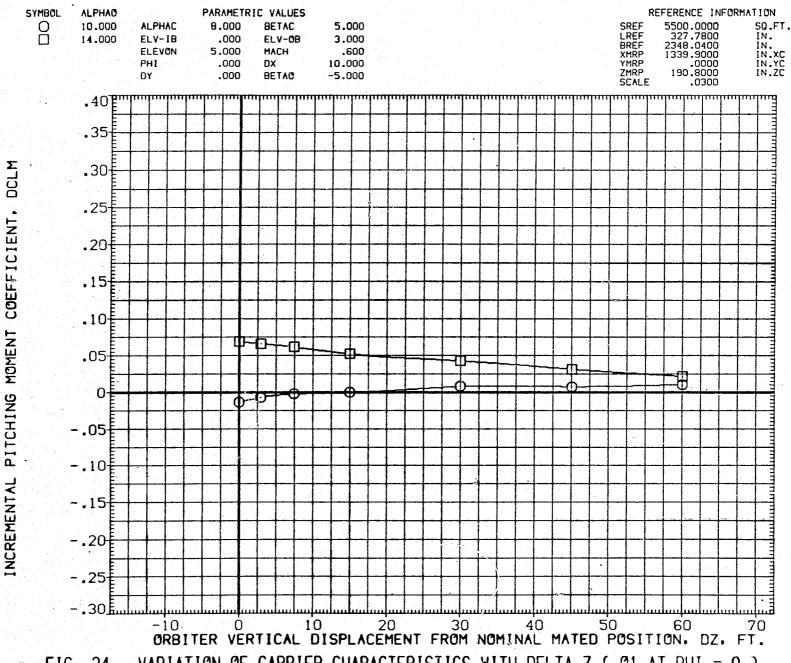


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

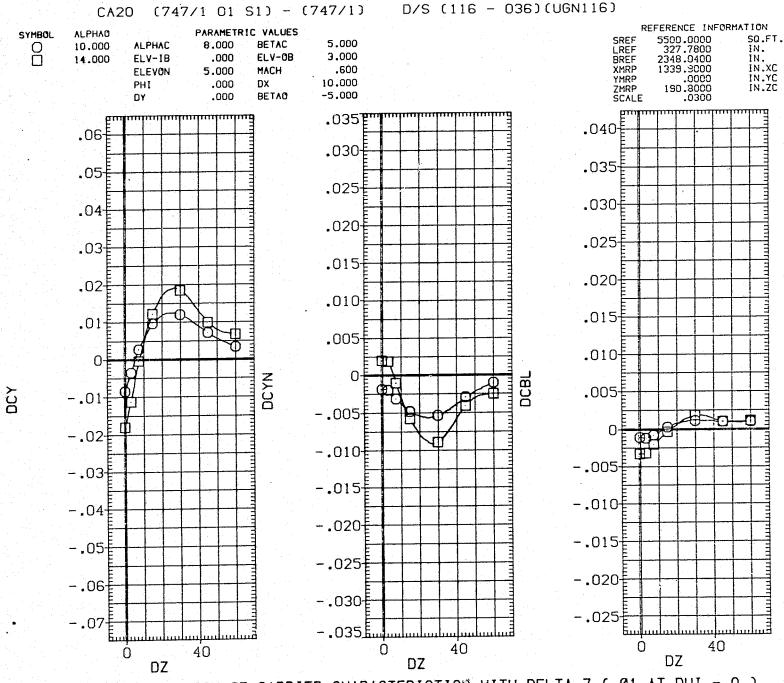


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 386

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24

DZ

40

0

DZ

40

DZ

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 388

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 389

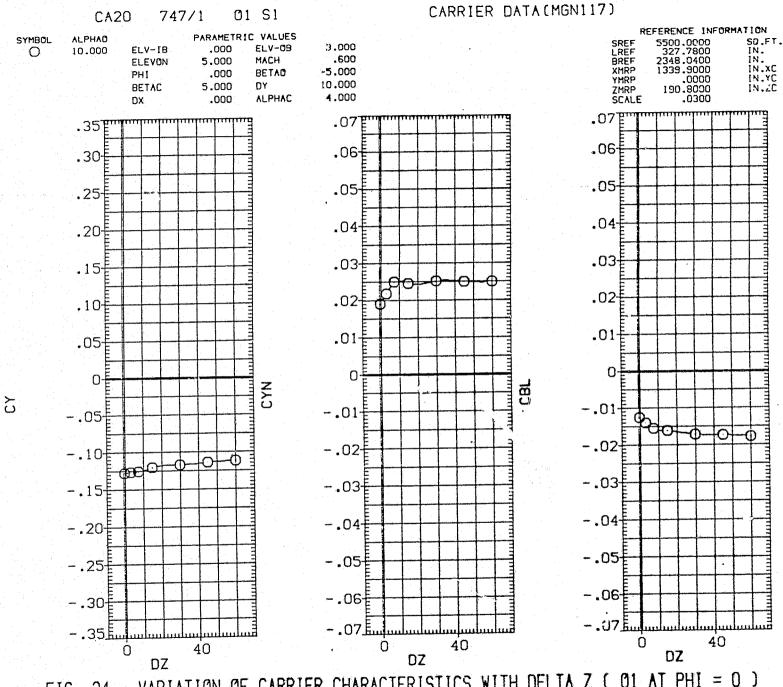


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 390

PAGE

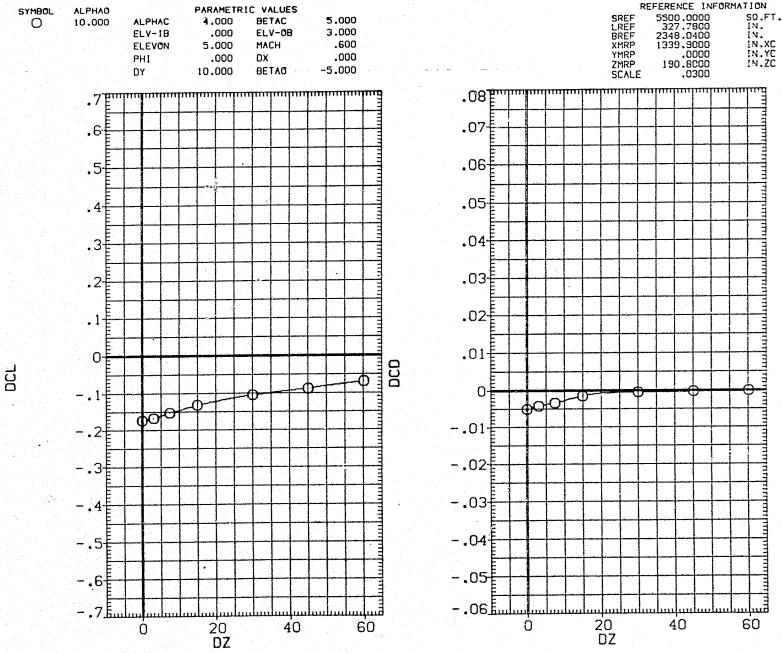


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 392

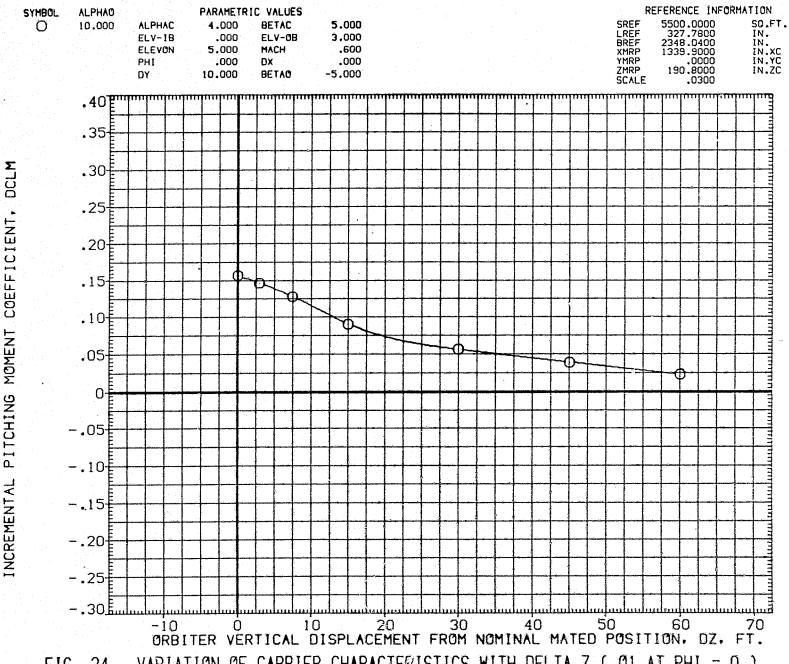
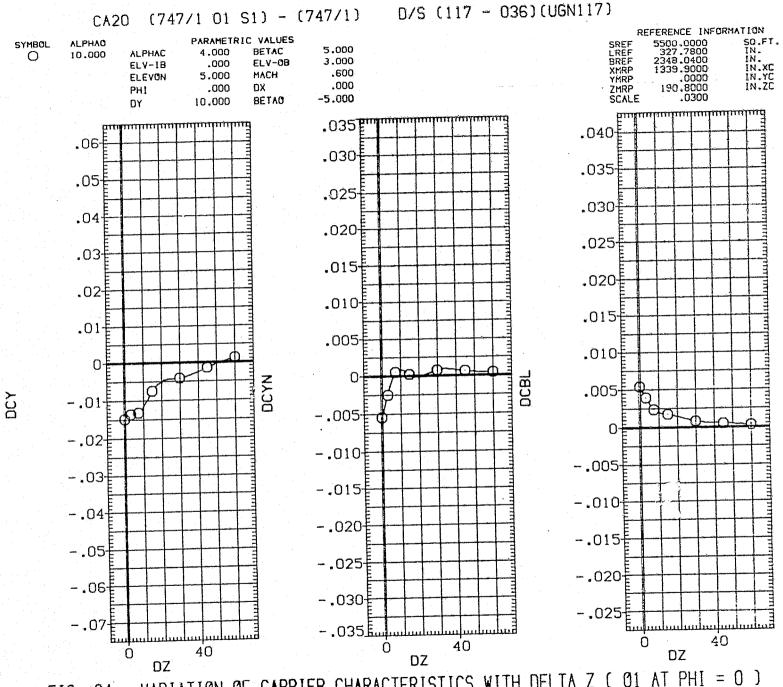


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

PAGE 393



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 394 PAGE

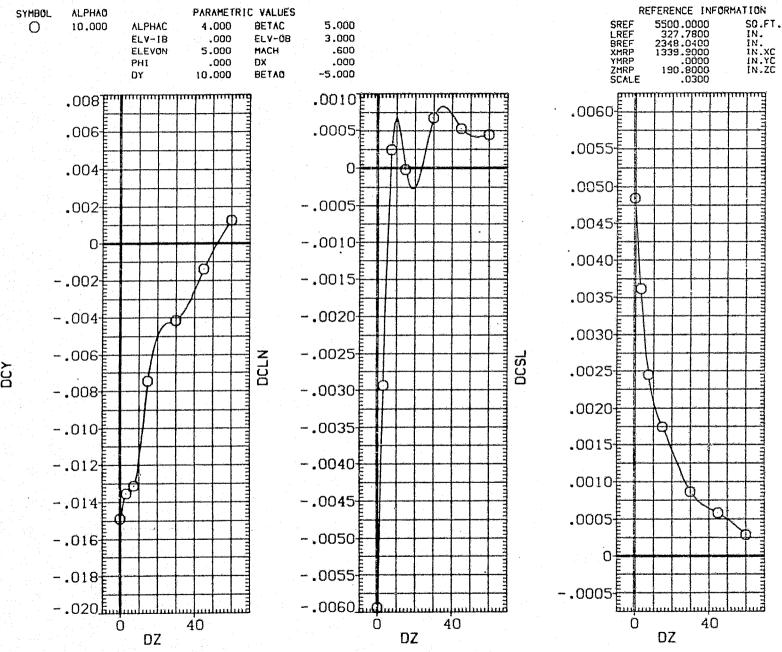


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 395

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 396

PAGE

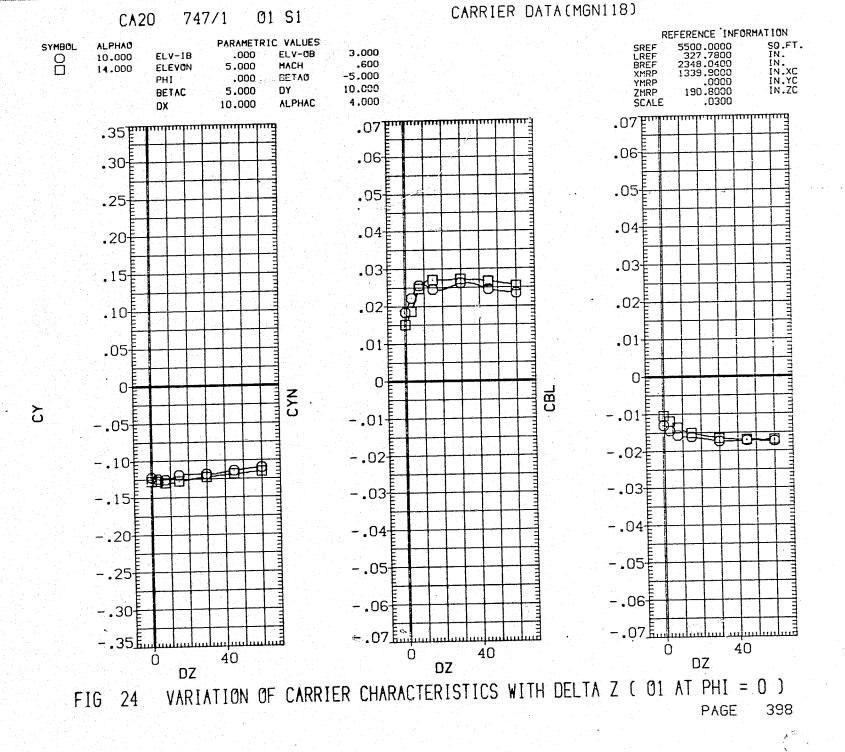


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 399

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 400

60

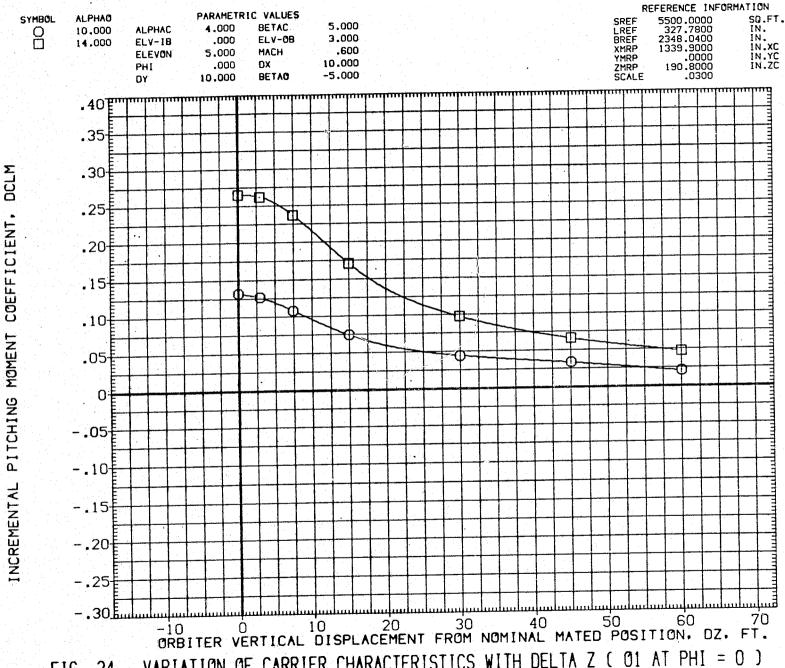
20 DZ

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-.06 <u>f</u>

20 **DZ** 40



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 401

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG PAGE 402

DZ VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG PAGE 403

DZ

0

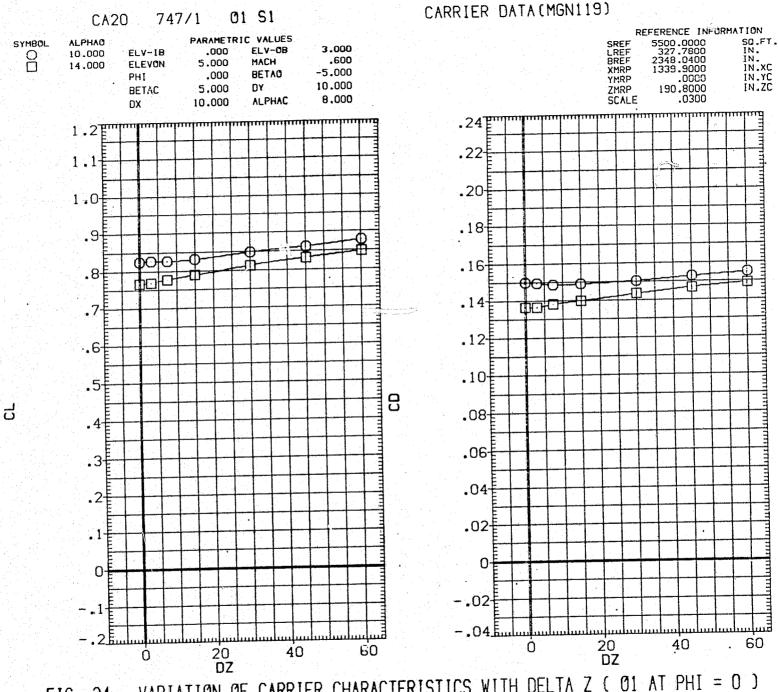
40

0

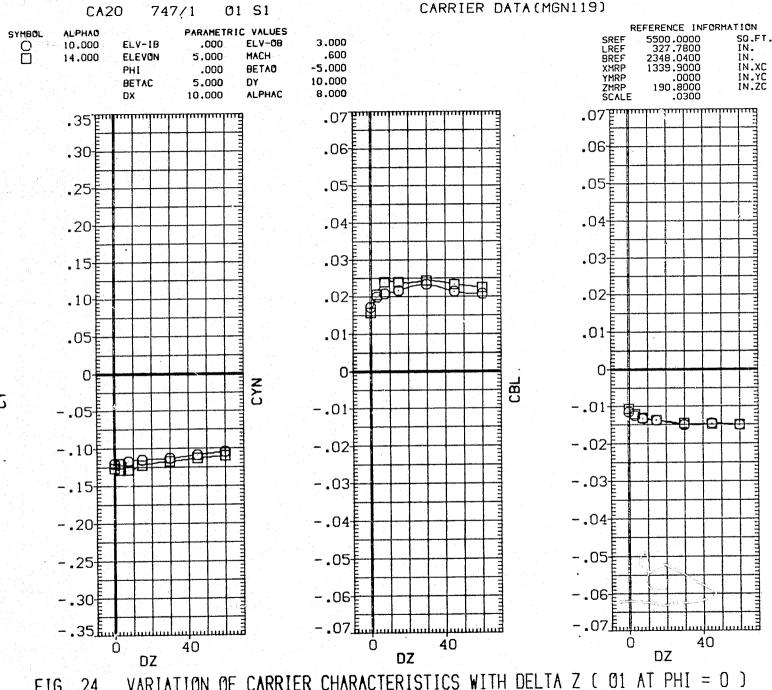
40

0

DZ.



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 404



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 406

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 407

DZ

0

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DZ

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DZ

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 408

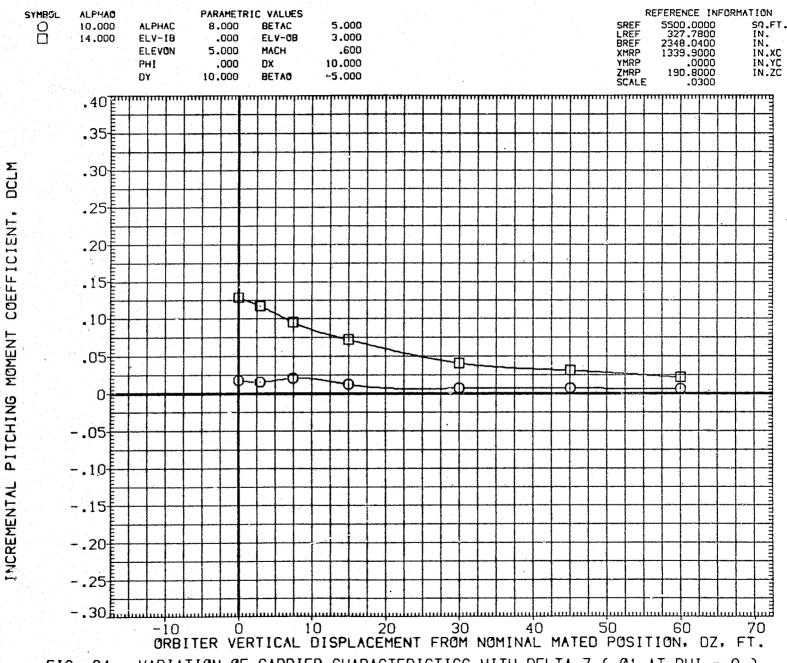


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

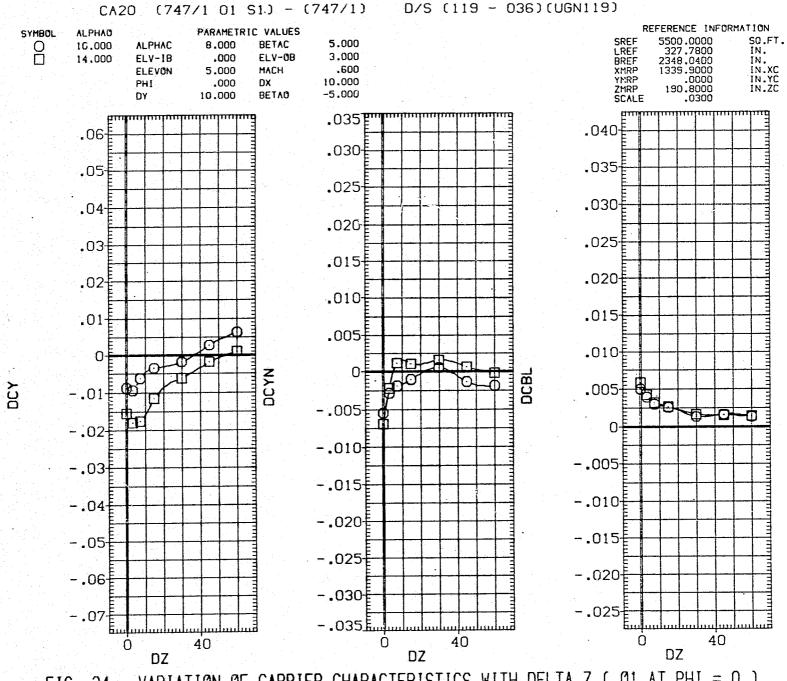


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 410

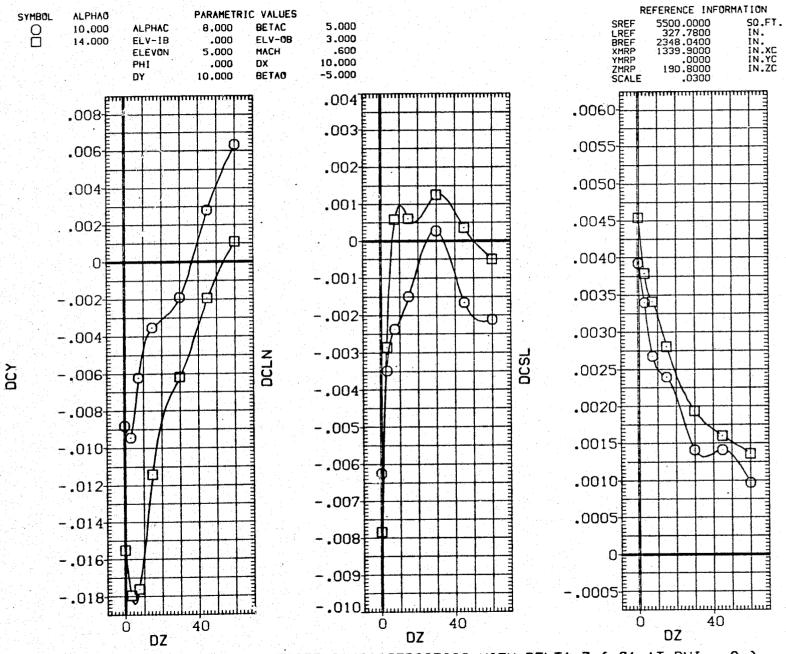


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 411

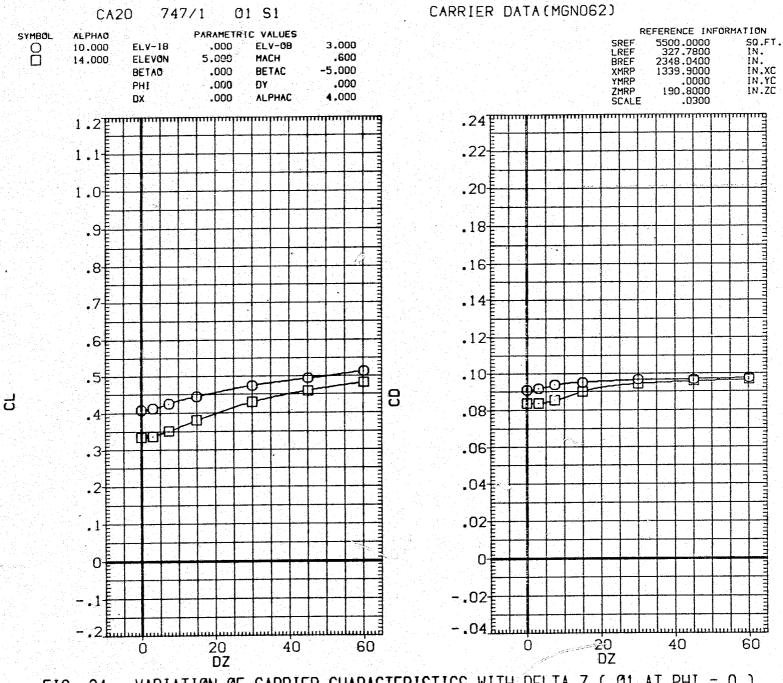
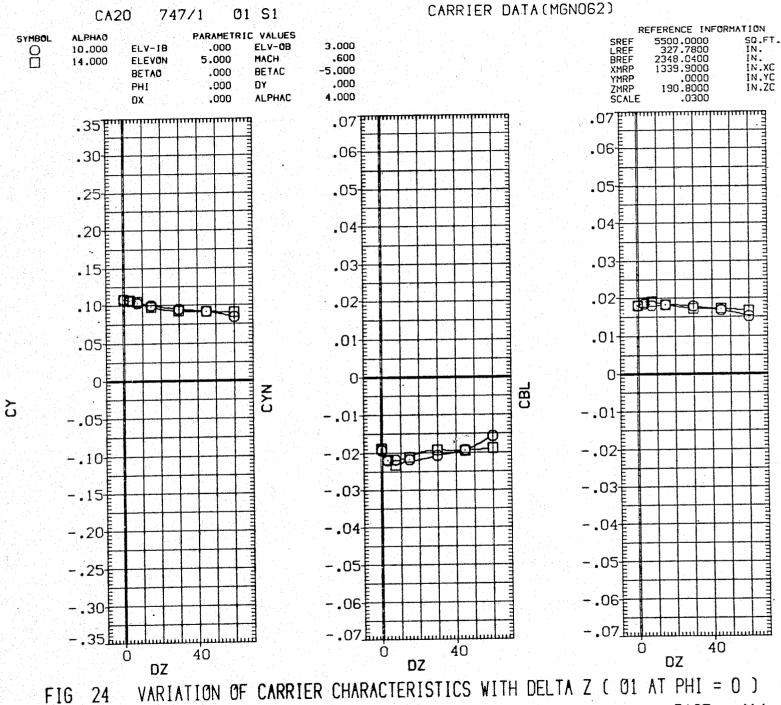


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 412

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



PAGE 414

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 415

DZ

40

0

DZ

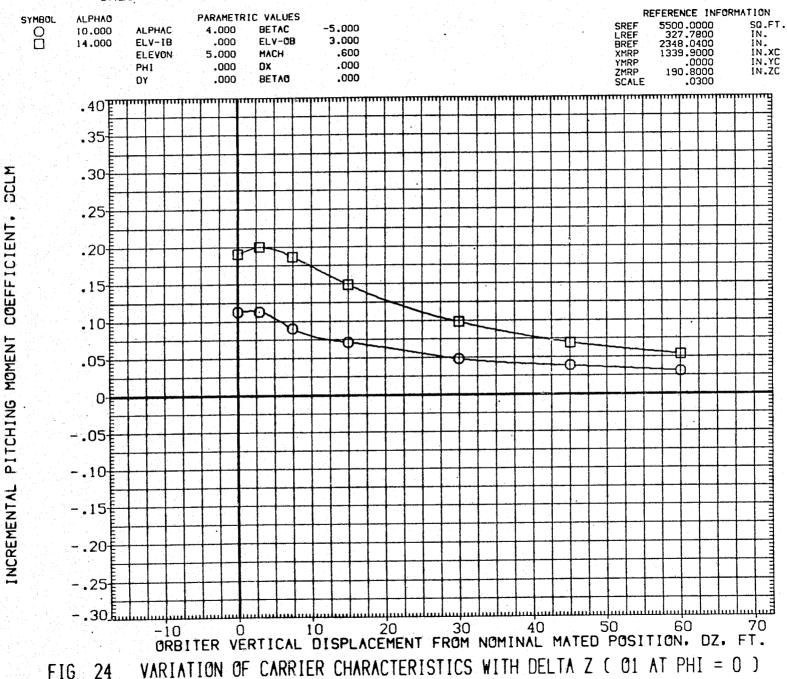
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 416



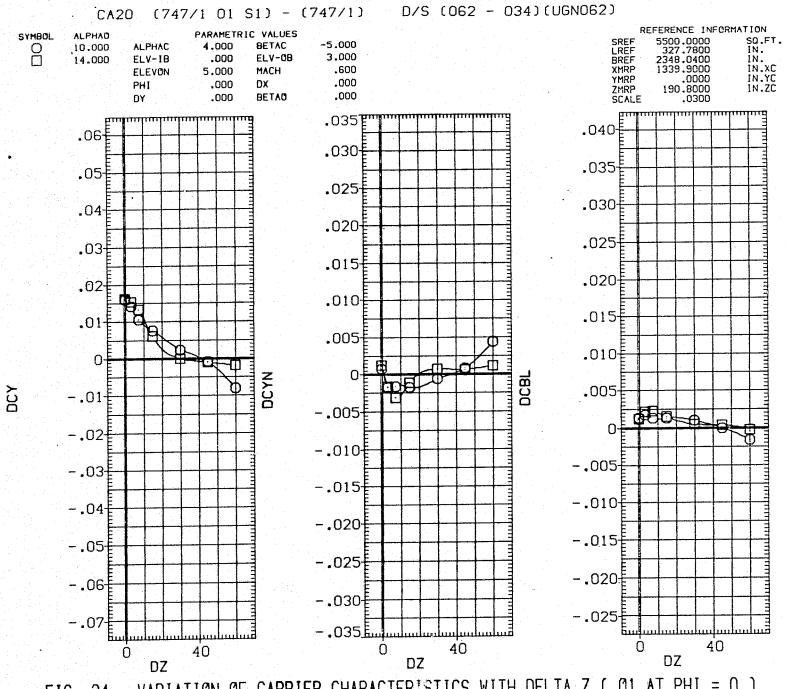
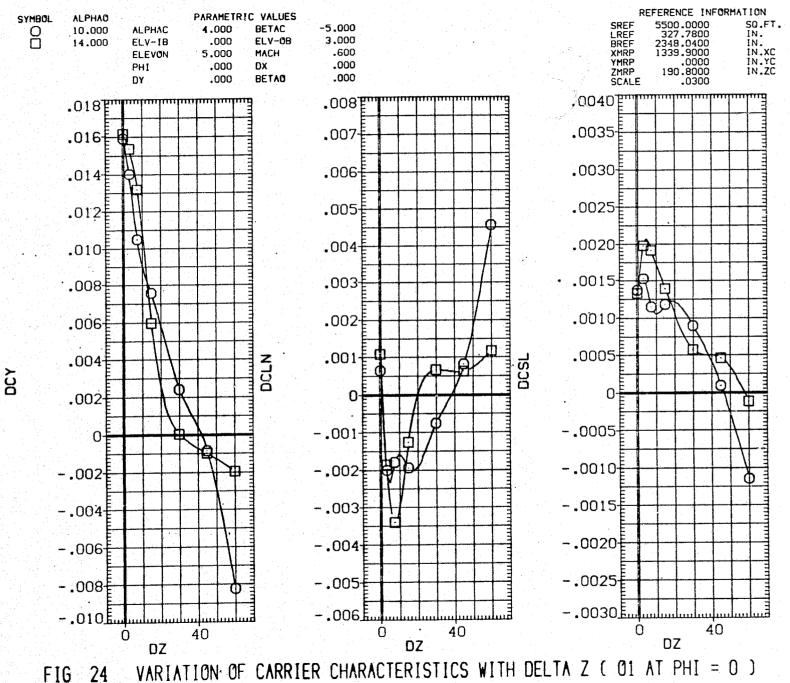


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 418



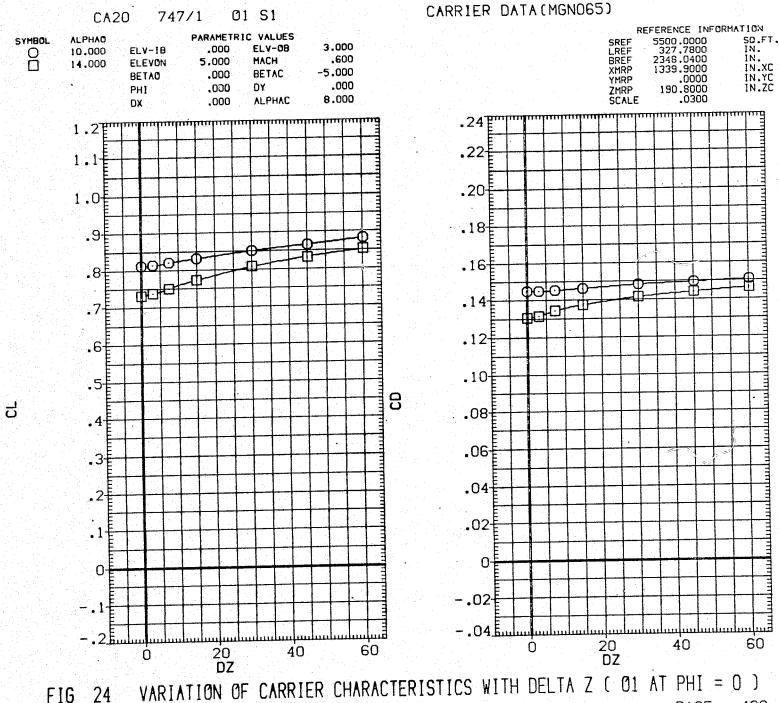


FIG 24 PAGE 420

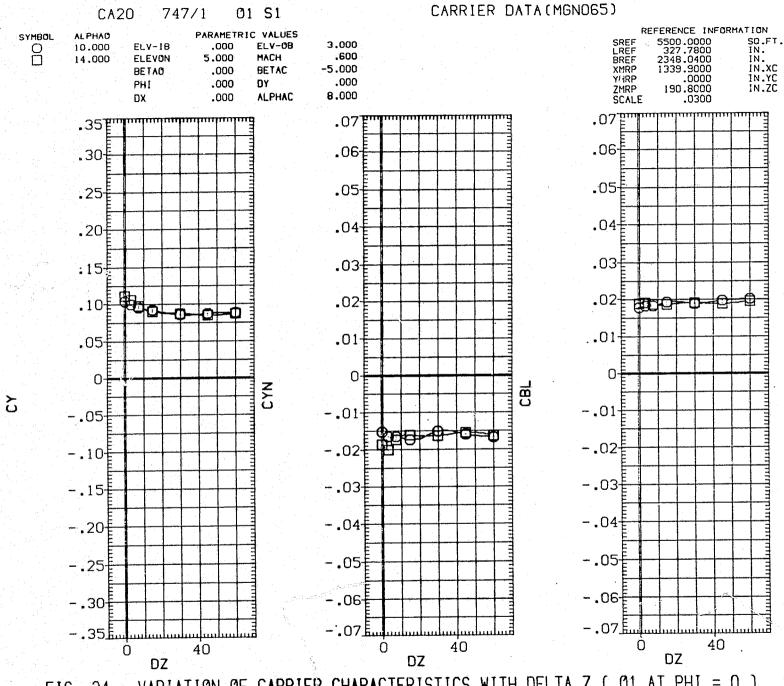


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

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DZ

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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 423

DZ

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DZ

ELV-18

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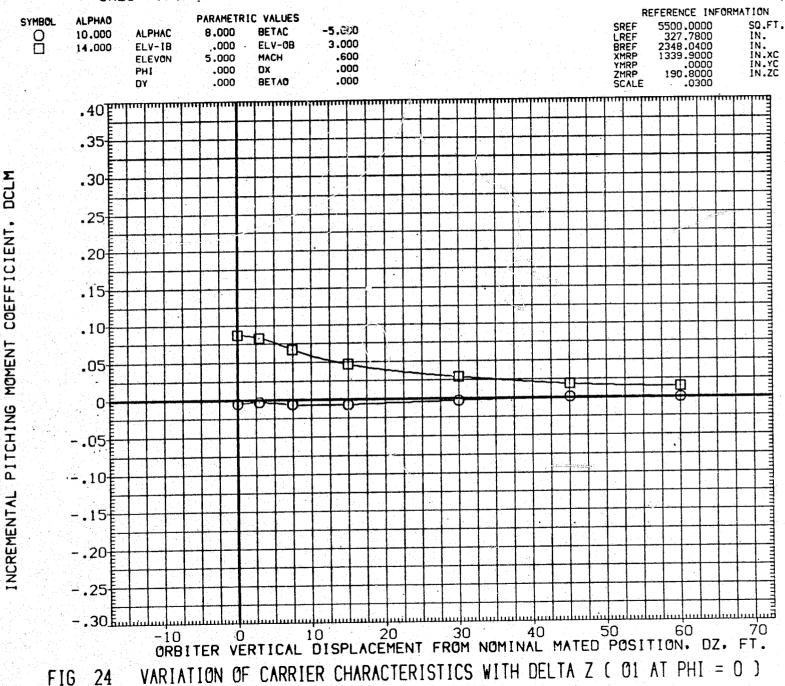
BETAO

PHI

DX

SYMBOL

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 424



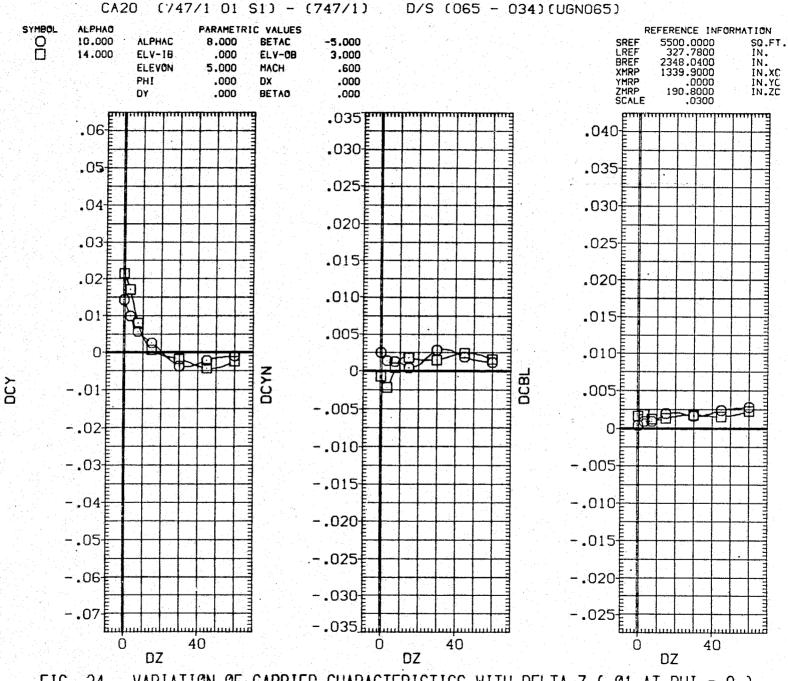


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 426

SYMBOL

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 $-.002\frac{1}{5}$

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DZ VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z FIG 24 PAGE.

DZ

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40

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-.0035基

.0008

.0006[

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DZ

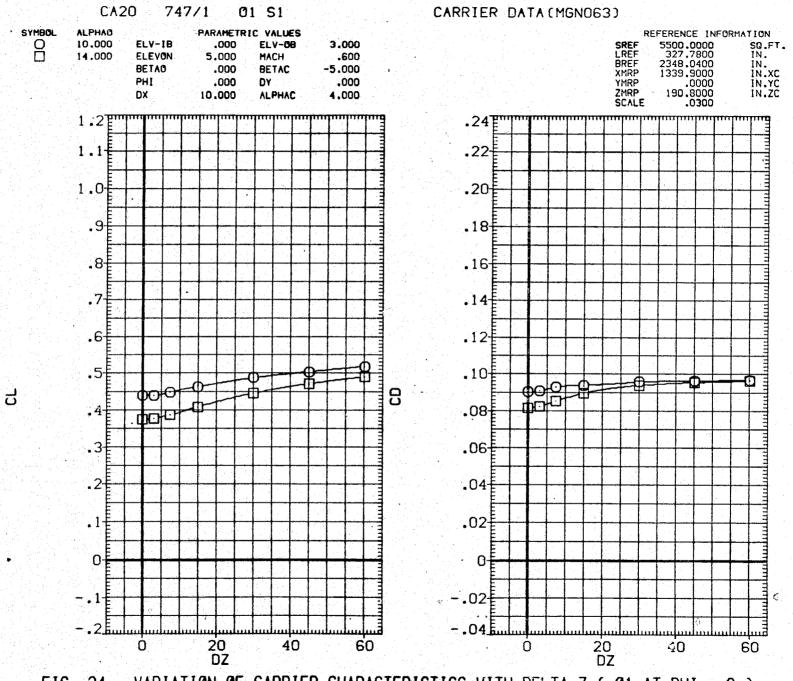


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 428

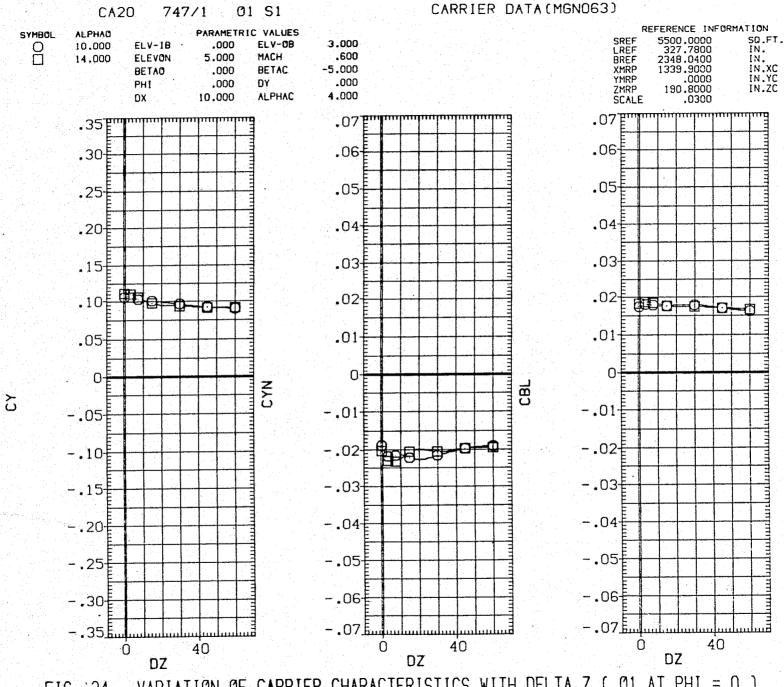


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 430

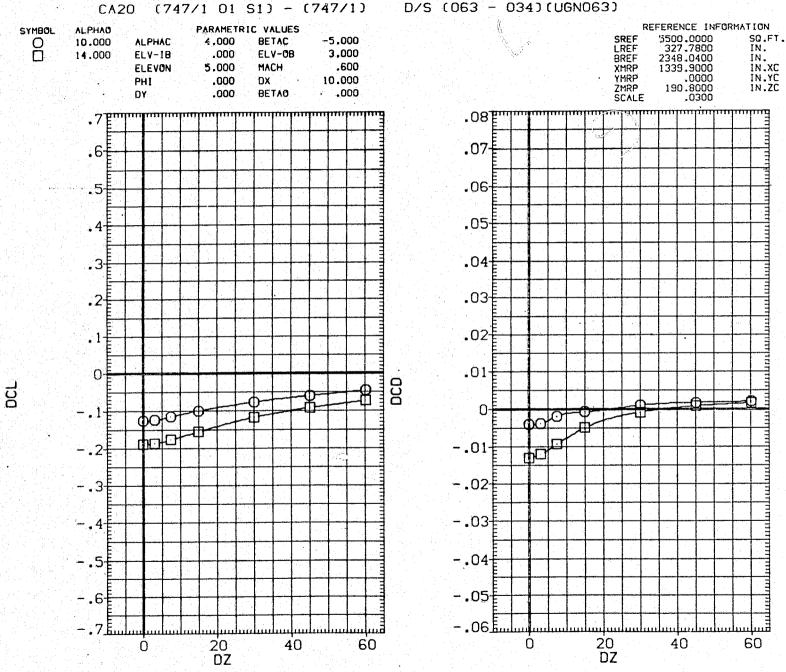


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 432

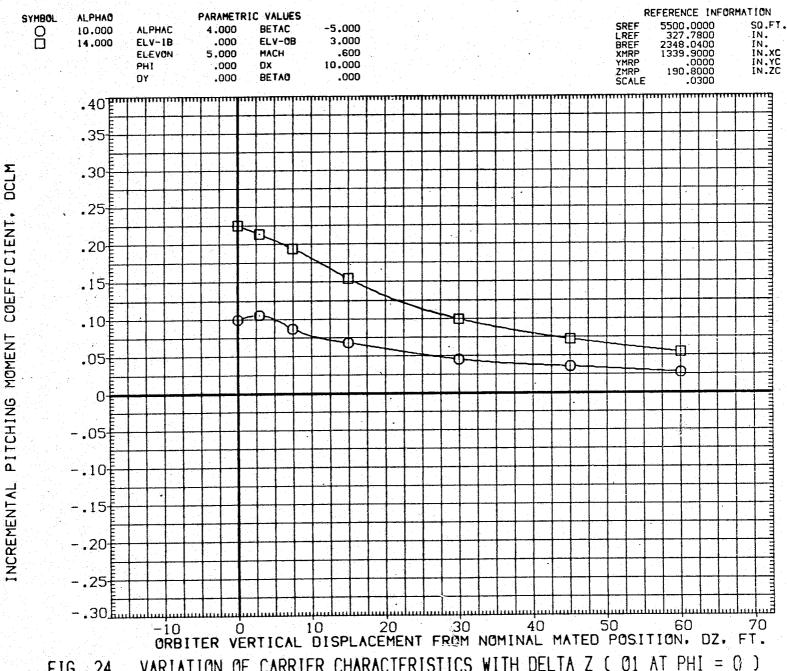
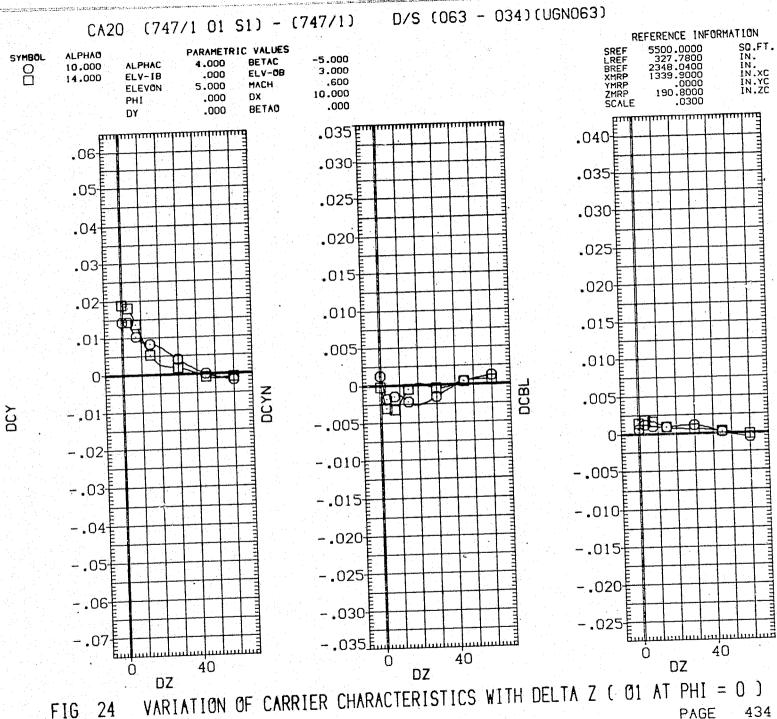


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) PAGE 433



24 FIG 434 PAGE

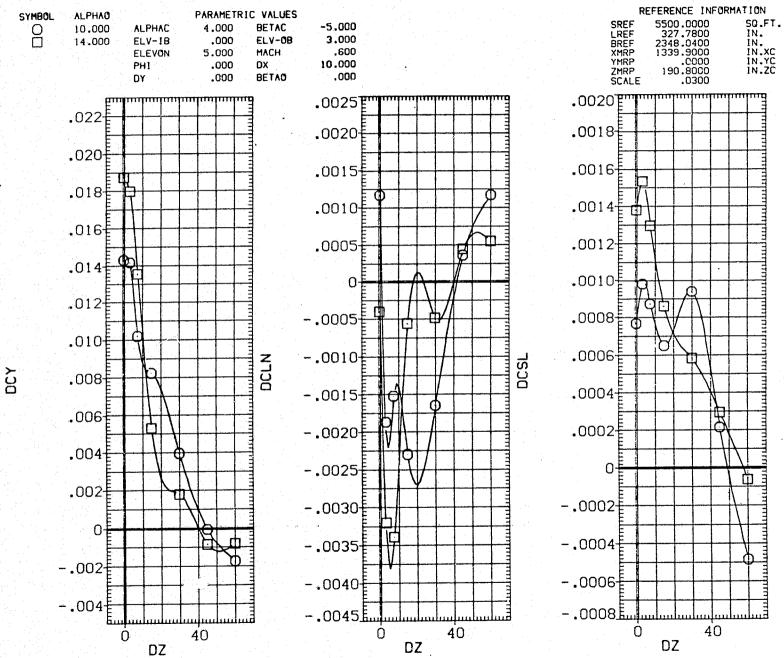


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 435

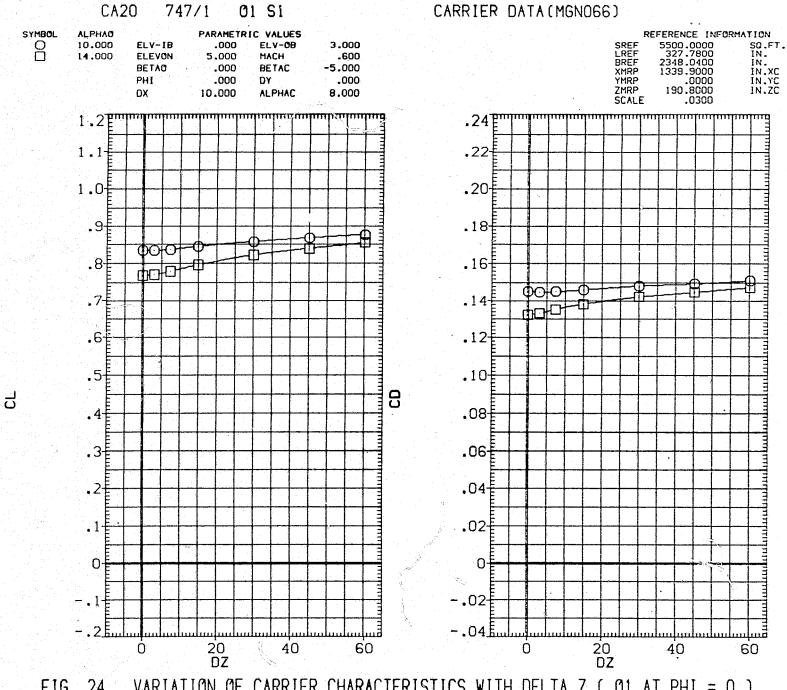


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 436

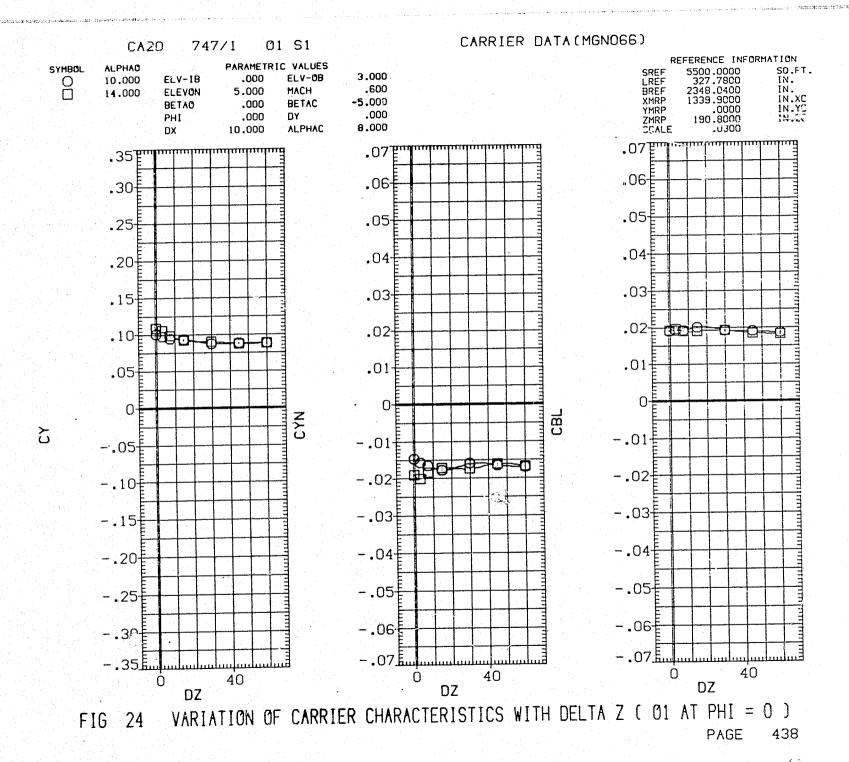


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 439

DZ

DZ

DZ

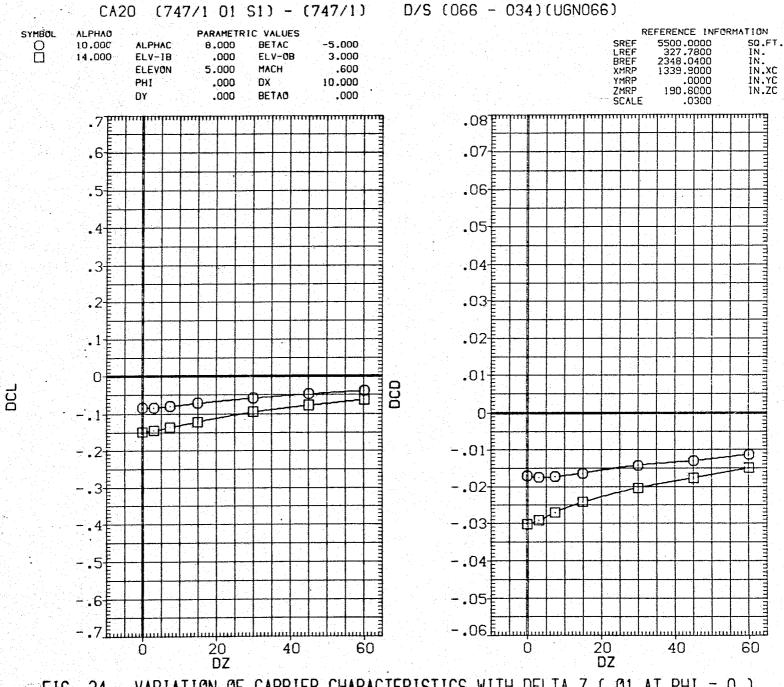


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 440

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA-Z (01 AT PHI = 0)
PAGE 442

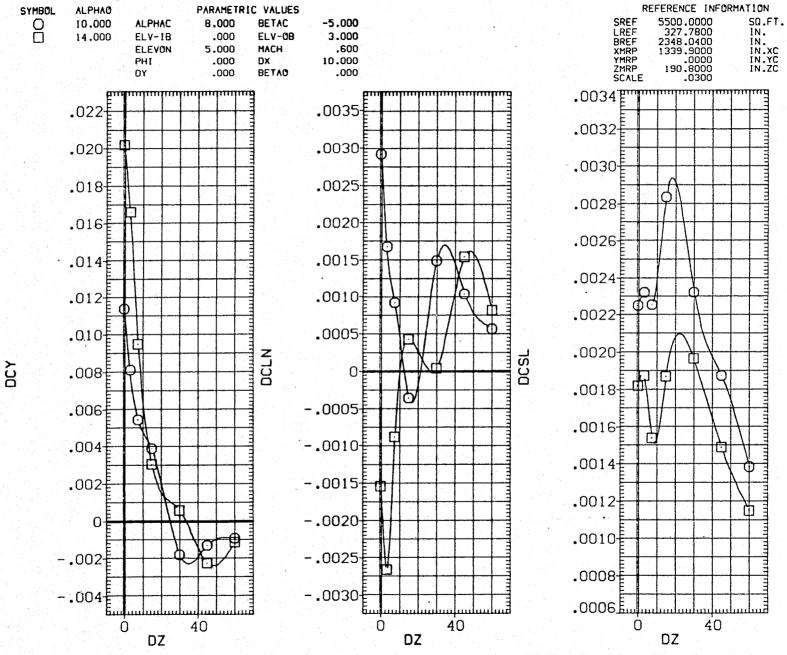


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

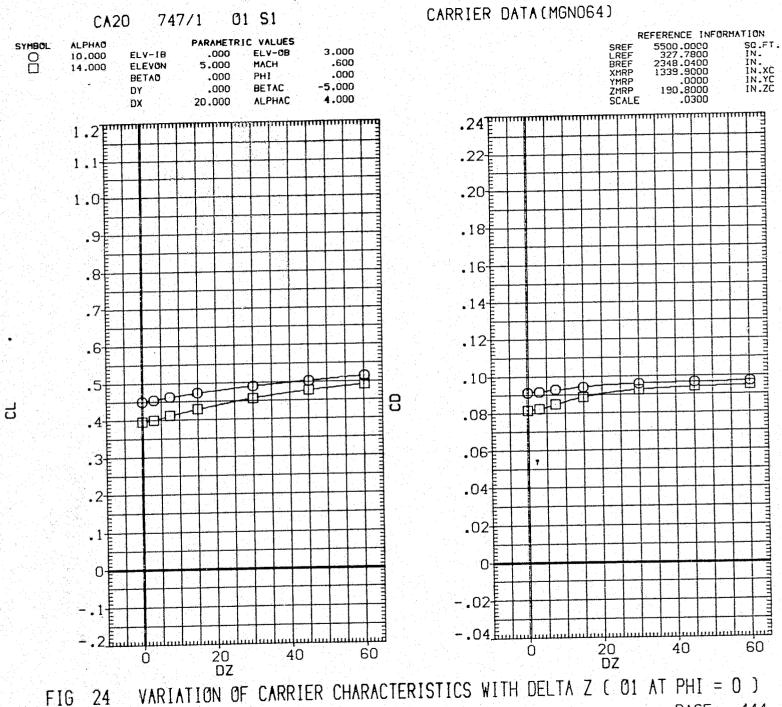
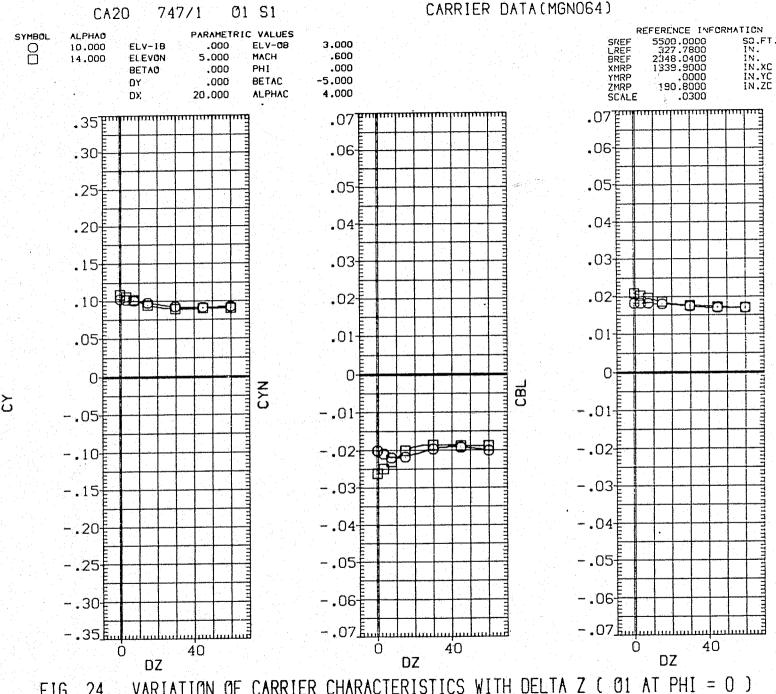


FIG 24 PAGE 444

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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG PAGE

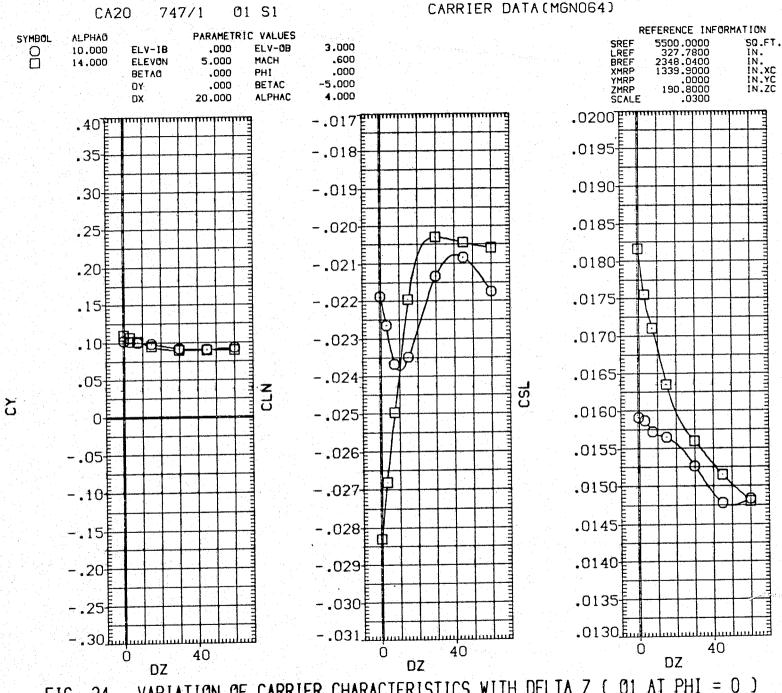


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 447

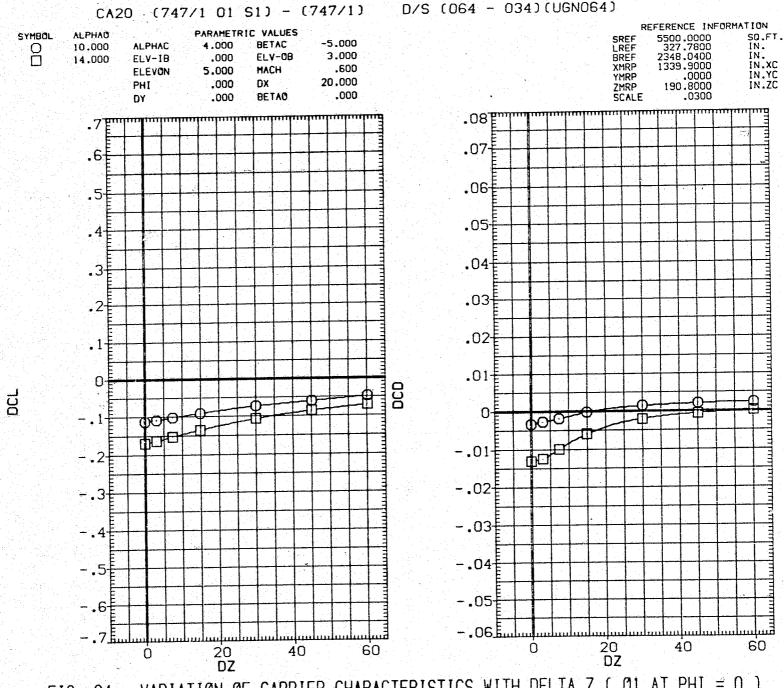


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 448

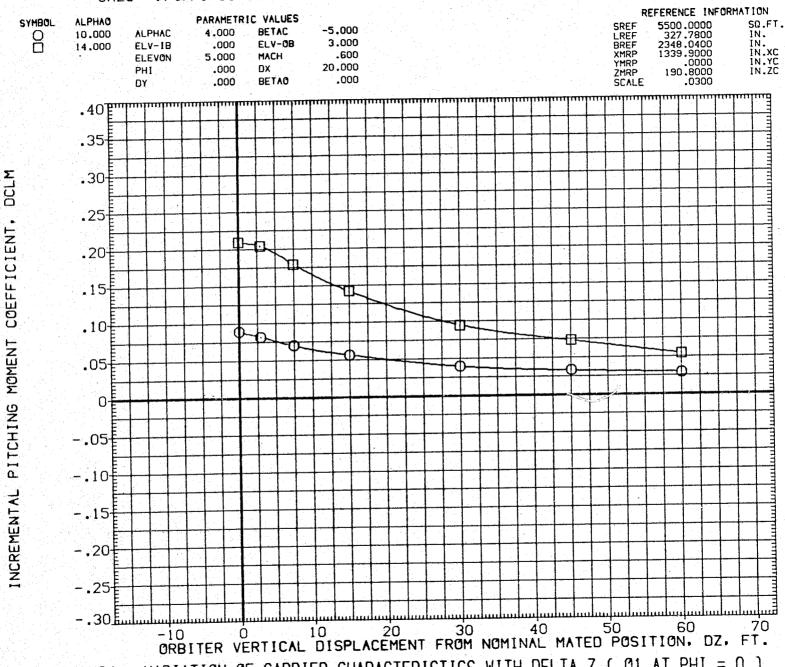
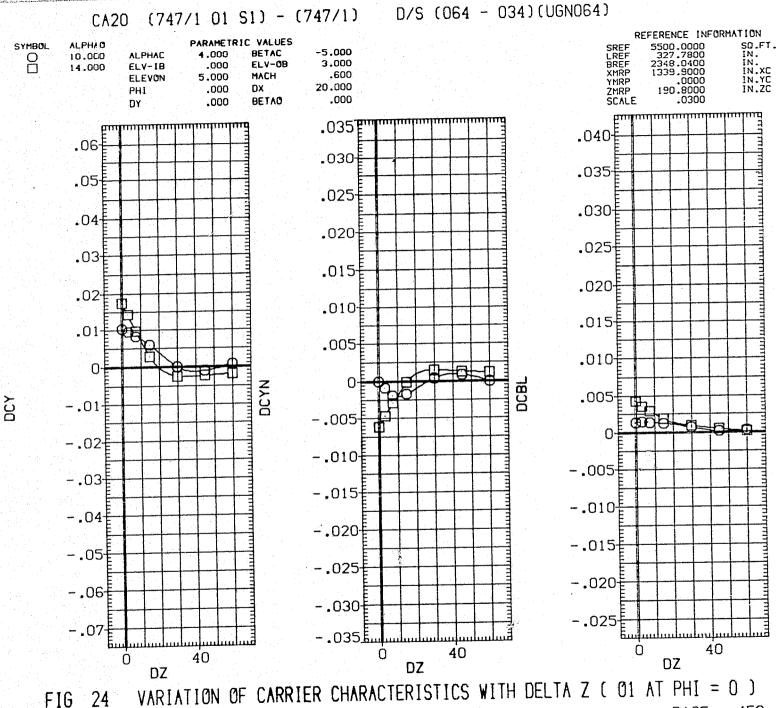


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



24 FIG **PAGE** 450

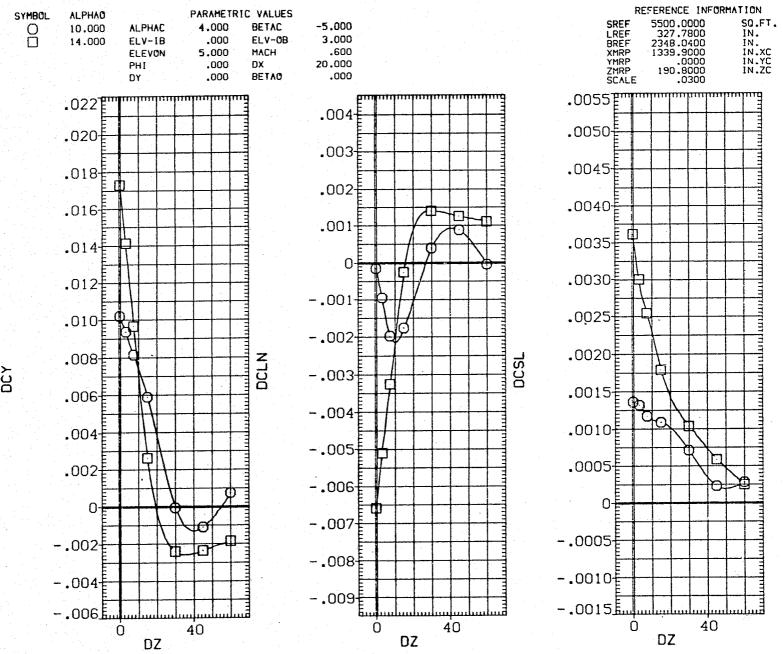


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 451

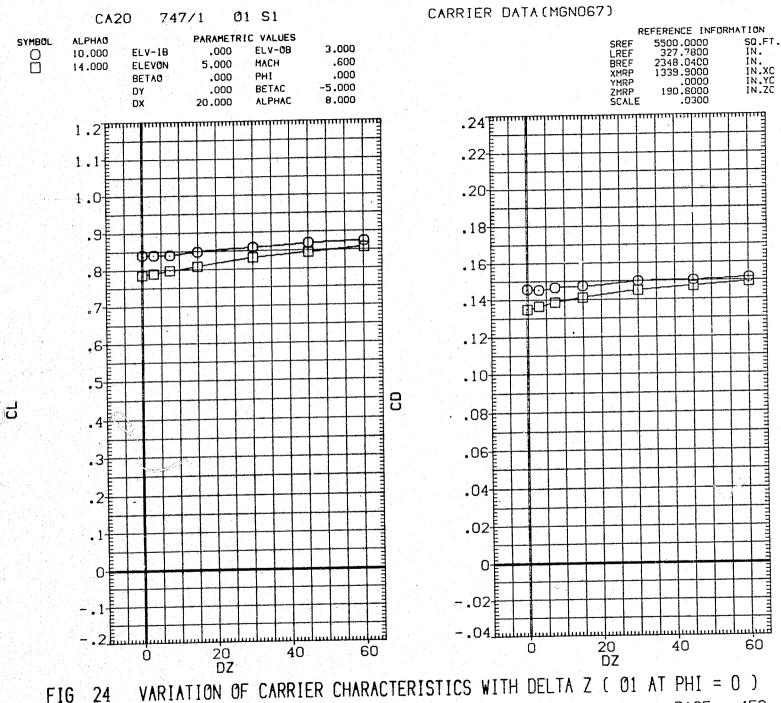


FIG 24 452 PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

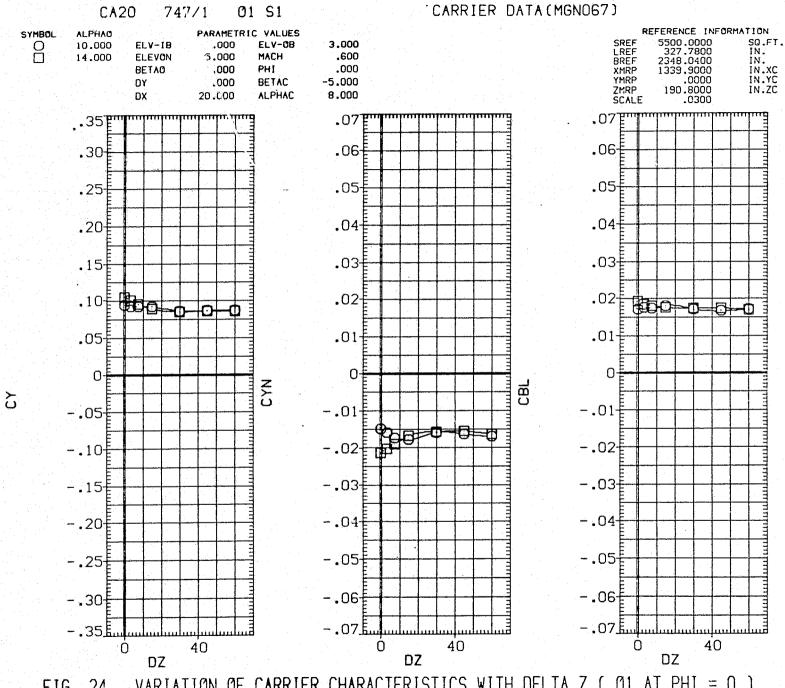


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 454

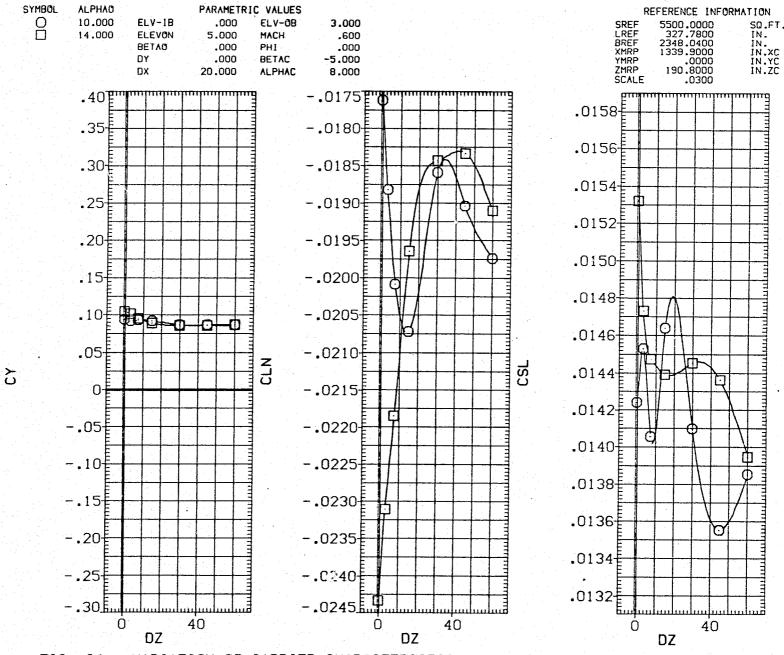


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 455

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 456

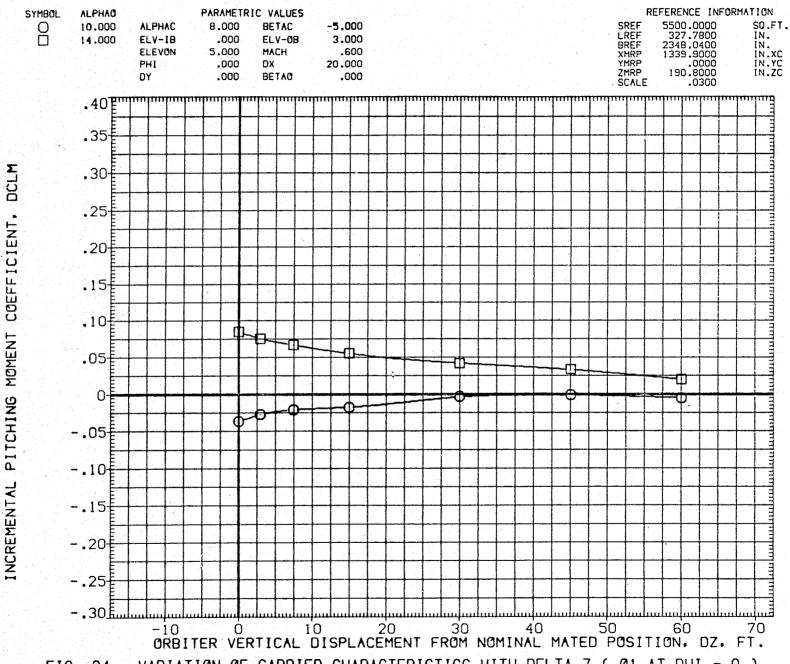
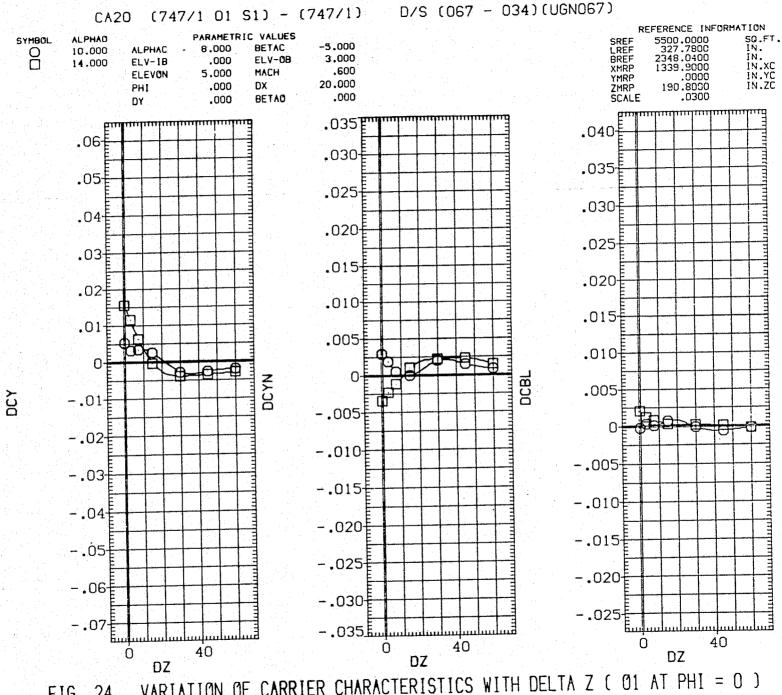


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 458

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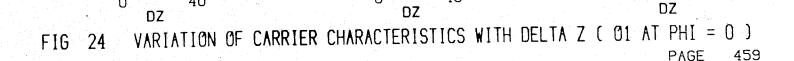
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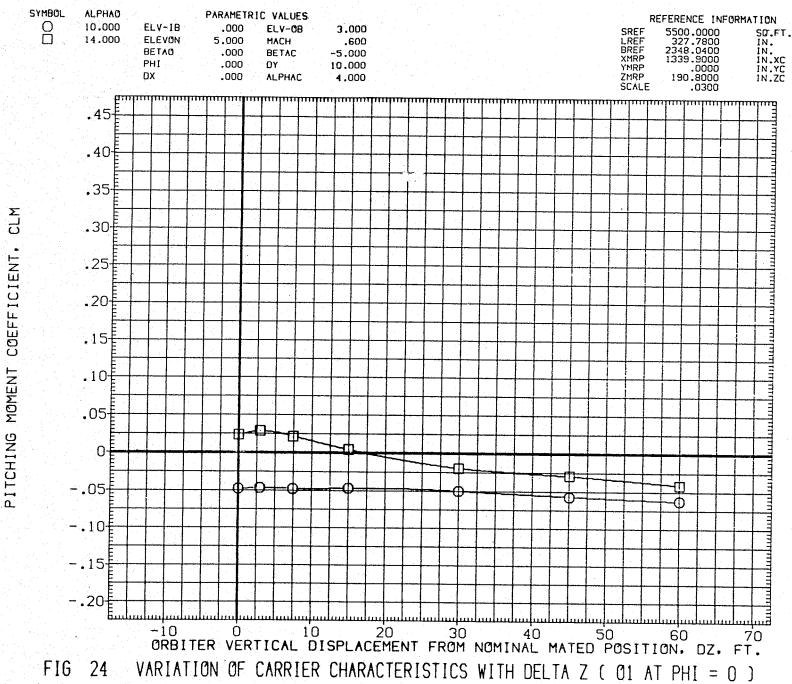
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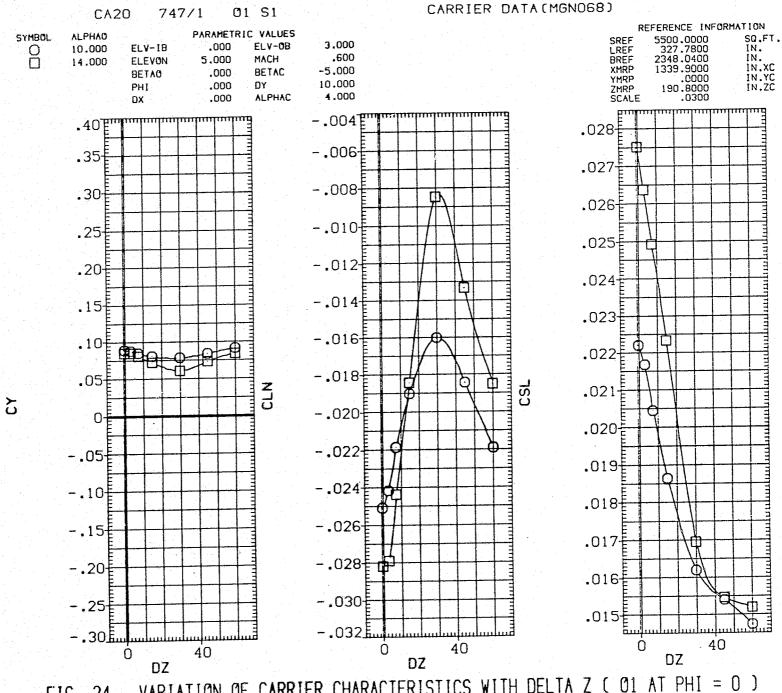
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FIG 24 PAGE 460



PAGE 461



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 463 PAGE

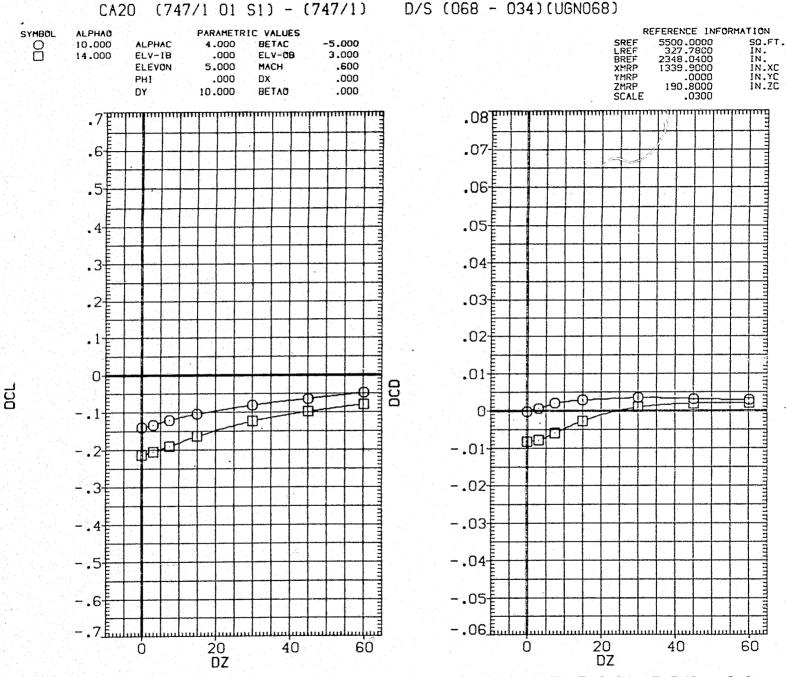


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 464

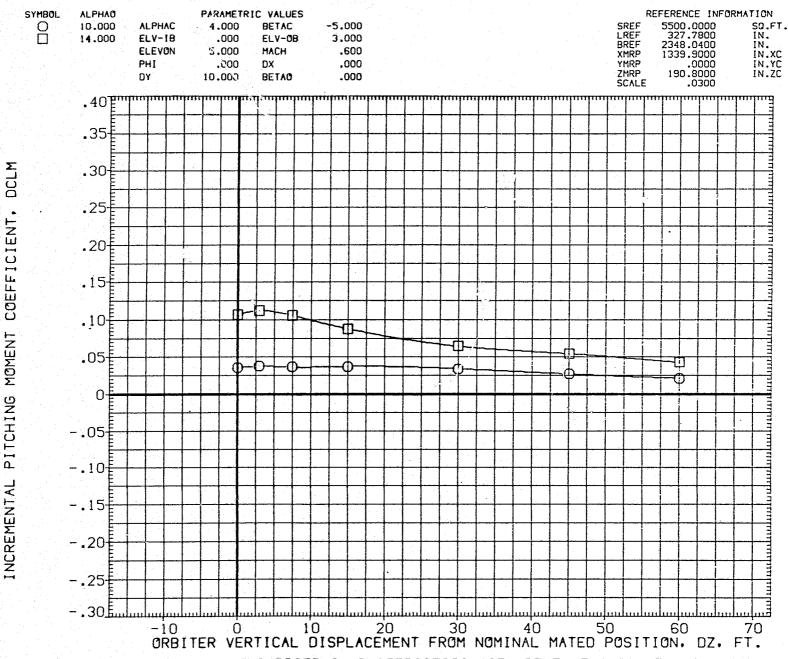
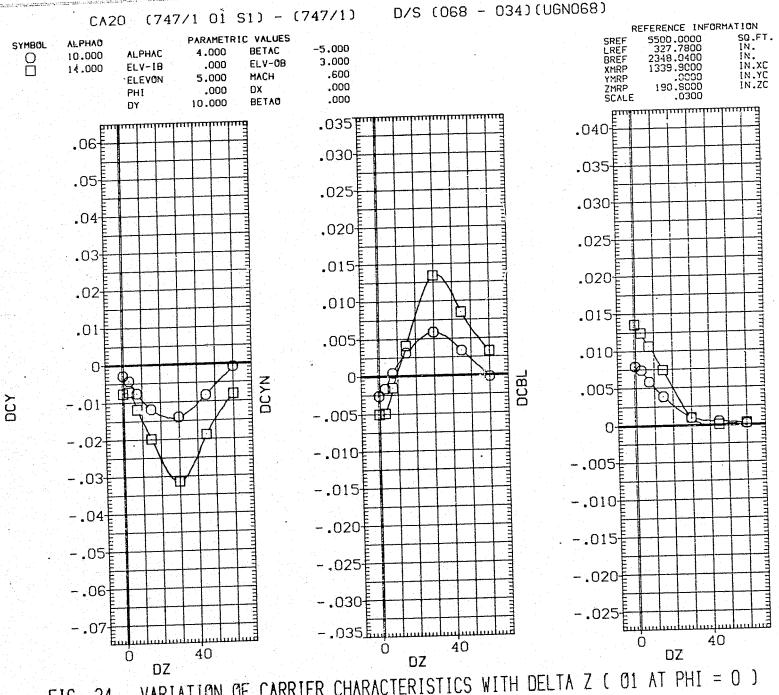


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 466 PAGE

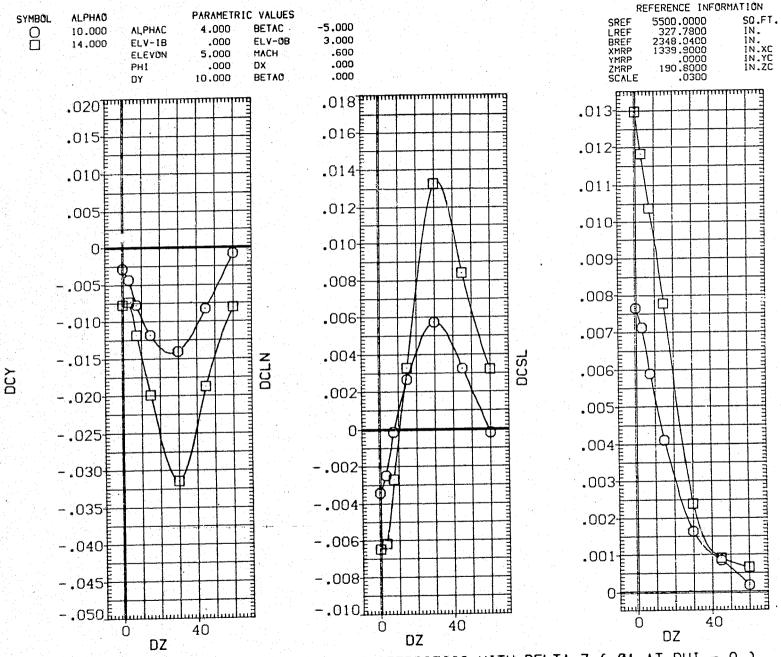
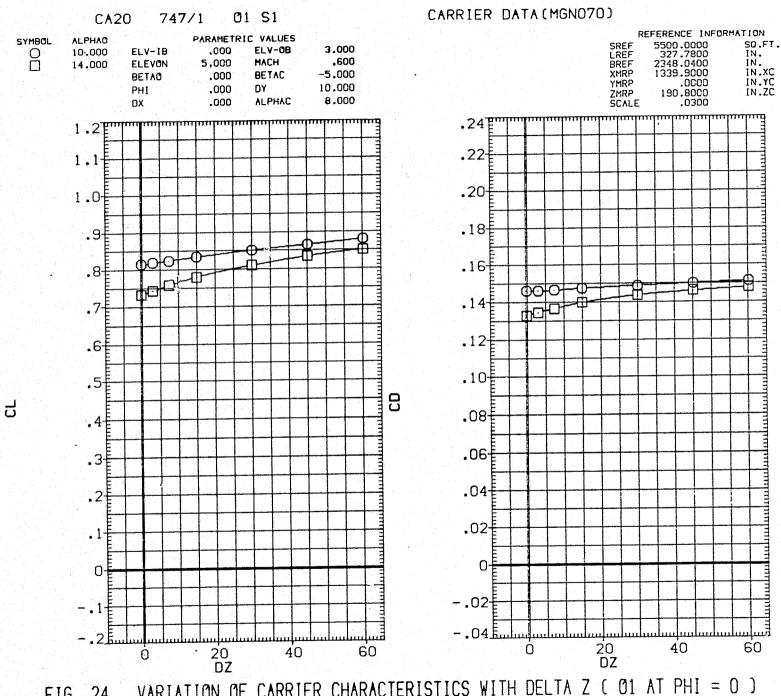


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 467



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG 468 PAGE

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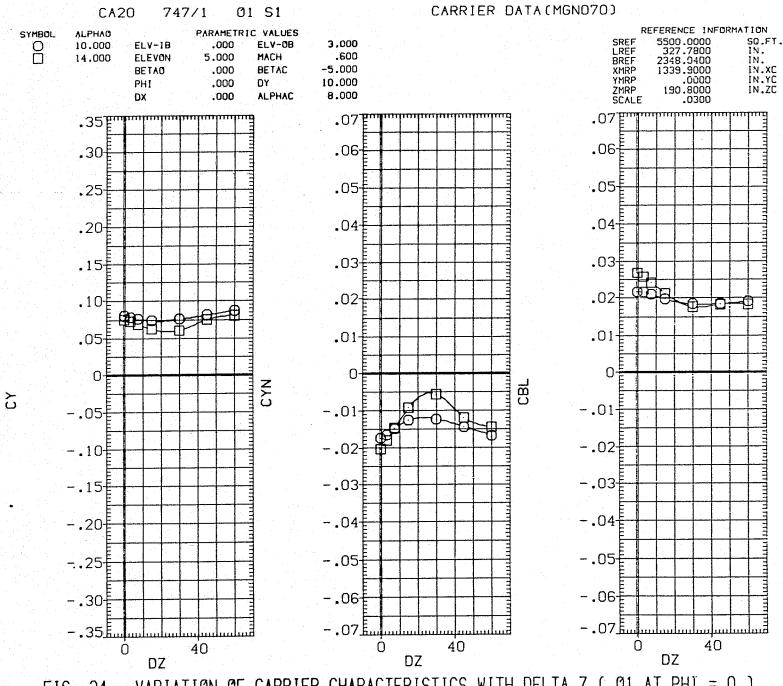


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 470

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 471

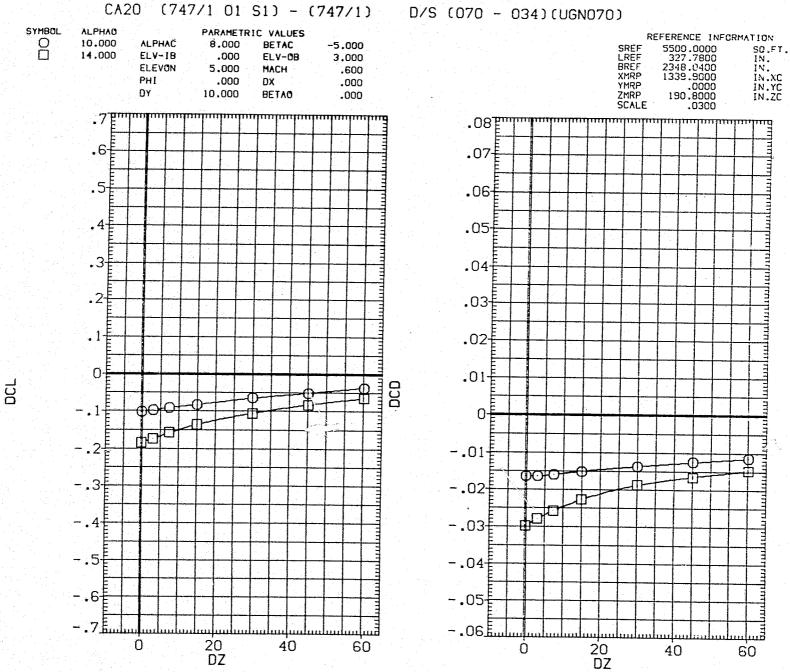
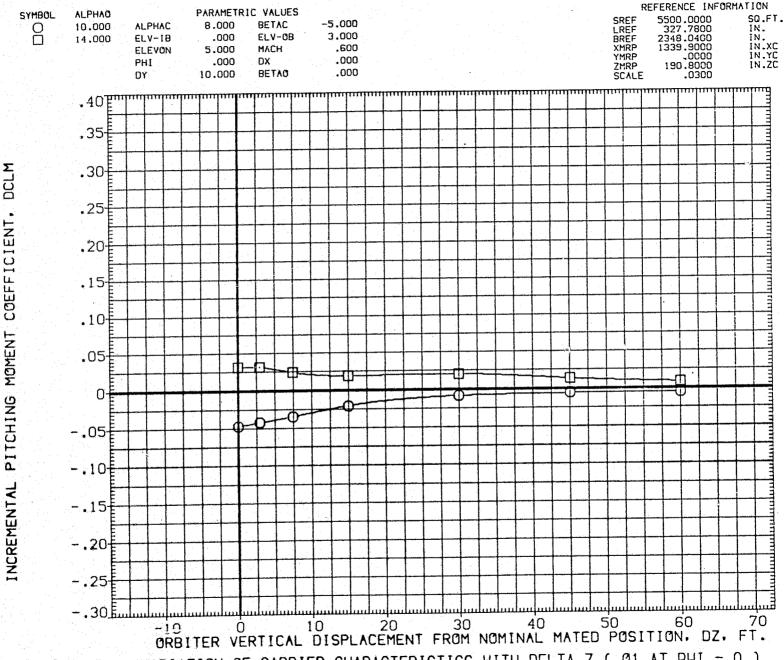


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 472



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 473 PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 474

DZ VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 475 PAGE

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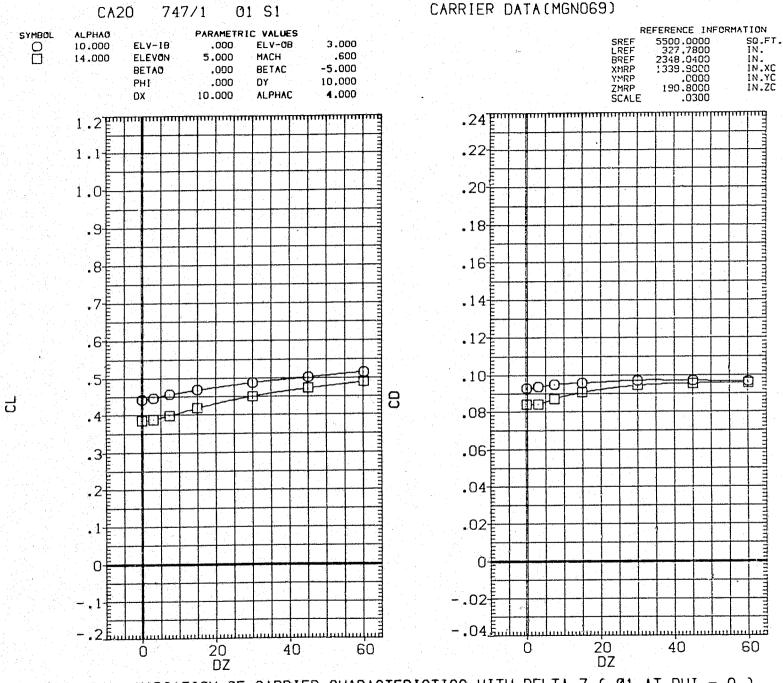
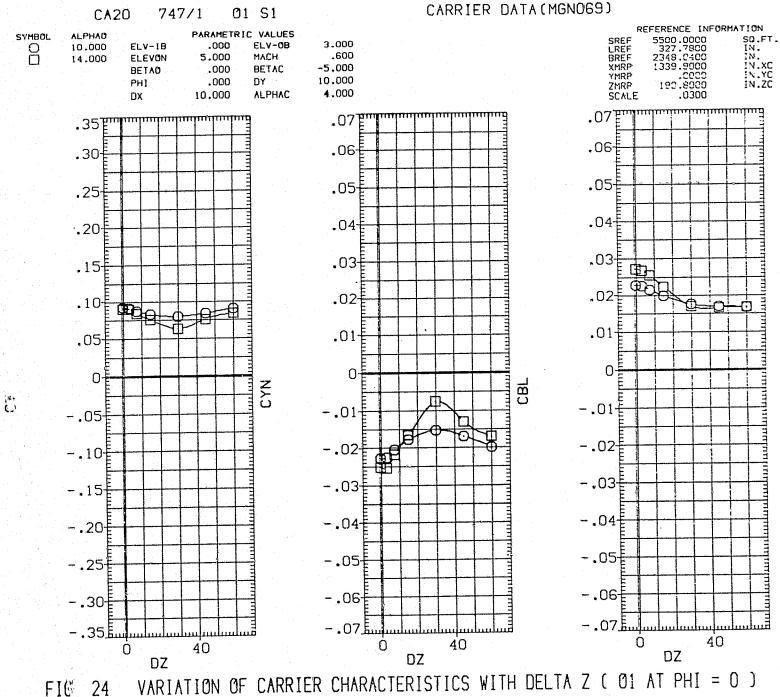


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 476



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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG 479 PAGE

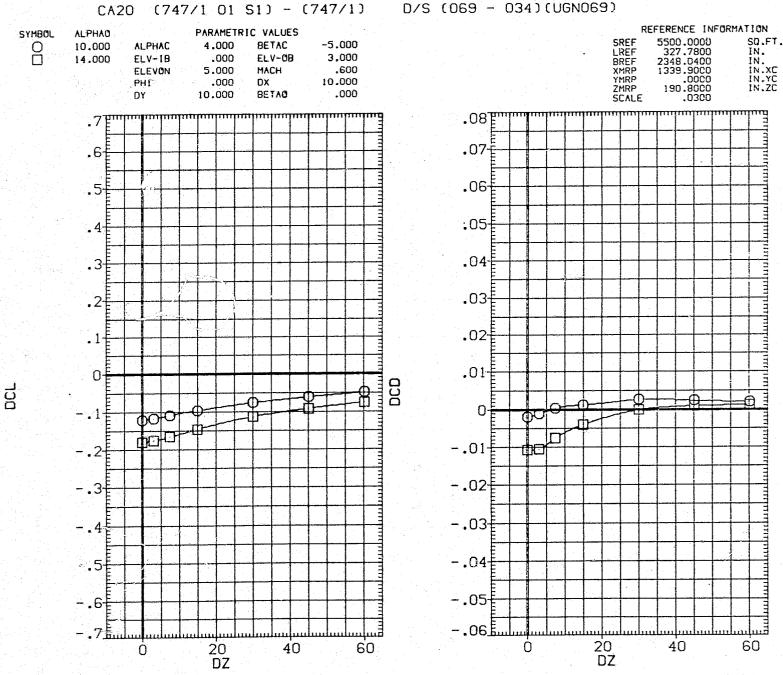
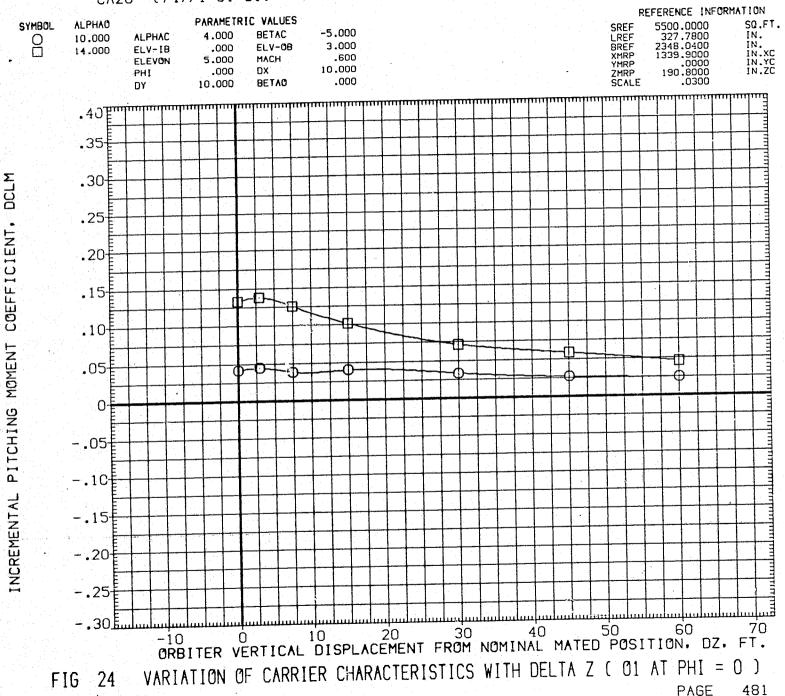
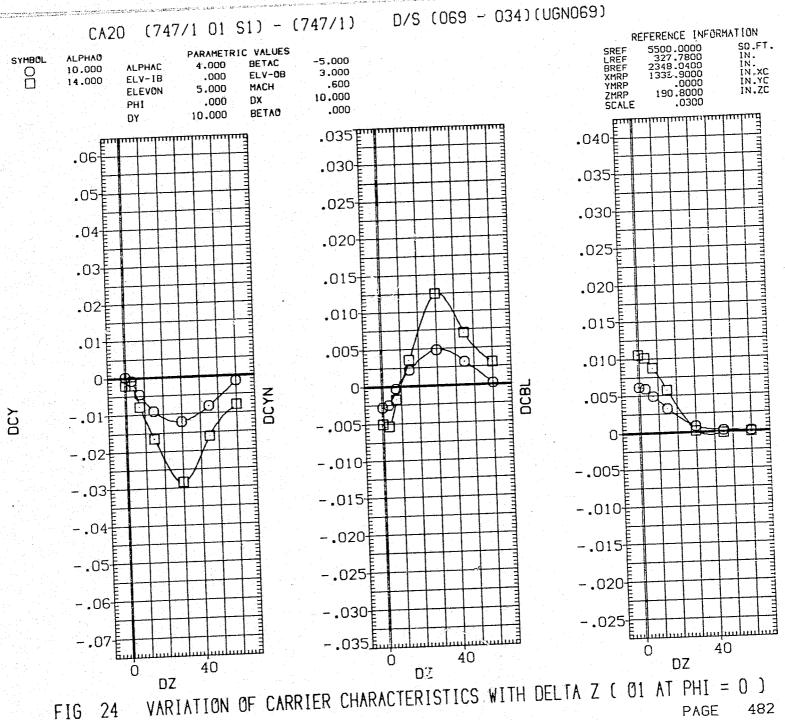


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)





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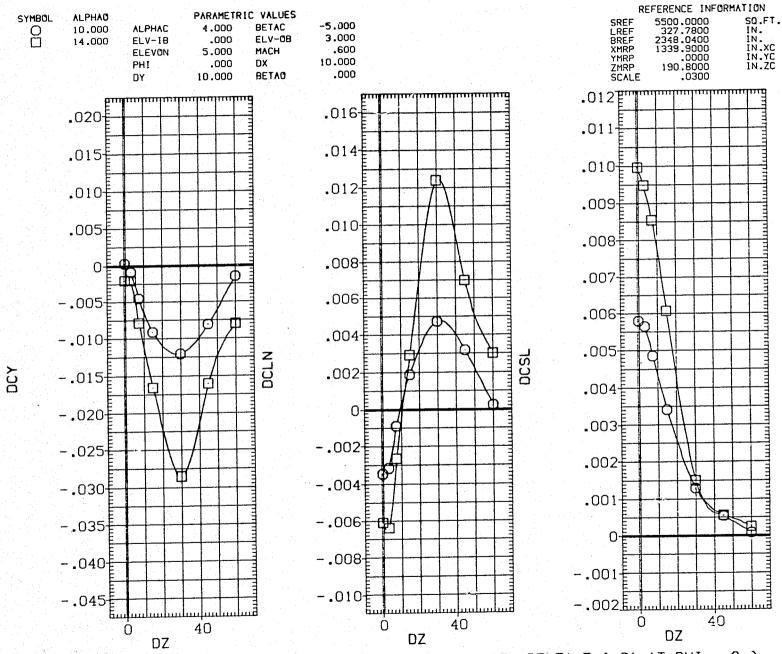
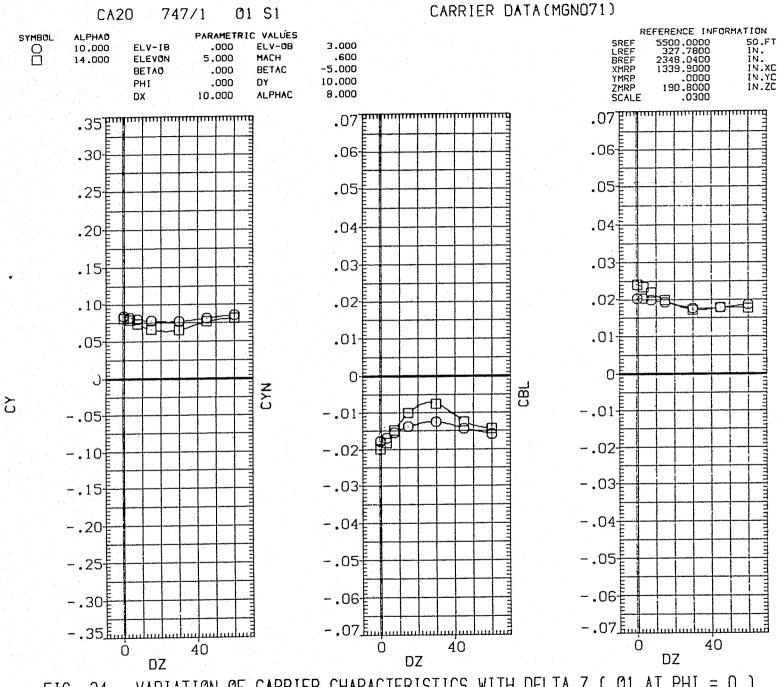


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

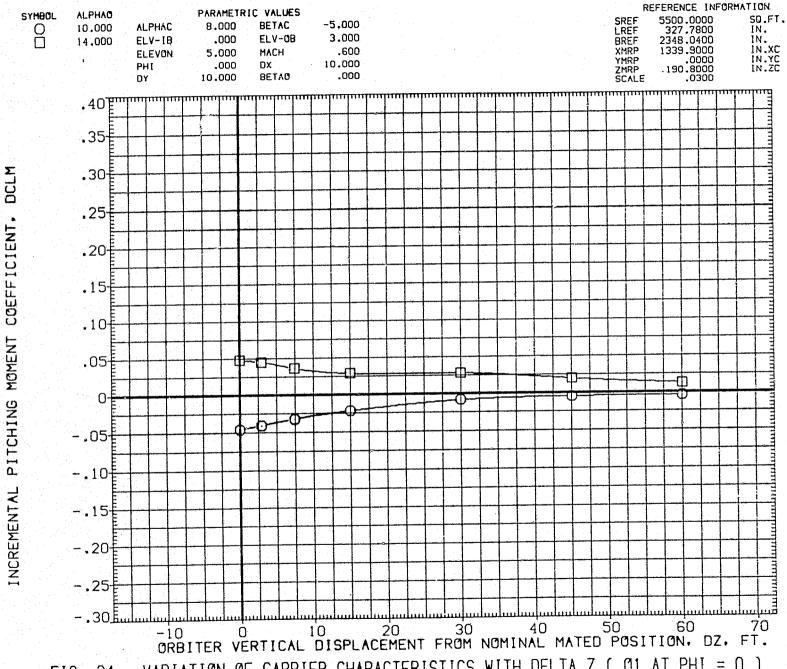
VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG PAGE 484



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG PAGE

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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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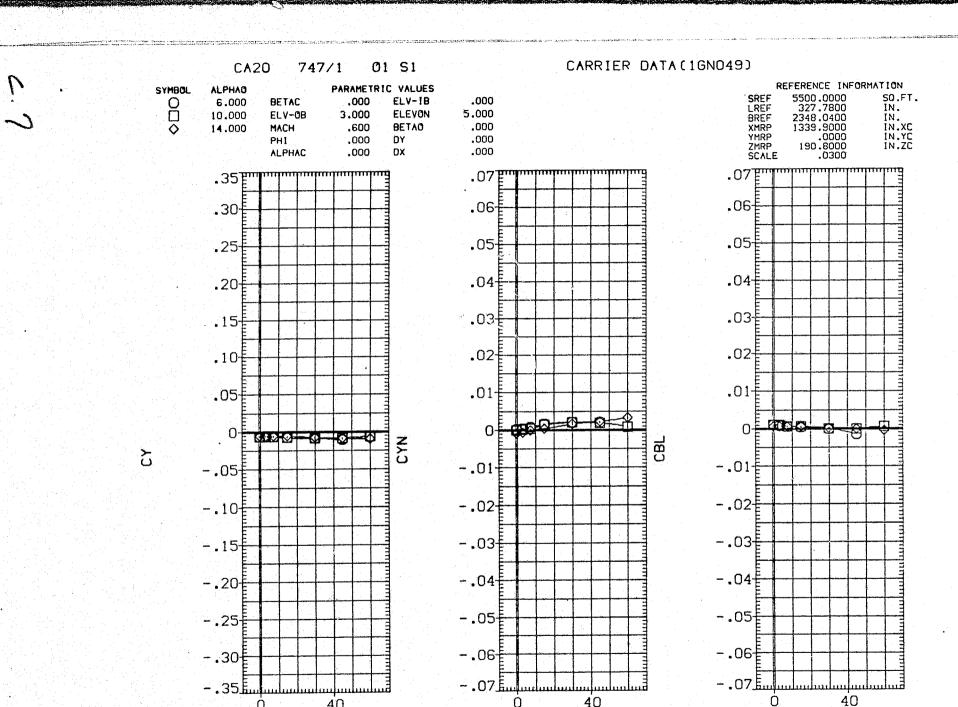
VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 490

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 494

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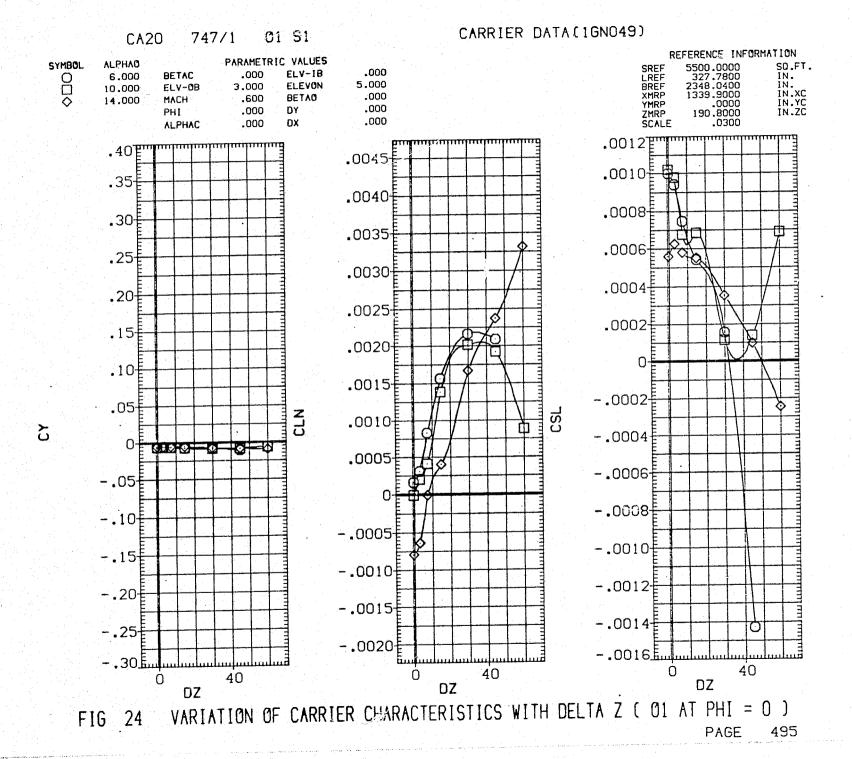
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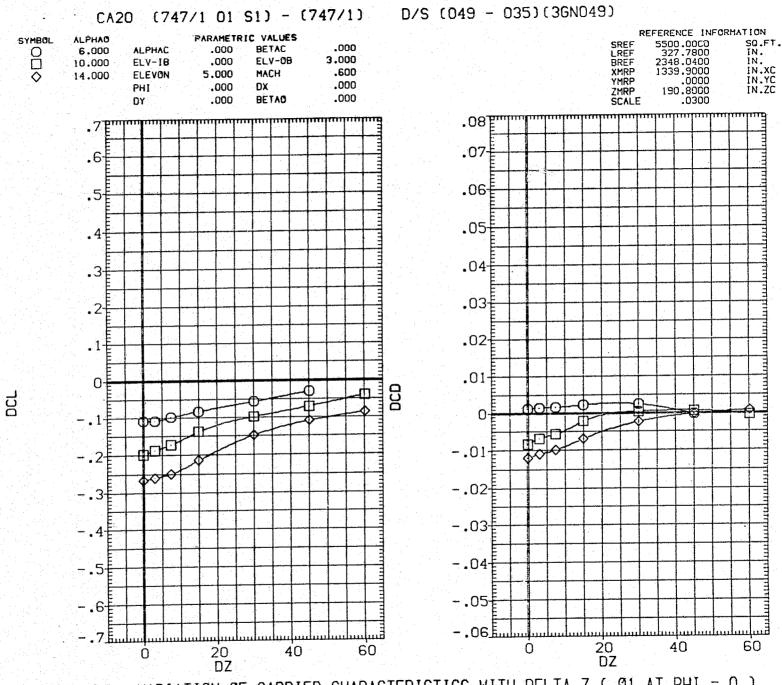


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 496

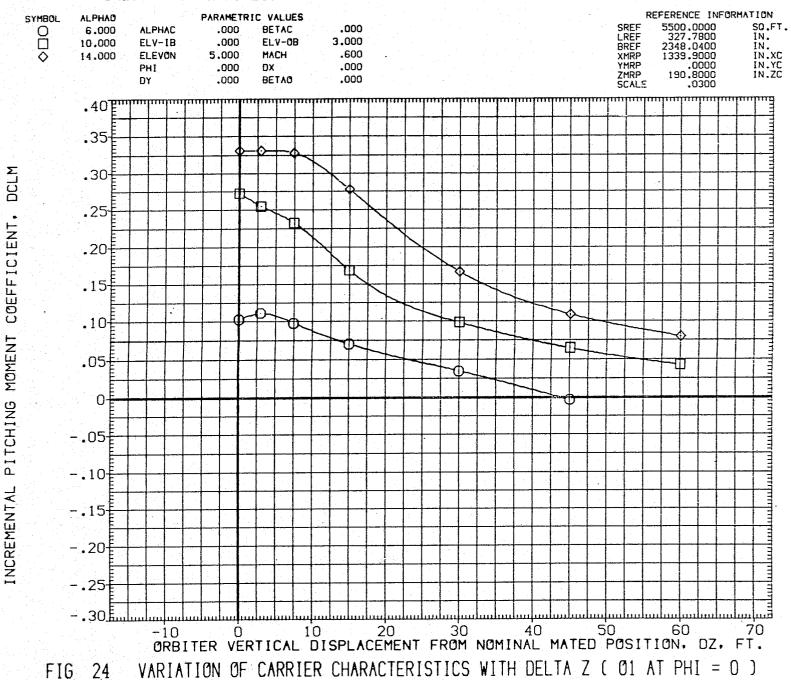


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 498

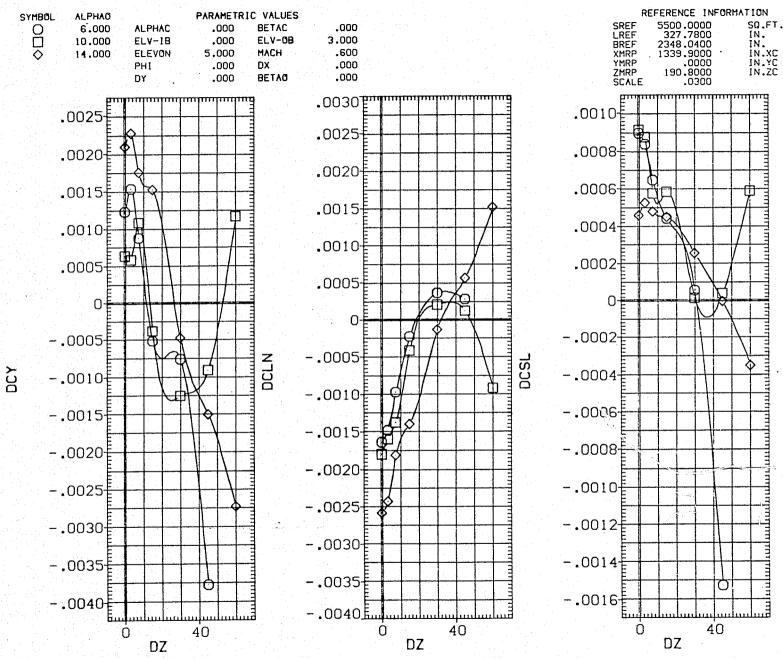


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 500

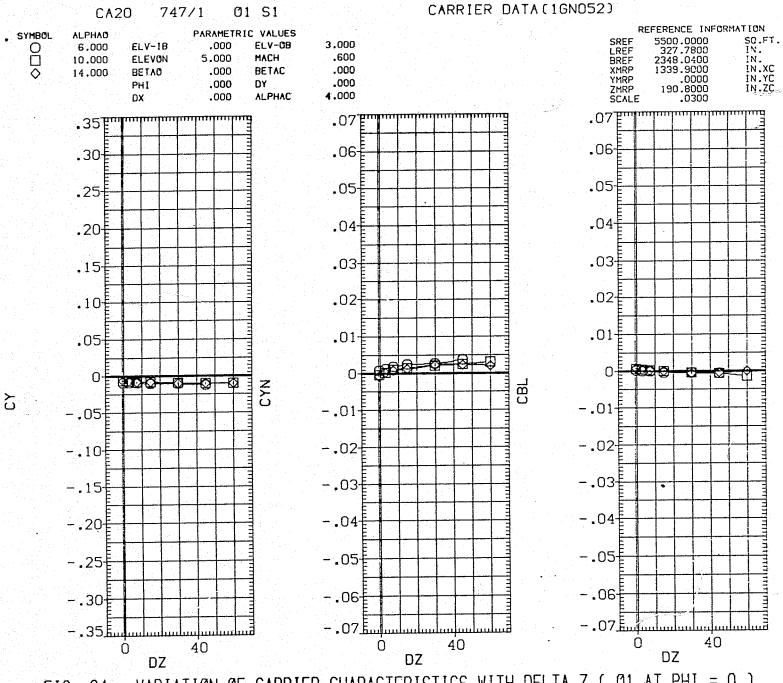


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 502

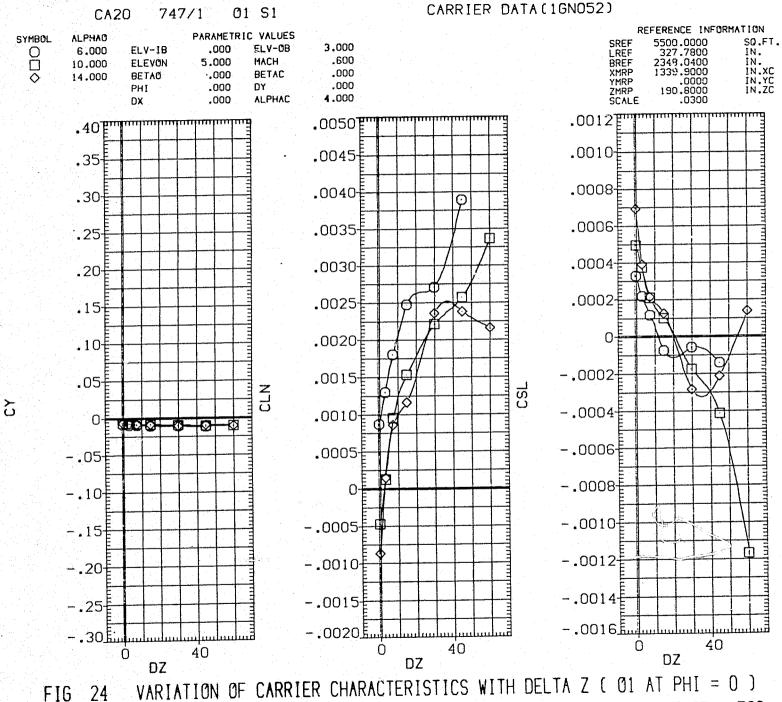


FIG 503 PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

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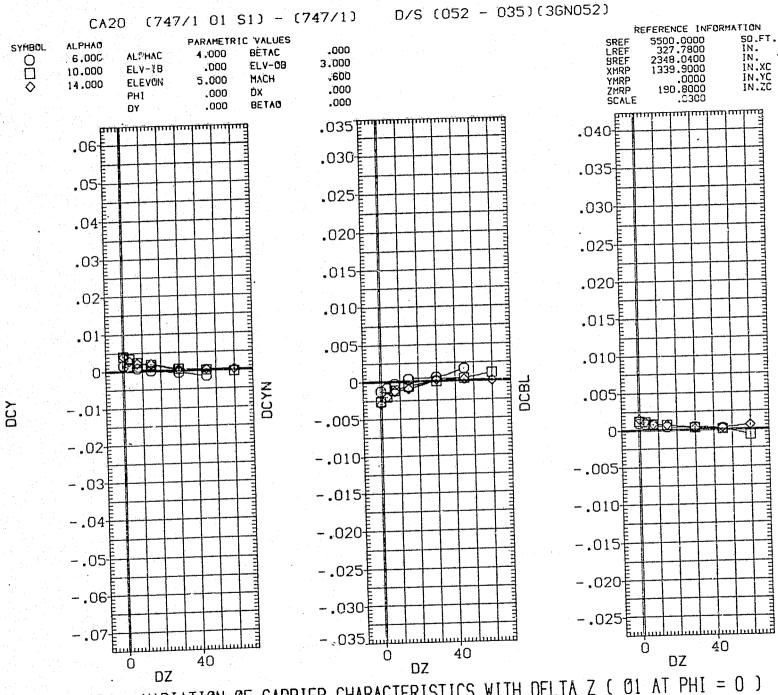
PAGE 504

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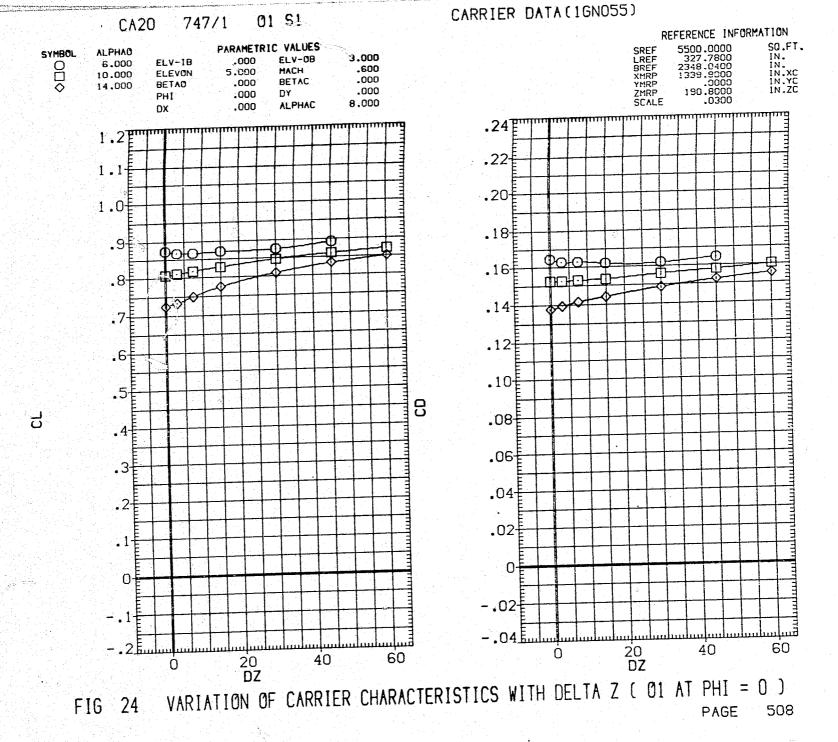
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 505



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0 24 FIG 506 PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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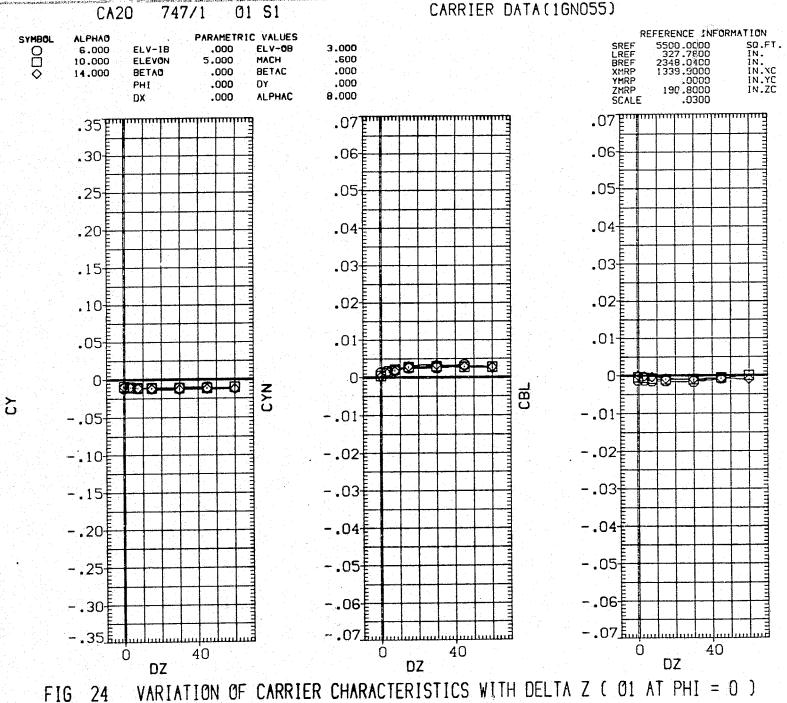


FIG 24 PAGE 510

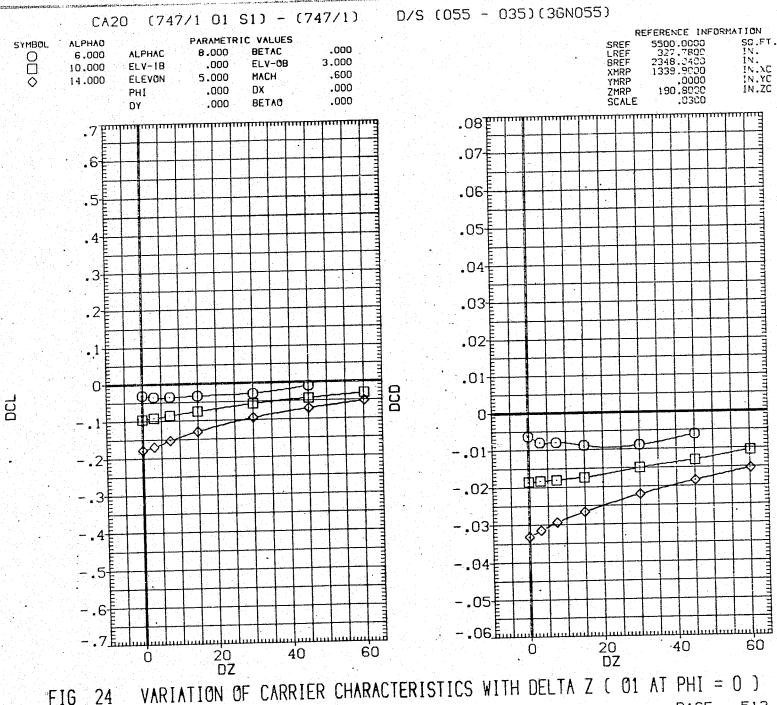


FIG 24 PAGE 512



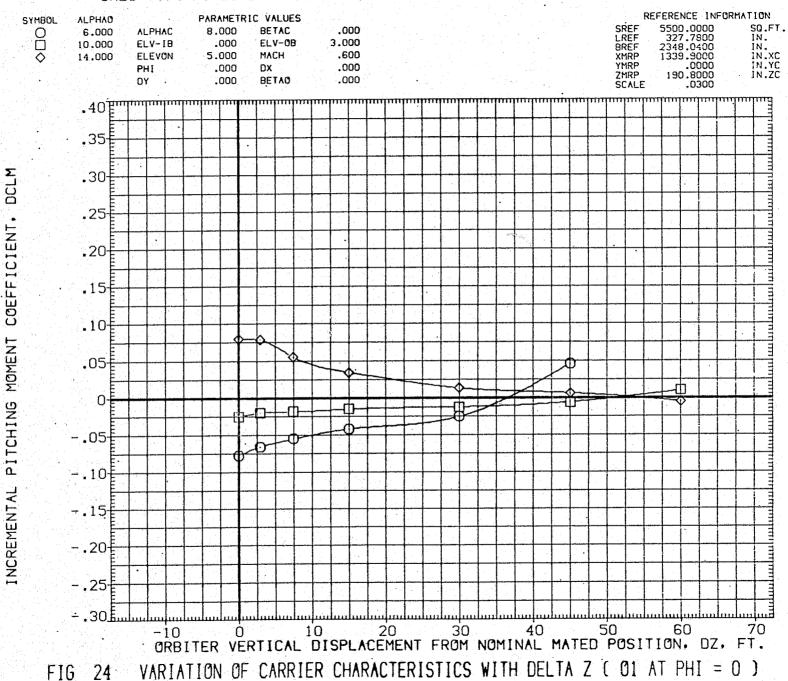


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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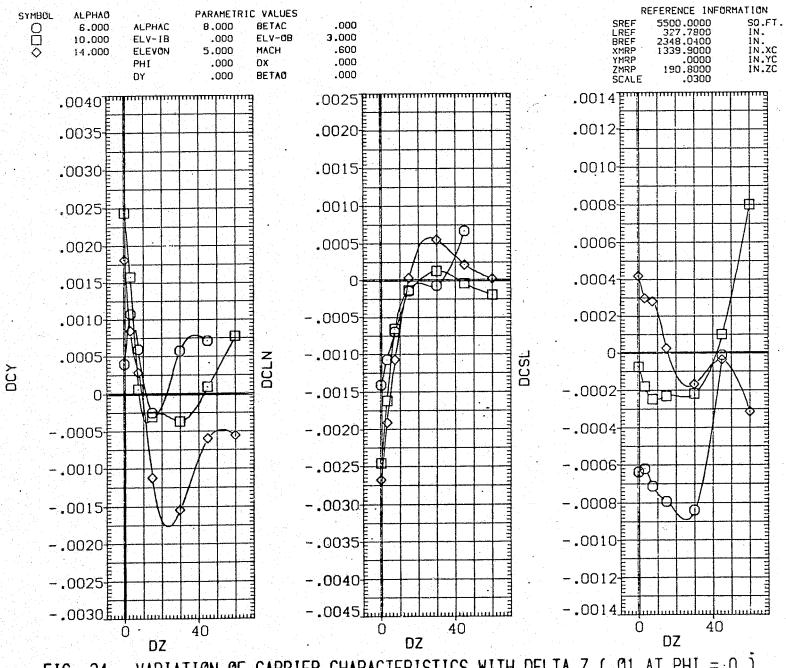


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 515

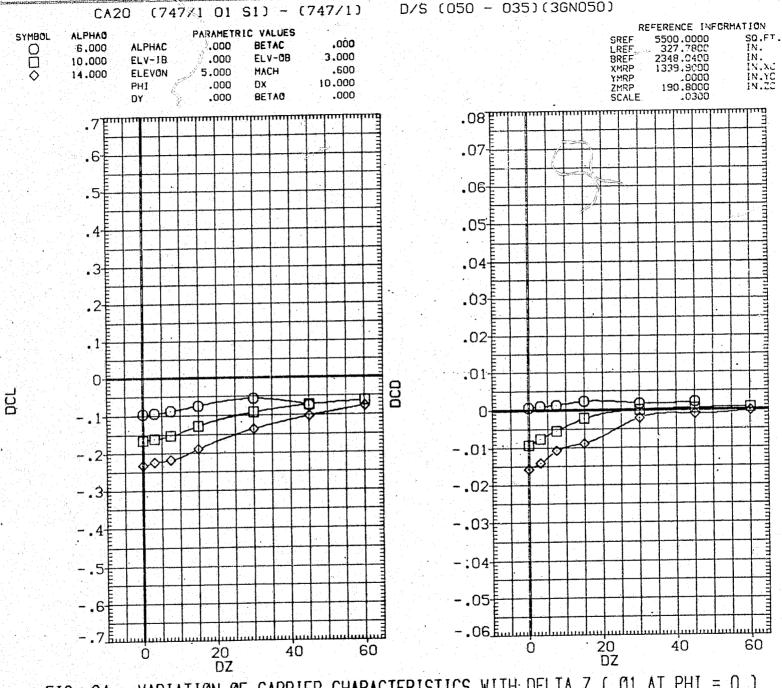


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 516

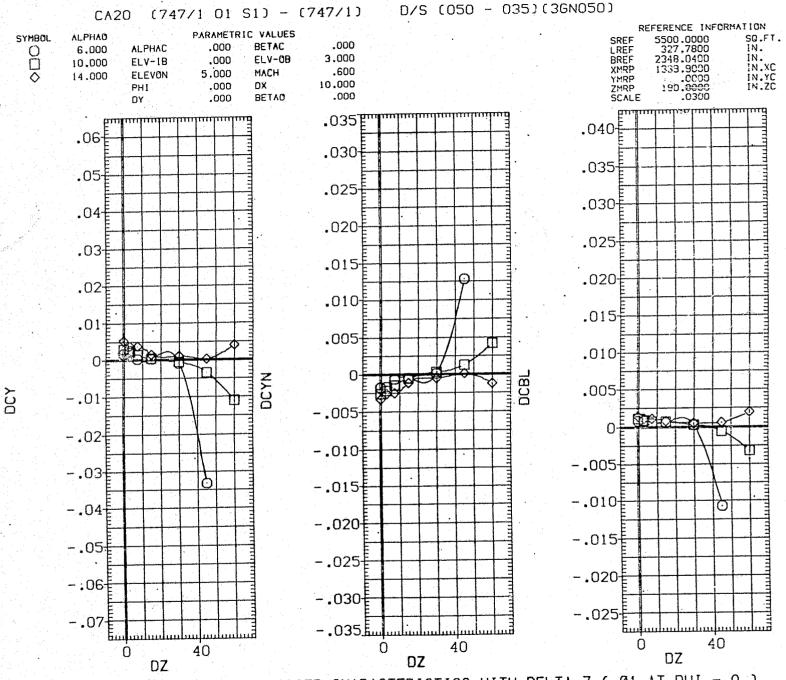


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 518

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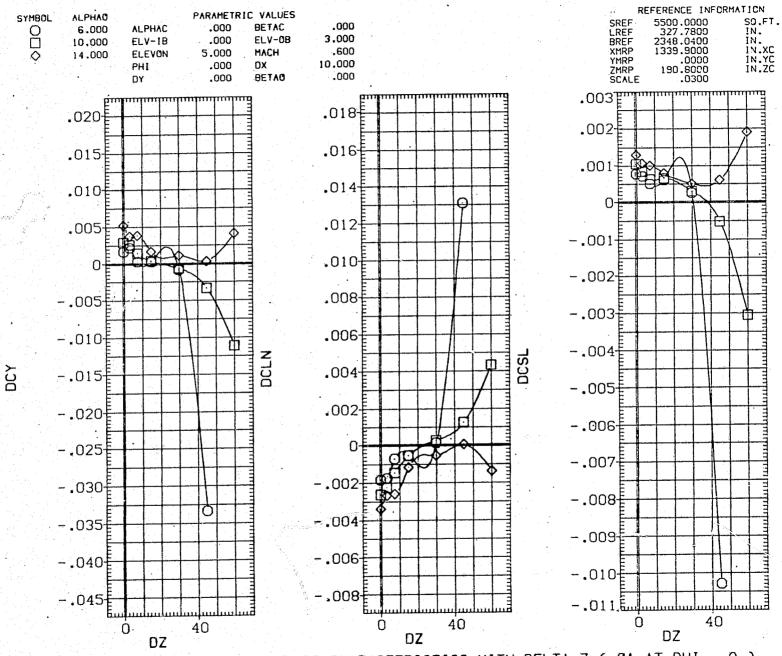


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 519

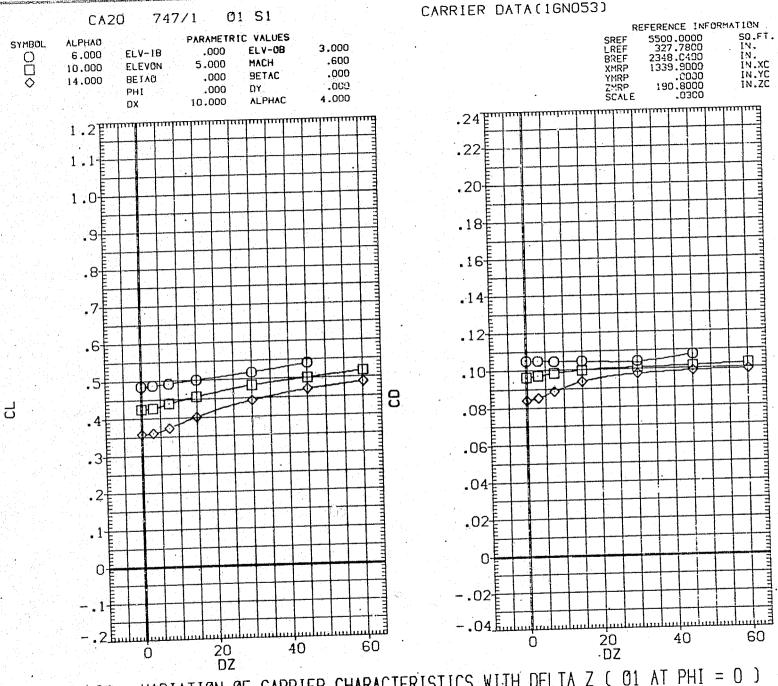


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 520

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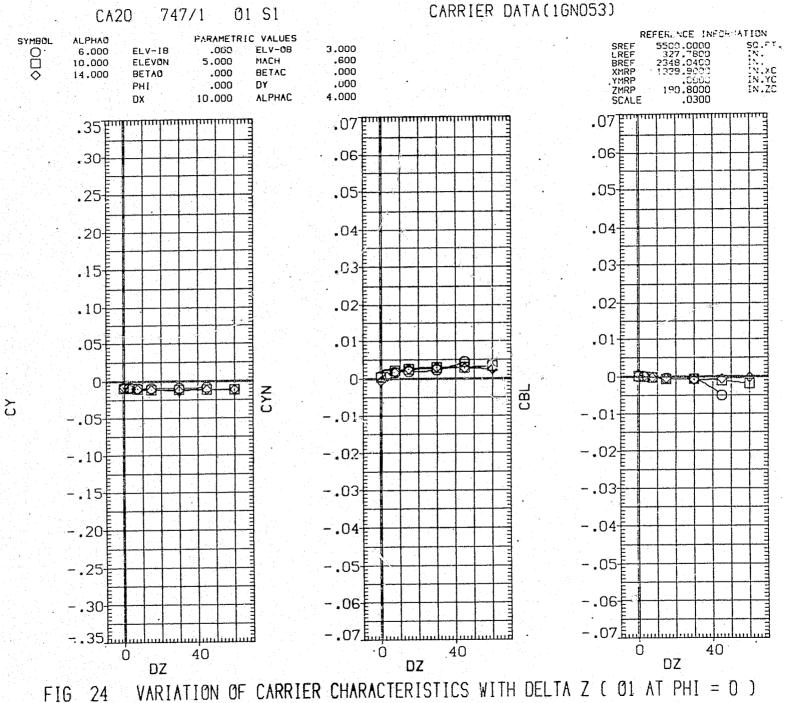


FIG 24 522 PAGE

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 525

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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

PAGE 526

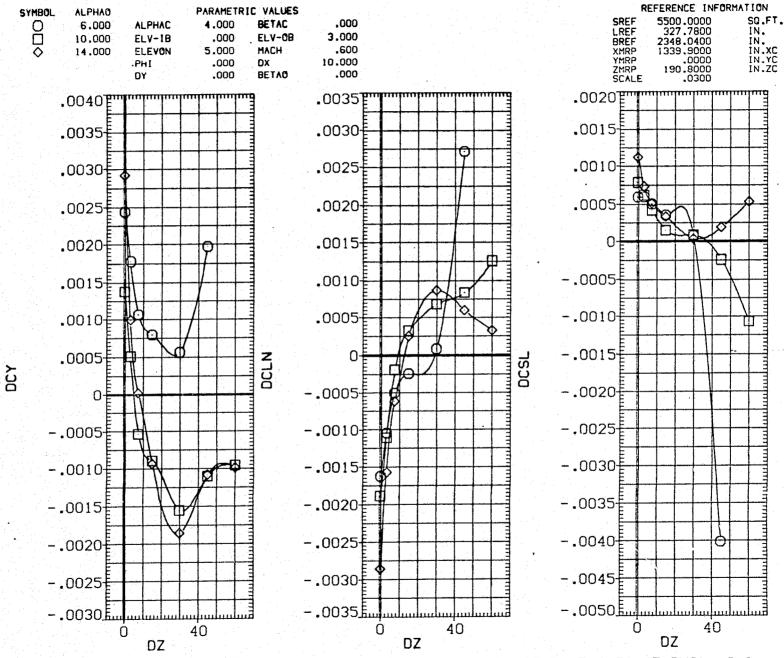
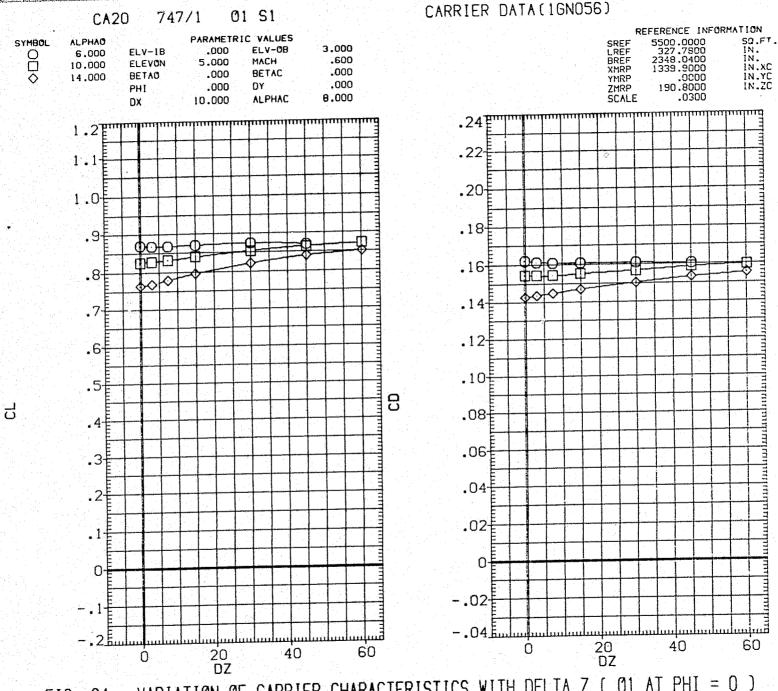


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0 FIG 528 PAGE

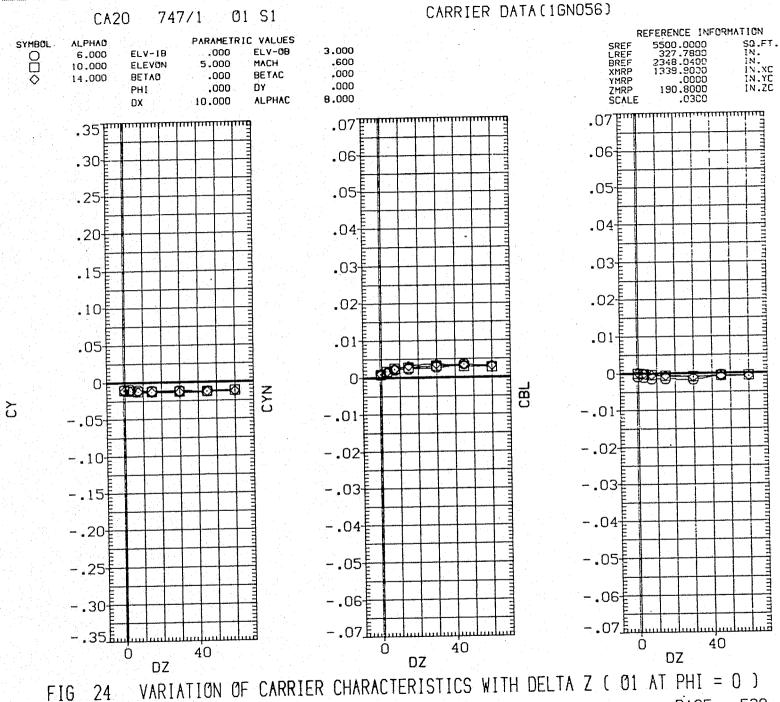


FIG PAGE 530

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 531

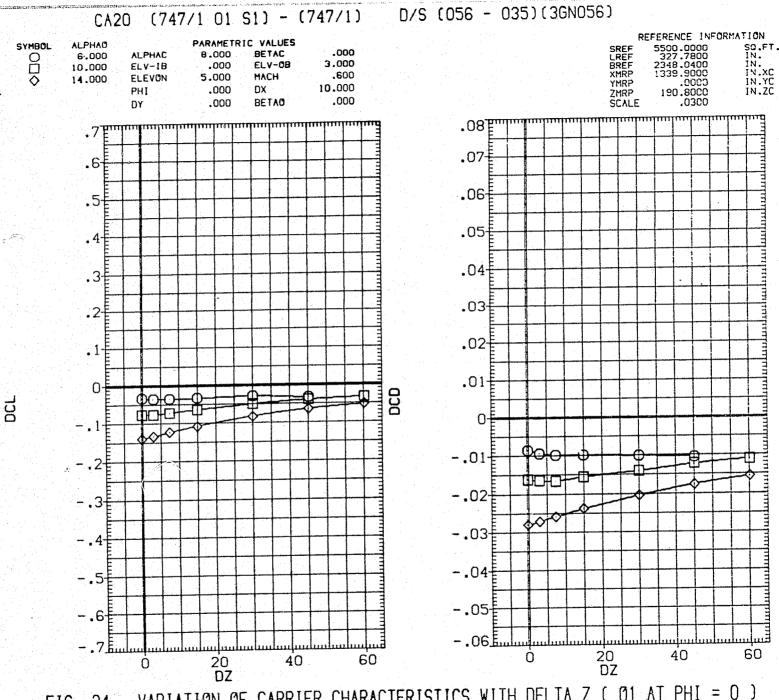
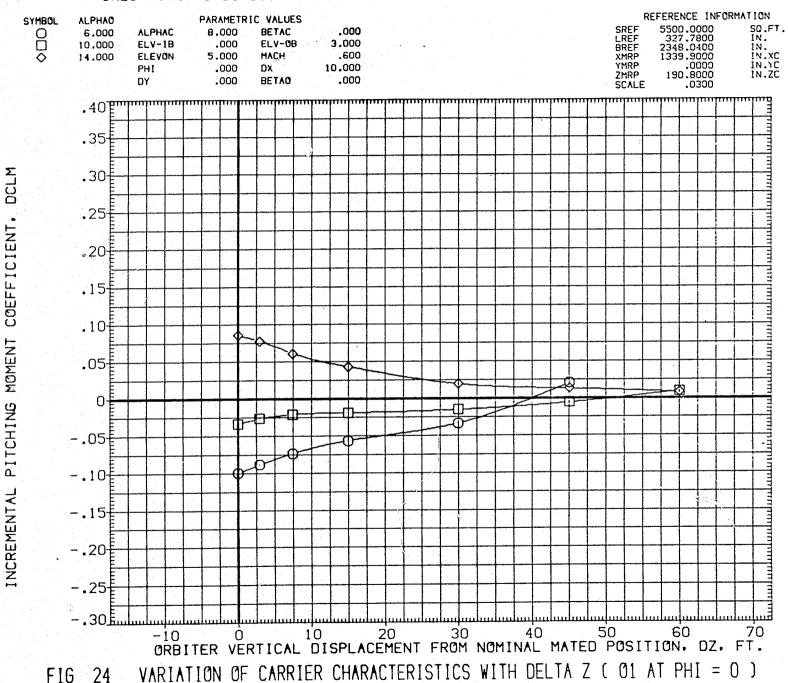


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 532



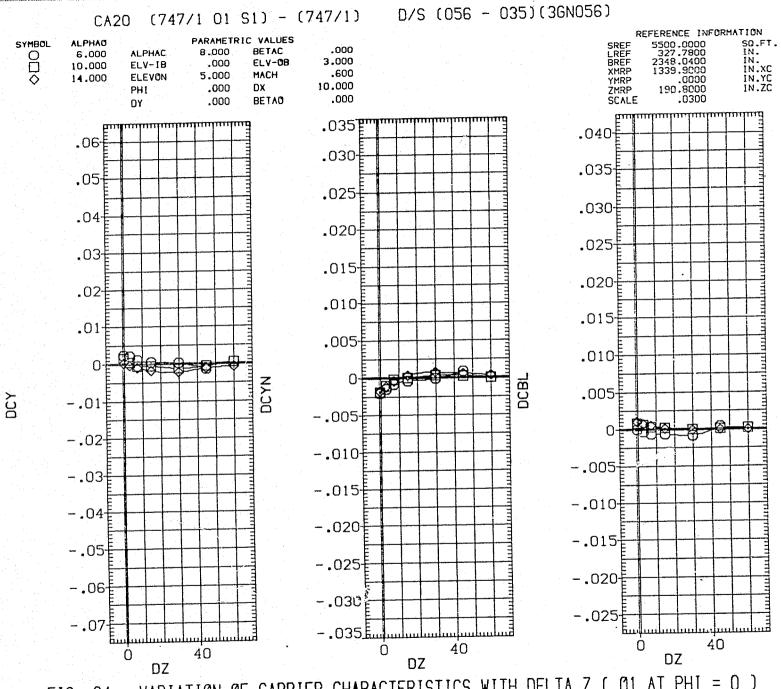


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

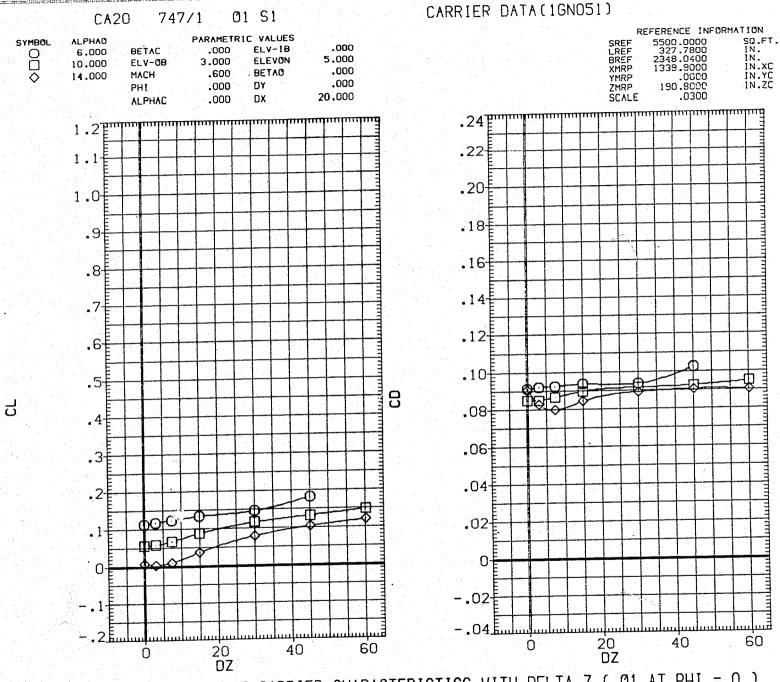


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 536

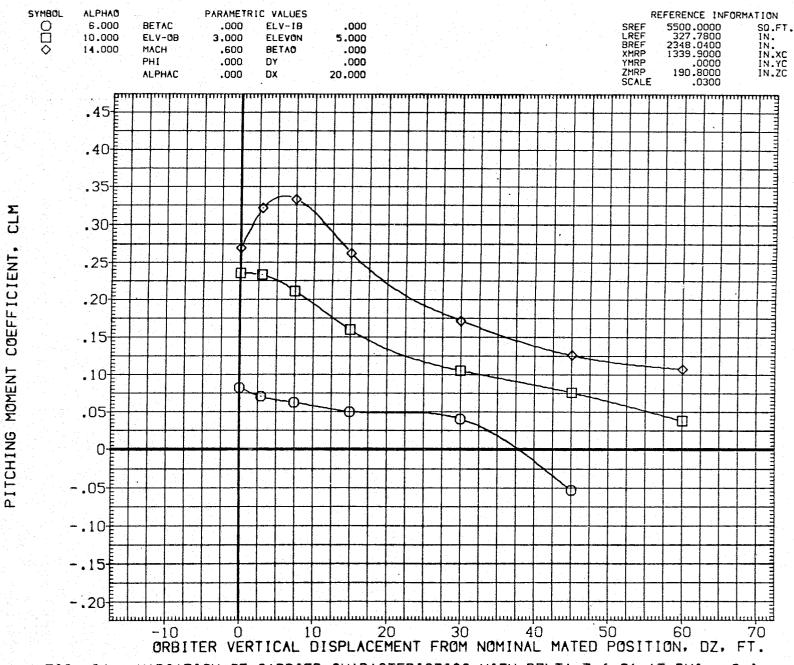
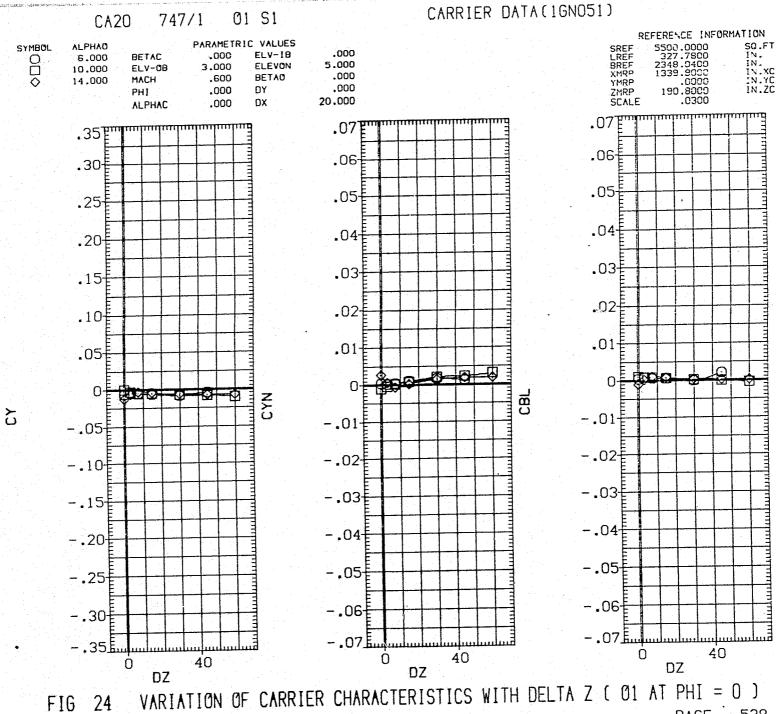


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



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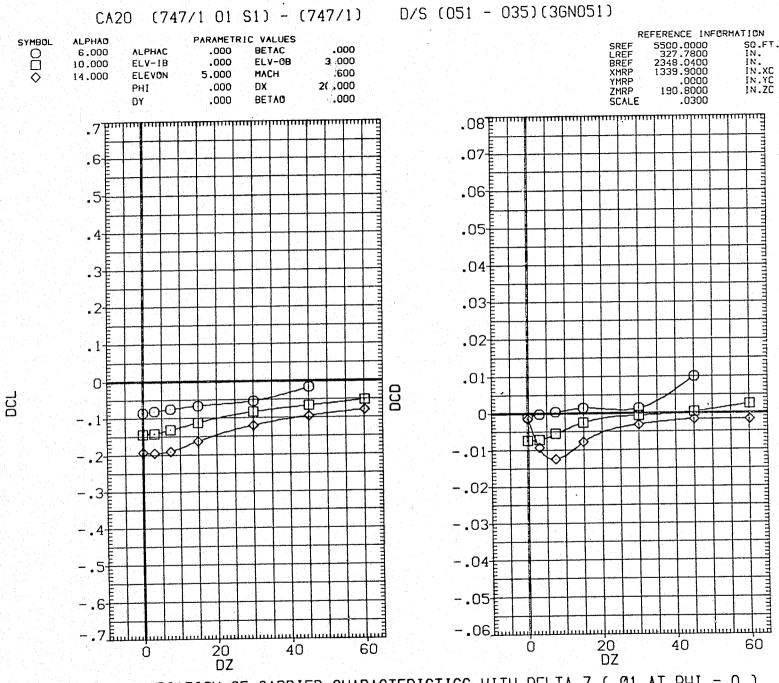


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 541

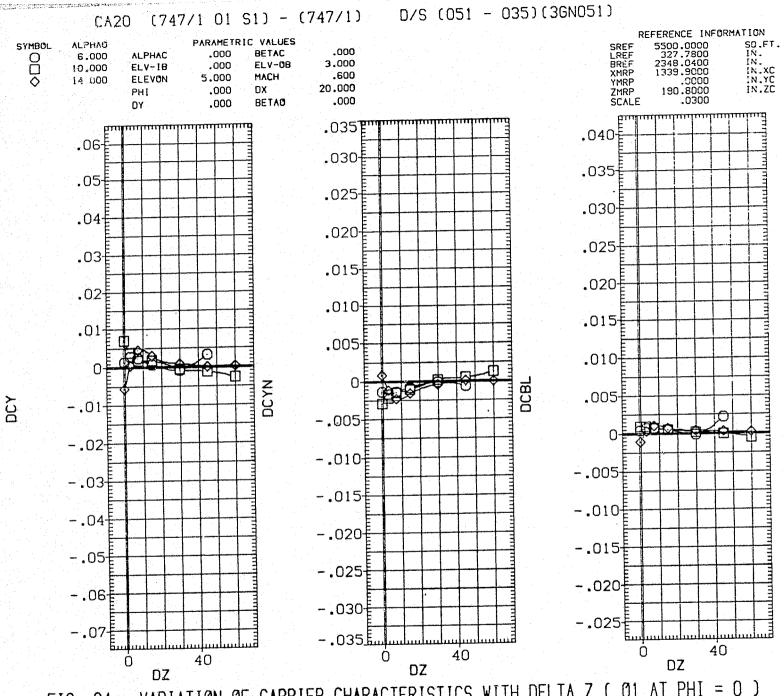


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 542

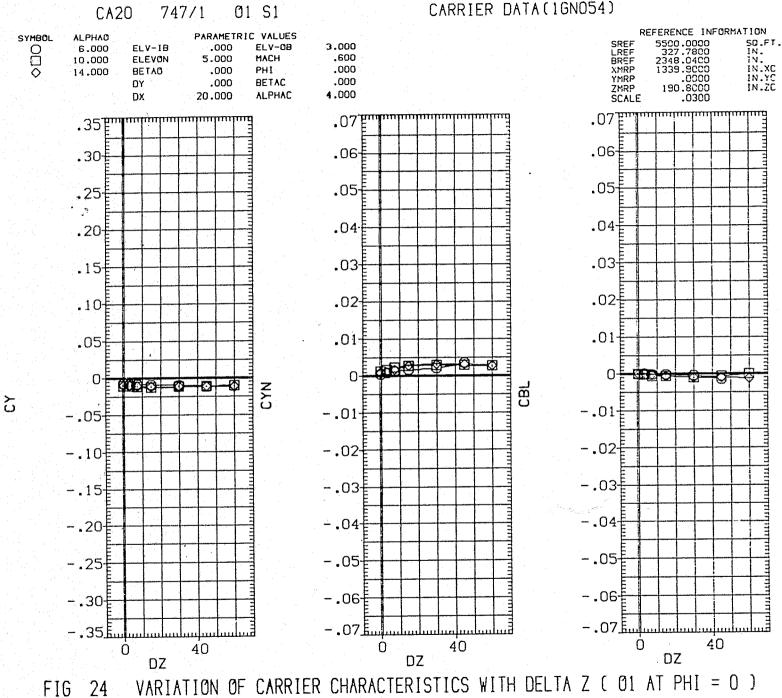
DZ VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG PAGE 543

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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)



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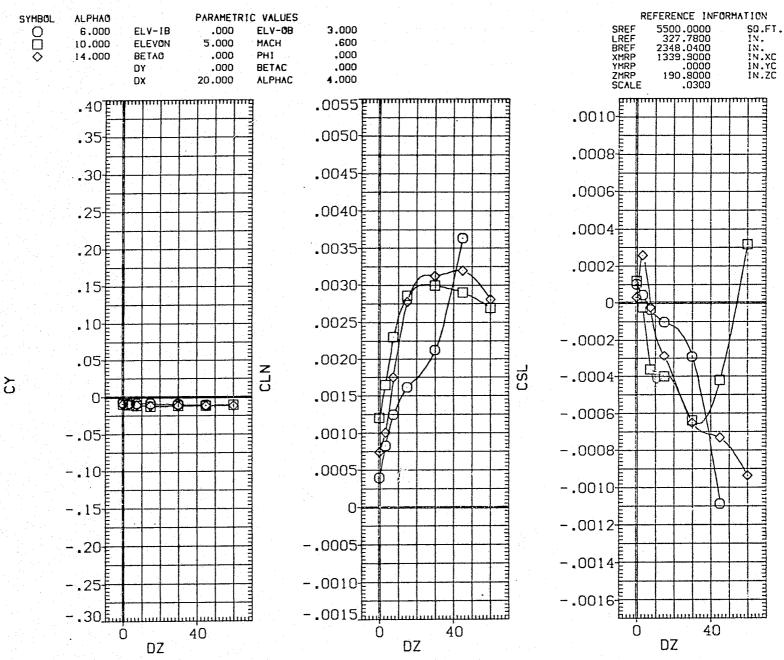


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 547

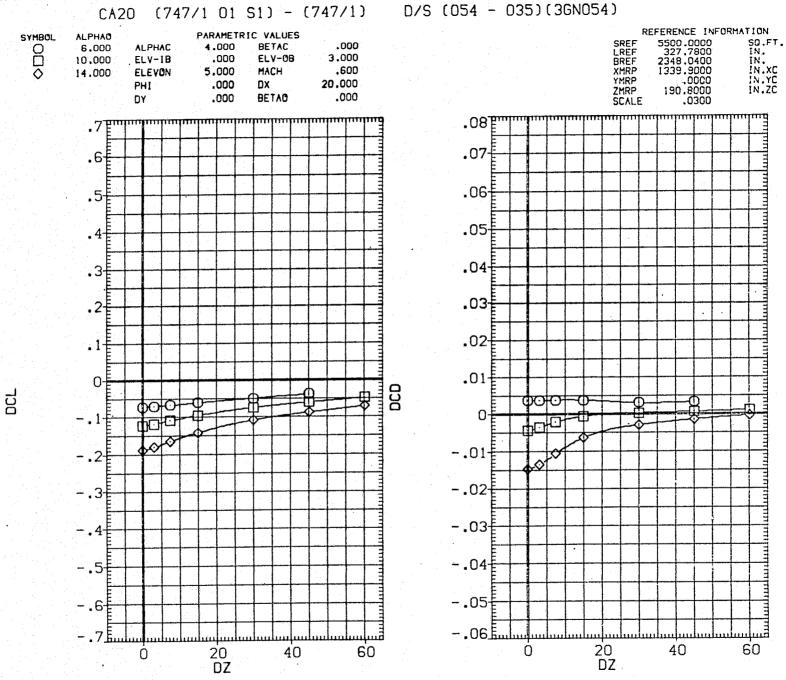
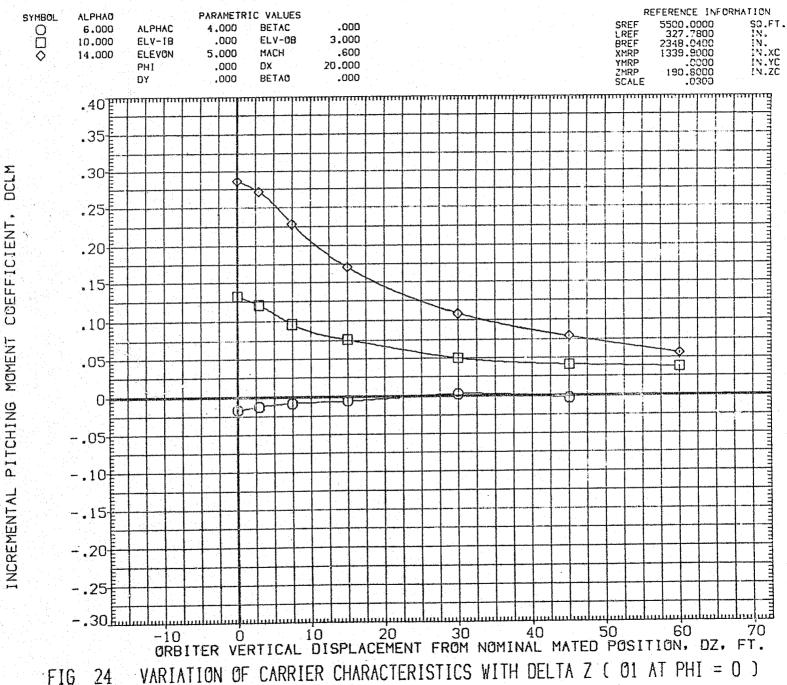


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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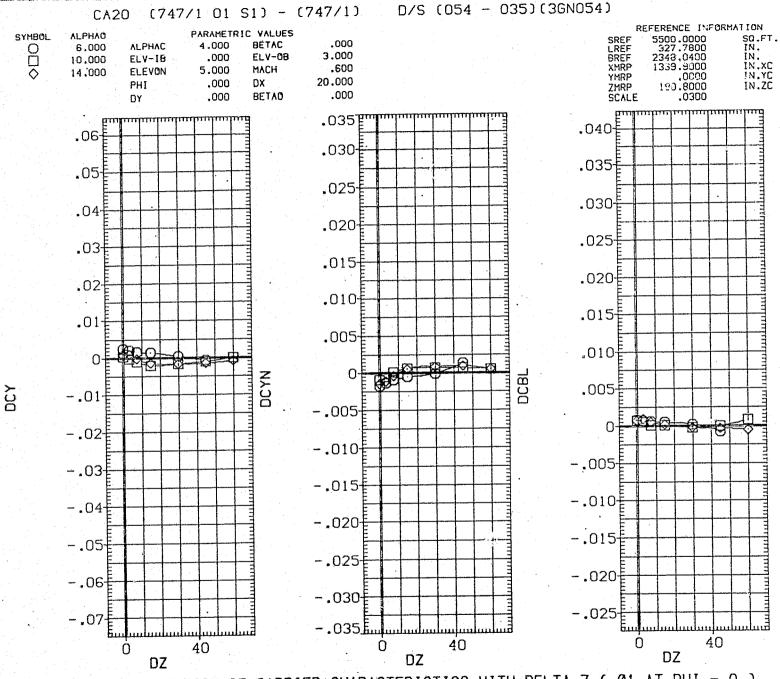


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 550

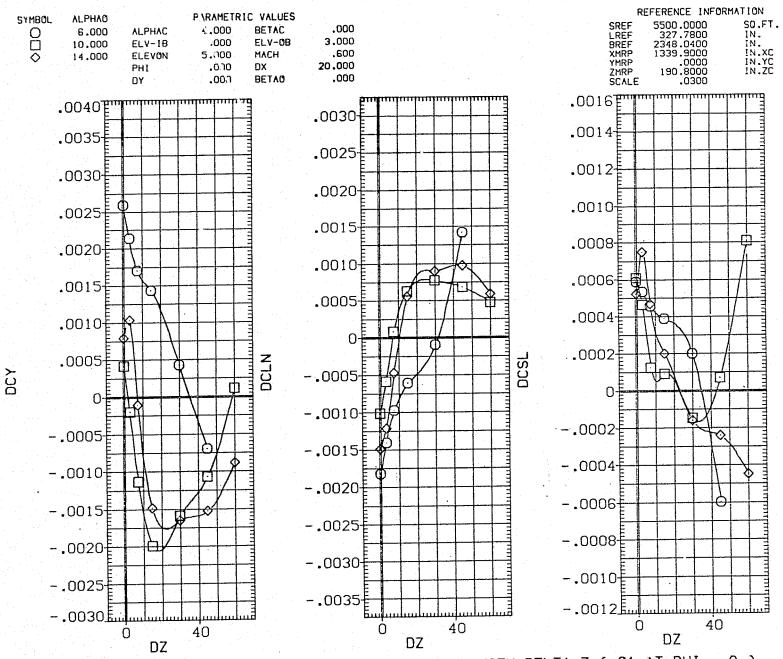


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 551

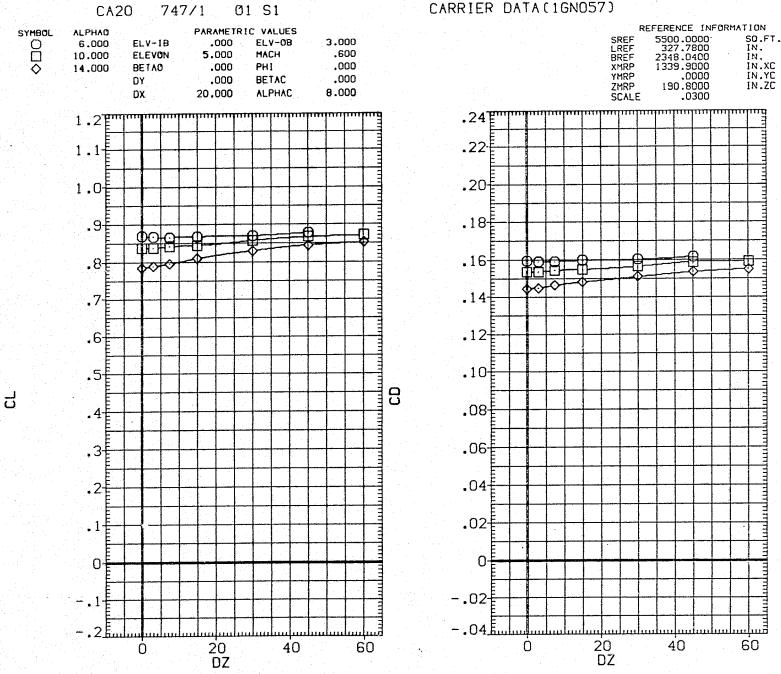


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 555 PAGE

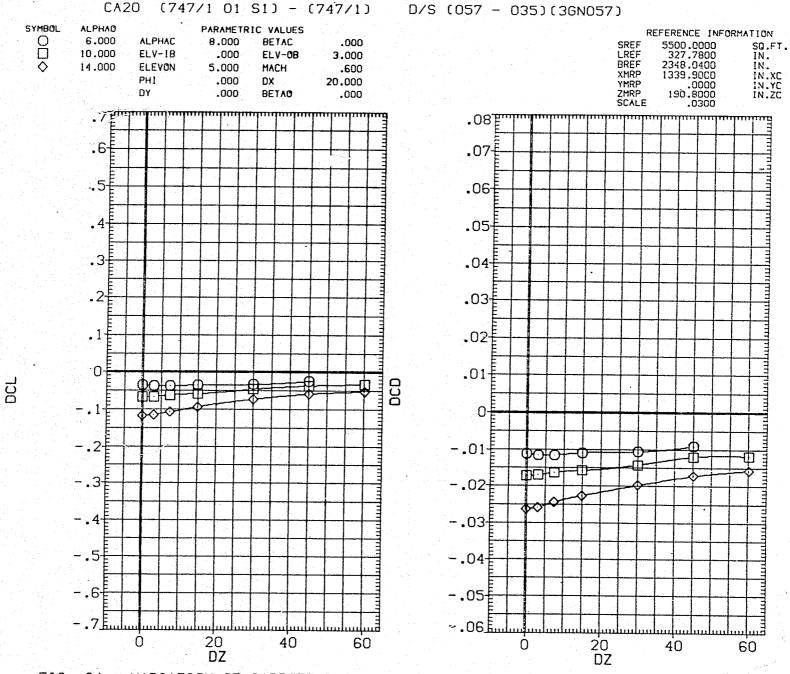
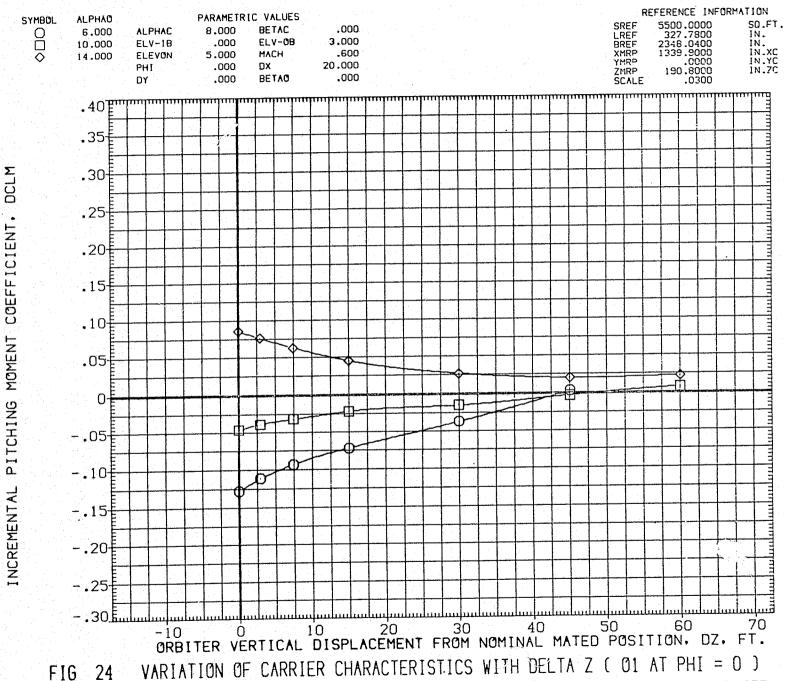


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 556



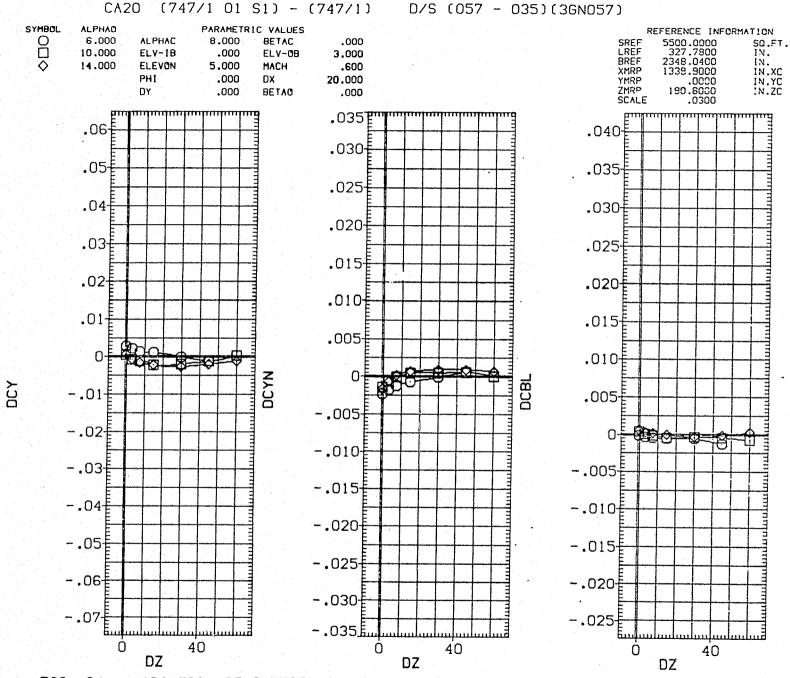


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 558

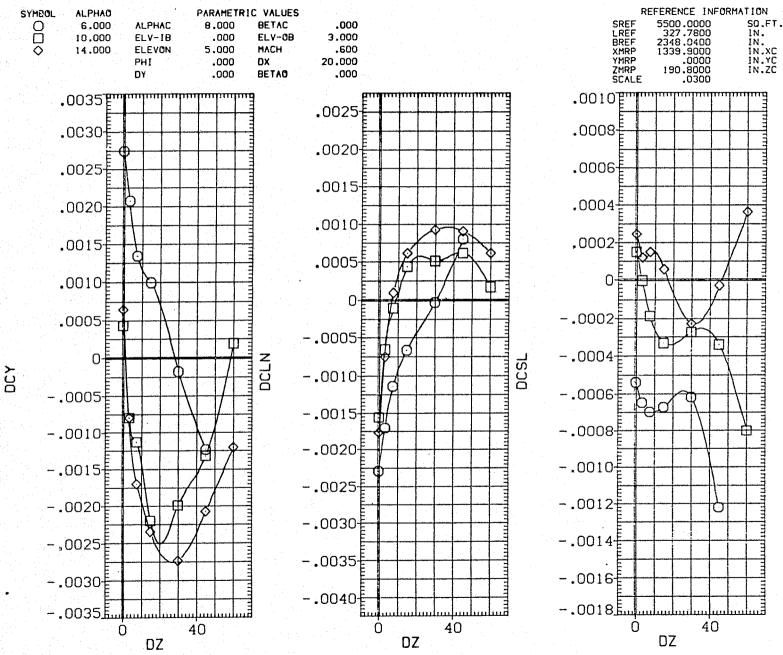


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 559

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 560

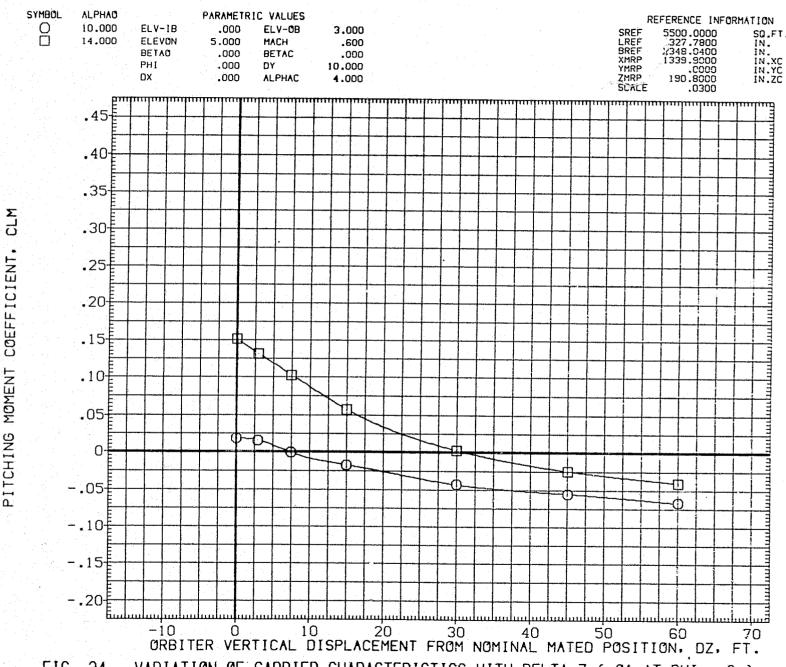


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

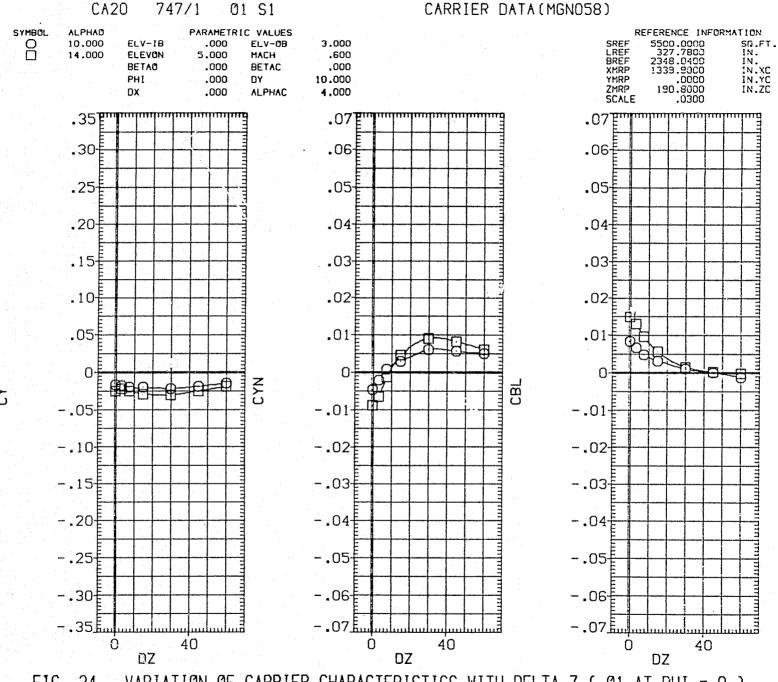
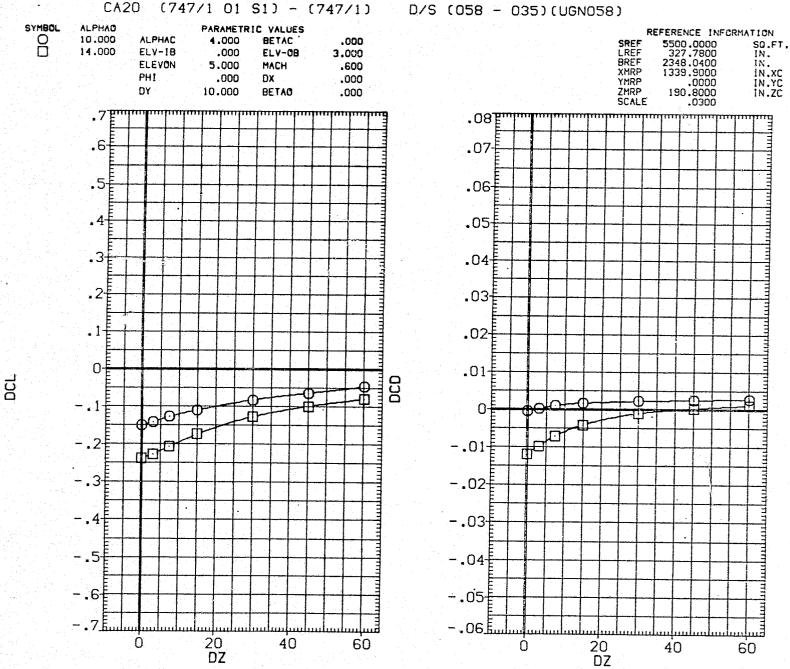


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 562

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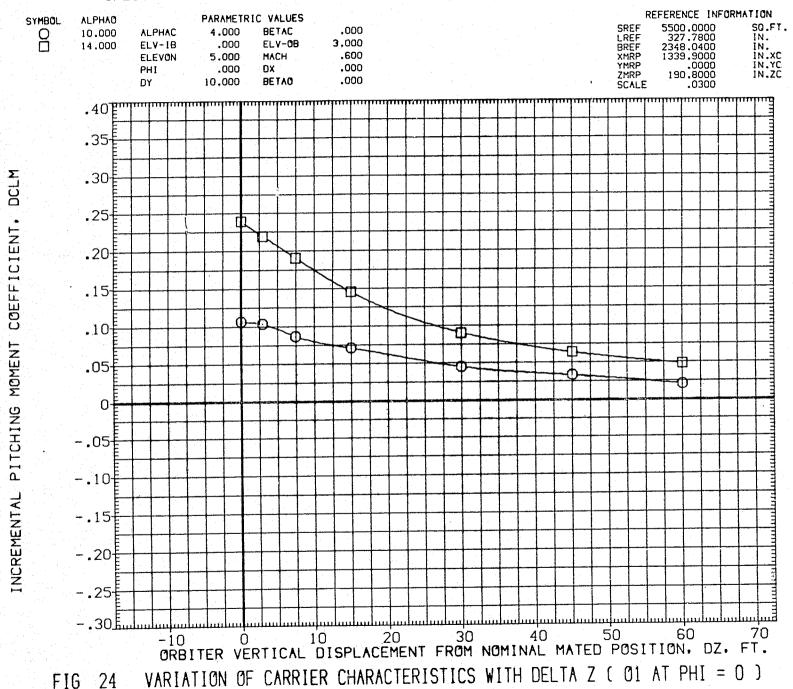


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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 564

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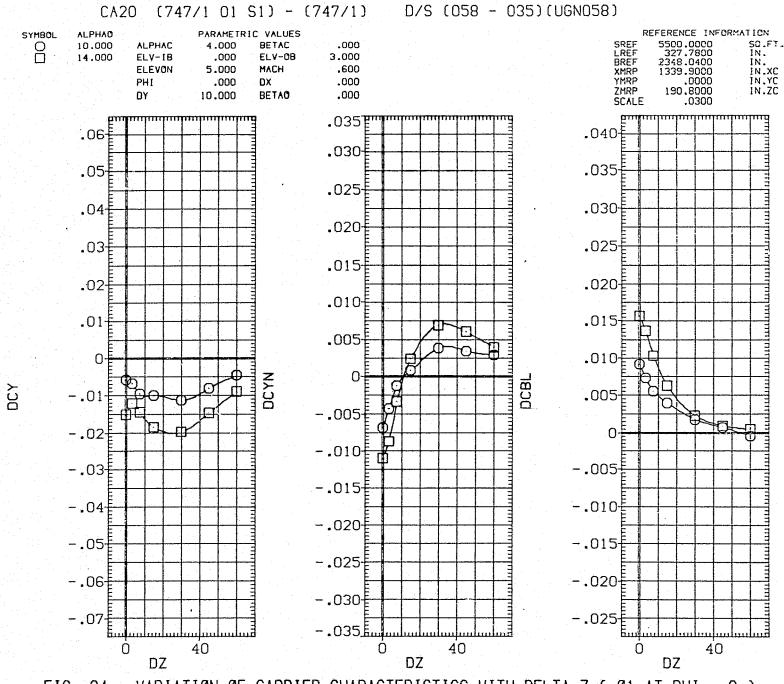


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 566

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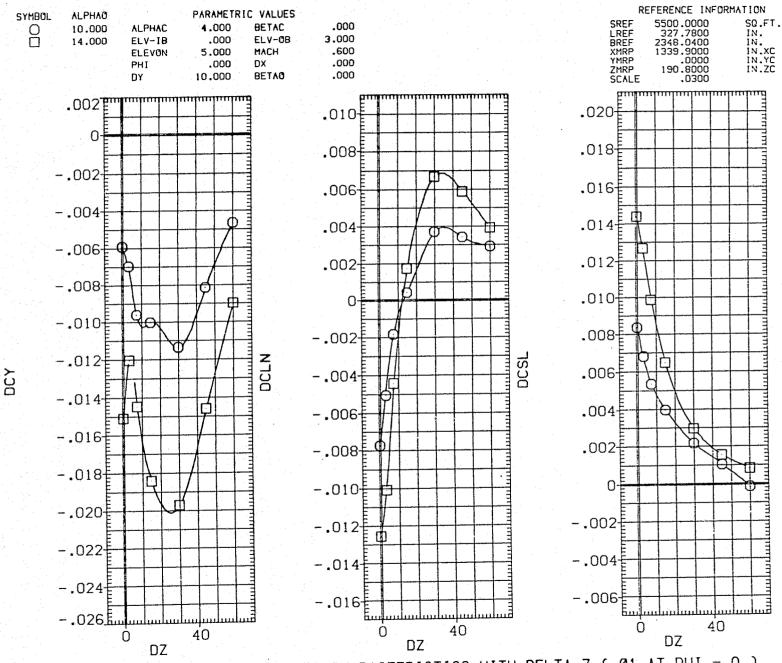


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 567

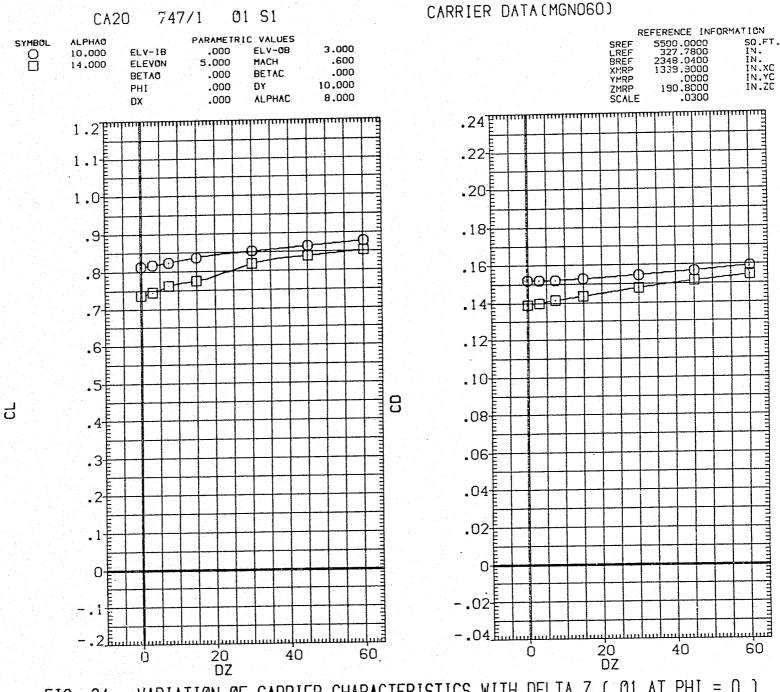


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 568

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 569 PAGE

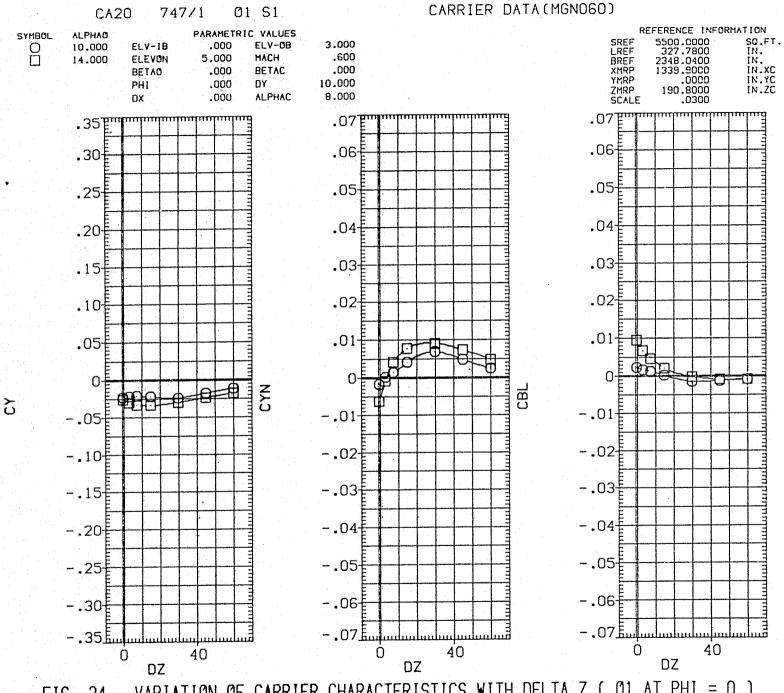


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 571

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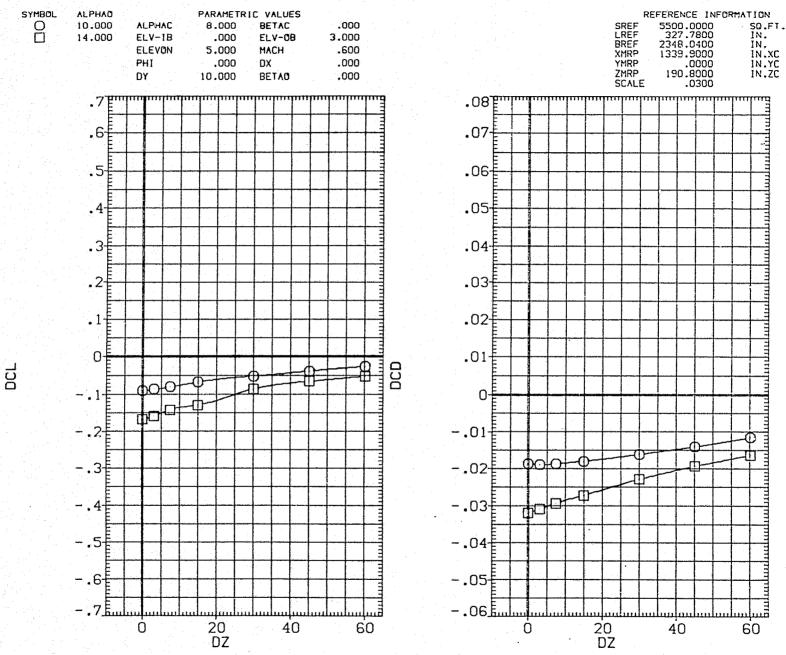
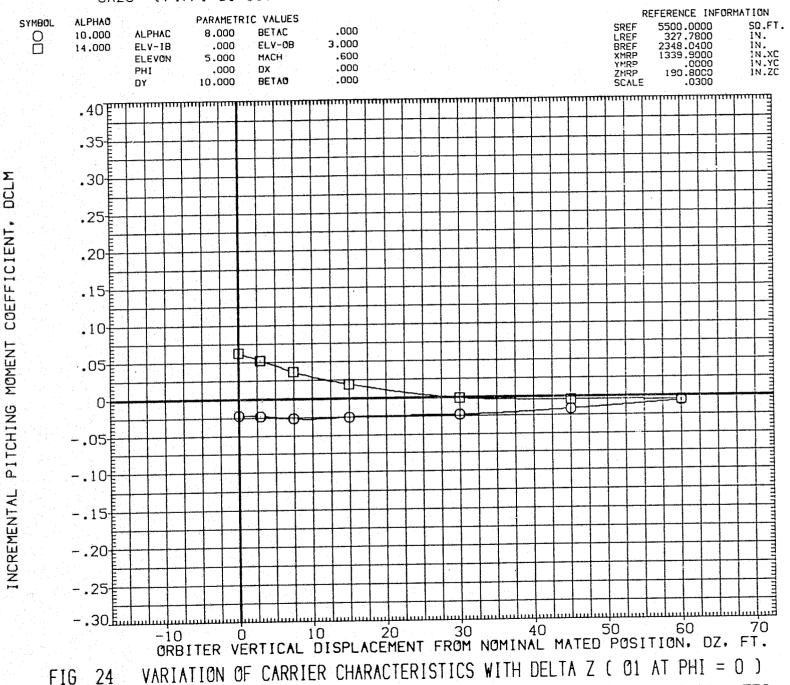


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 572



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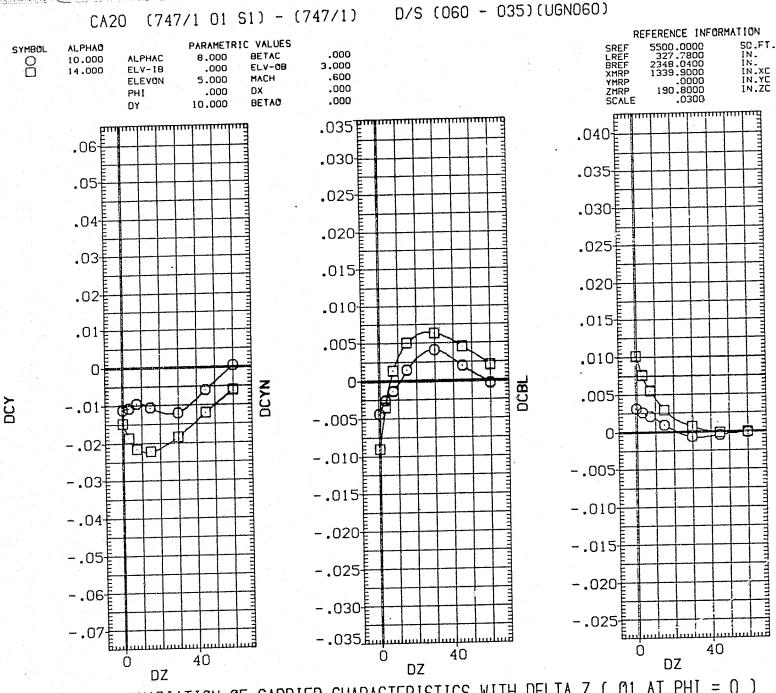


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 574

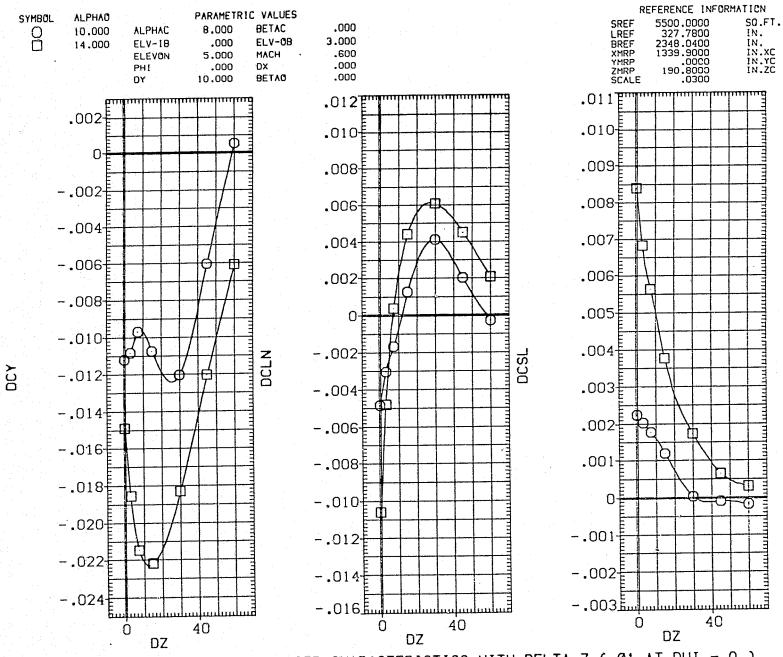


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 575

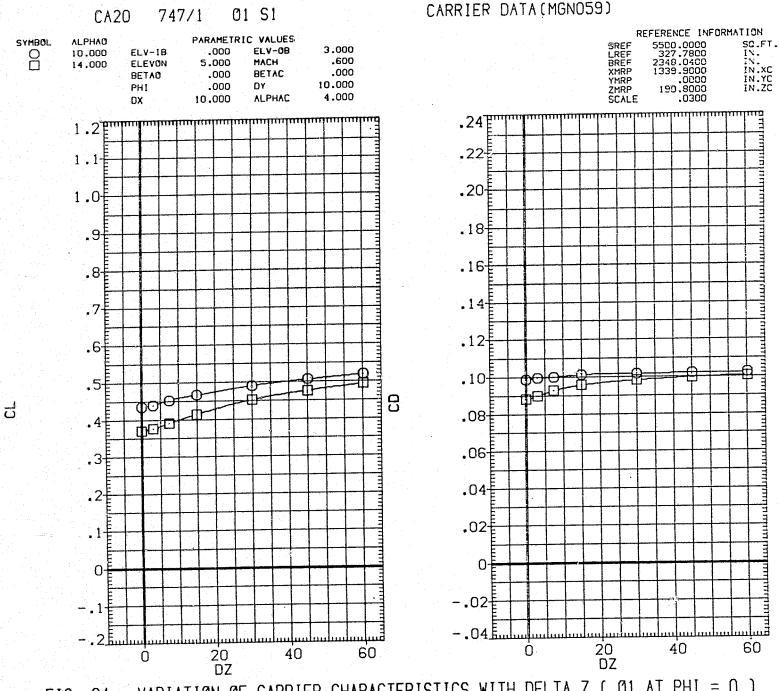


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 576

FIG 24

PAGE 577

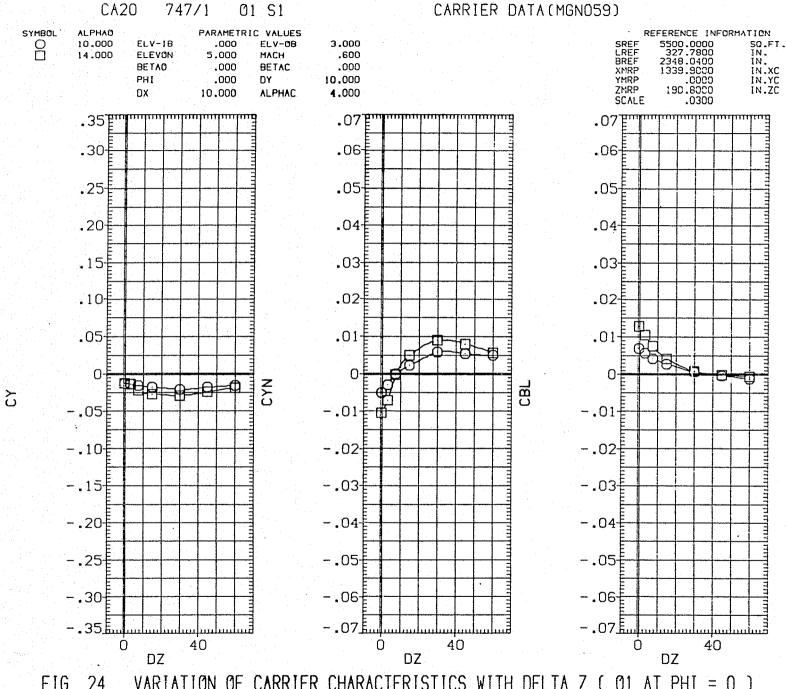


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 578

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 579

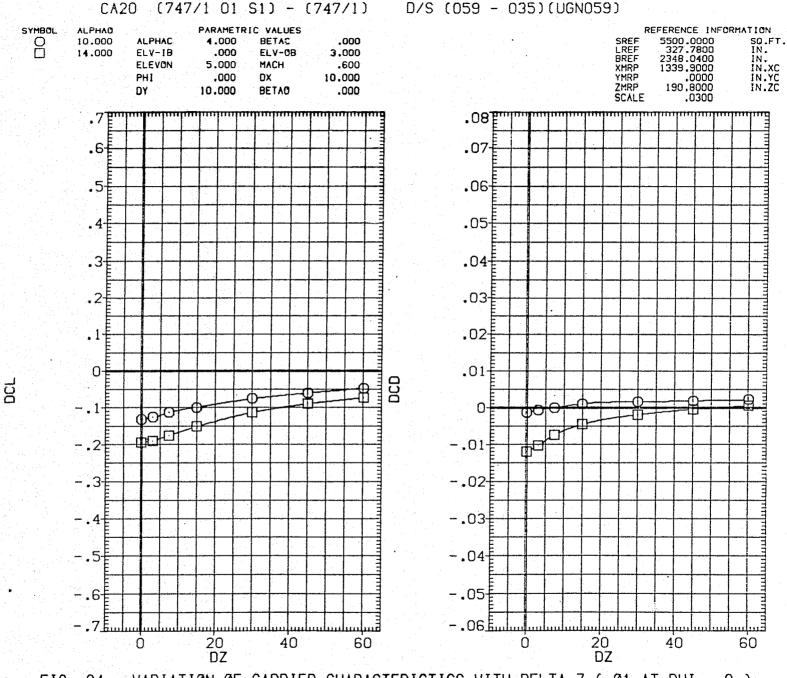


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 580

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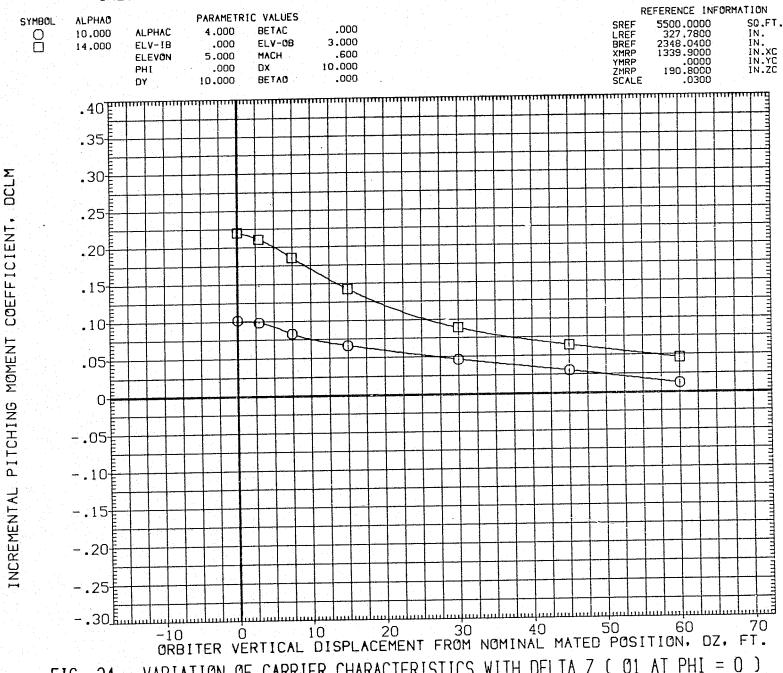


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 581

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 582

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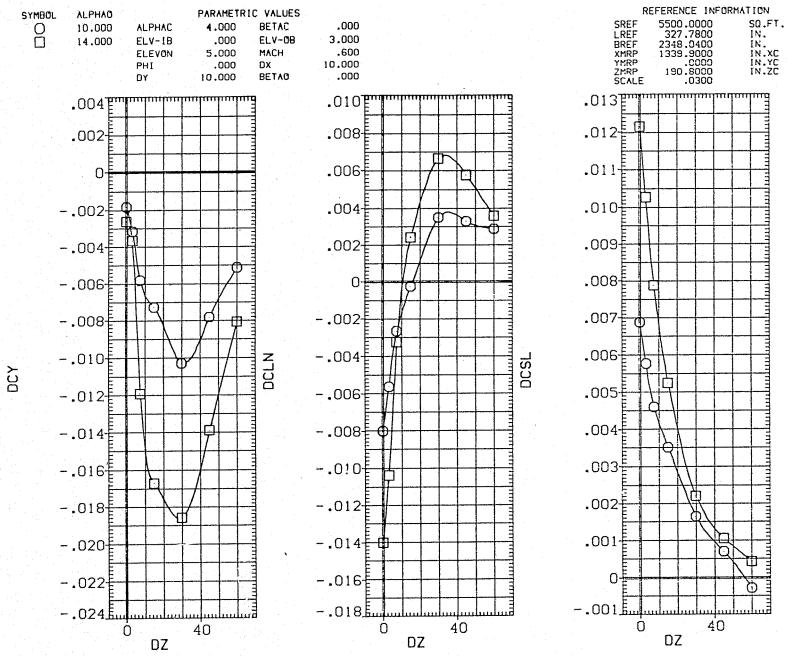


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 583

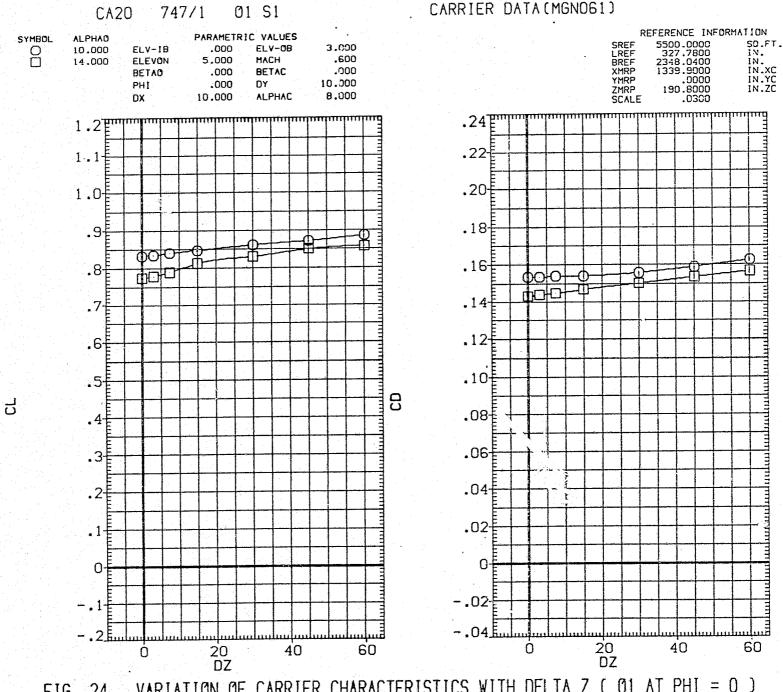


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 587

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 588

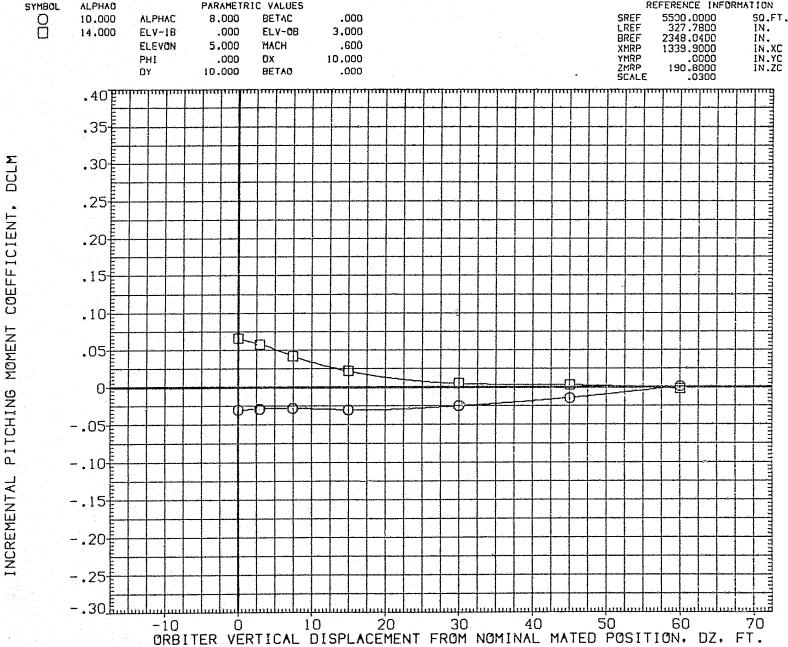
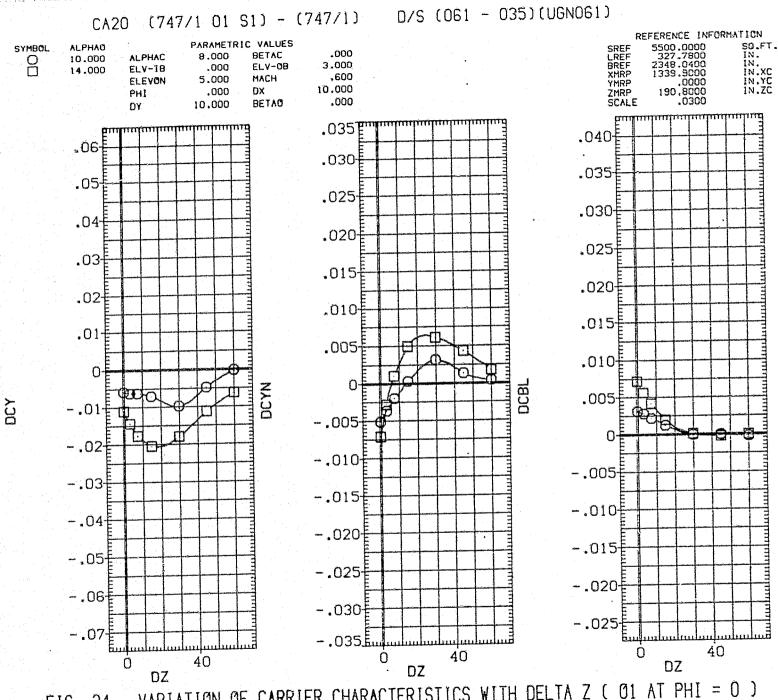
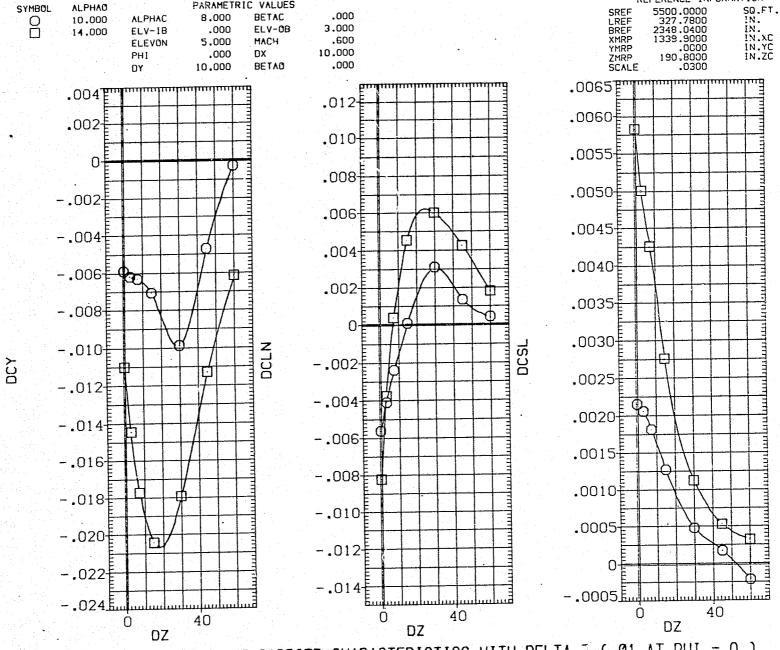


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 589



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) 24 FIG PAGE 590

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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA 2 (01 AT PHI = 0) FIG 24 591 PAGE

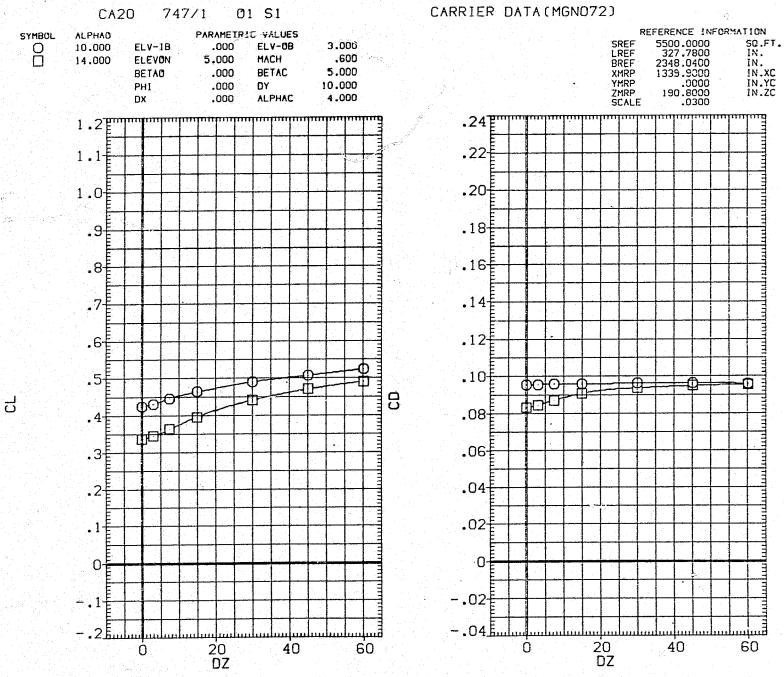
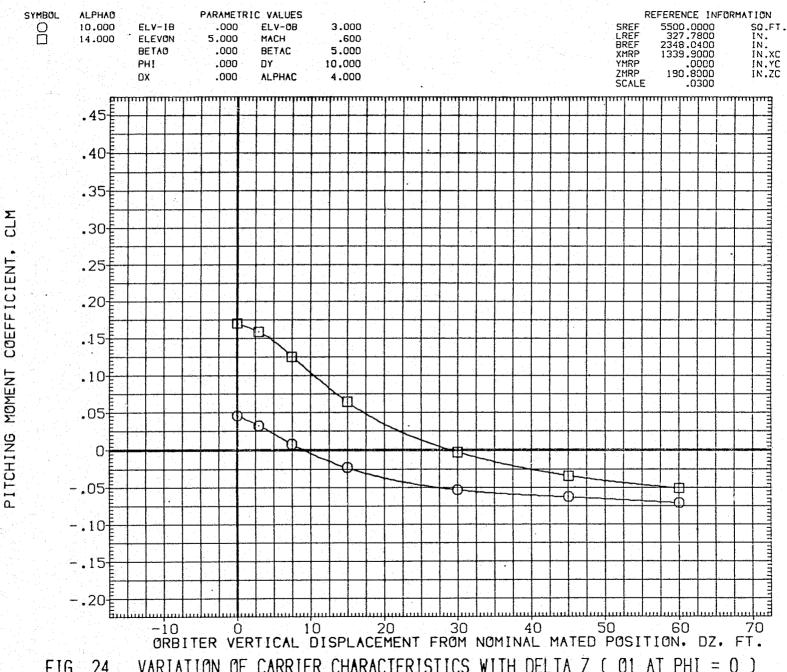
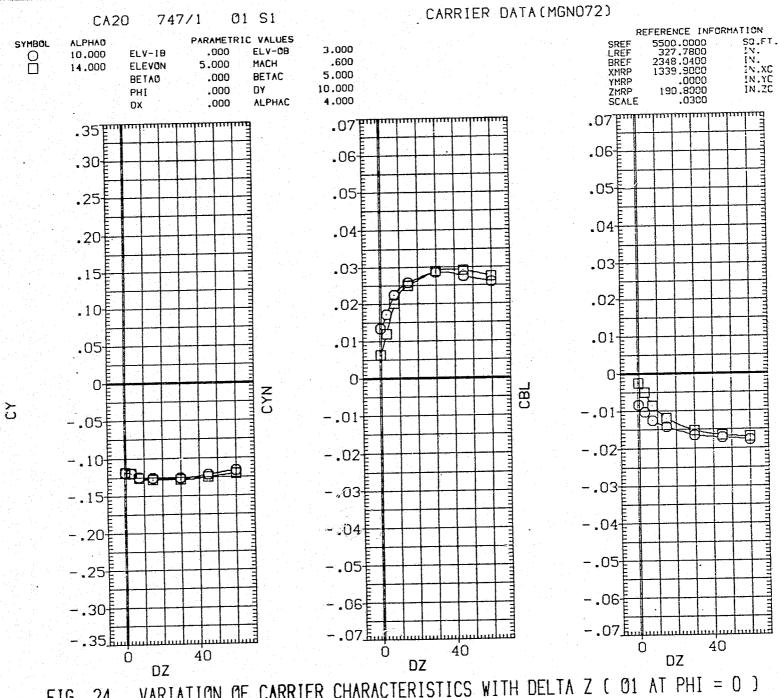


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 PAGE 594

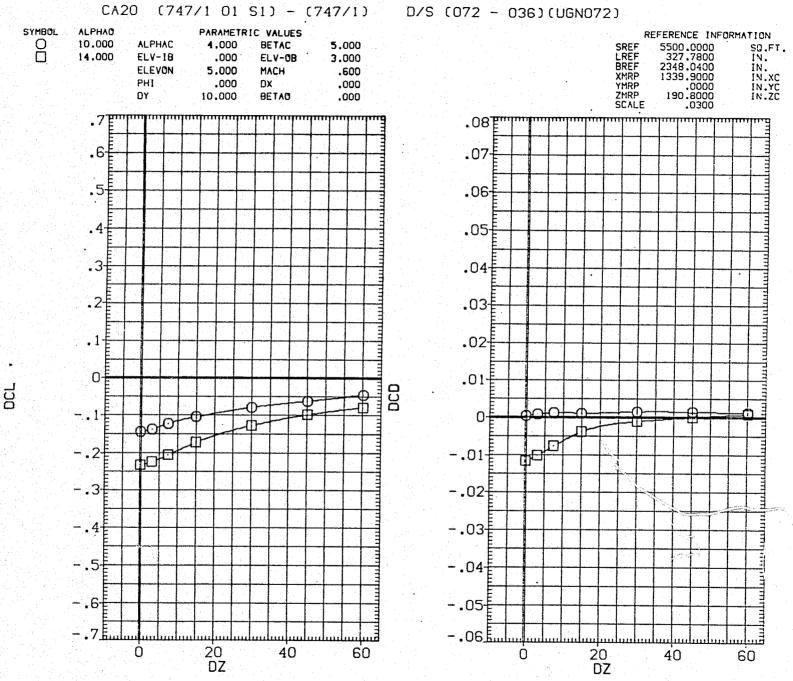
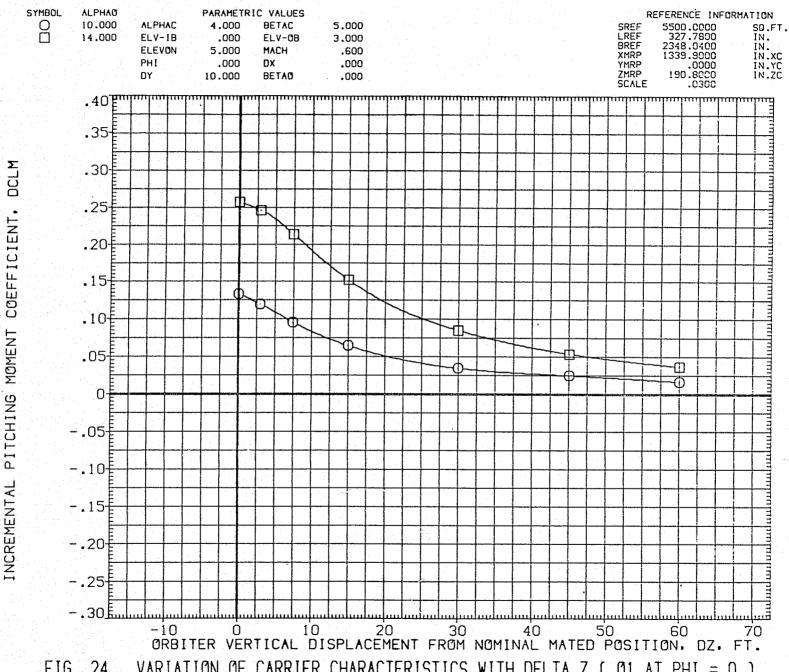


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 598

FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 599

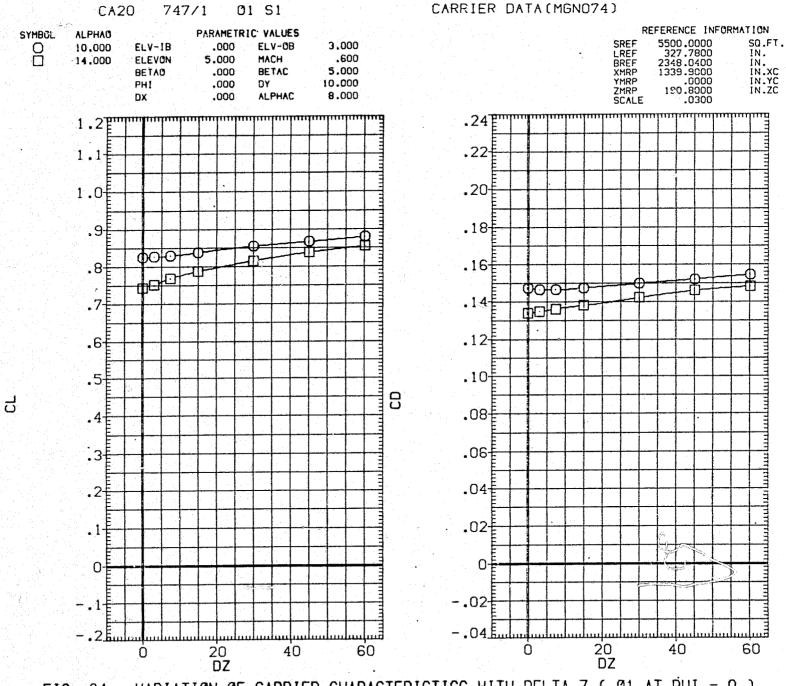
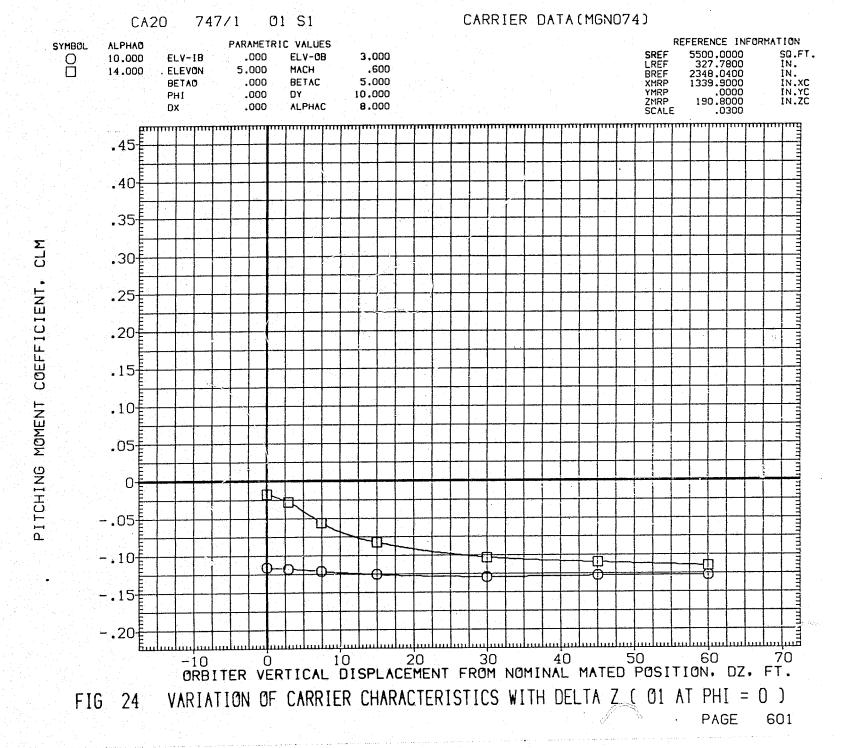


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 600

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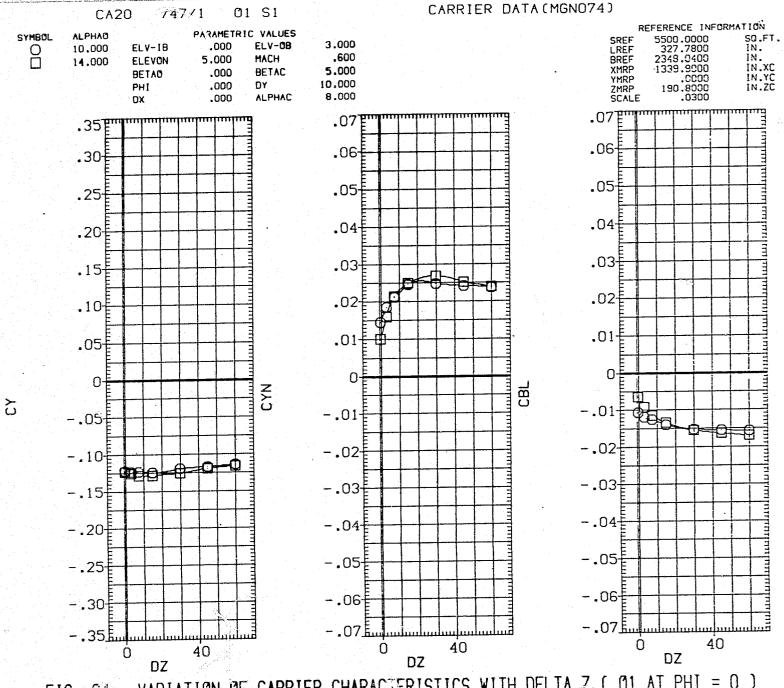


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 602

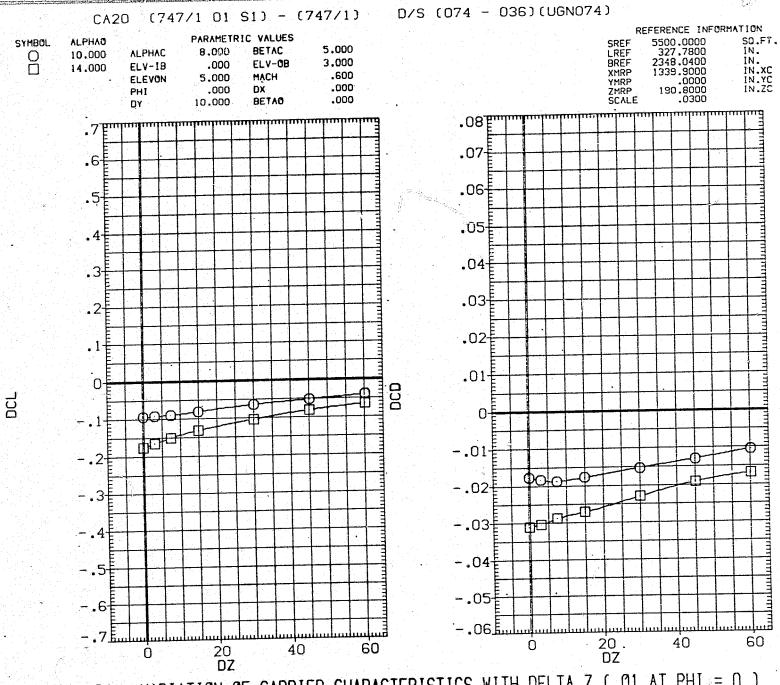
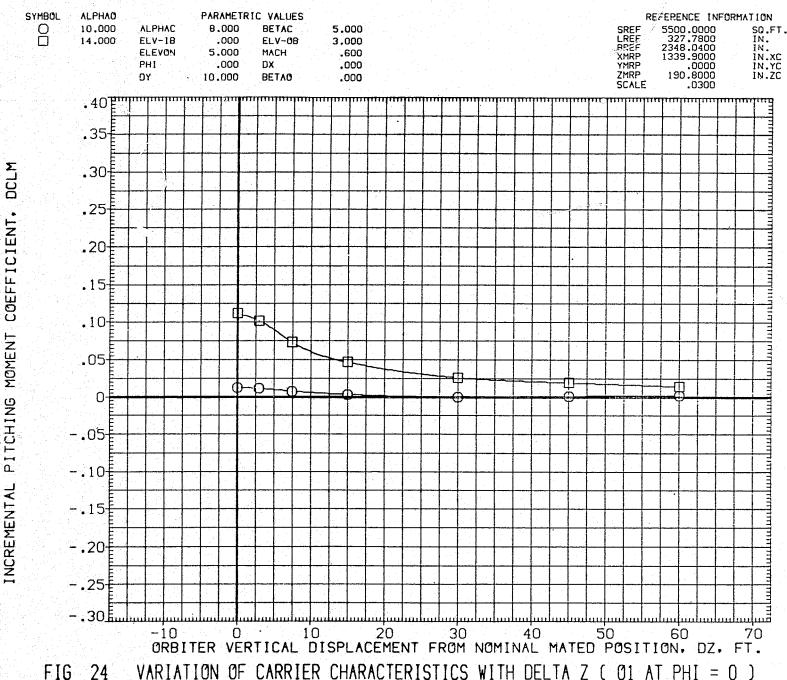


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 604



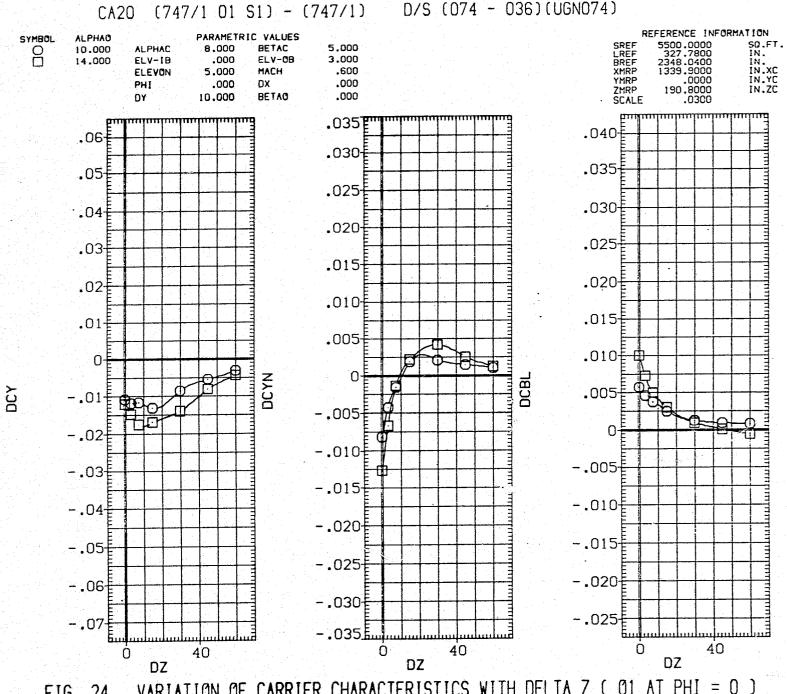
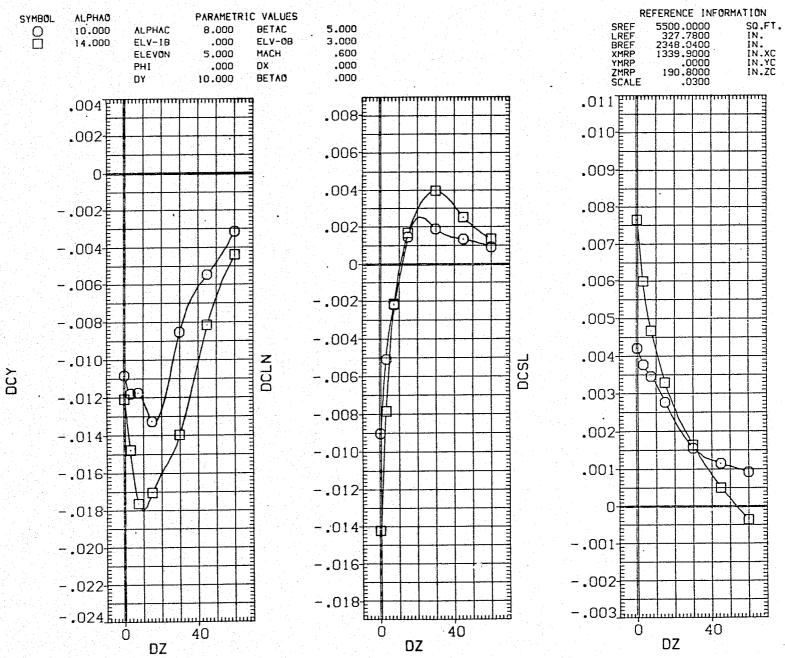
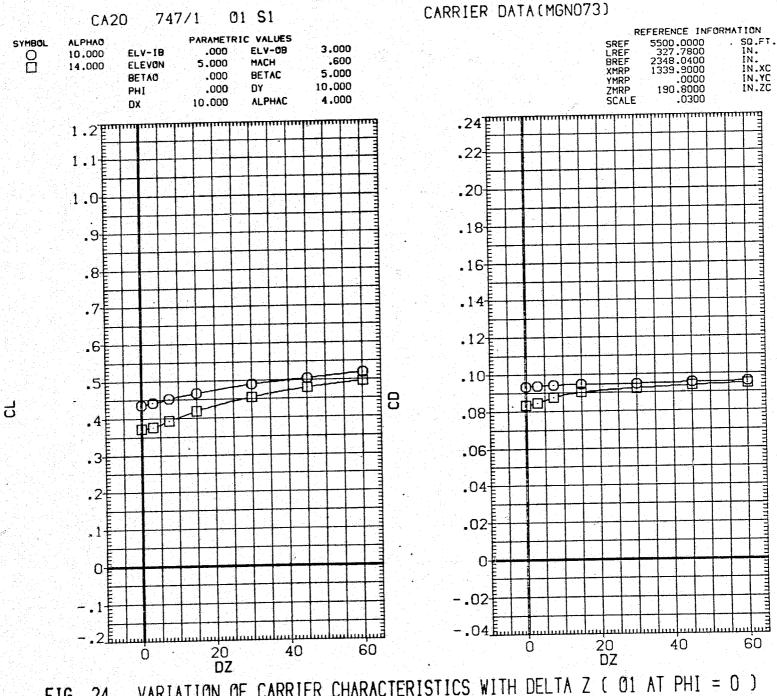


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0) FIG 24 607 PAGE



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = FIG 24 608 PAGE



FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 609

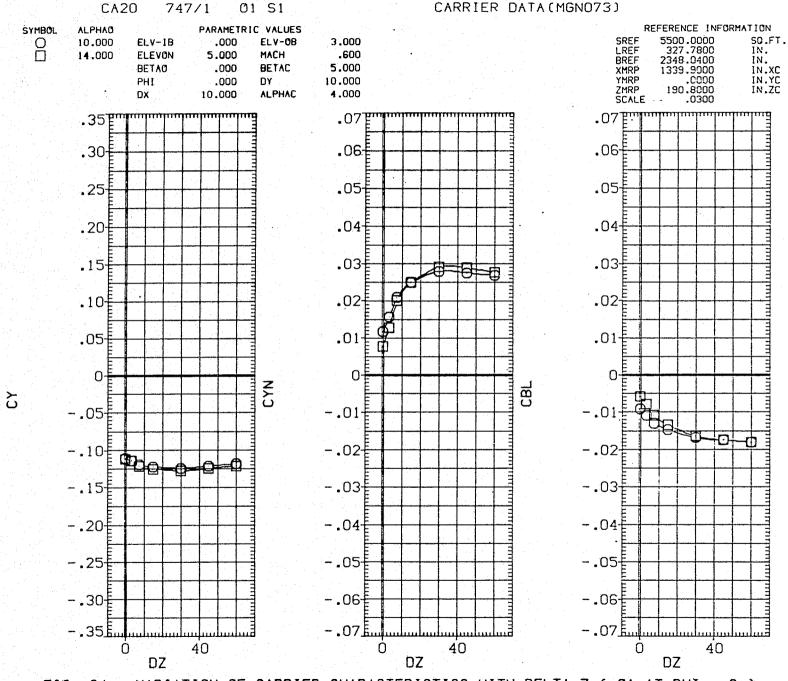


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 610

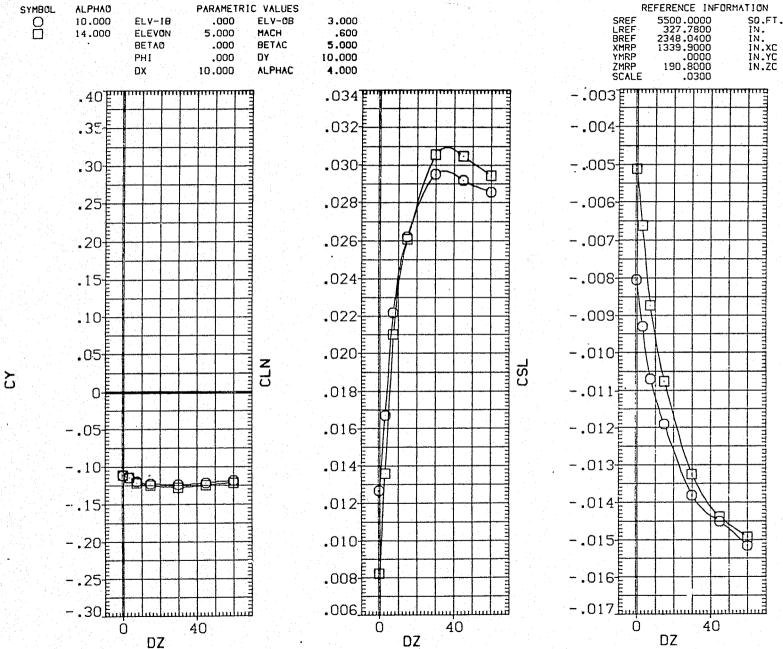


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

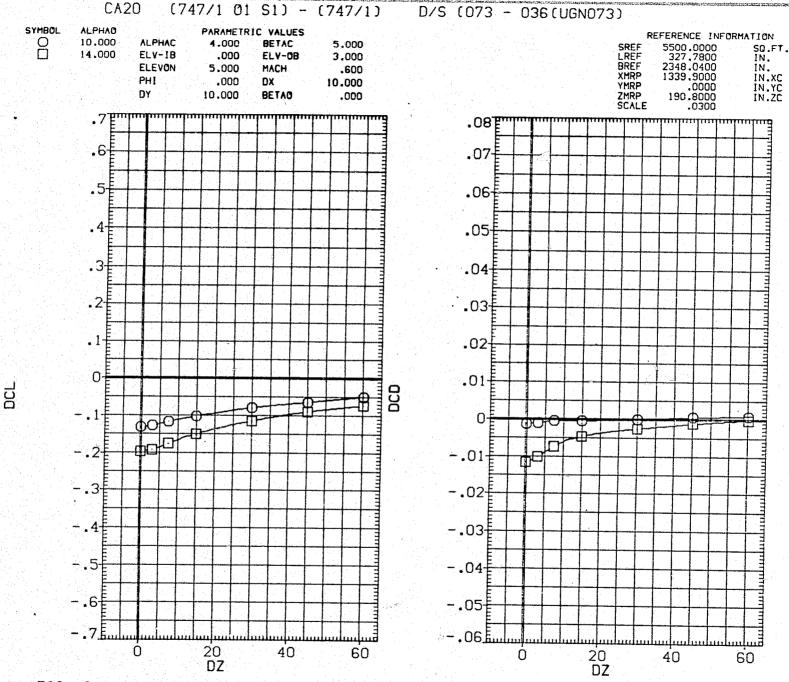


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 612

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FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 614

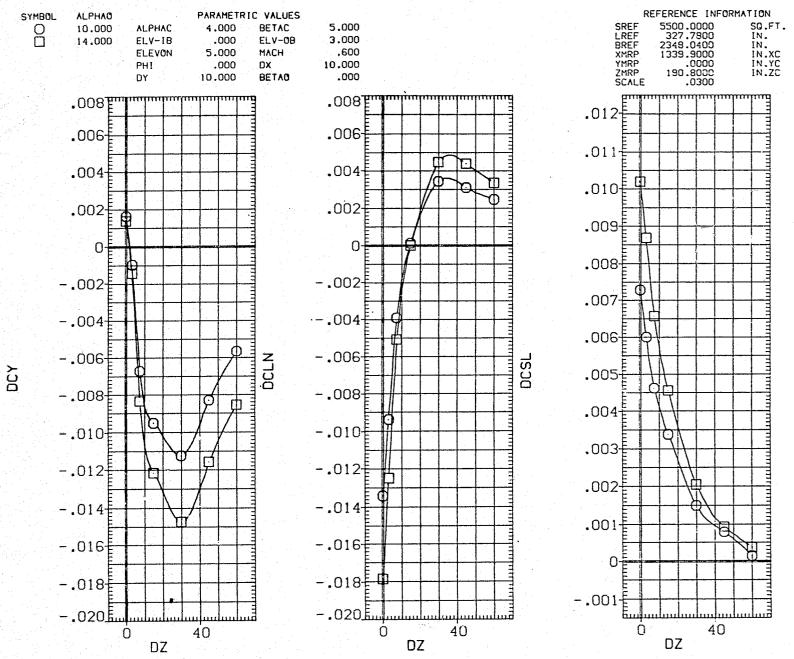


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 615

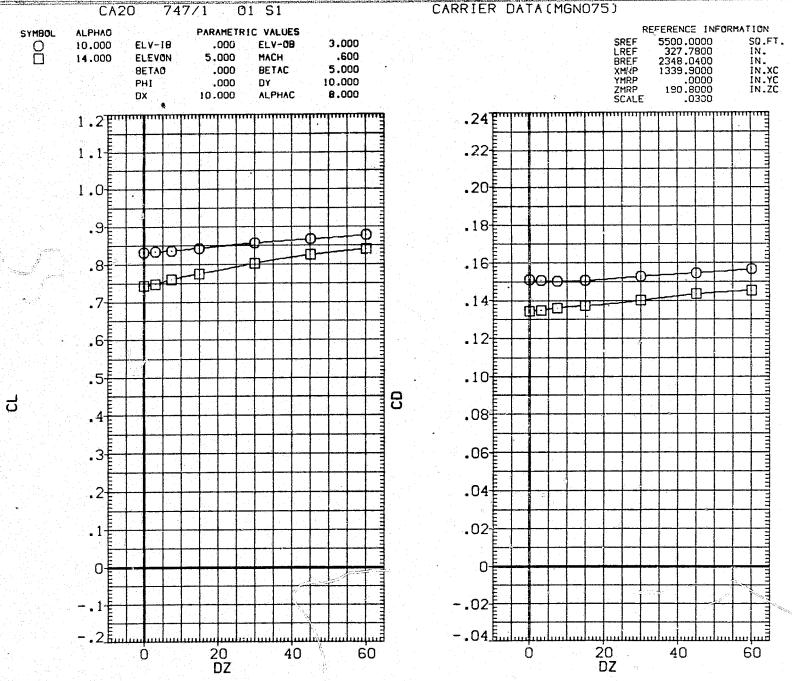
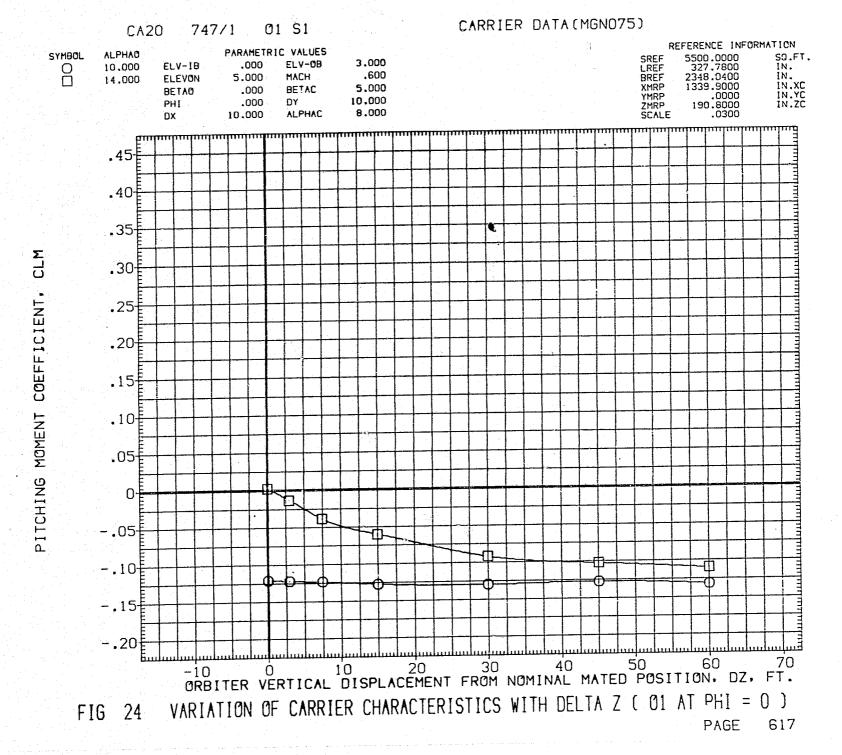


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 616



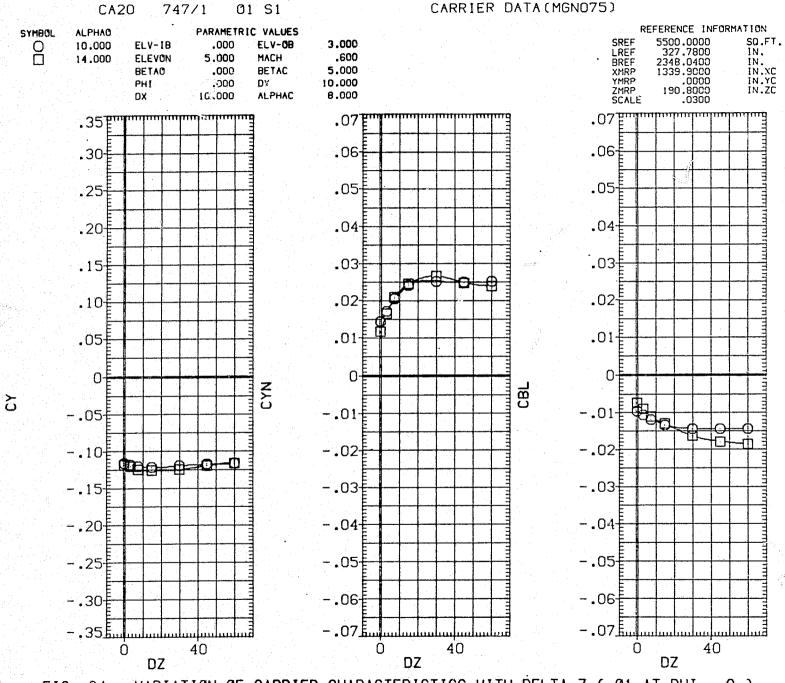


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 618

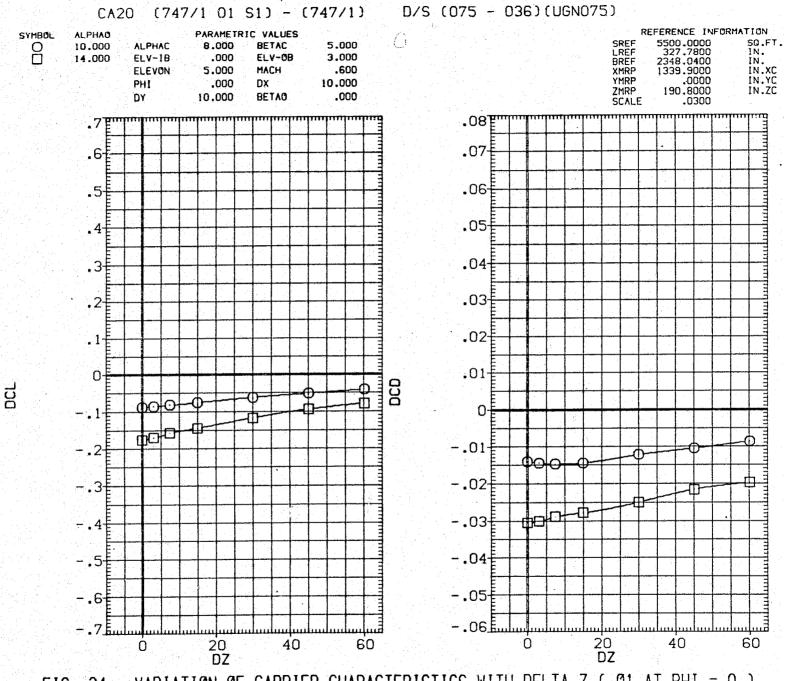


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 620

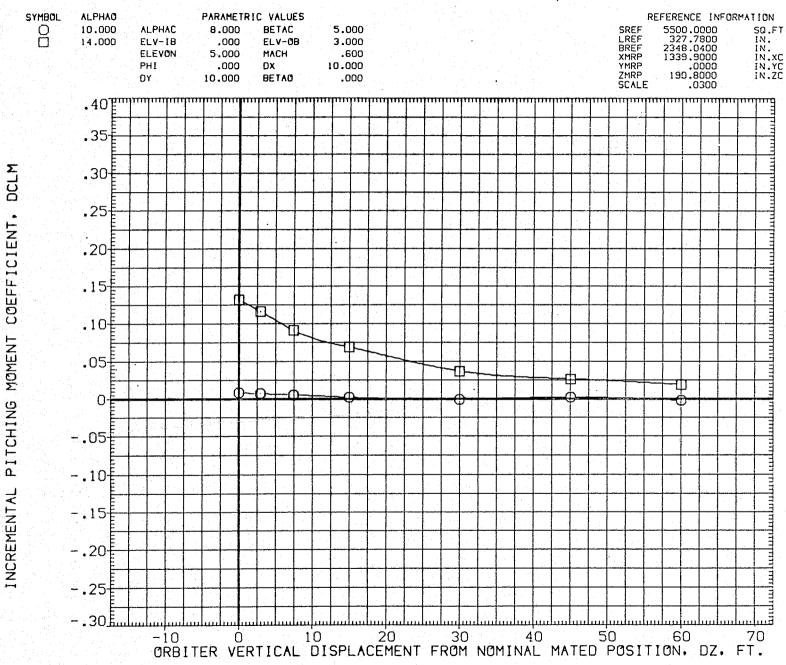


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

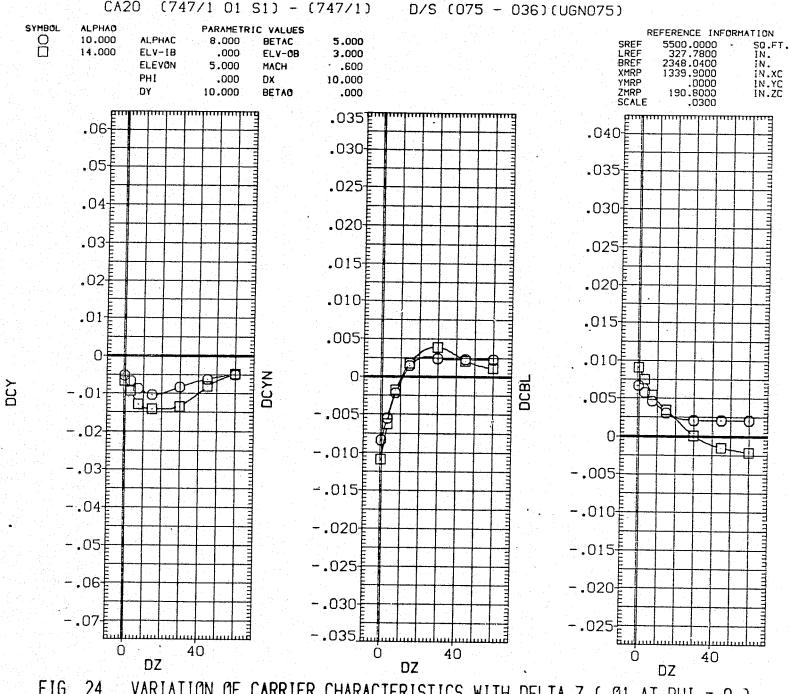


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)
PAGE 622

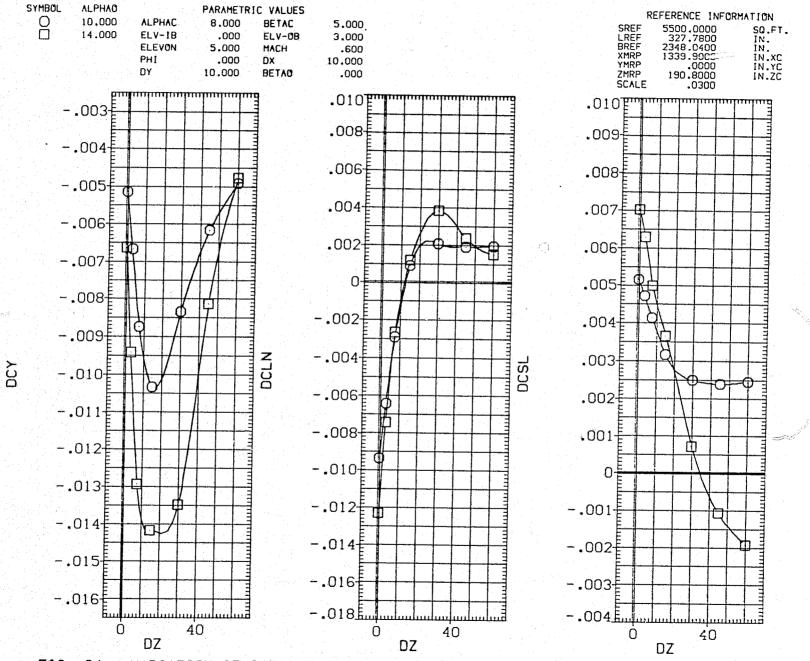


FIG 24 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 0)

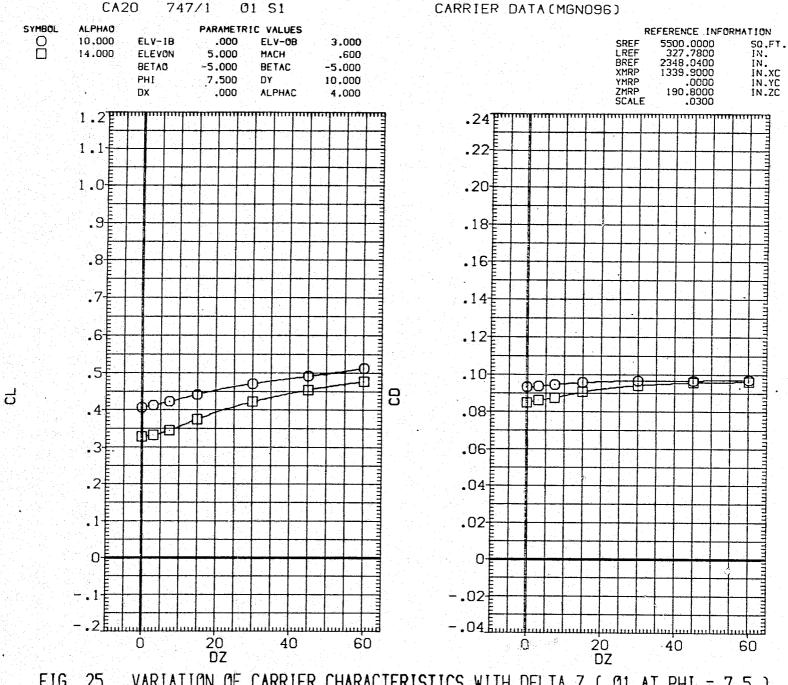


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 624

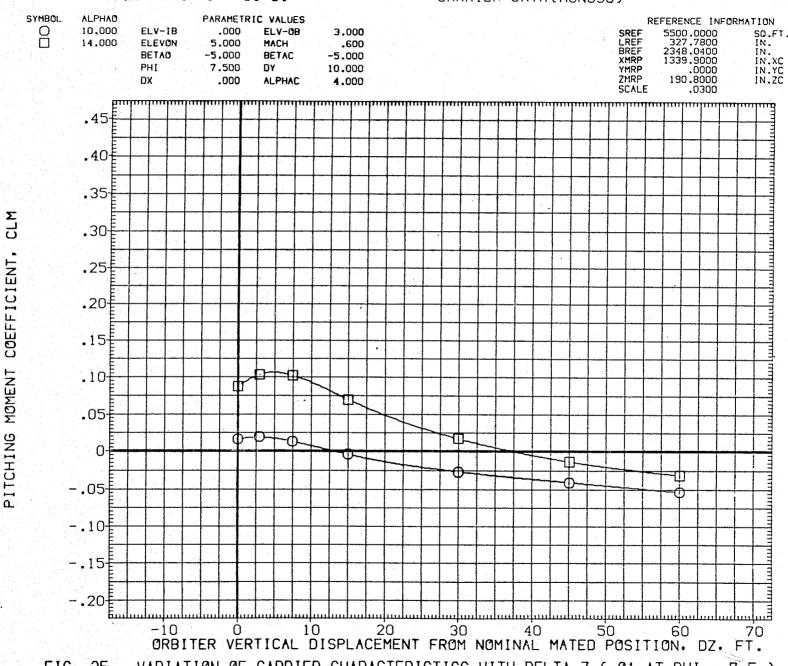


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

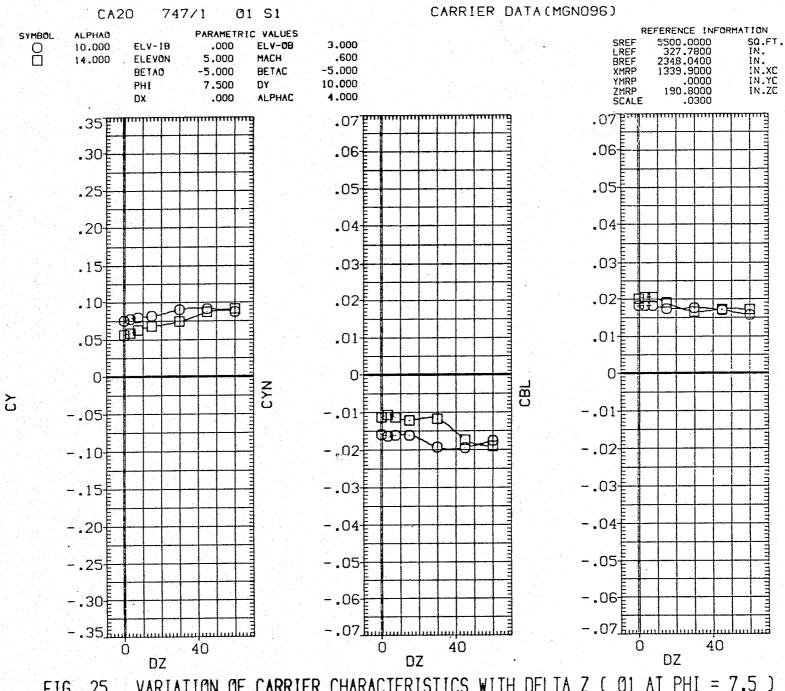


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 626

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 25 627 PAGE

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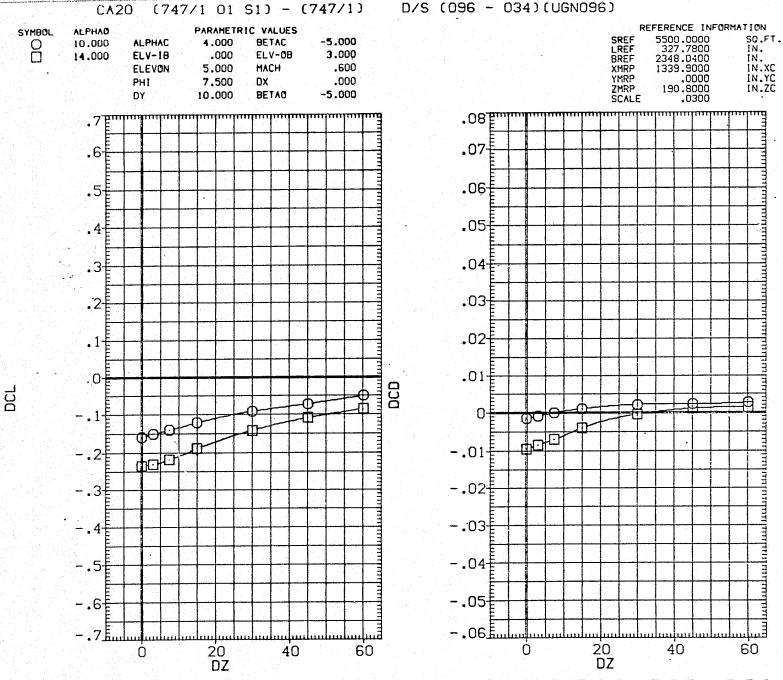


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 628

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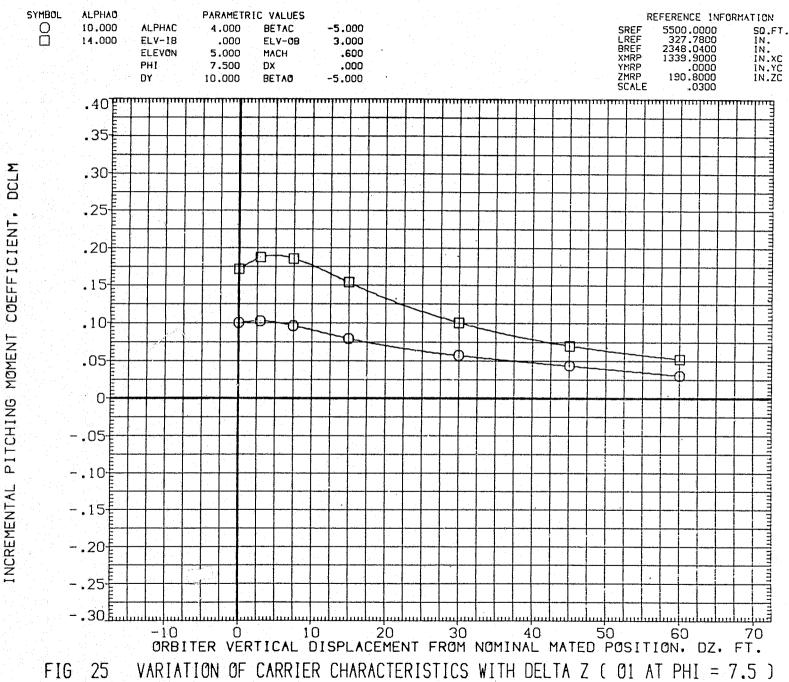


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 630

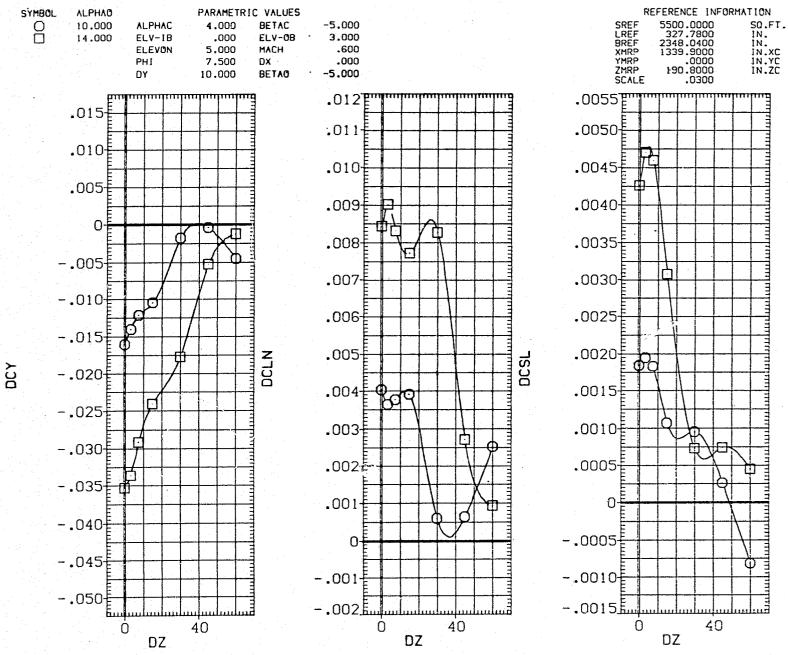


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 631

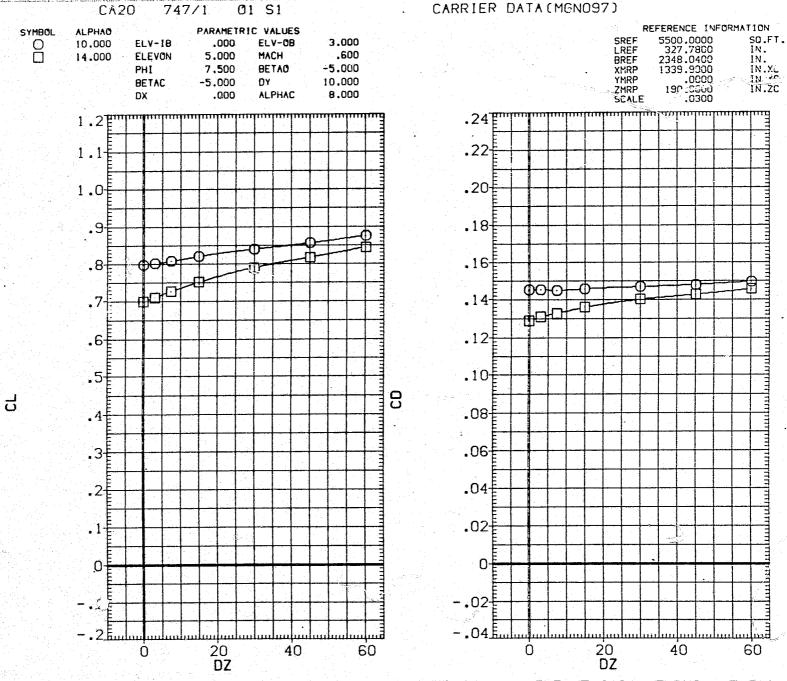


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 632

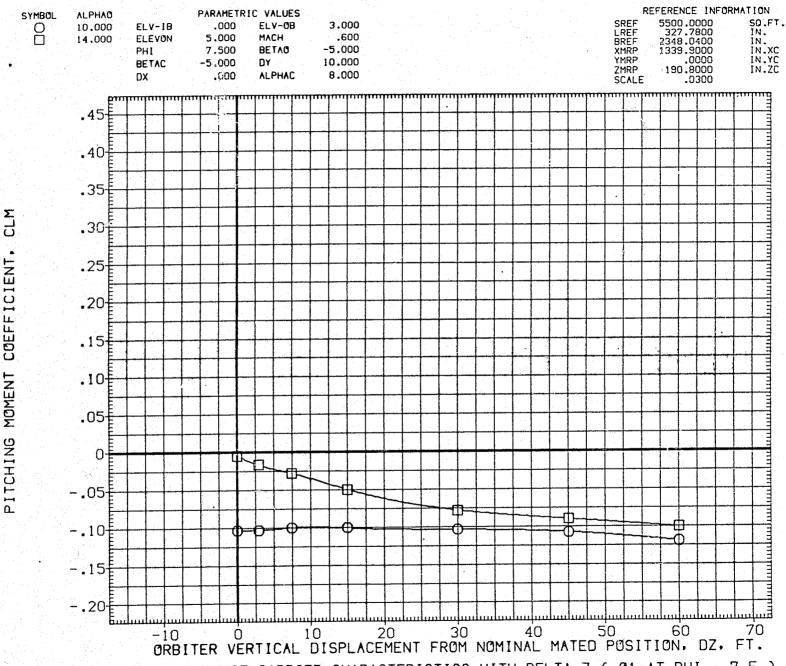


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

PAGE 633

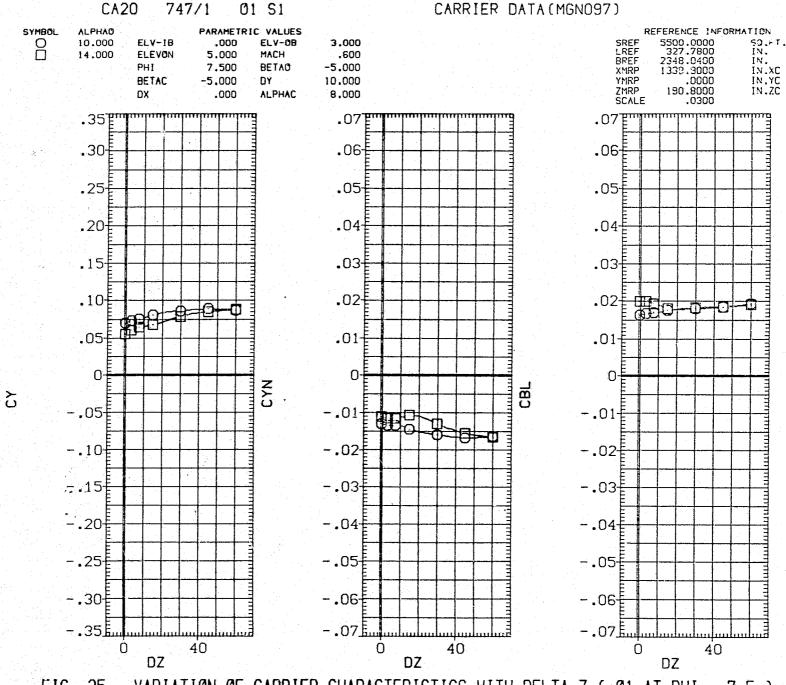
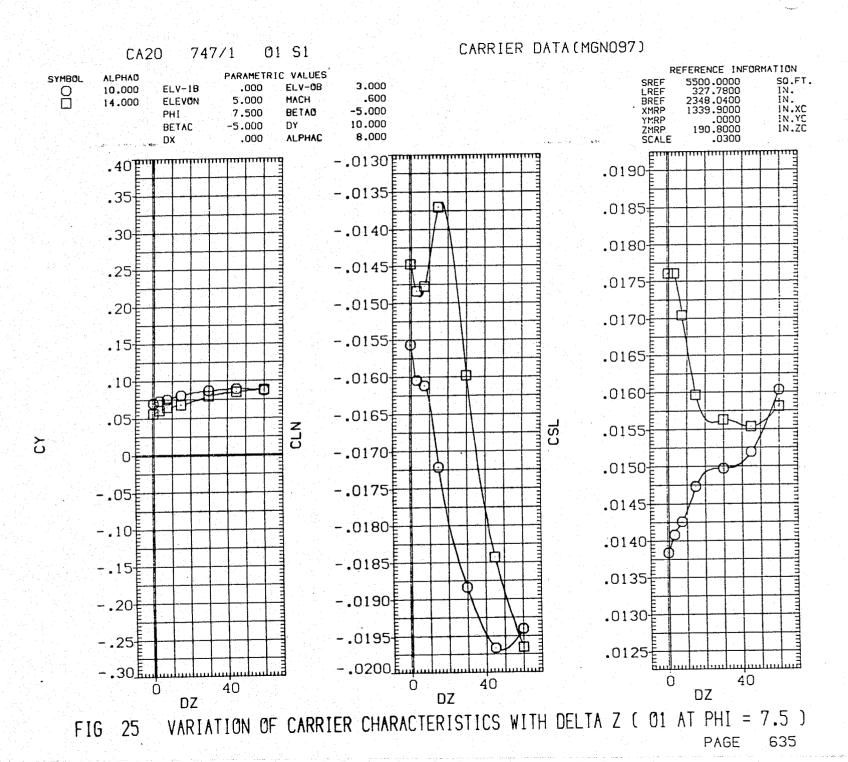
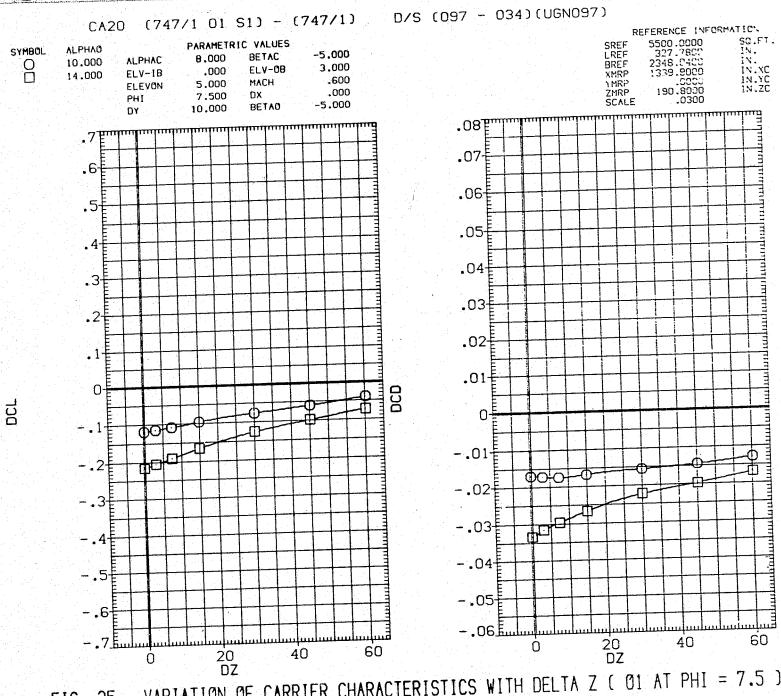


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 634

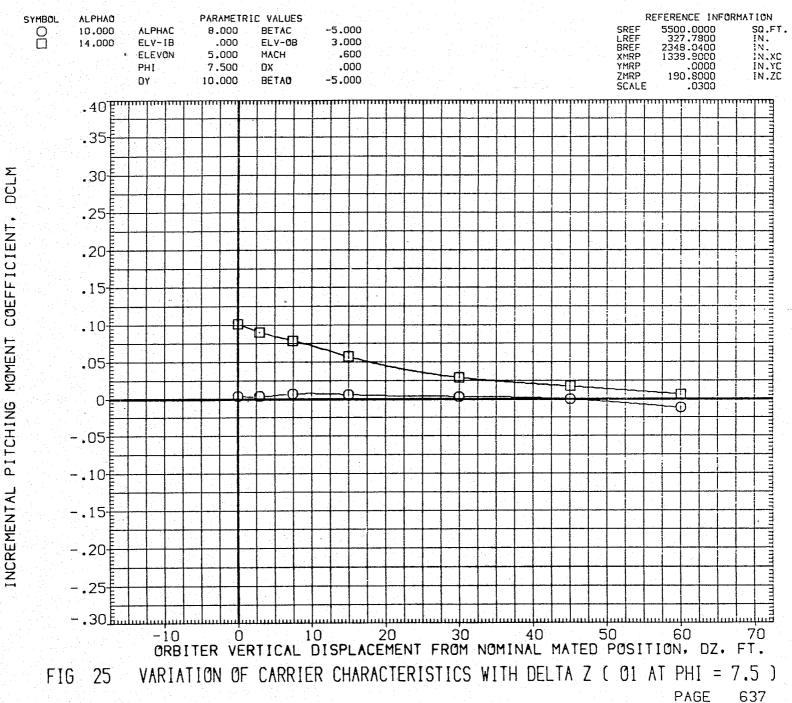


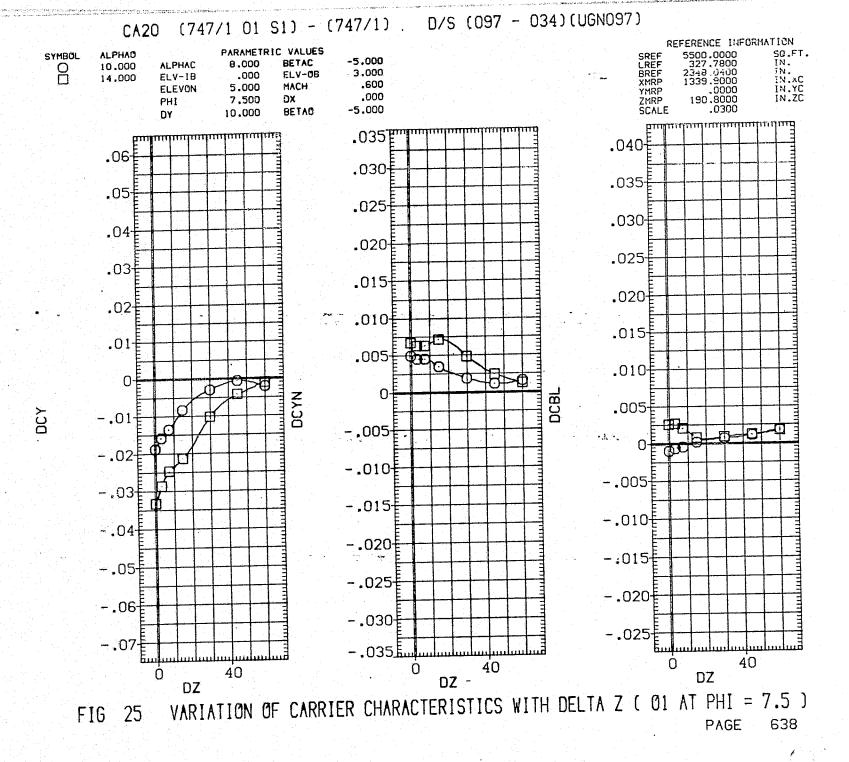


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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 25 FIG PAGE

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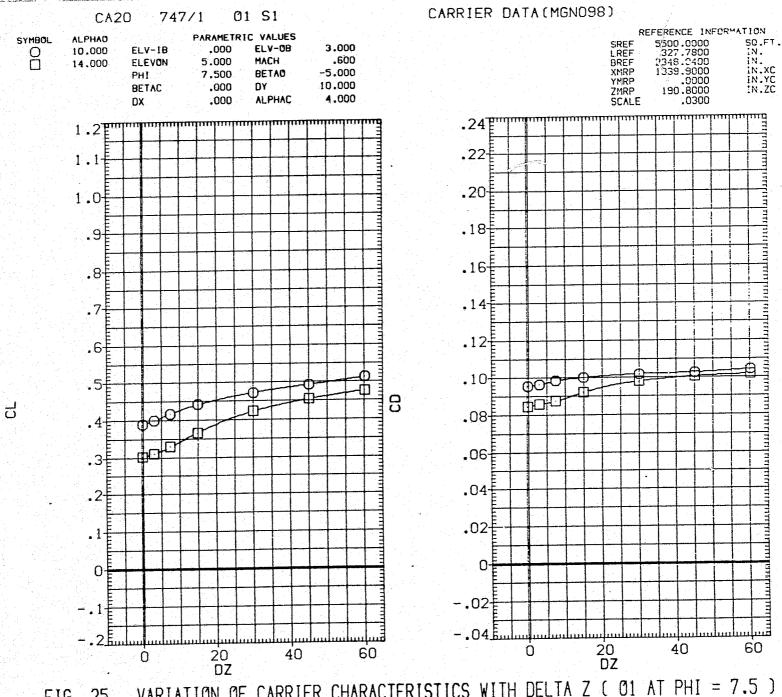
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 639 PAGE

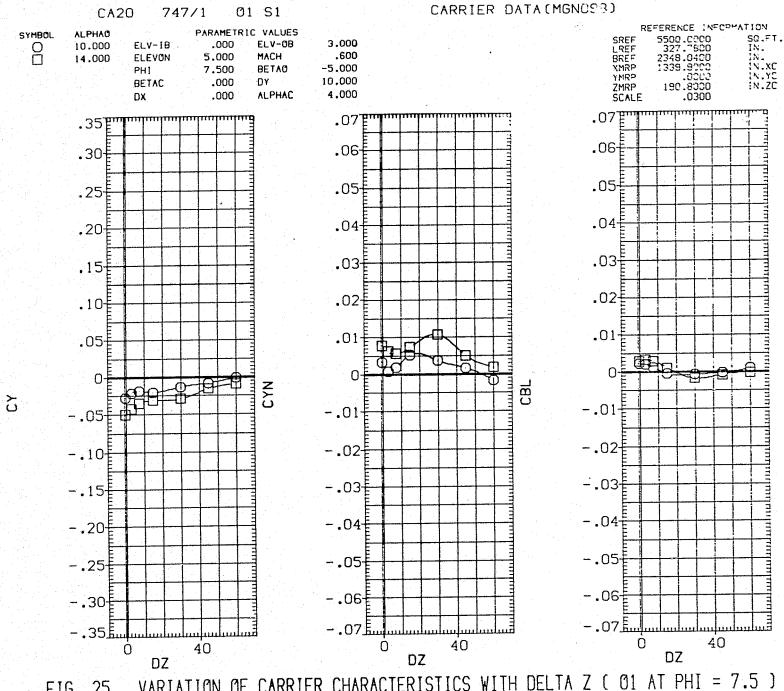
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 25 FIG PAGE 640

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) PAGE 641



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 25 FIG PAGE 642

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 643

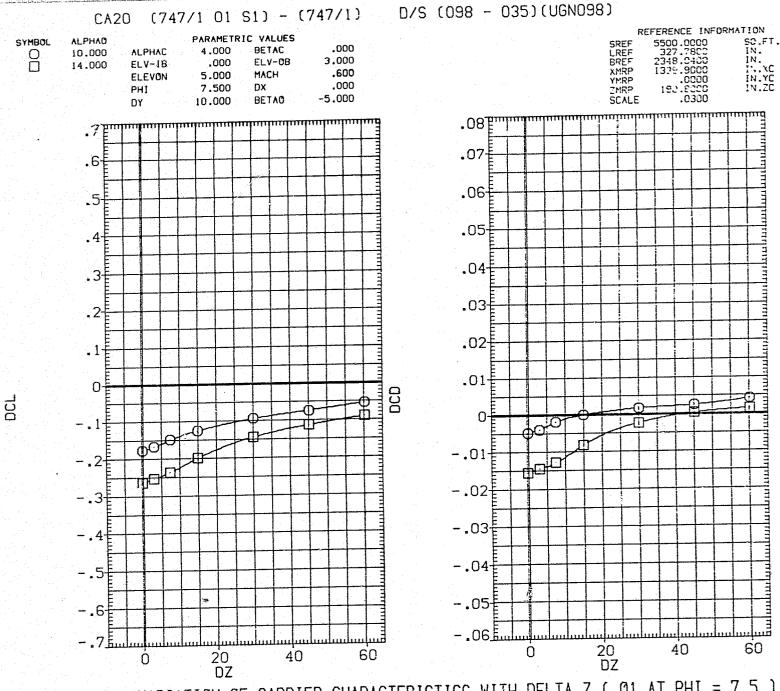


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 644

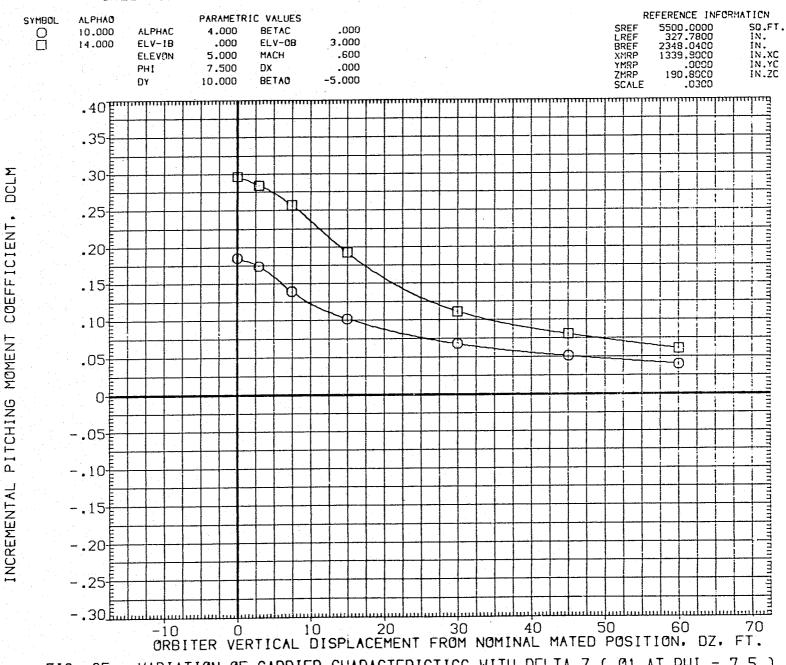


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

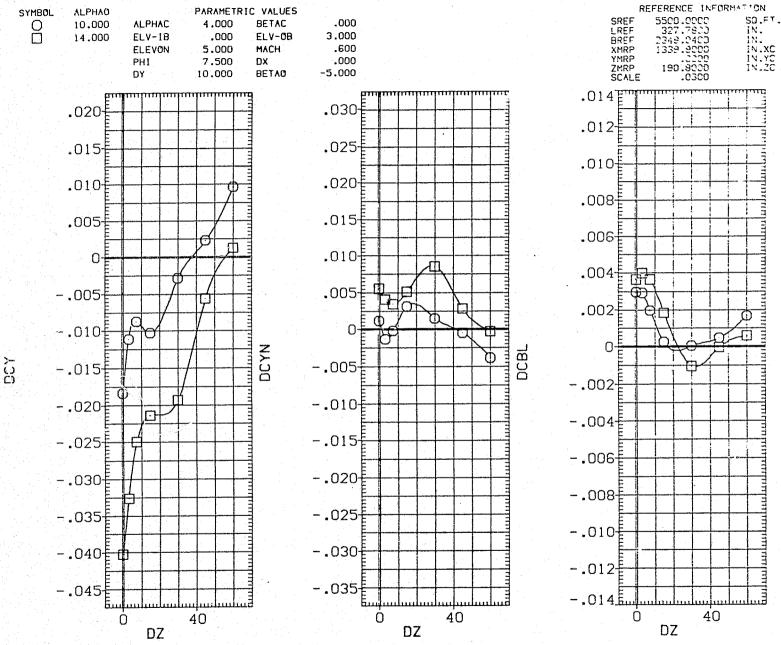


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

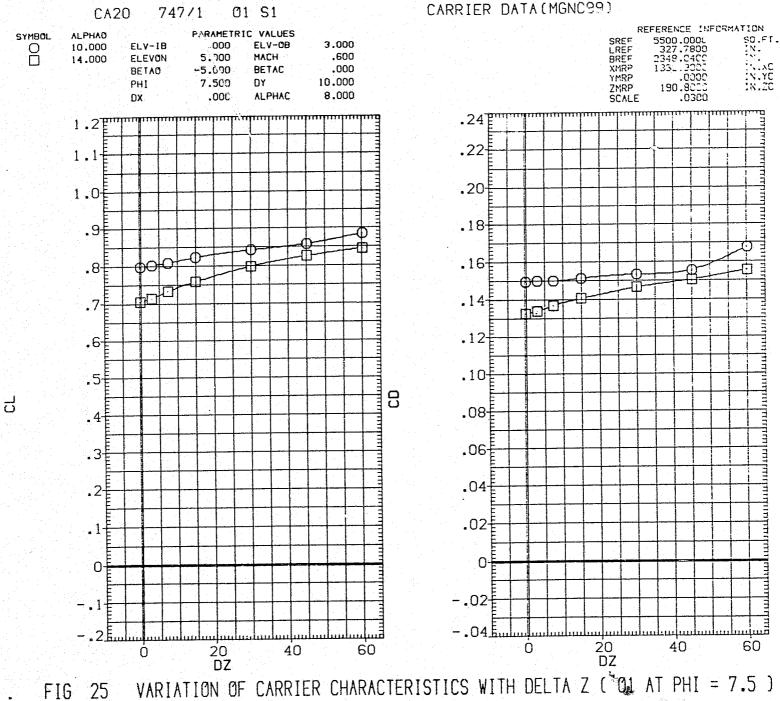


FIG PAGE 648

PAGE 649

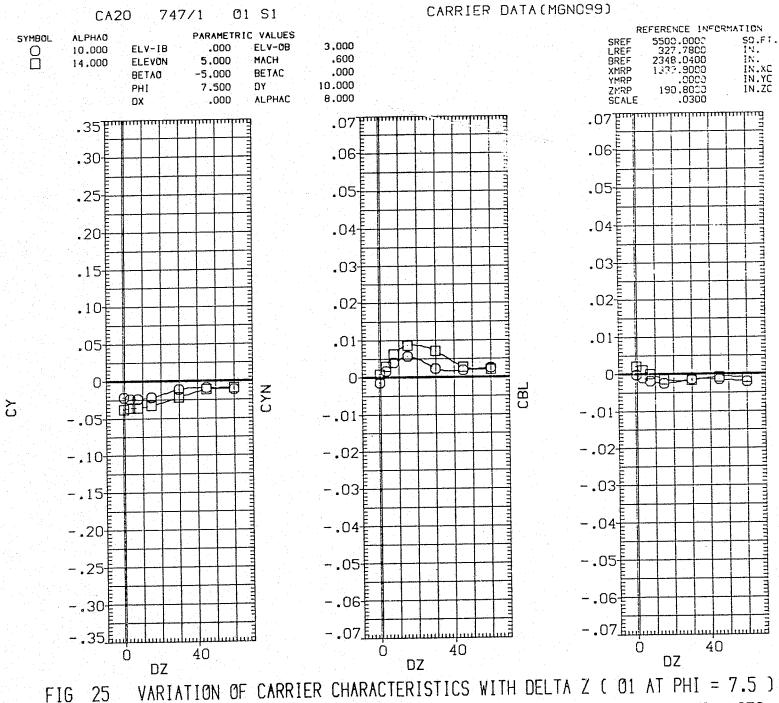


FIG PAGE 650

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 651

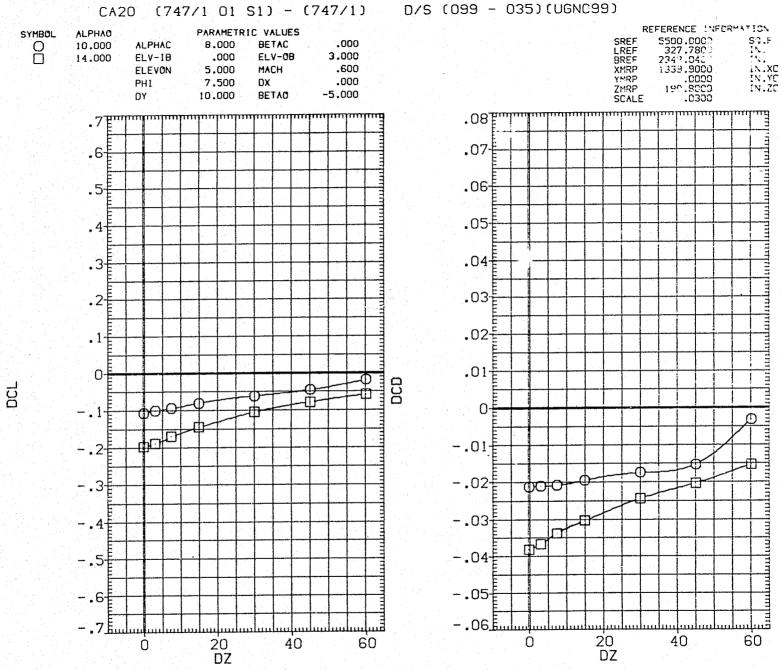
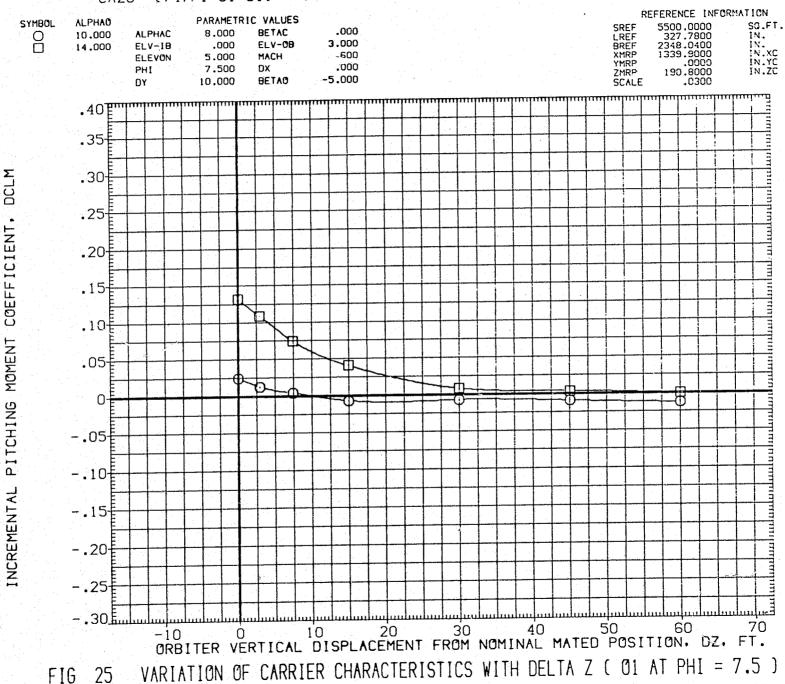


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 652



PAGE

653

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 654

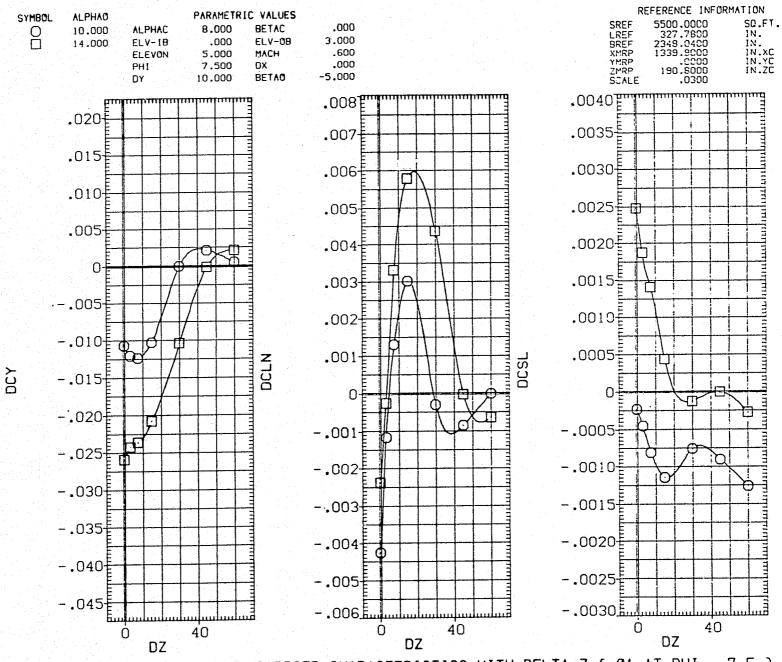


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 655

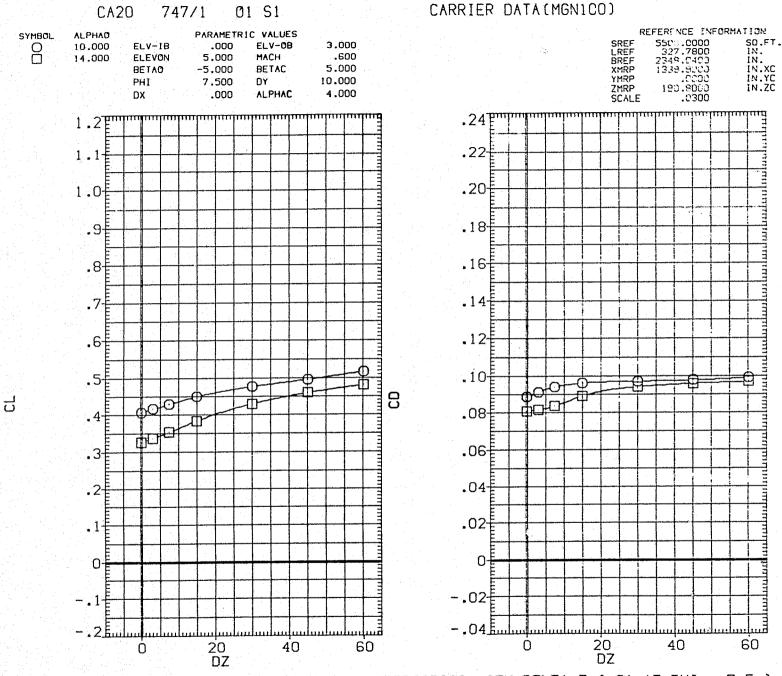
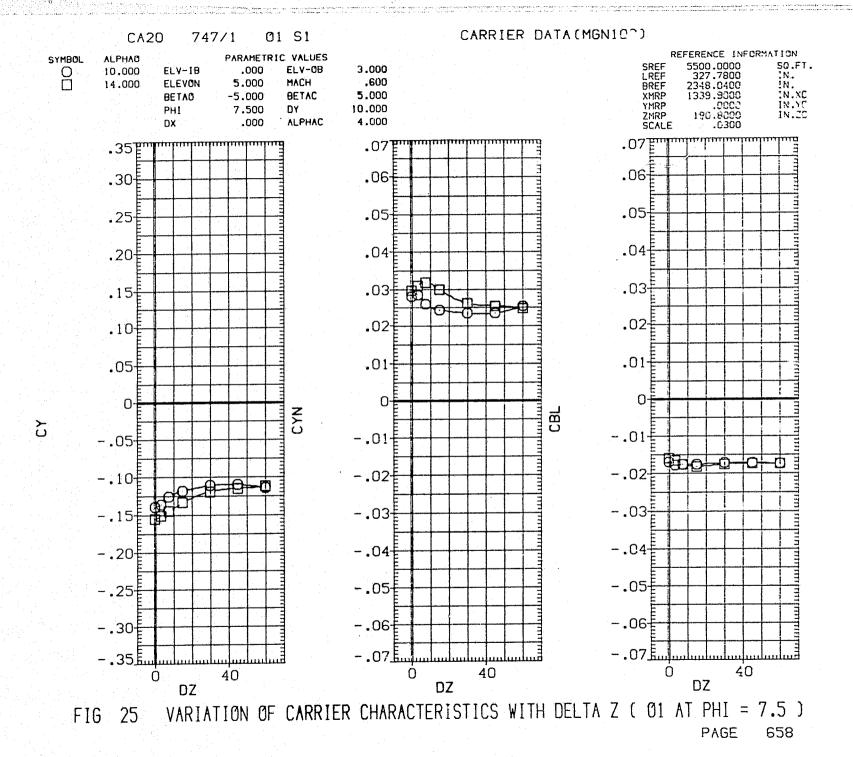
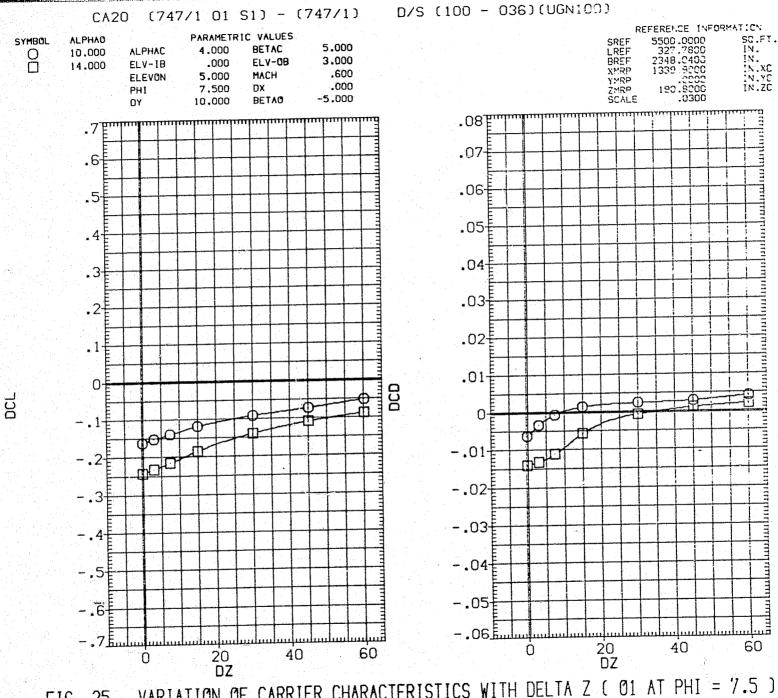


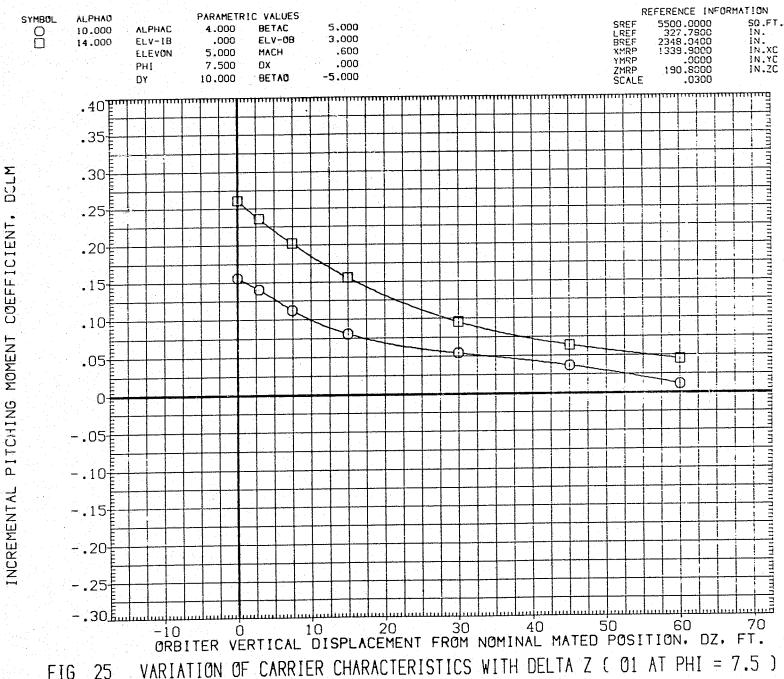
FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 657



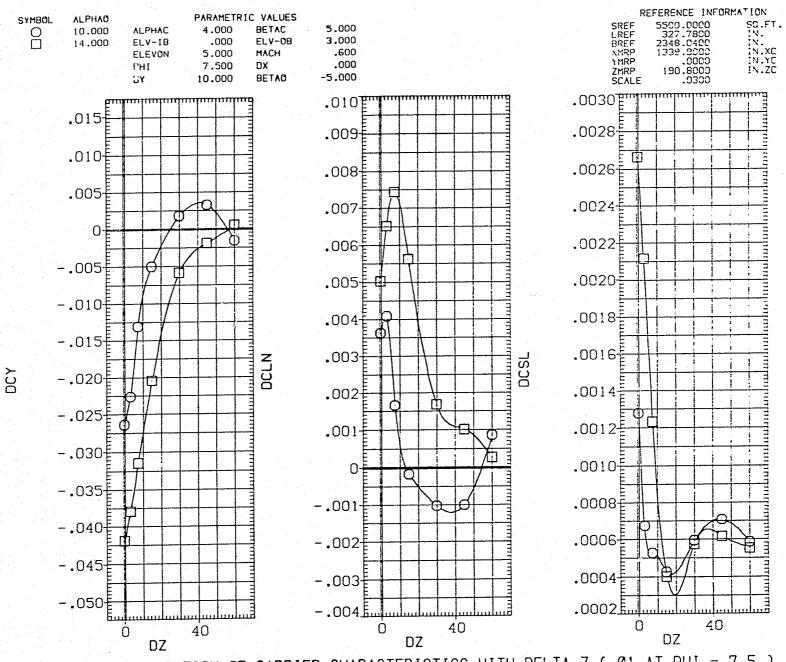


VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 25 660 PAGE



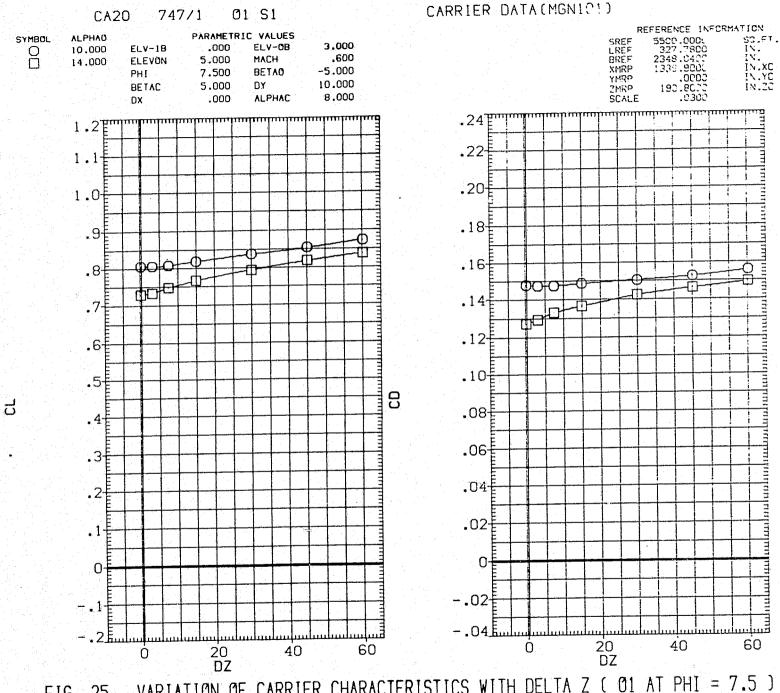
VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 661 PAGE

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)



5 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

FIG



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 664 PAGE

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE G65

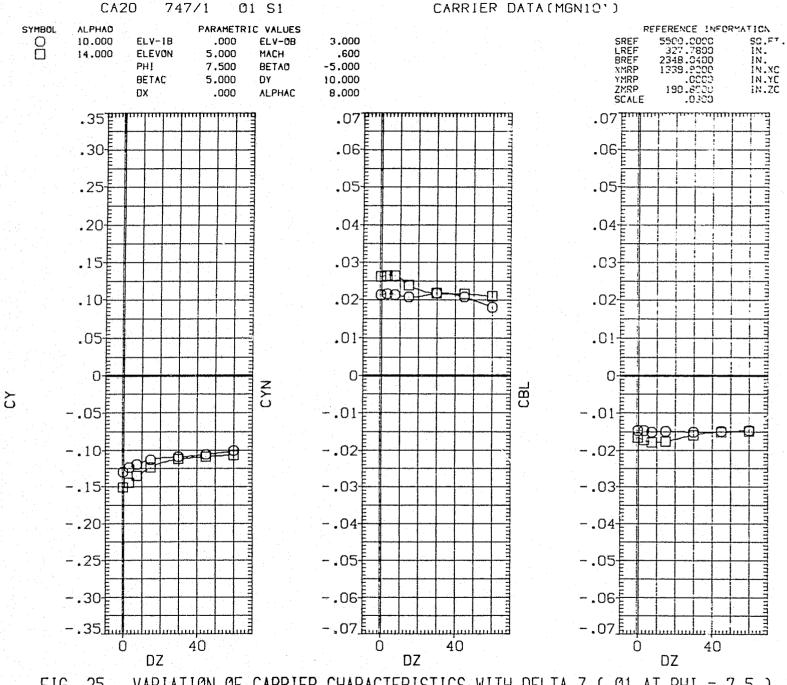


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 666

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

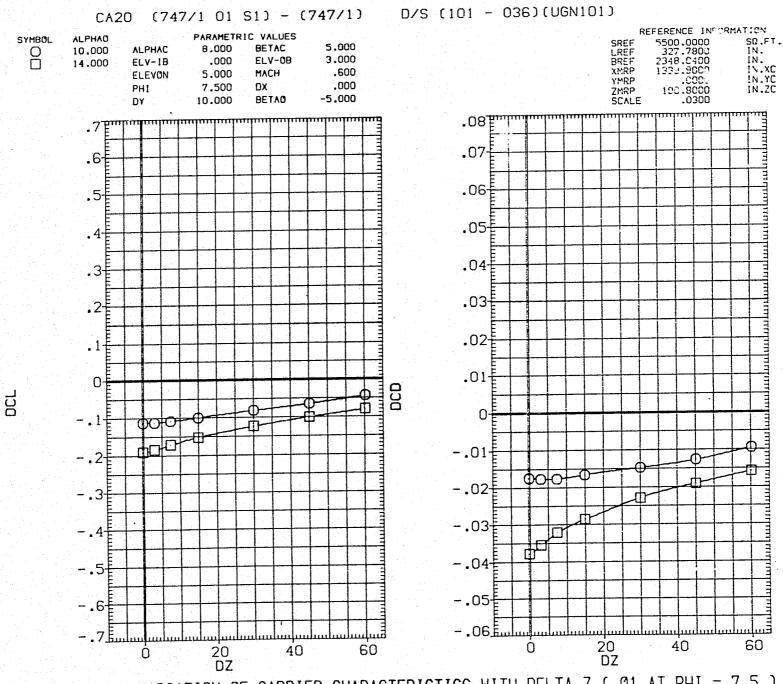


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 668

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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 669

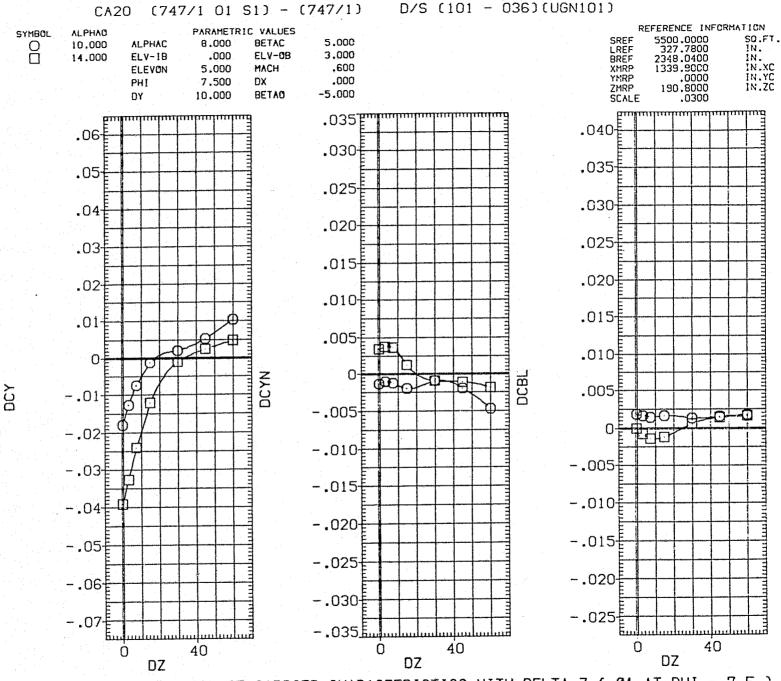


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

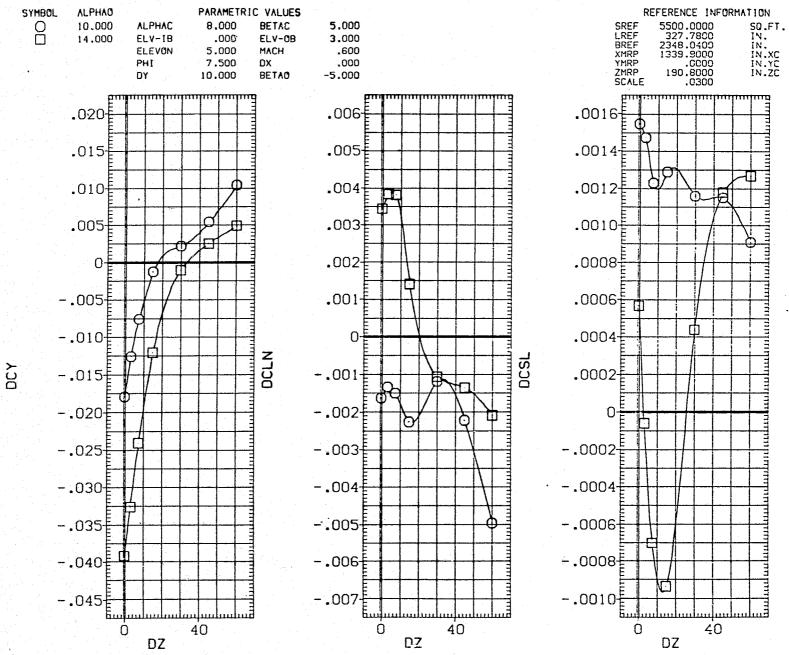


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 671

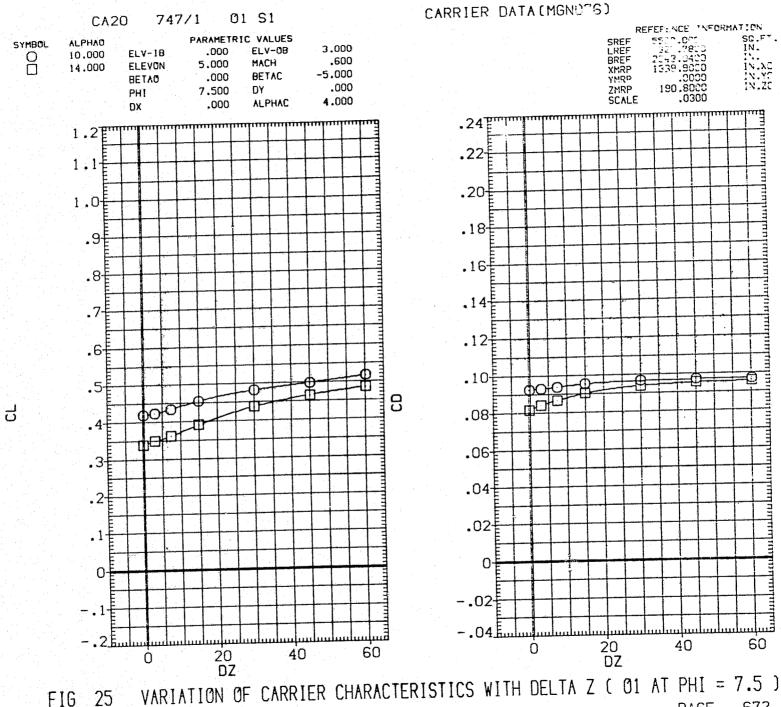
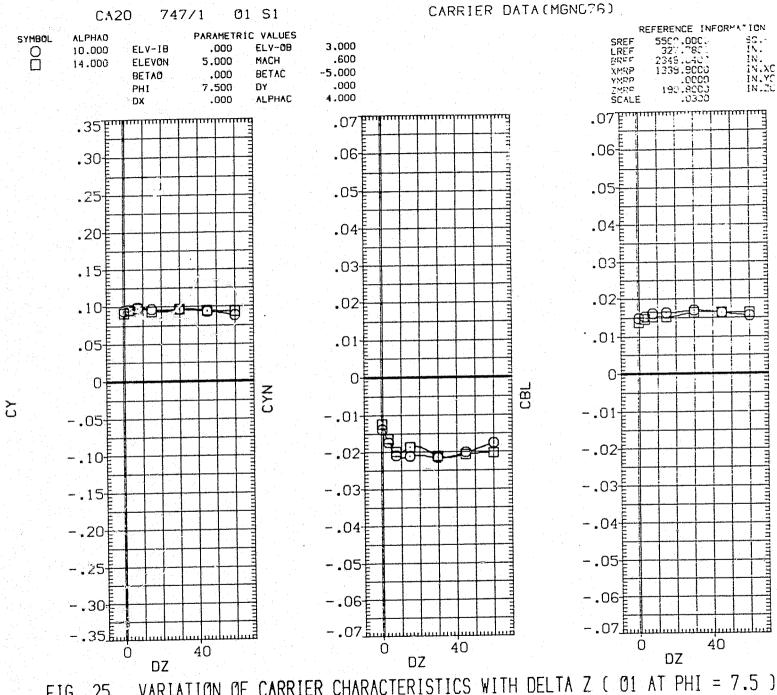


FIG 672 PAGE

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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG PAGE 674

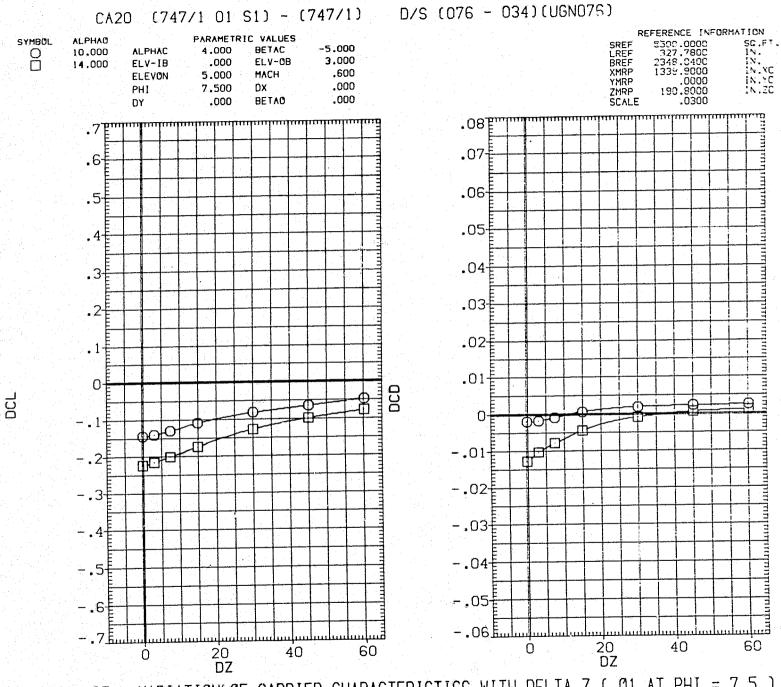


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

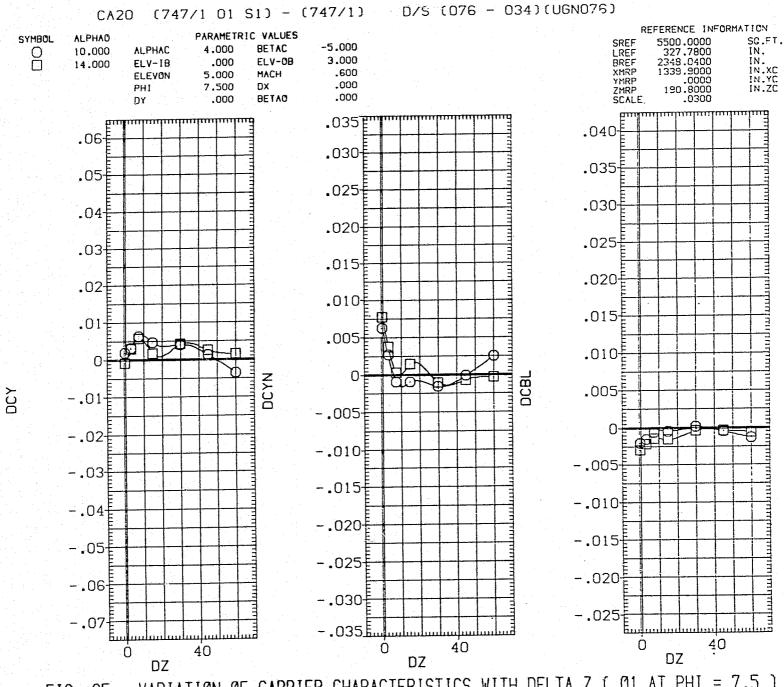


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 678

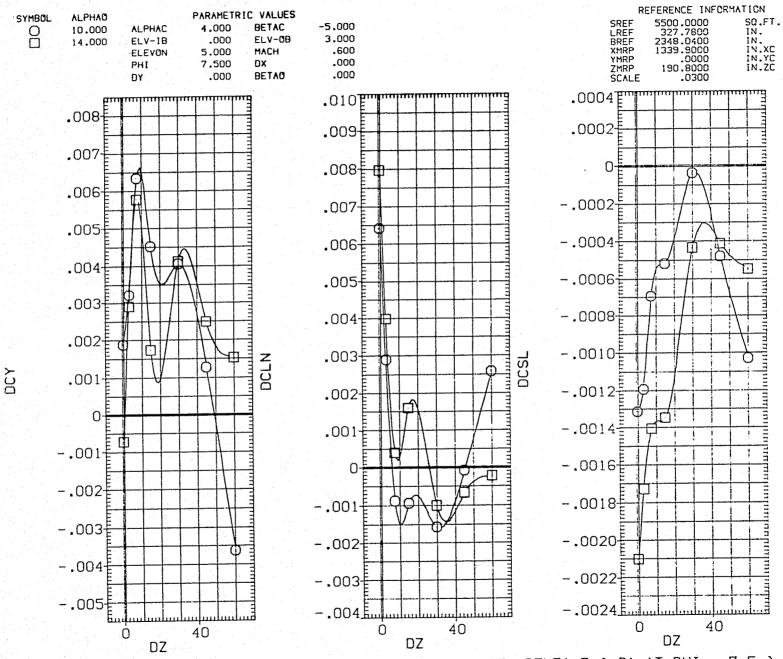


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

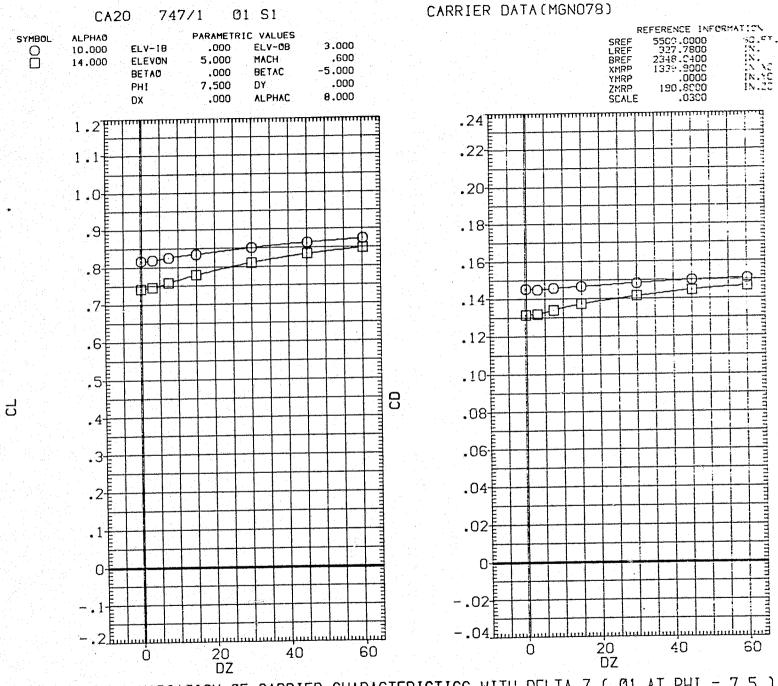


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

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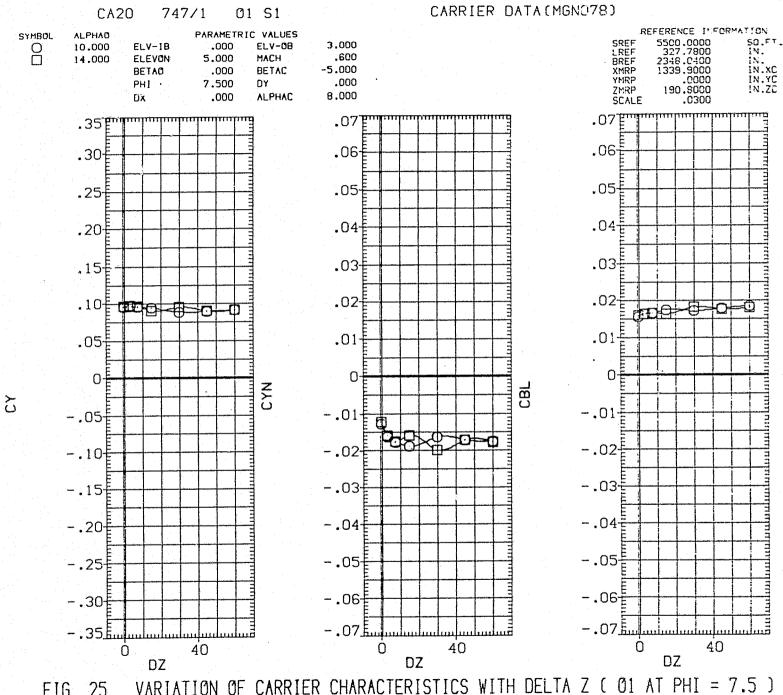
MOMENT

ITCHING

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG

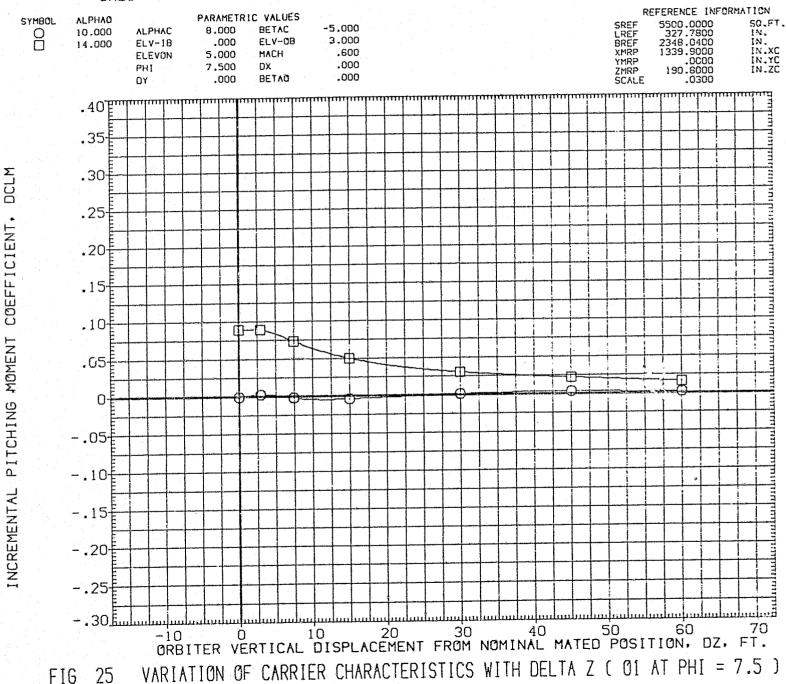
ORBITER VERTICAL DISPLACEMENT FROM NOMINAL MATED POSITION, DZ, FT.

681



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 25 FIG 682 PAGE

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 684

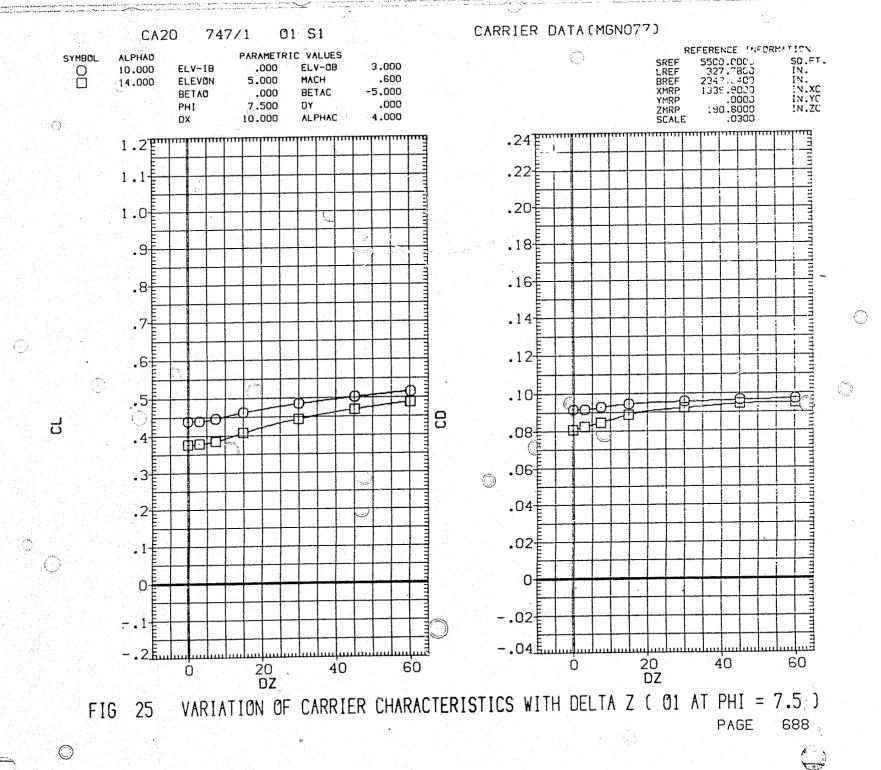


685

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 686

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG PAGE 687

DZ



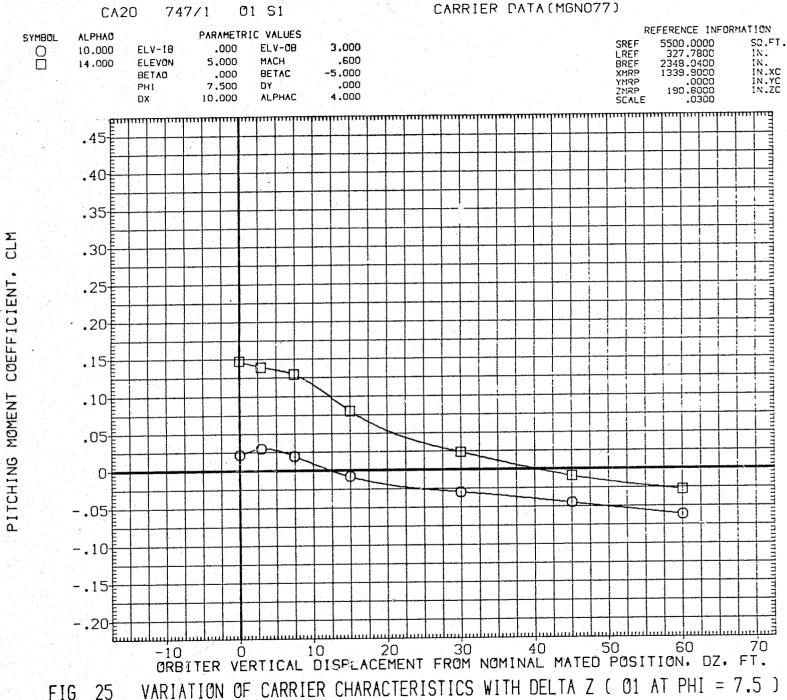


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 689 PAGE

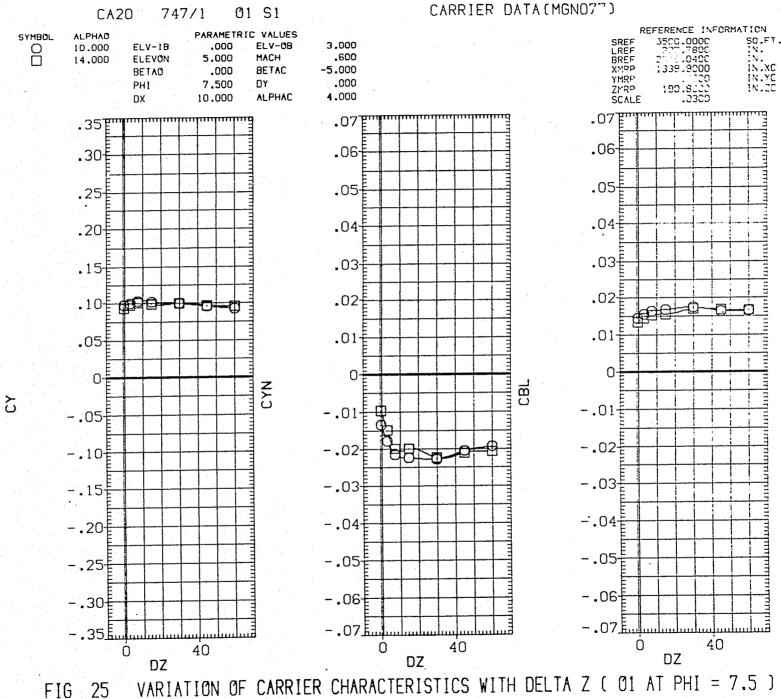


FIG PAGE 690

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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 692

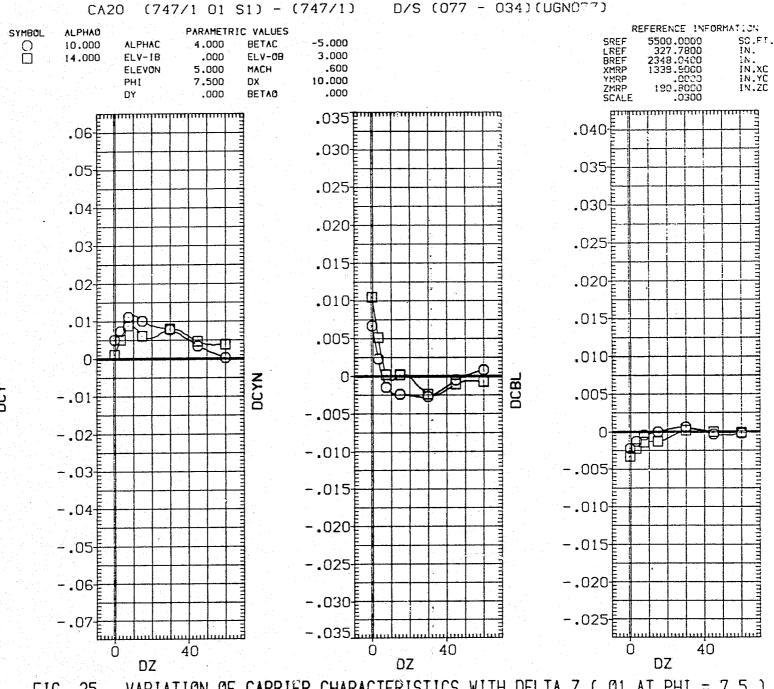


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 694

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

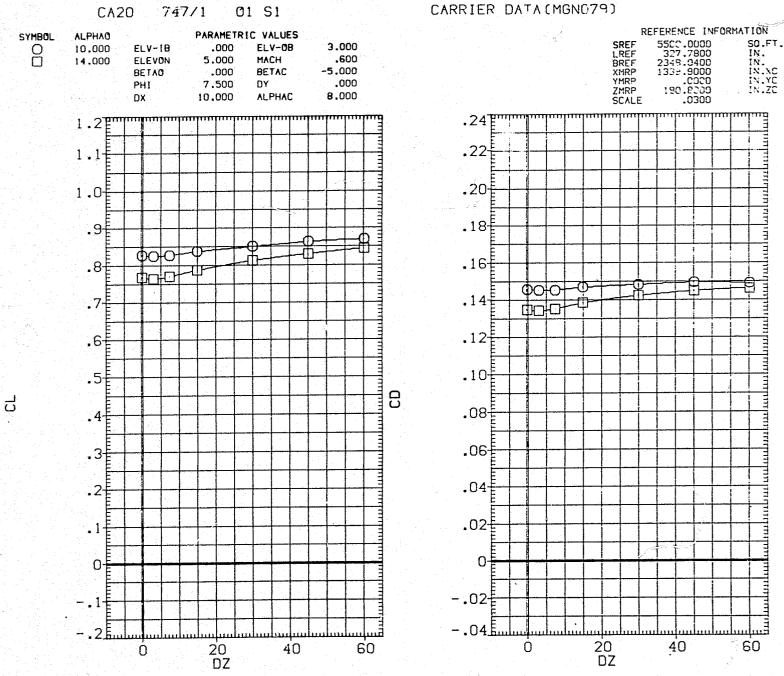


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

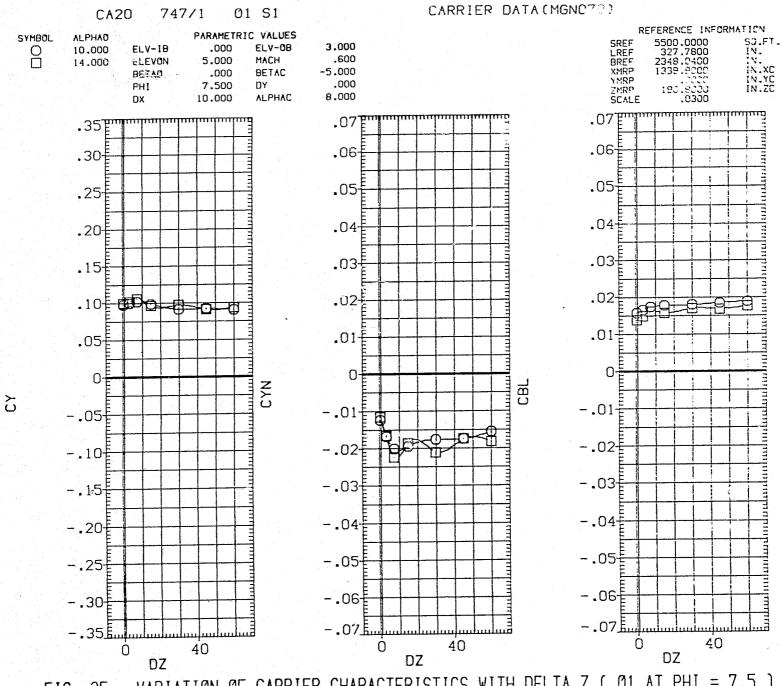


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

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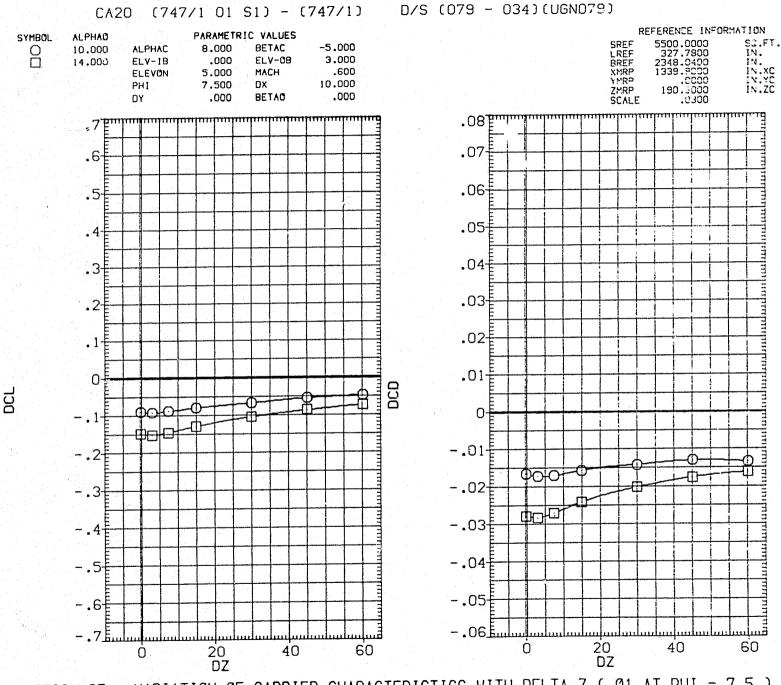
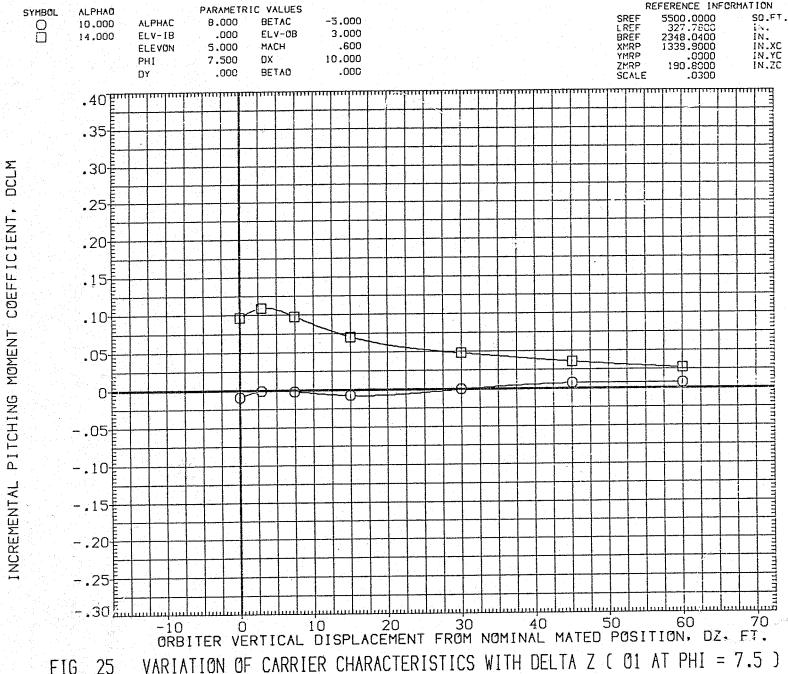


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 25 PAGE 701

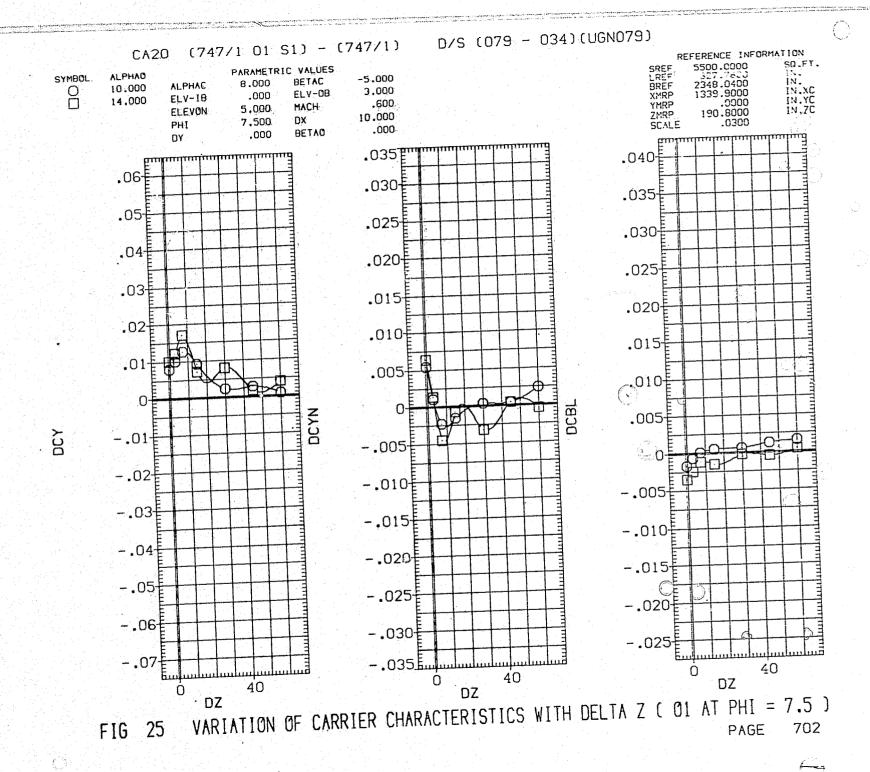


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

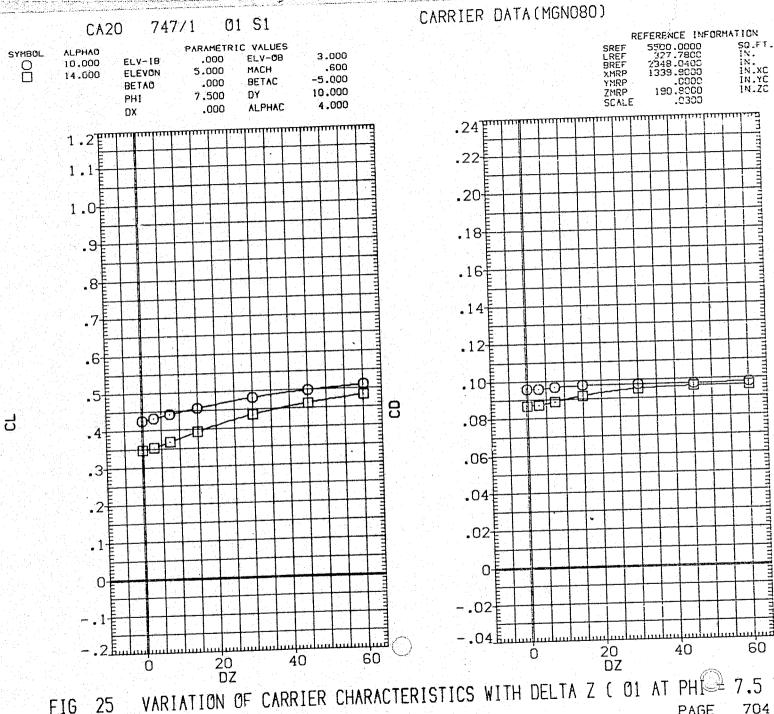


FIG PAGE

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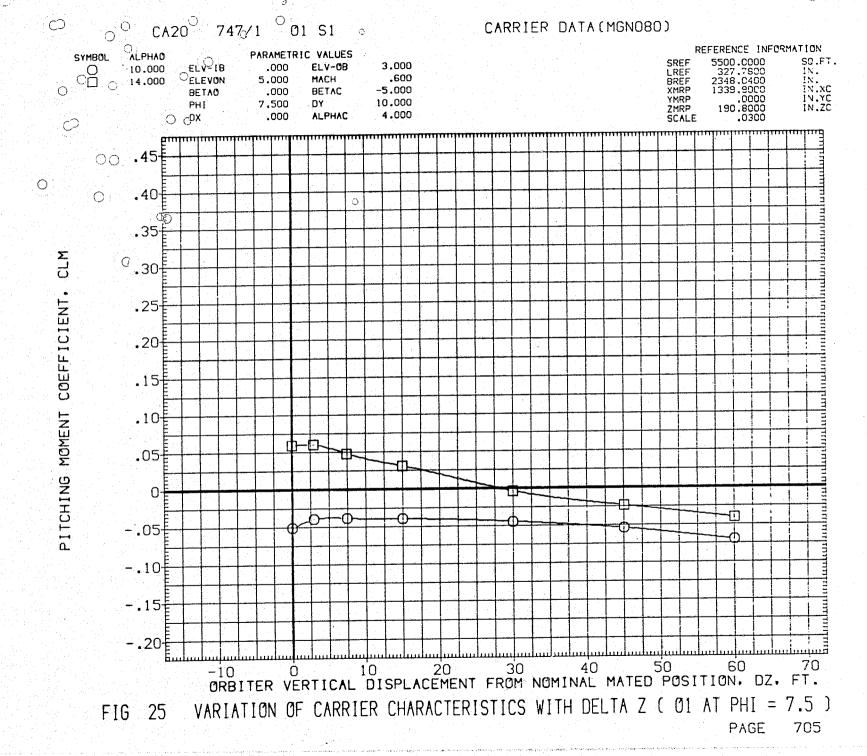


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 706

PAGE

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 708

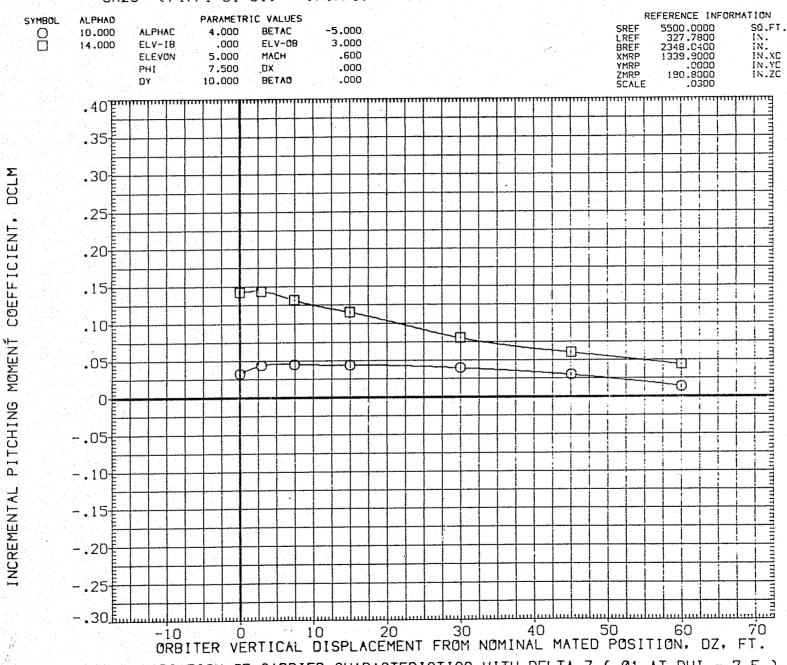


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

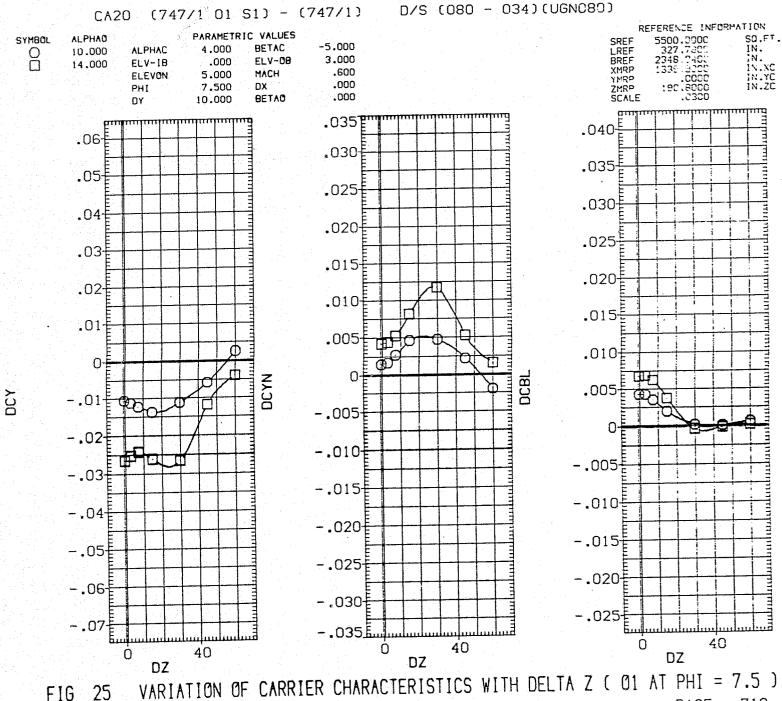


FIG PAGE 710

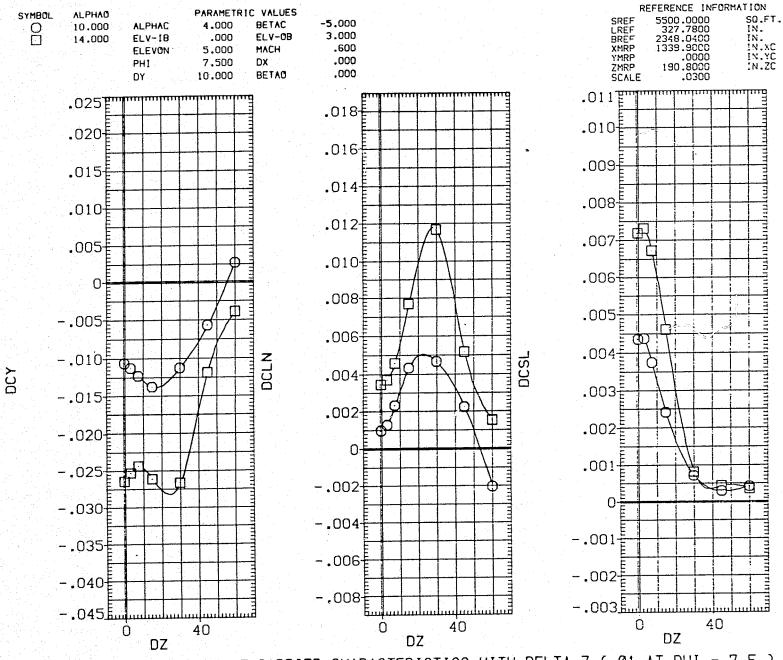


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 711

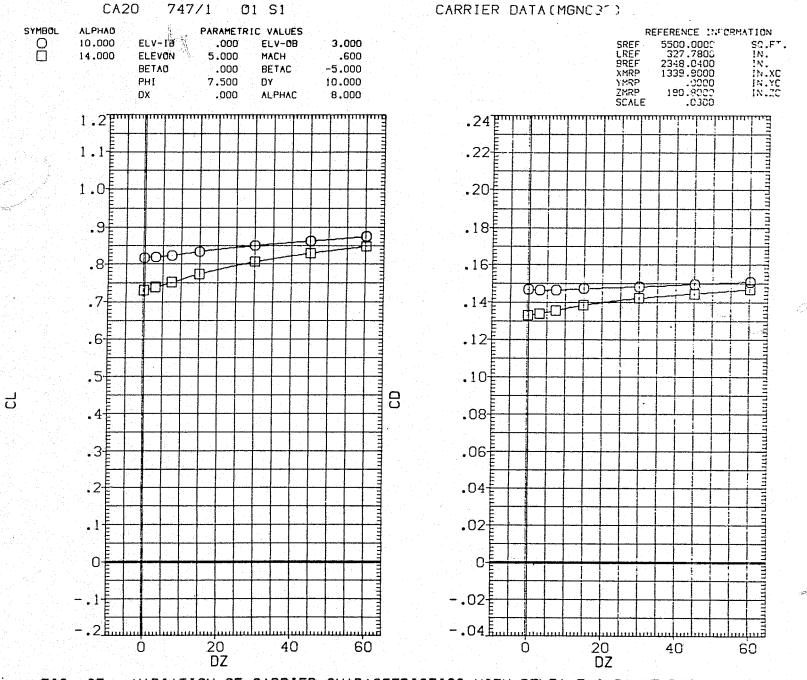
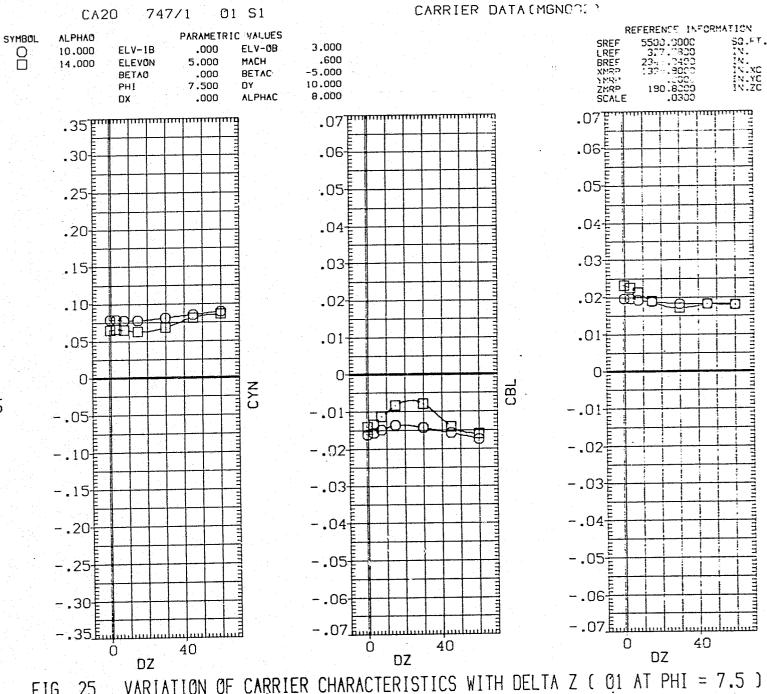


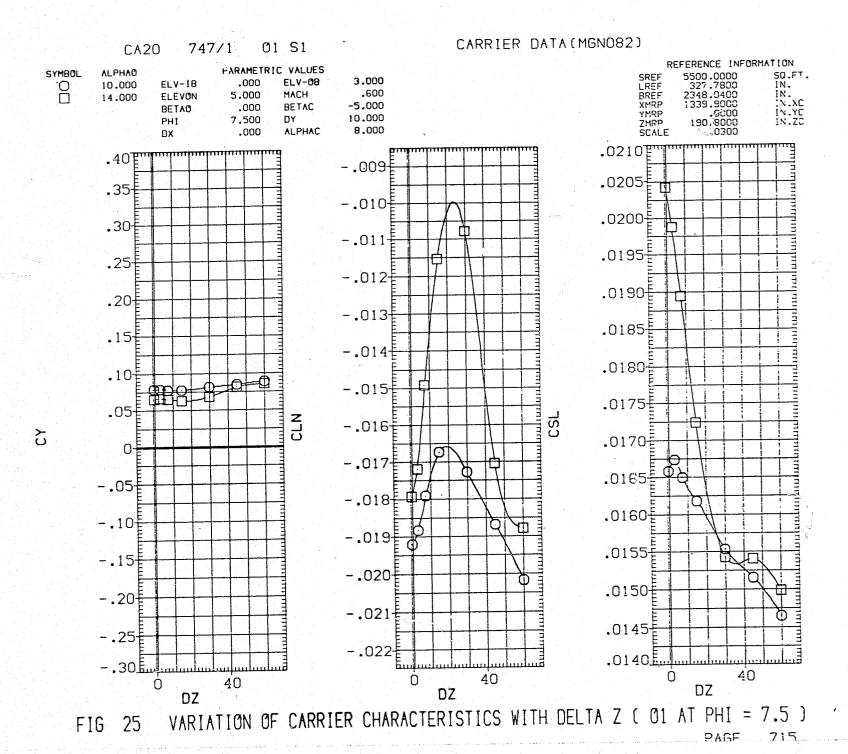
FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 712

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

PAGE



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 25 714 PAGE



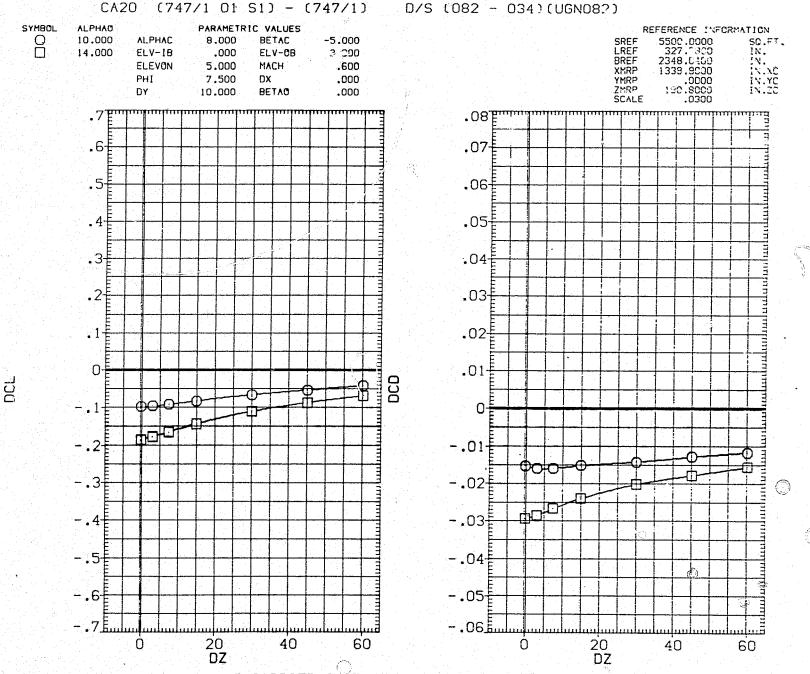


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 716

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

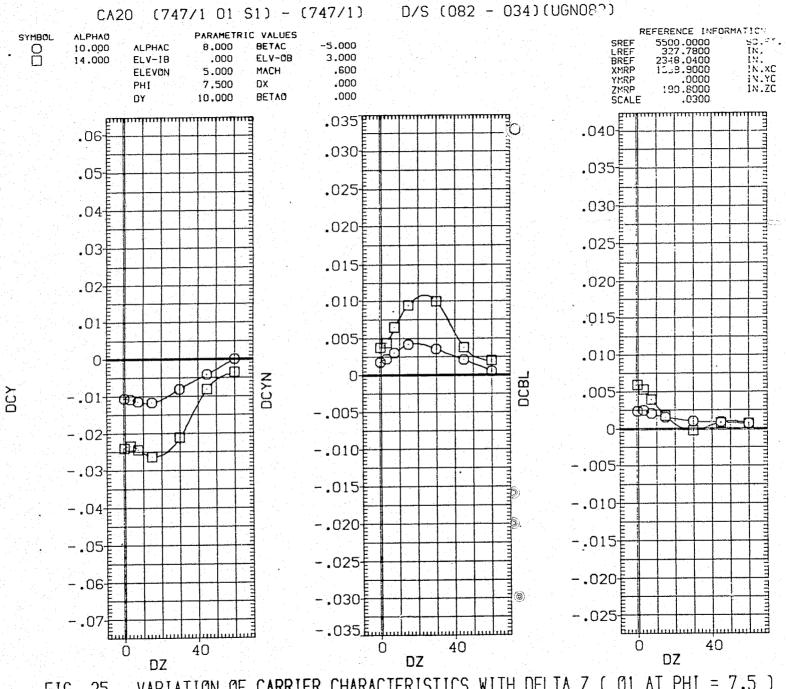


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 718

SYMBOL

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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 719

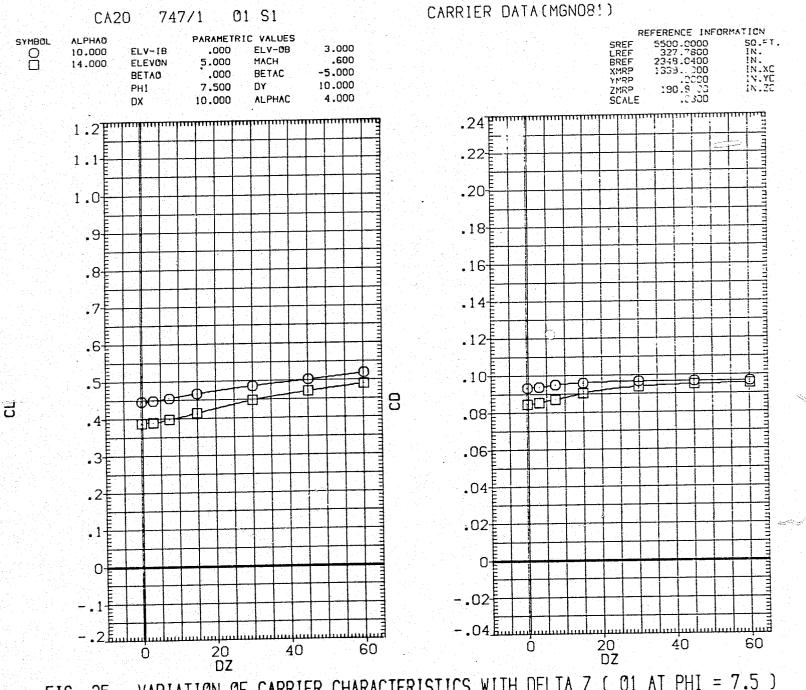
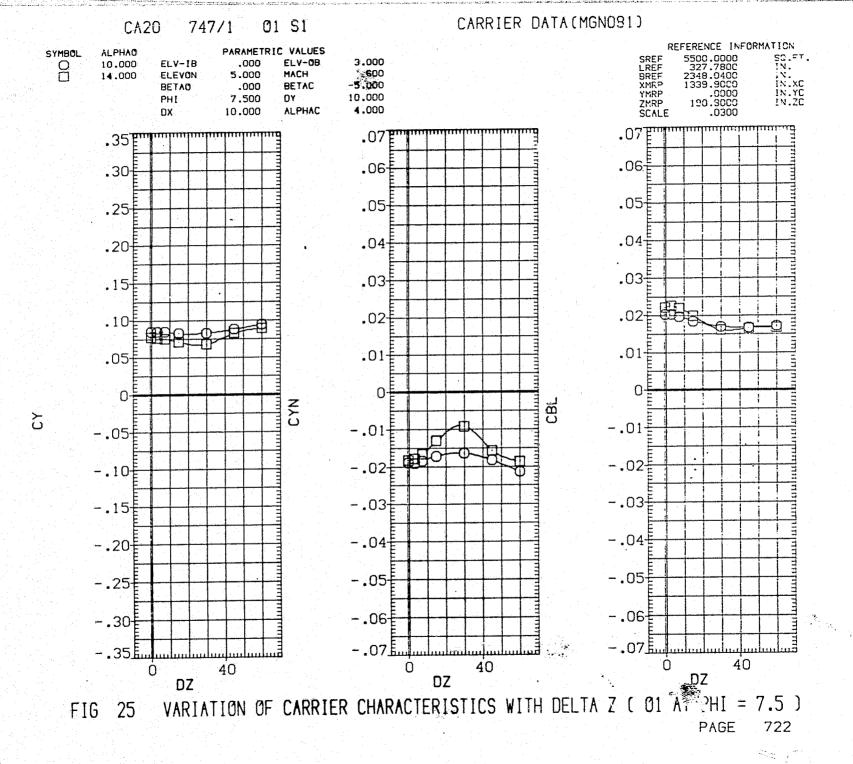
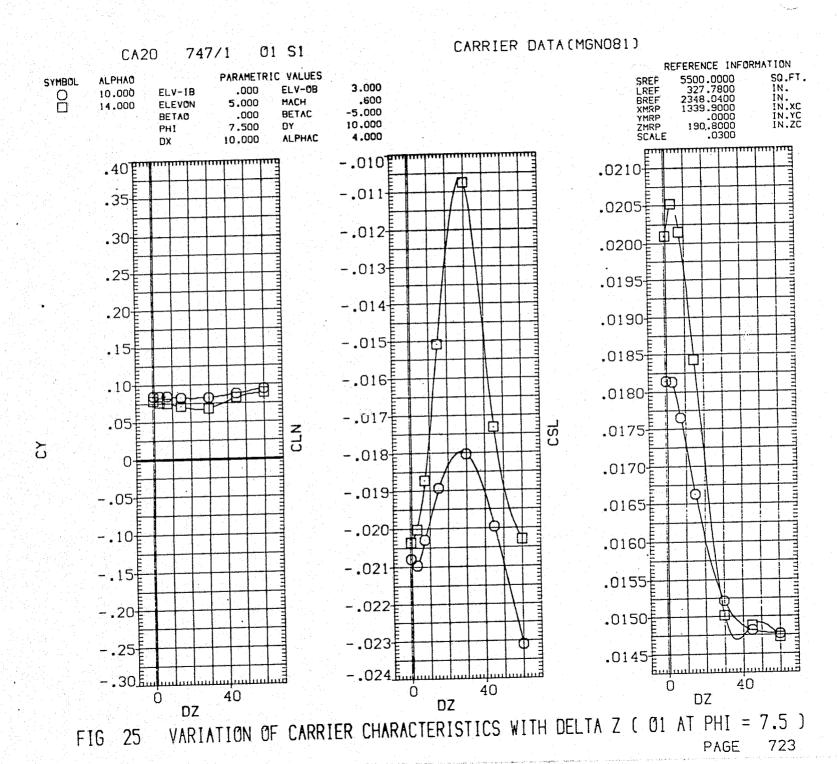


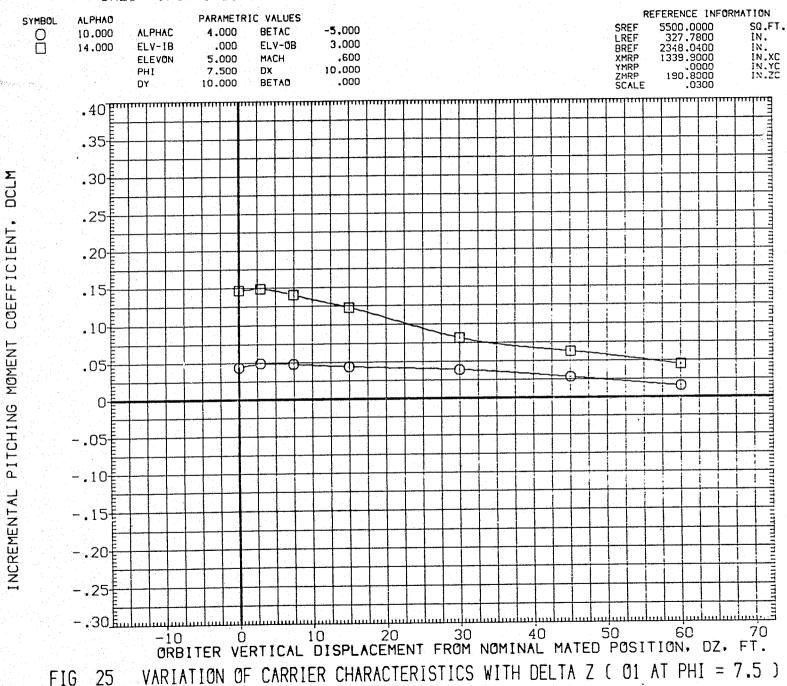
FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 720

PAGE





VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) PAGE 724



PAGE

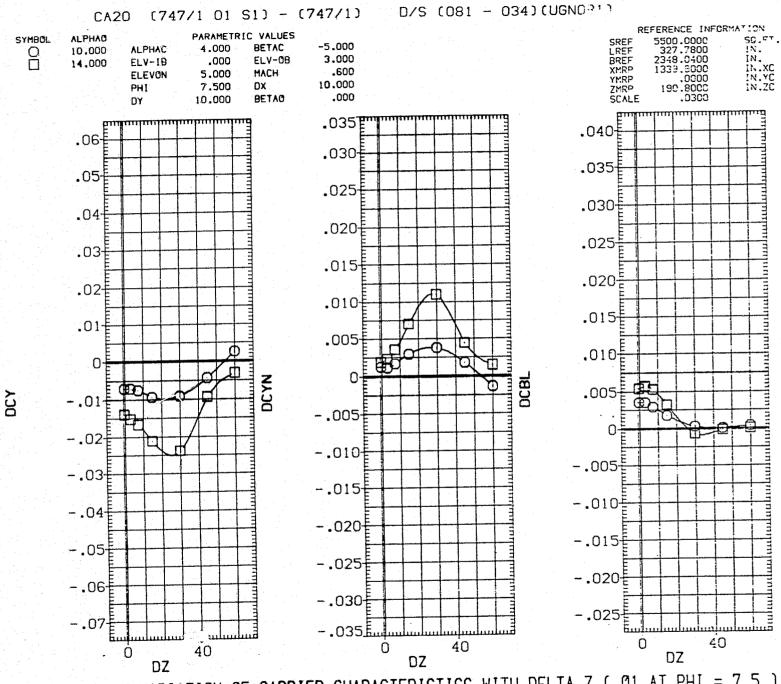


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

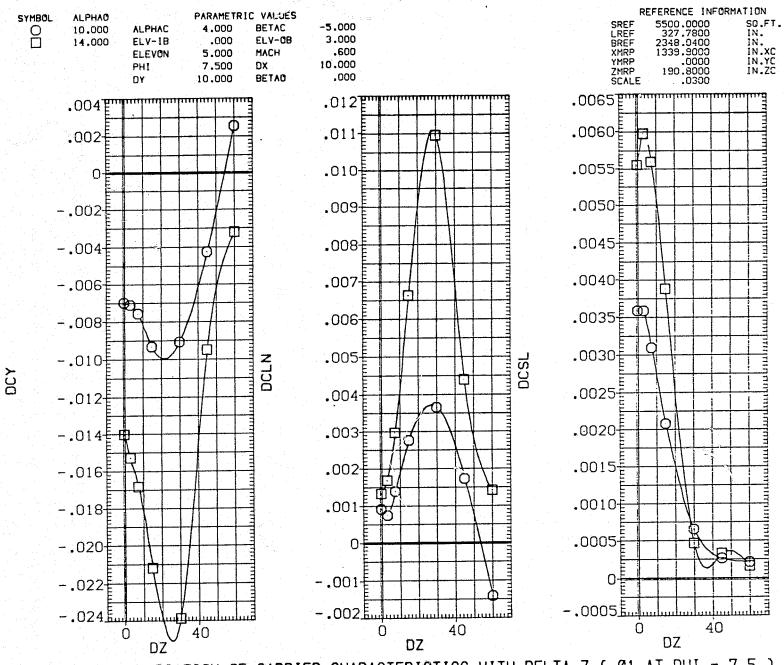


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 727

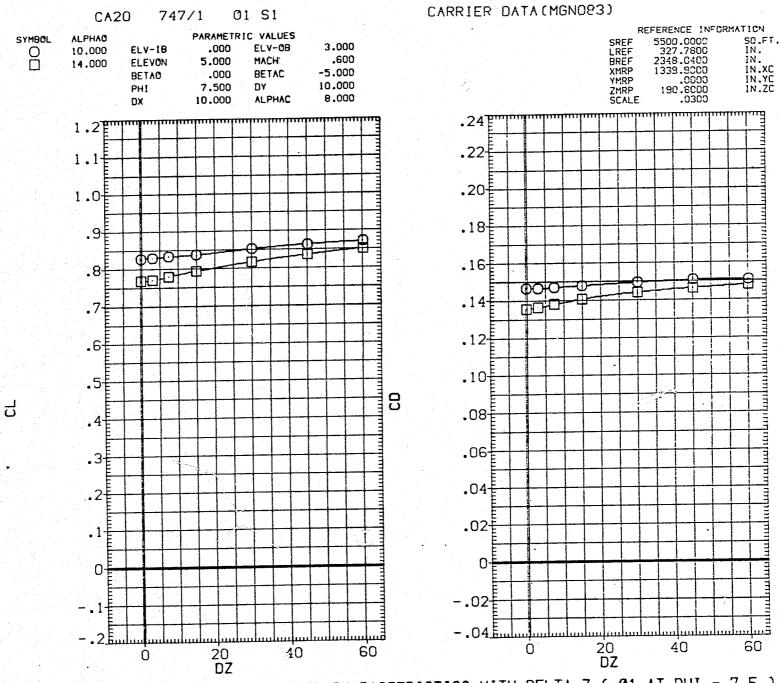
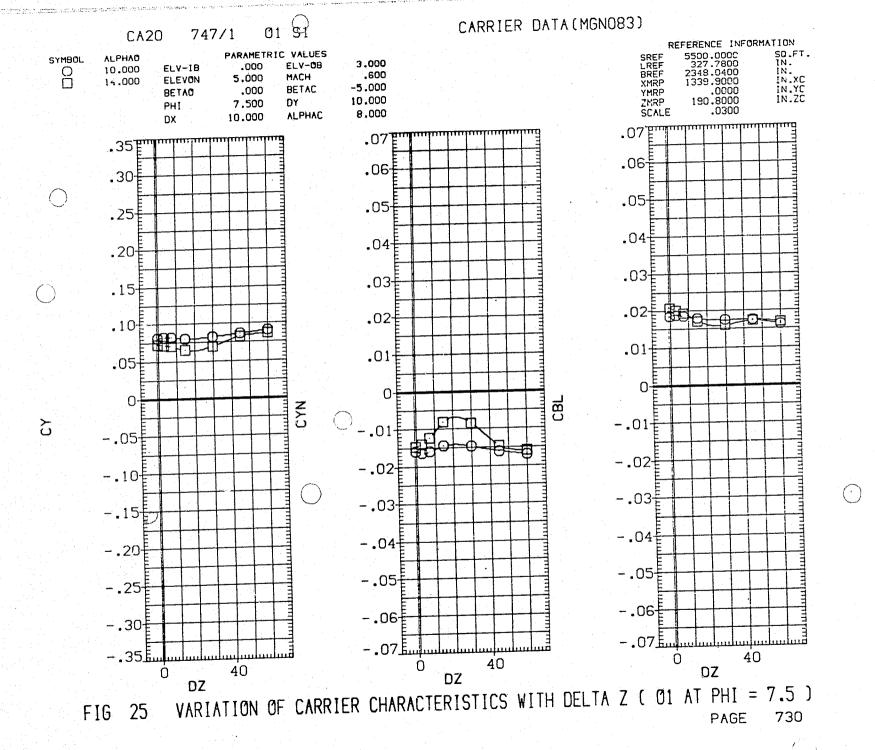


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 728

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)



PAGE

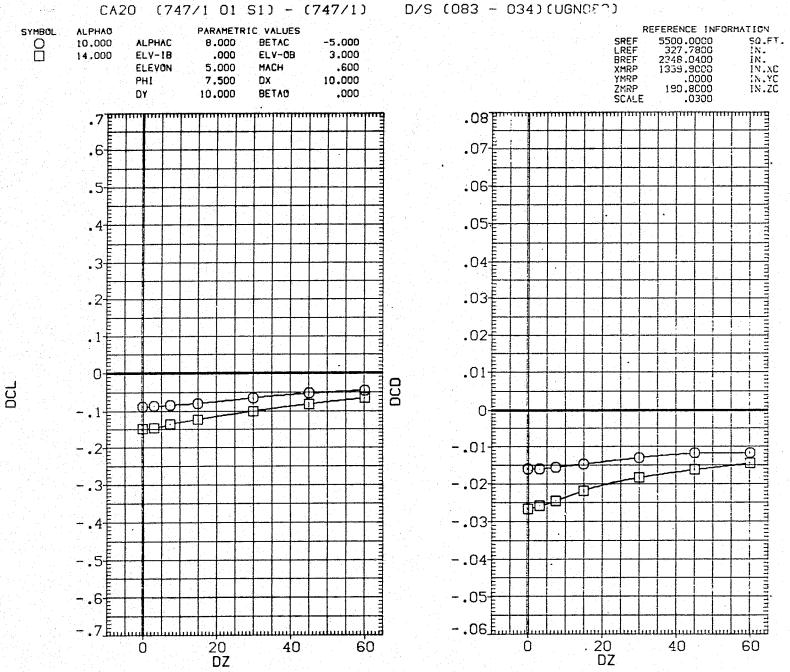
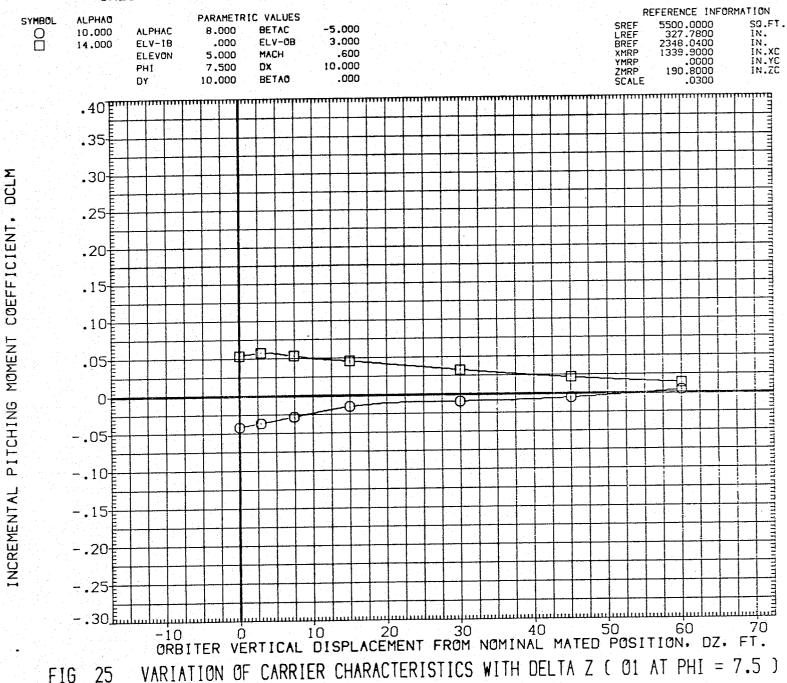


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 732



PAGE

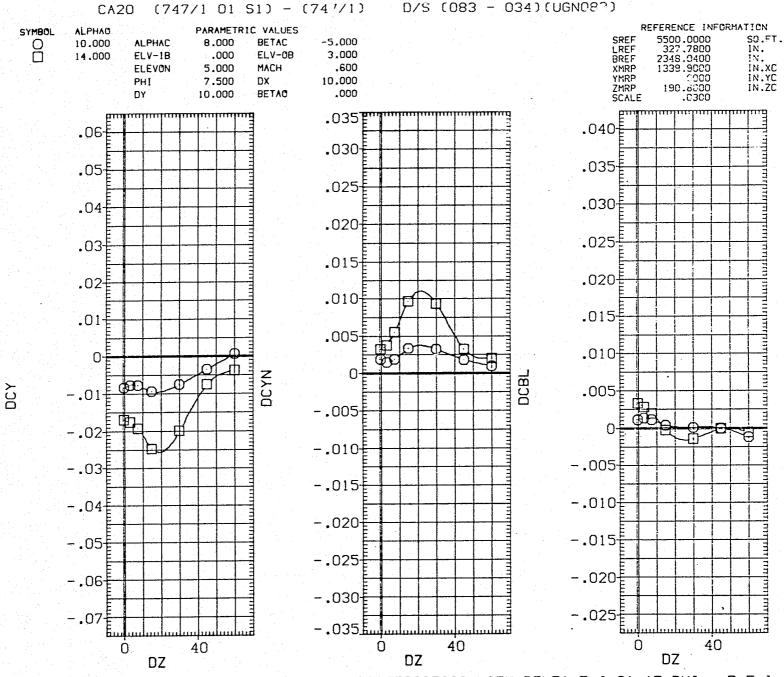


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) PAGE 735

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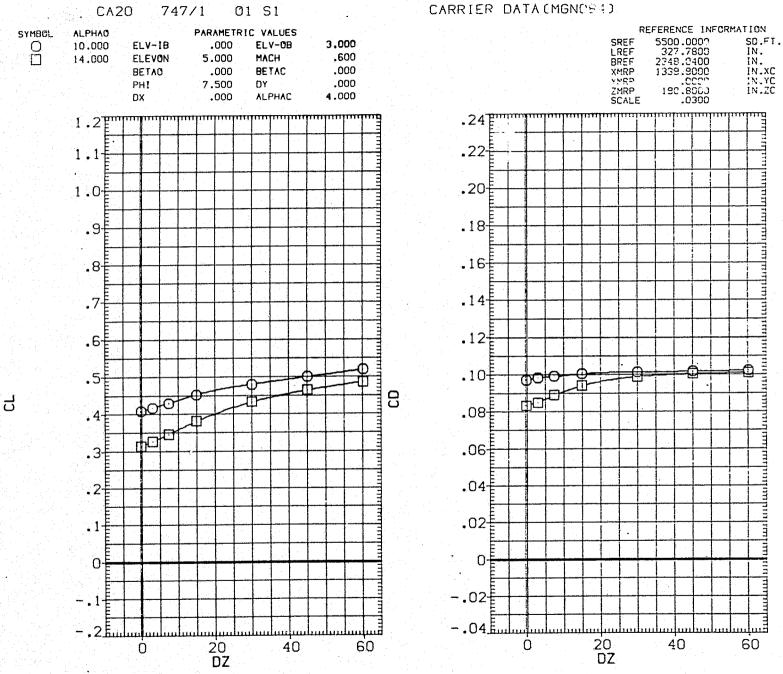


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 736

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

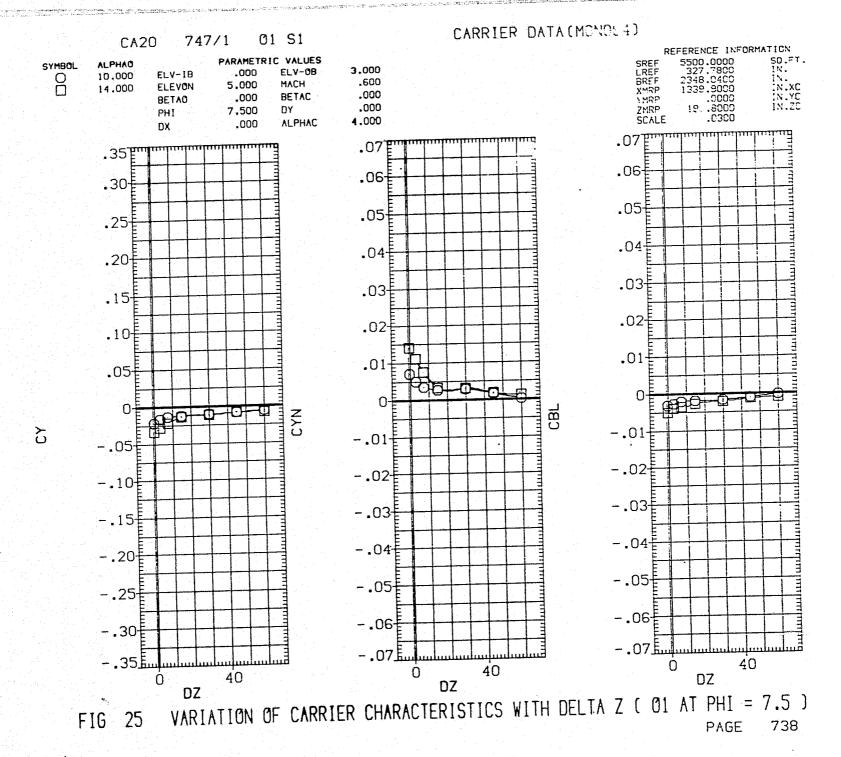


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 739

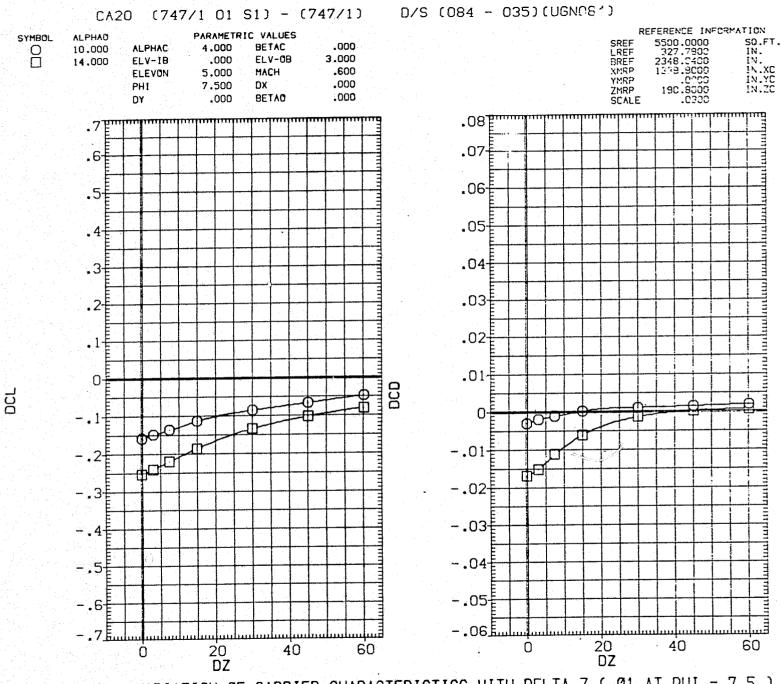
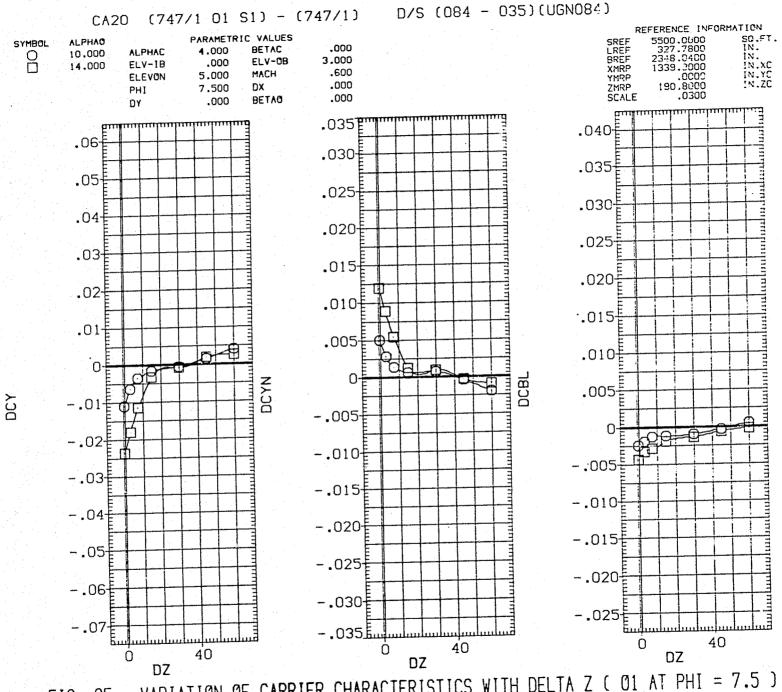


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 741



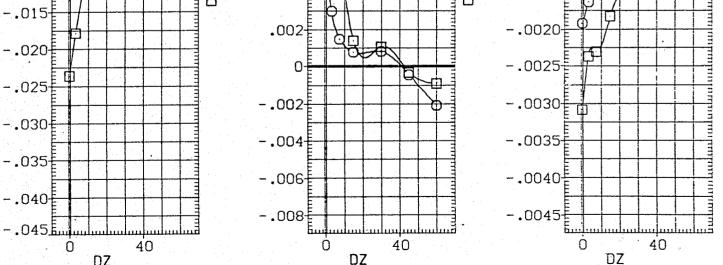
VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 742 PAGE

SYMBOL

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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG PAGE 743

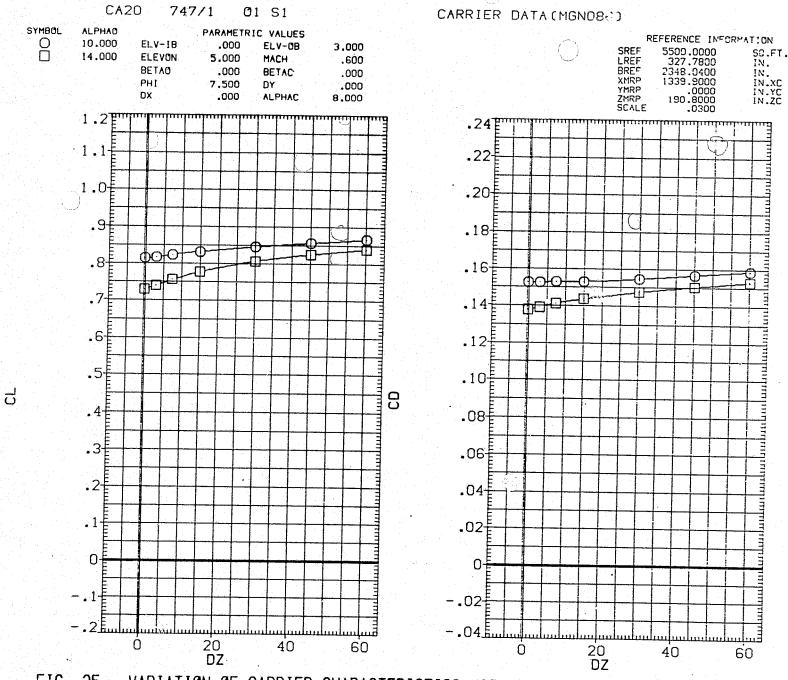


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 744

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 745

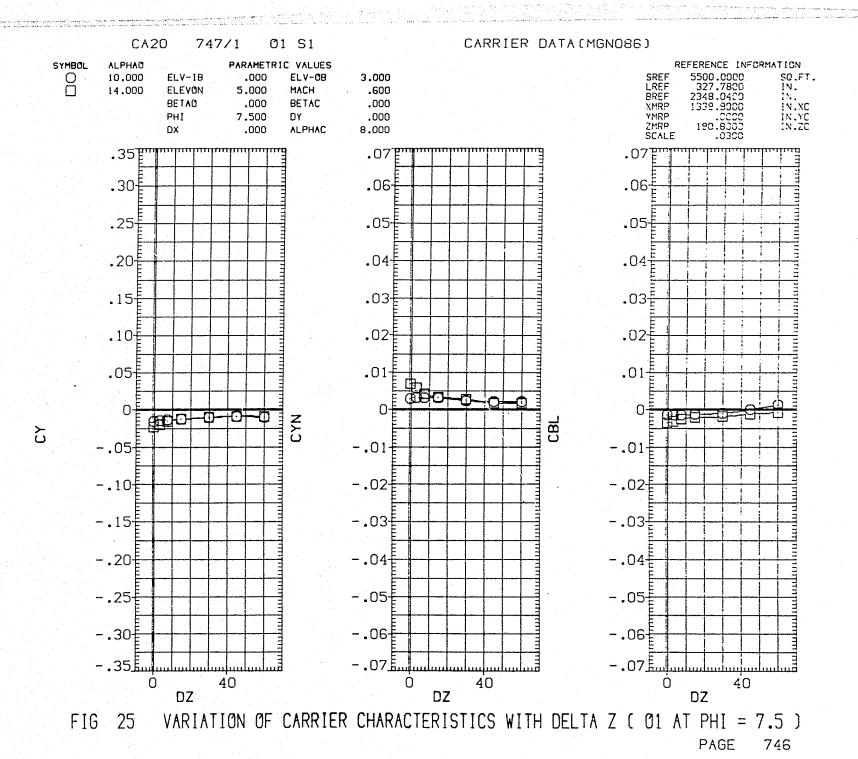
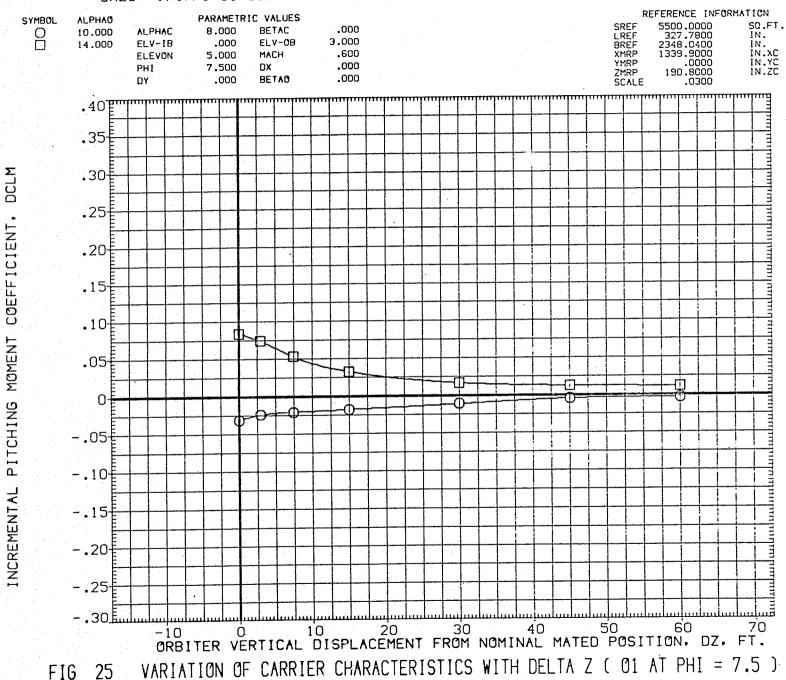


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 747

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 748





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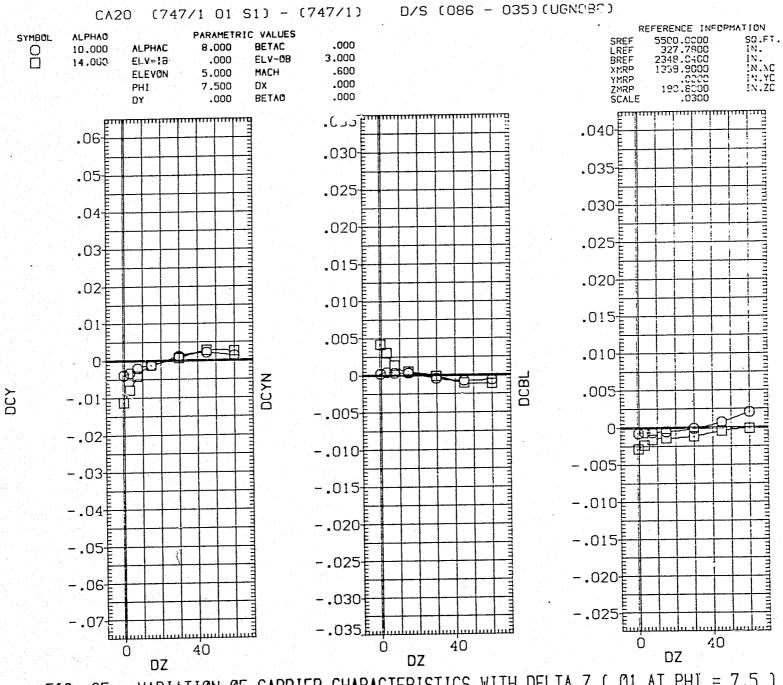


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PH) = 7.5)

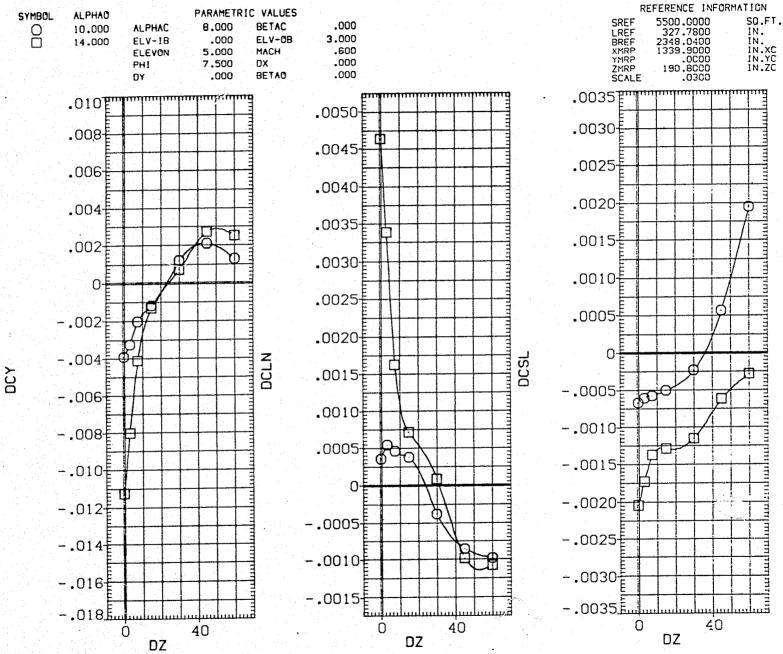


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 752

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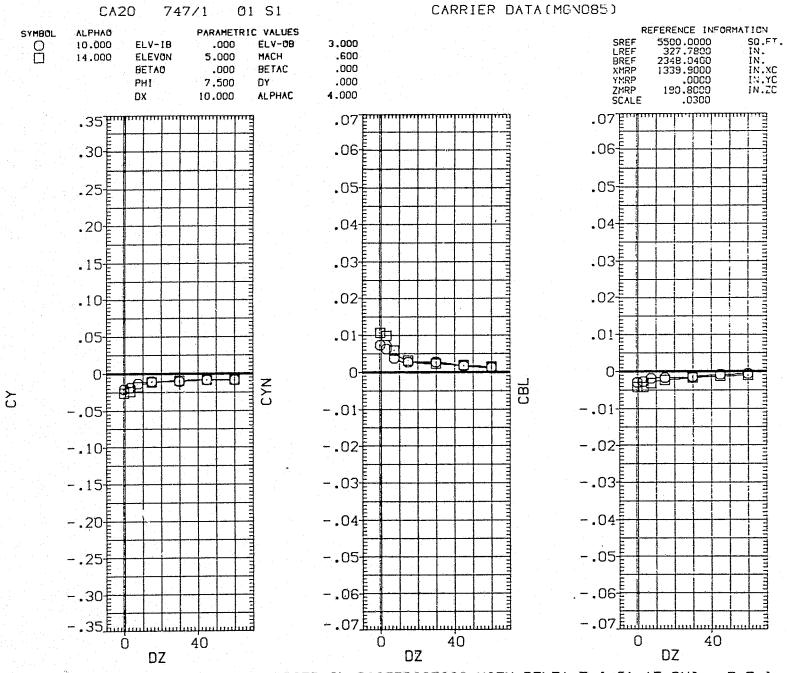


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 754

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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG PAGE 755

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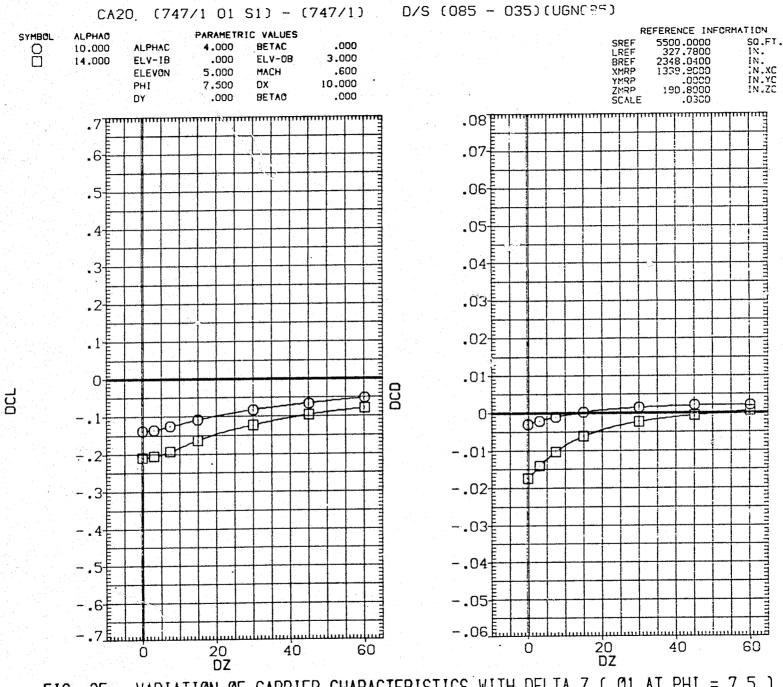
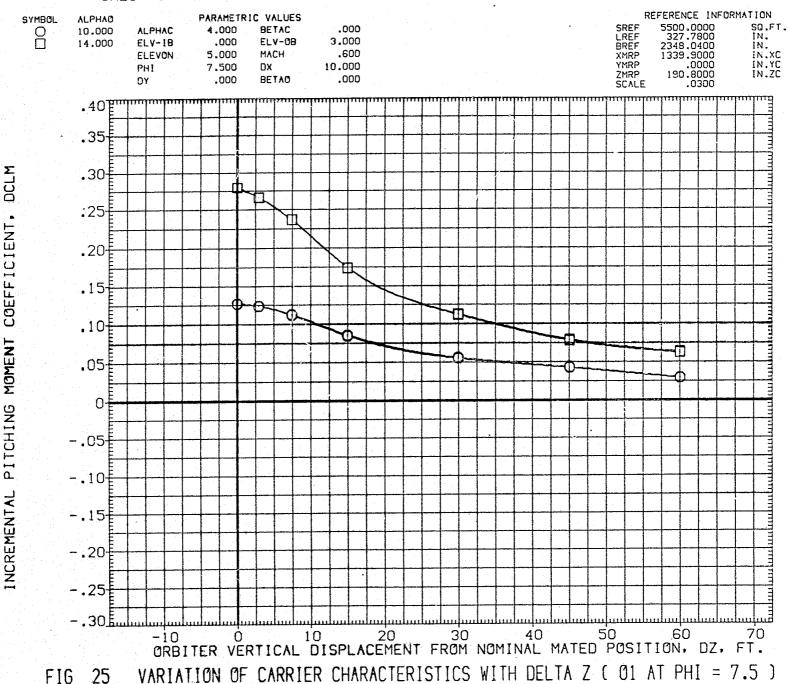


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

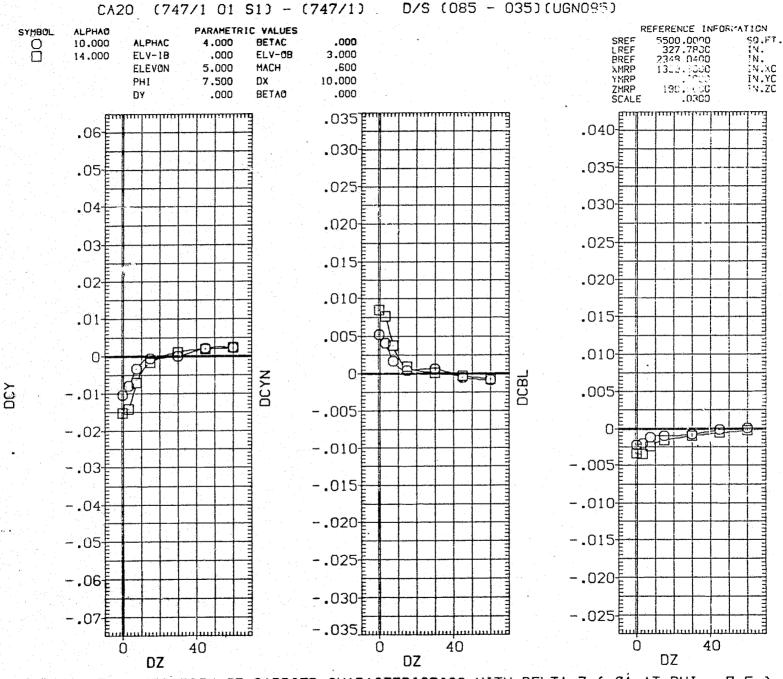


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 758

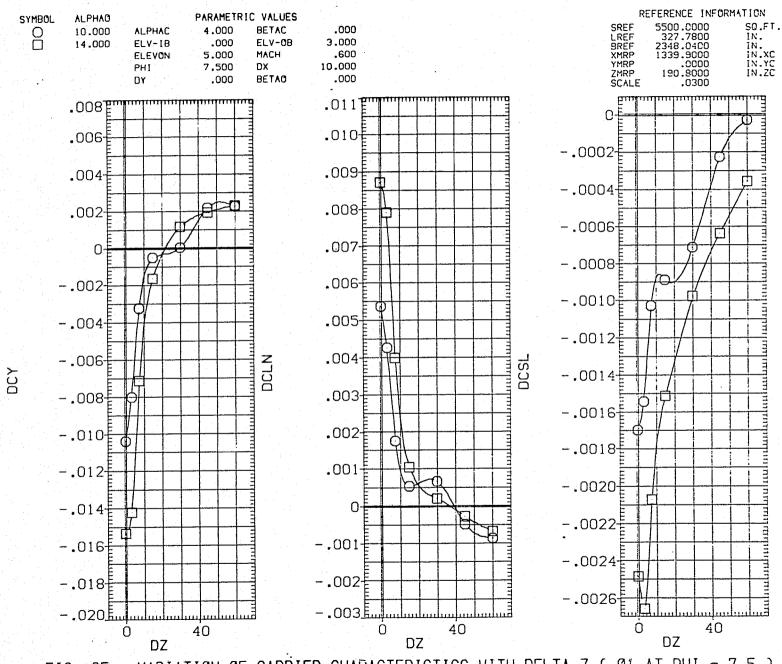


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 759

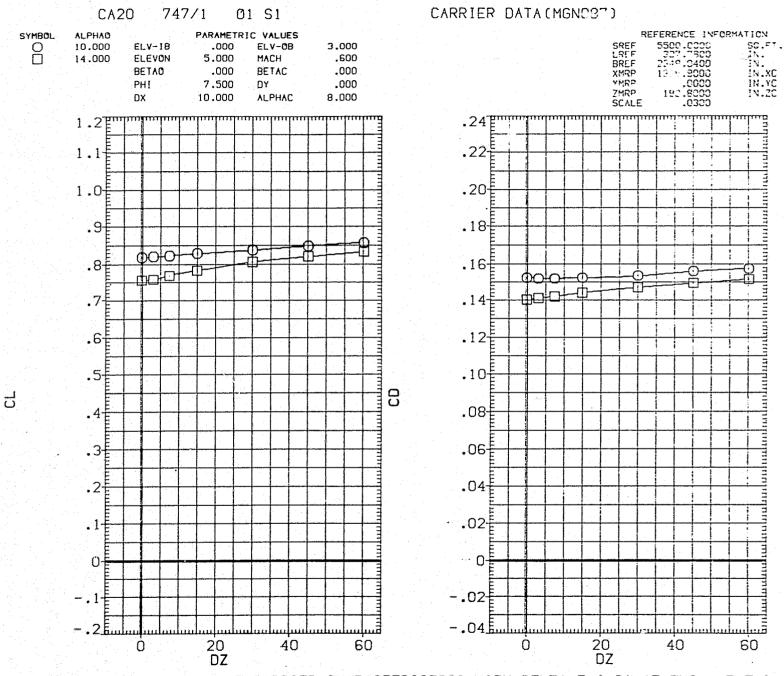
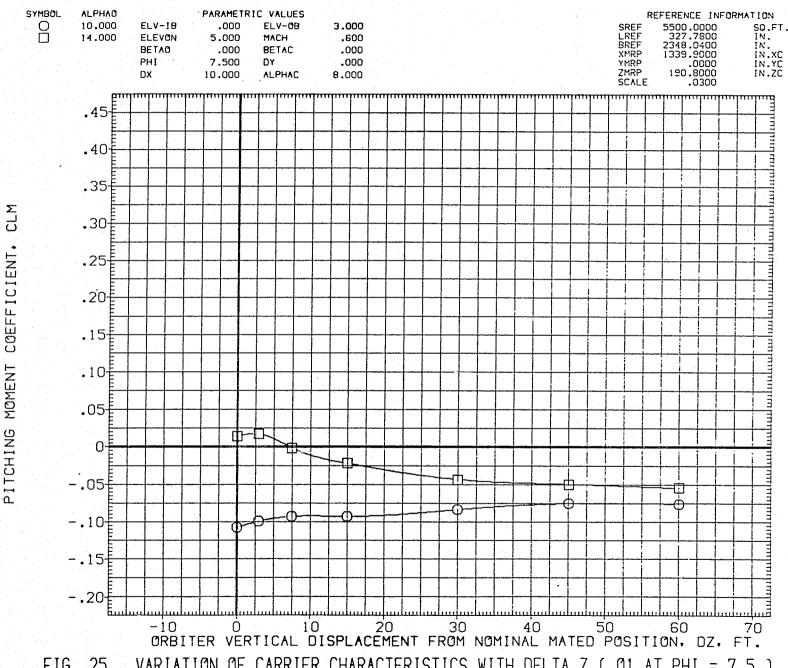


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 25

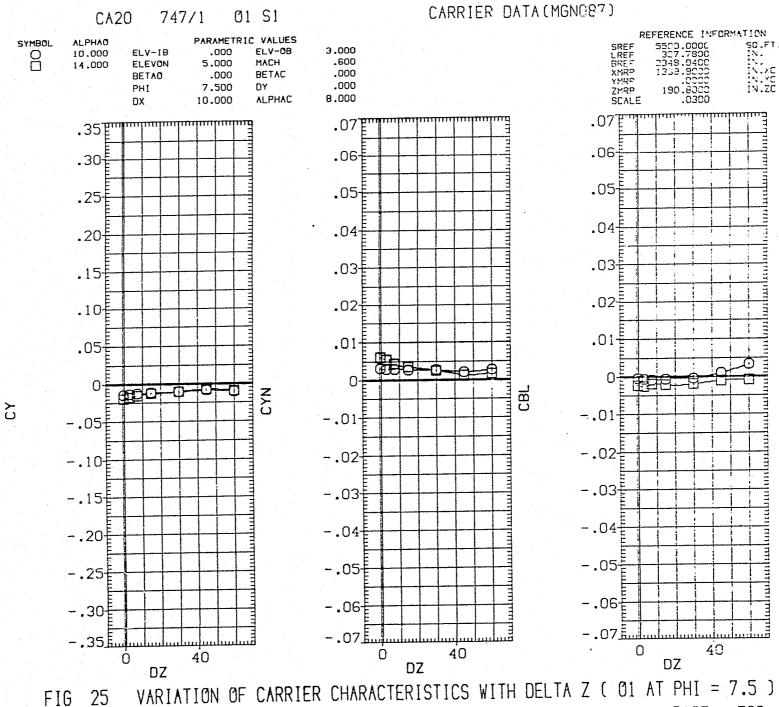


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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 763

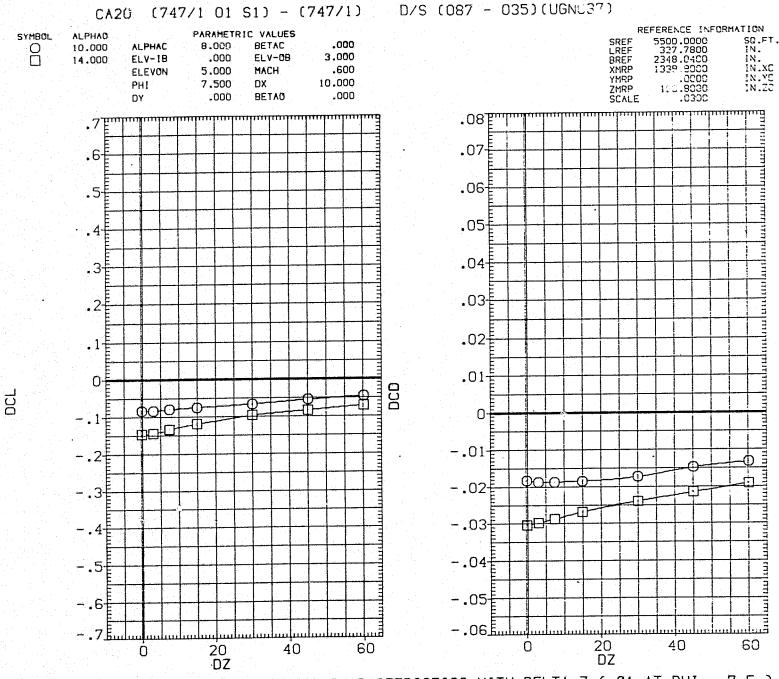


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

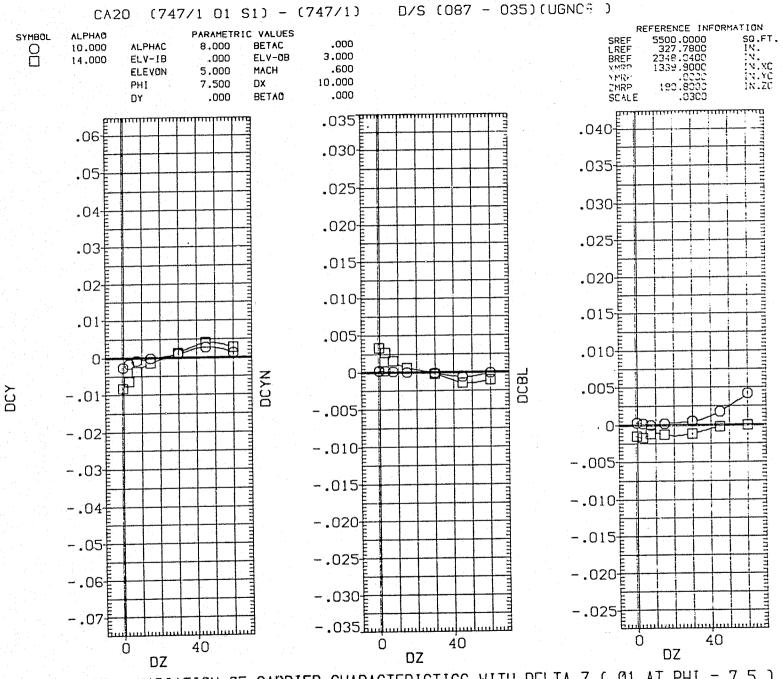


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 766

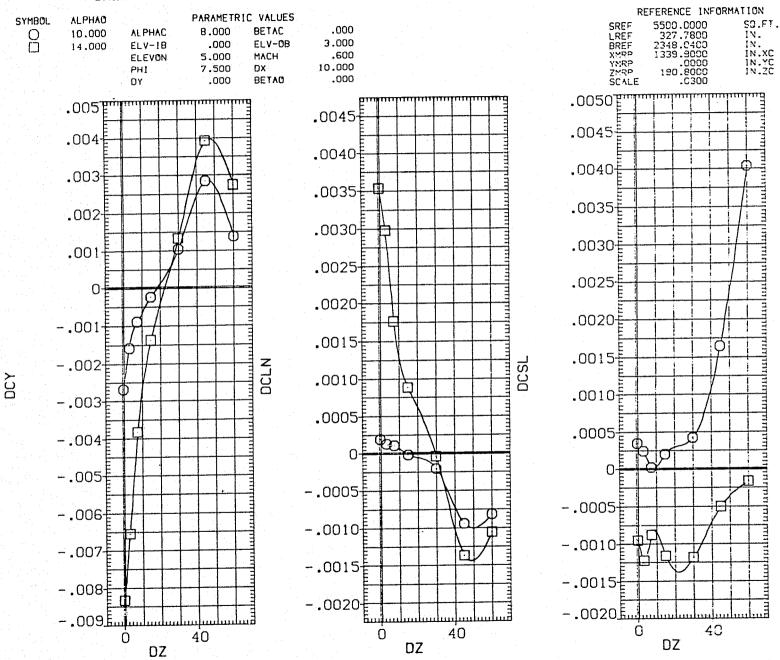


FIG 25 : VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

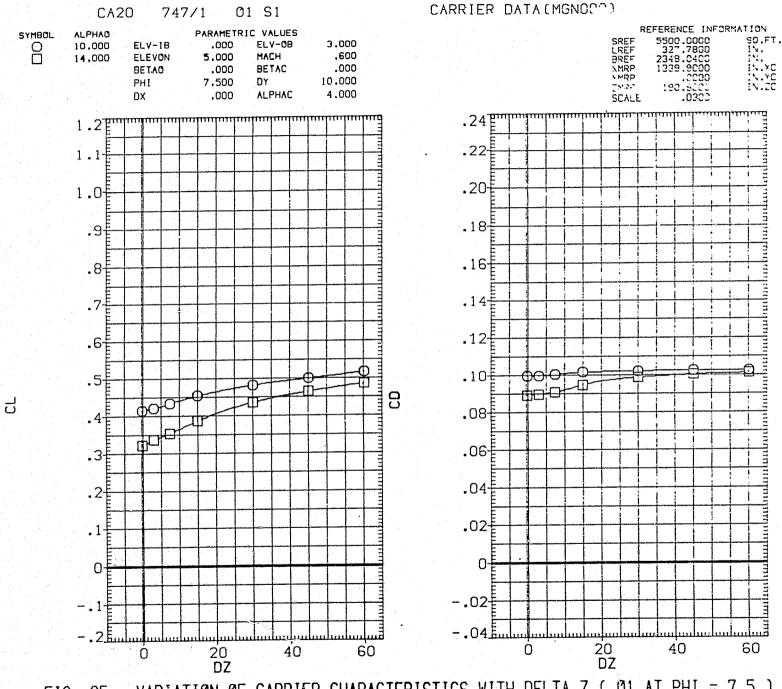
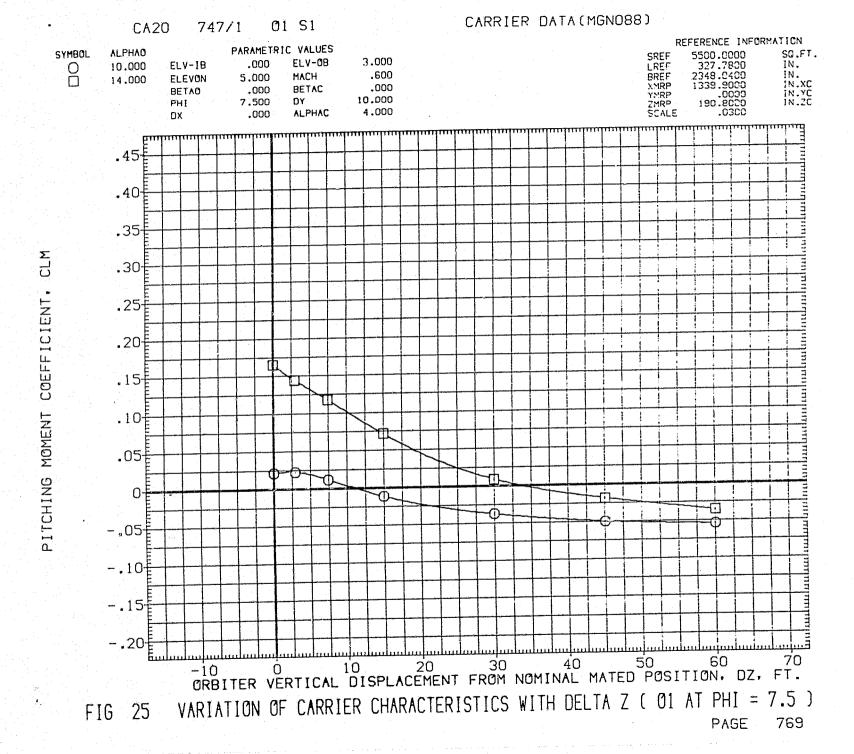


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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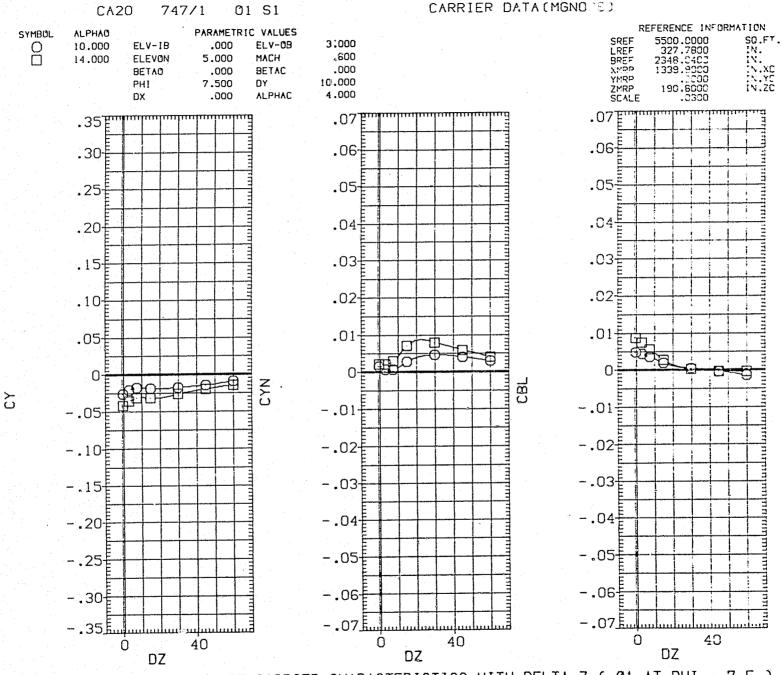


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

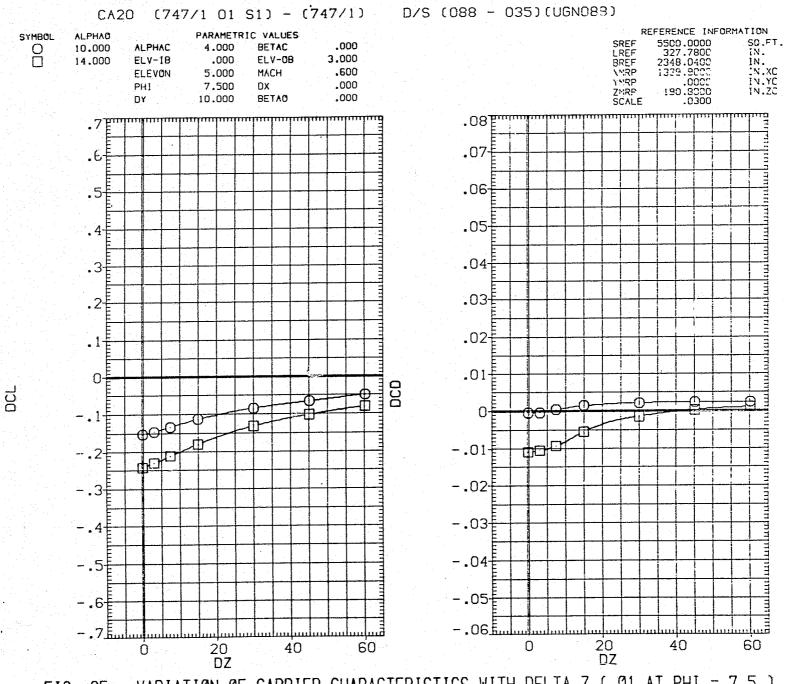


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 772

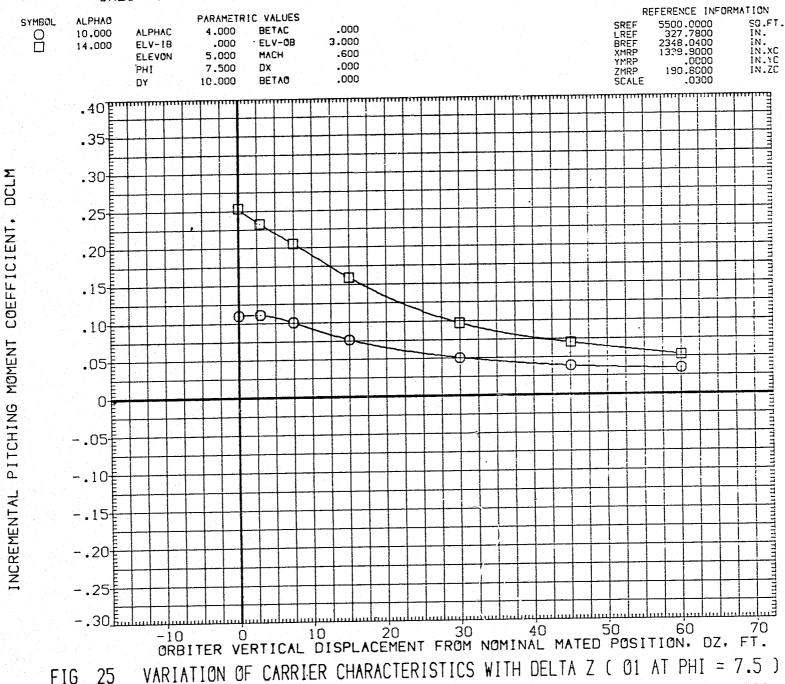


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 774

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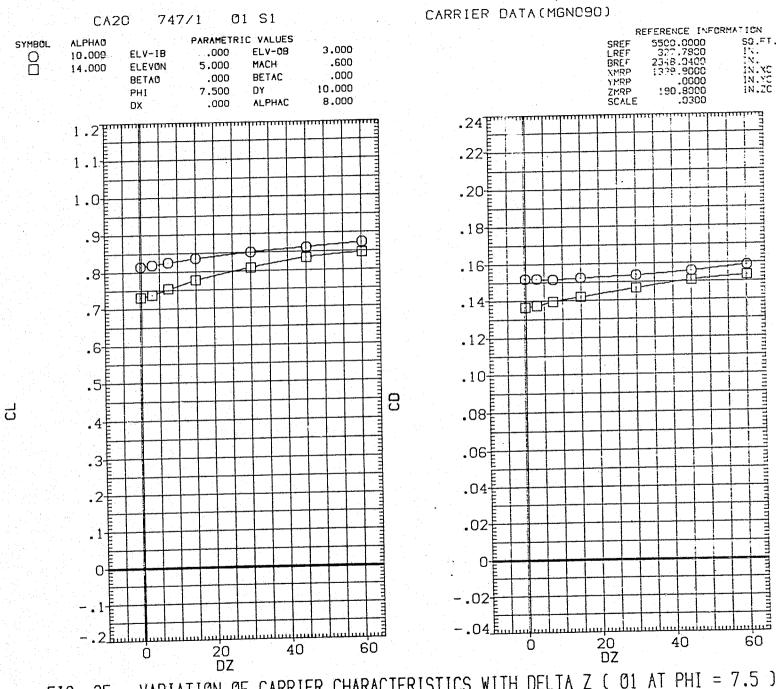
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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

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DZ

DZ



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 25 FIG 776 PAGE

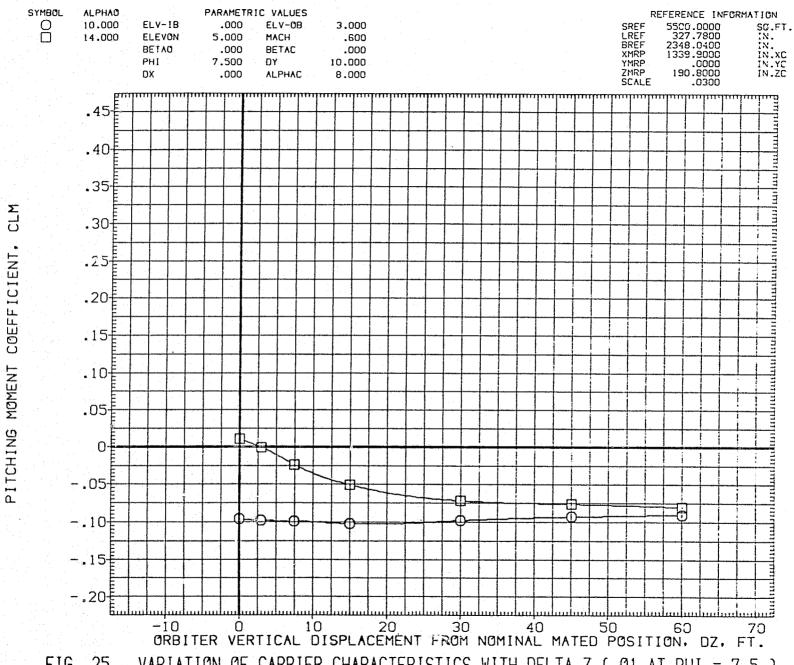


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 777

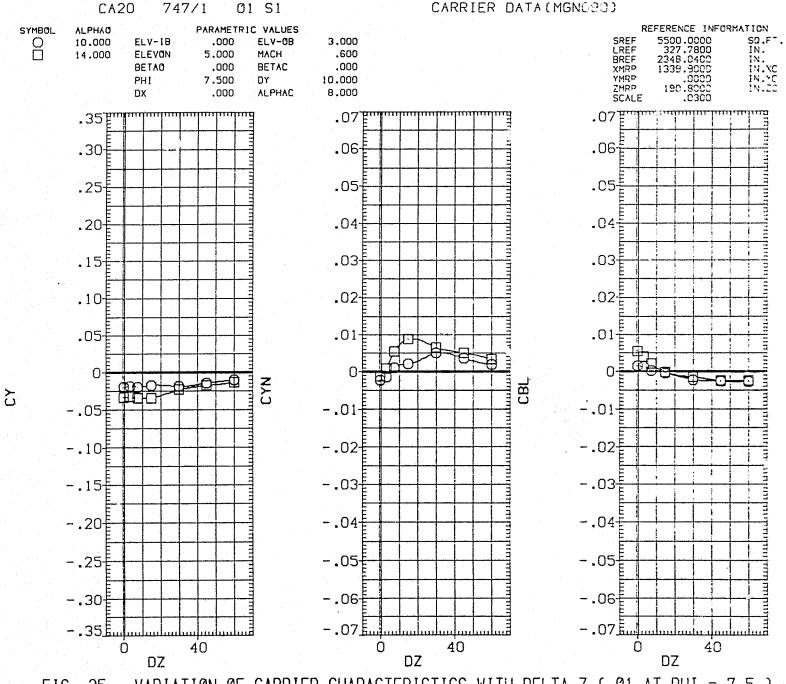


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 778

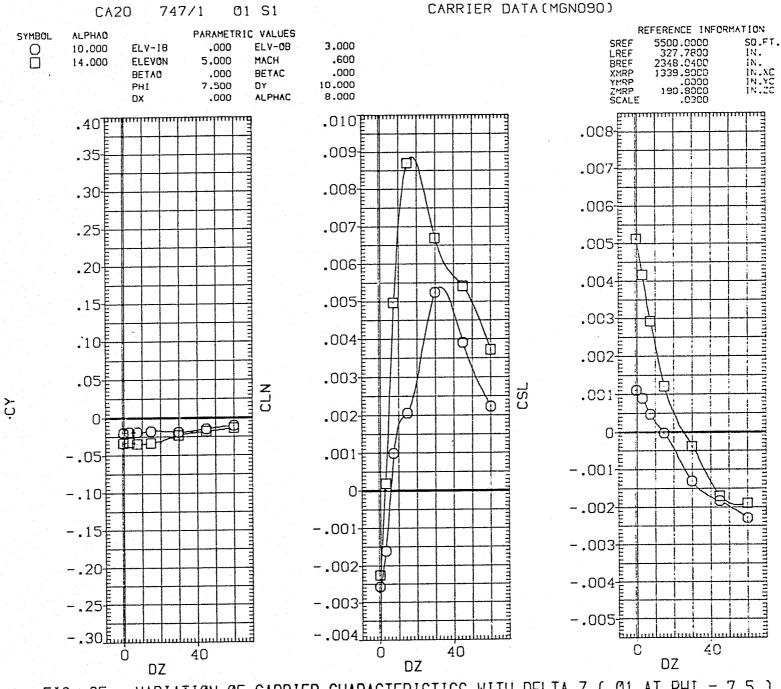


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 780

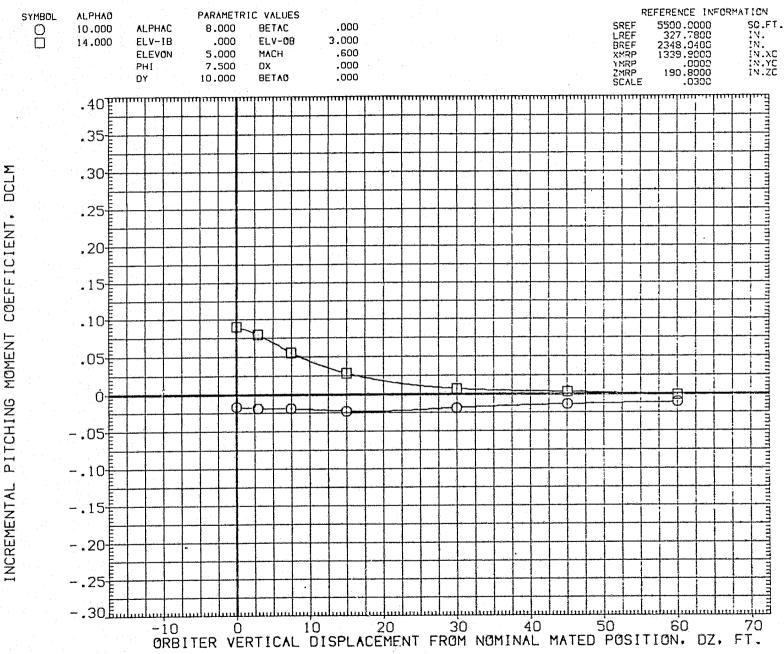


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 781

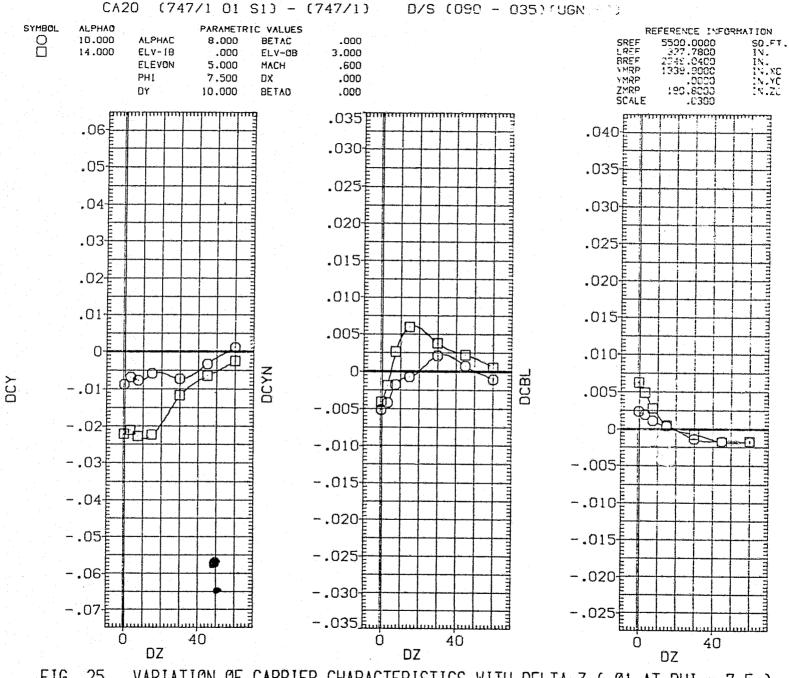


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 782

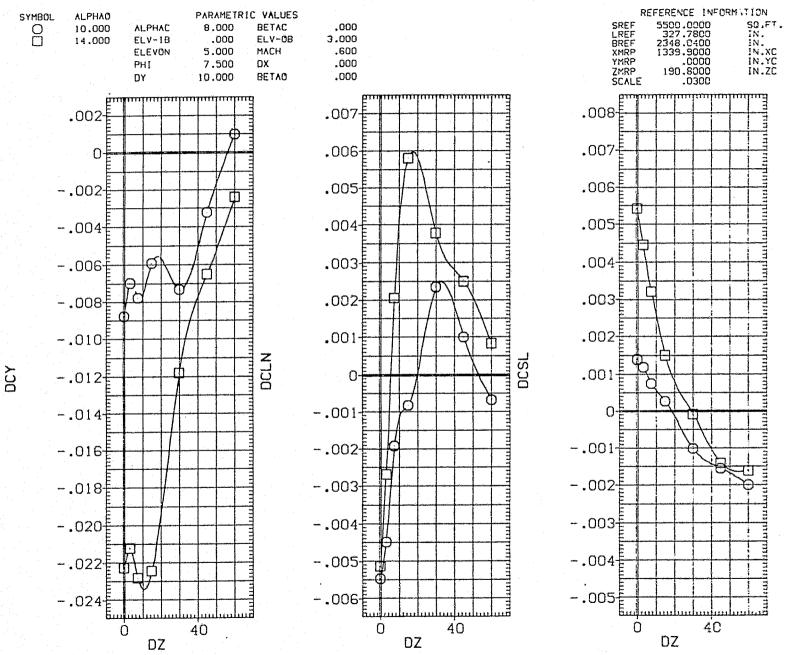


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

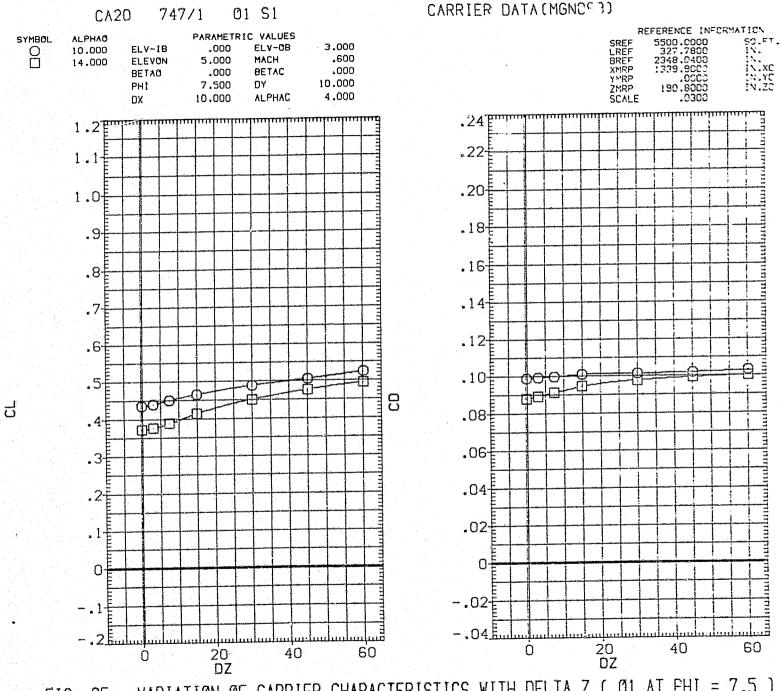
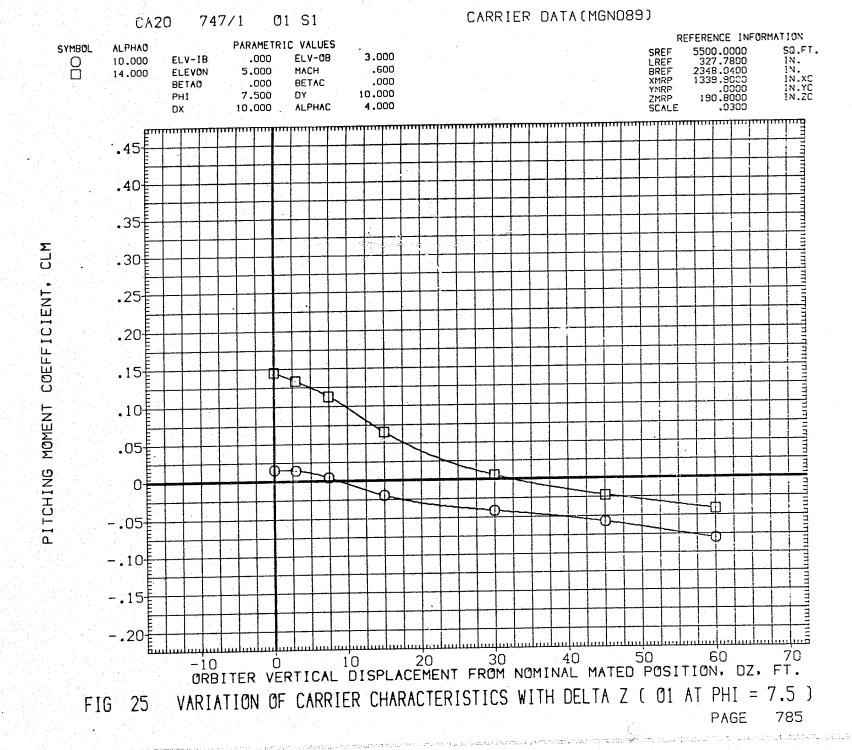


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT FHI = 7.5)
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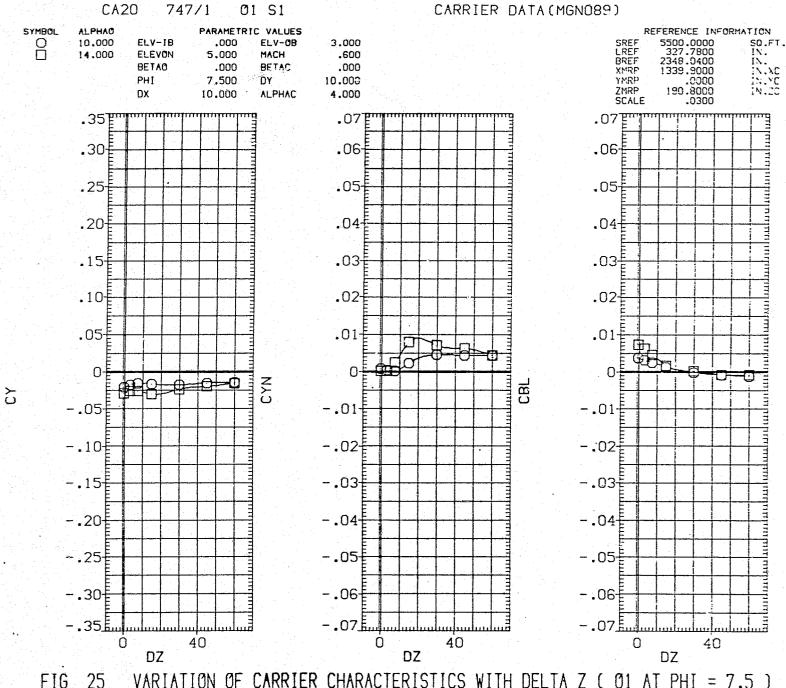


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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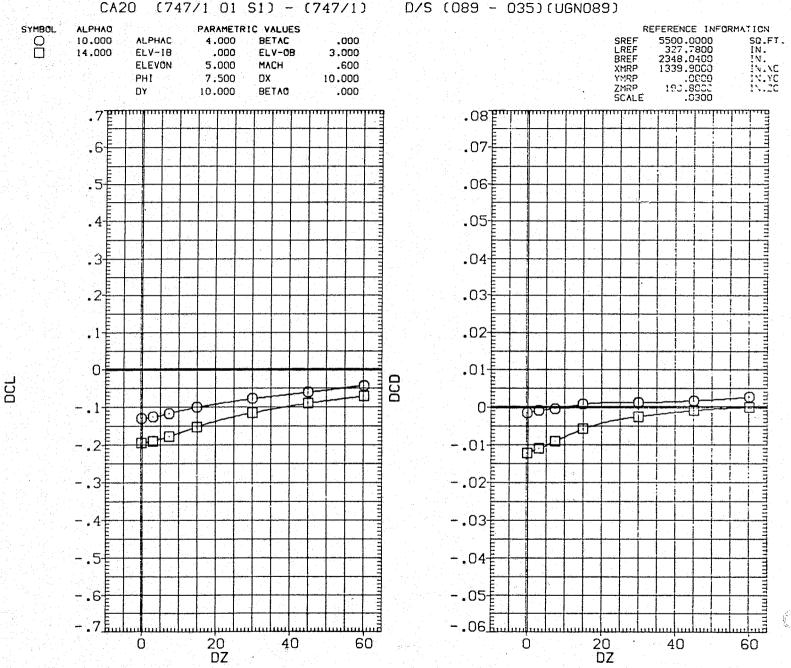
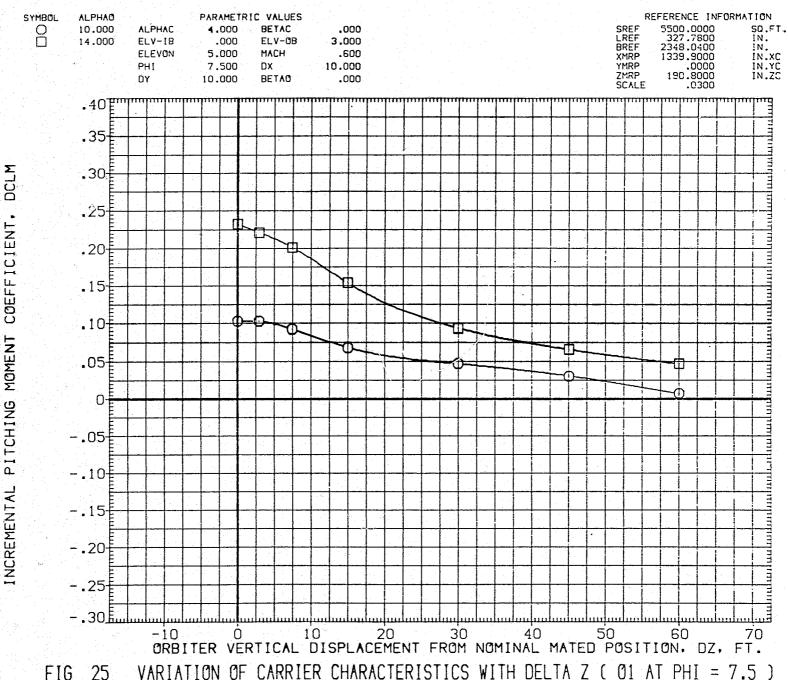


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 788



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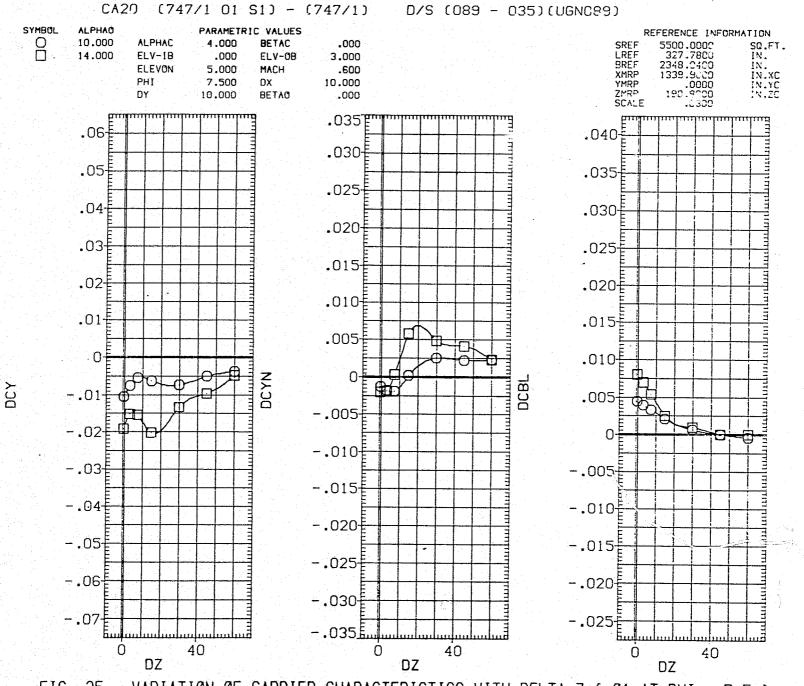


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 790

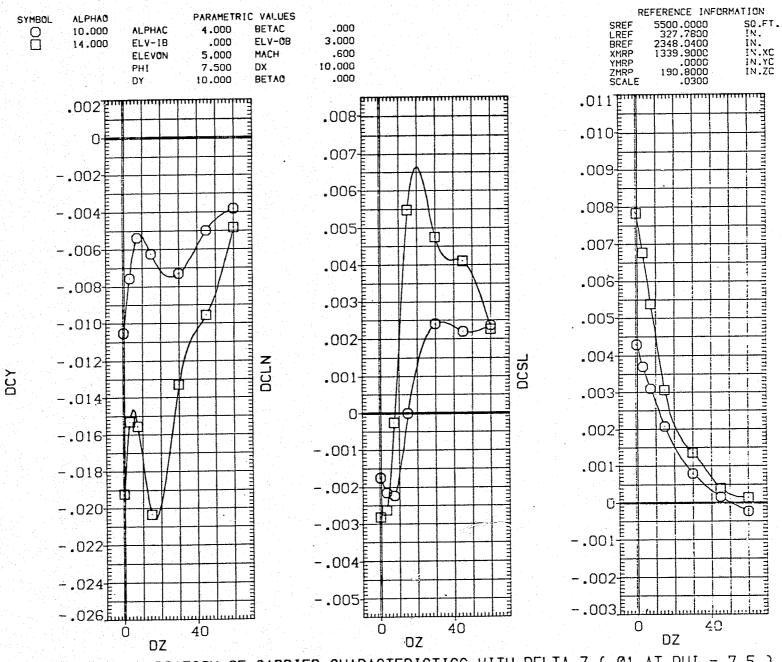


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 791

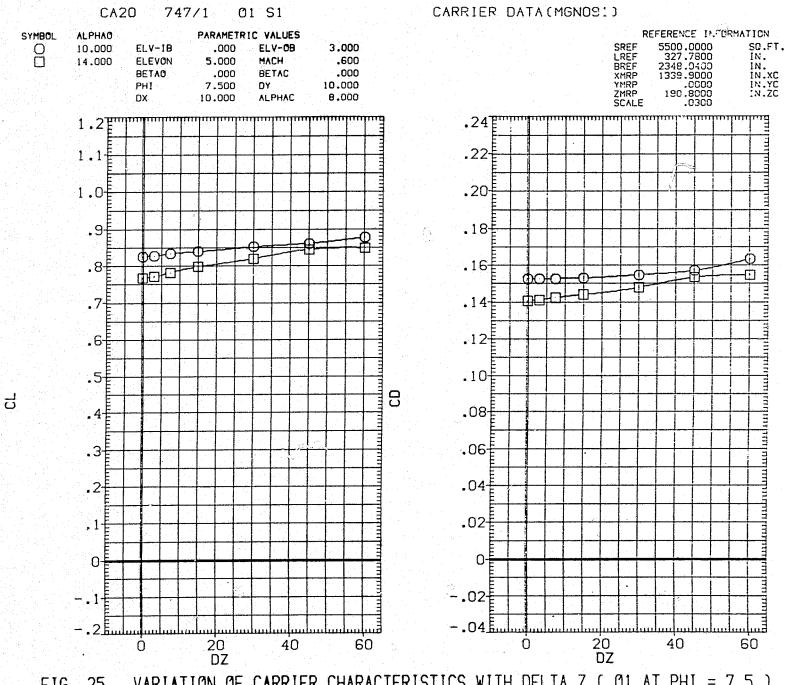
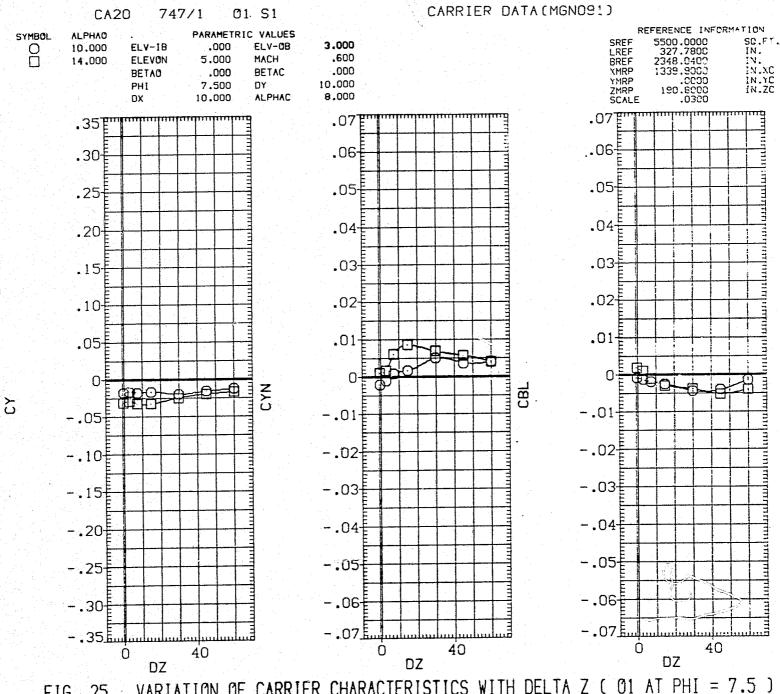


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 794 PAGE

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 795

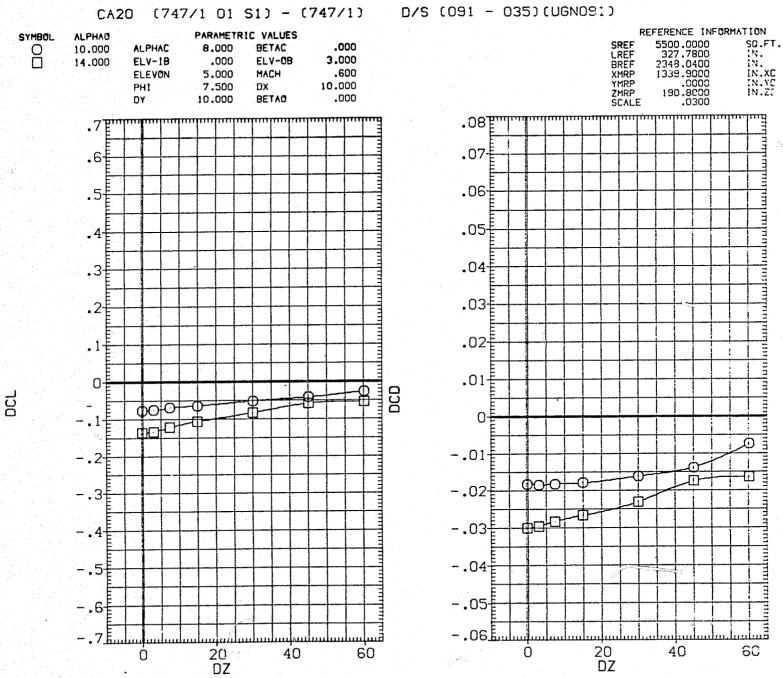


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 798

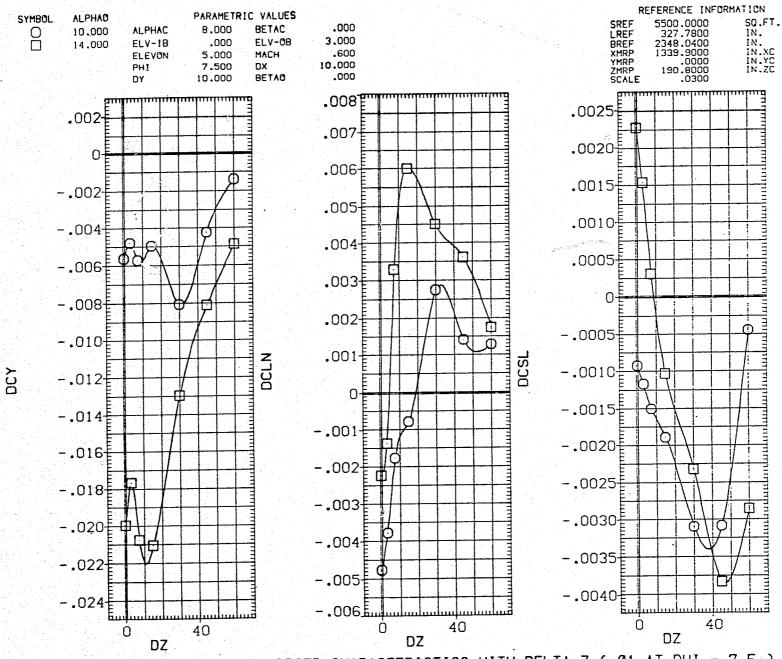


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 799

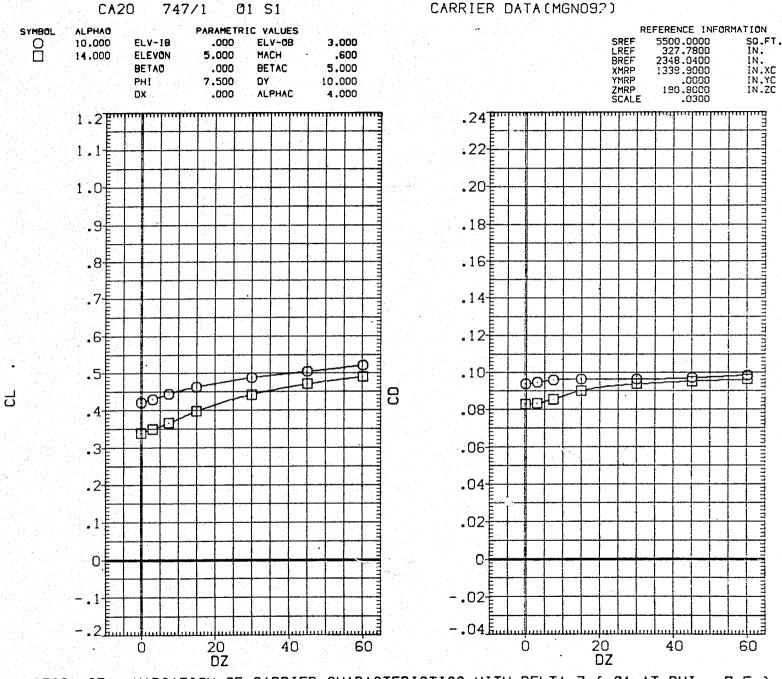


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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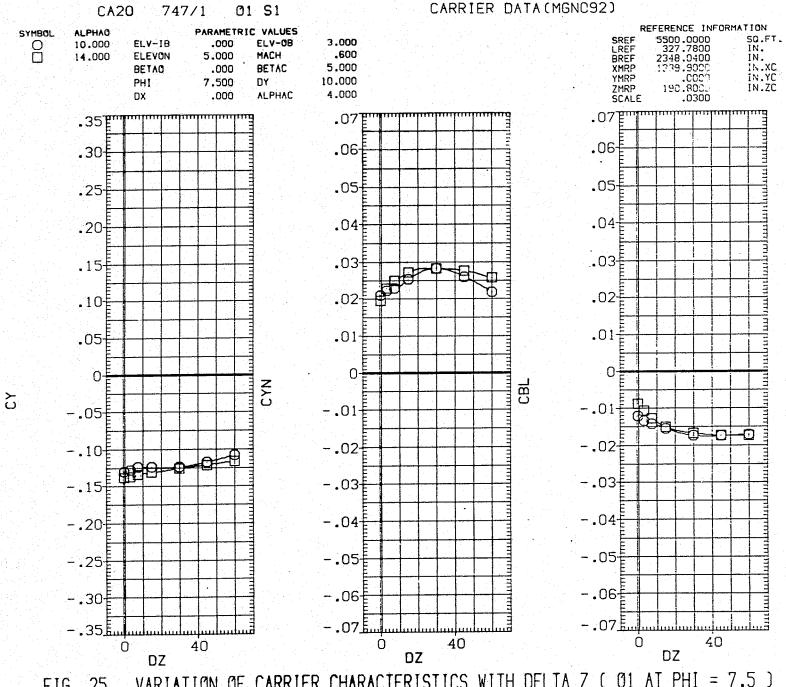


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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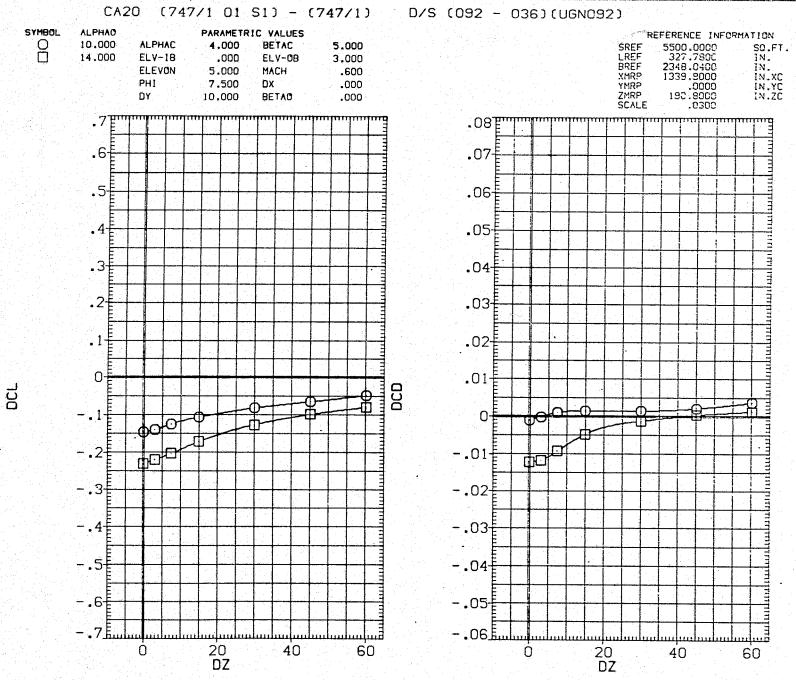
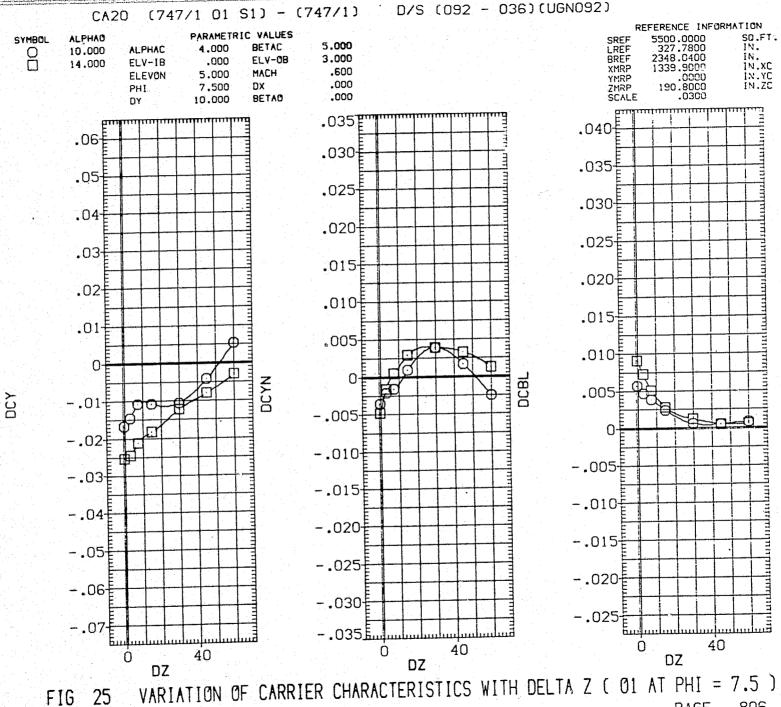
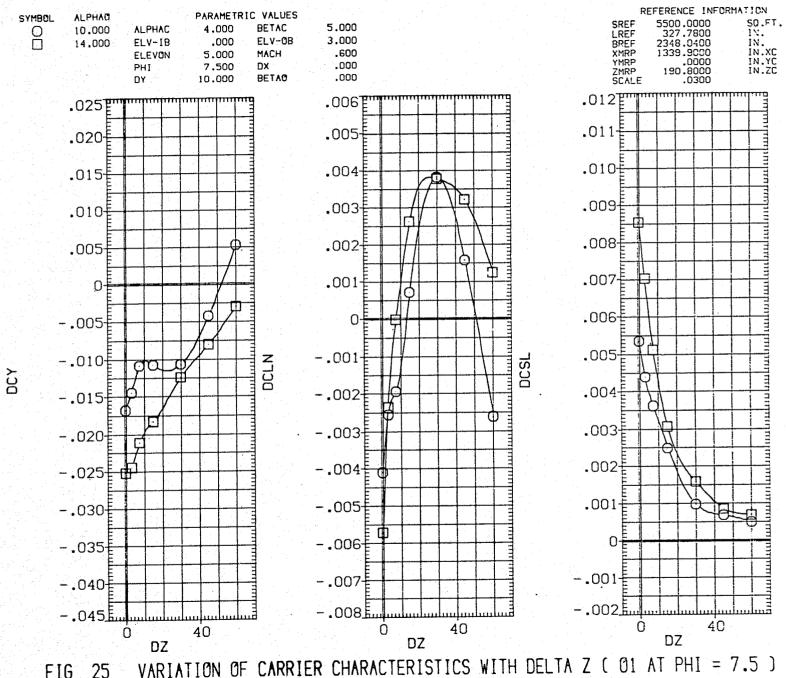


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 804

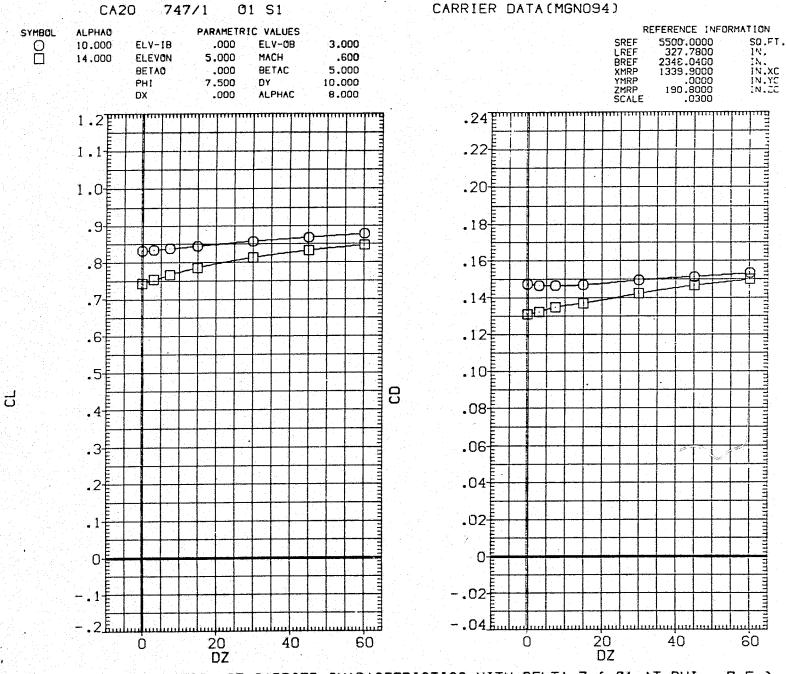
FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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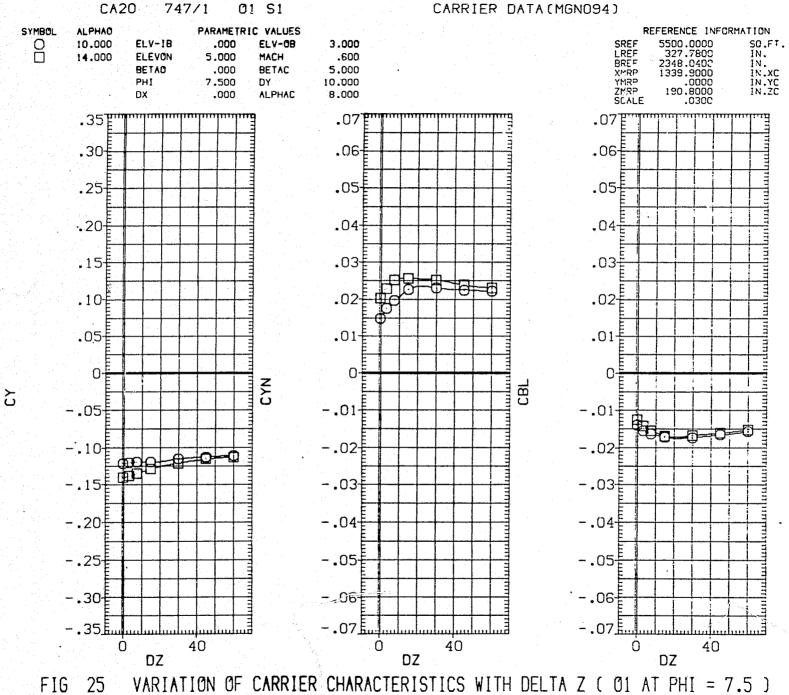


VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) 25 PAGE 807



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FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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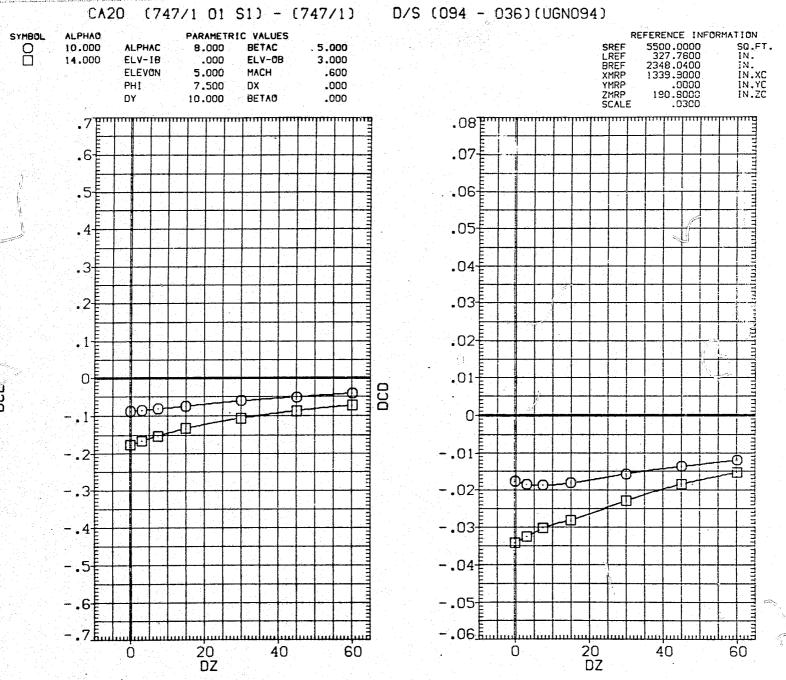


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 812

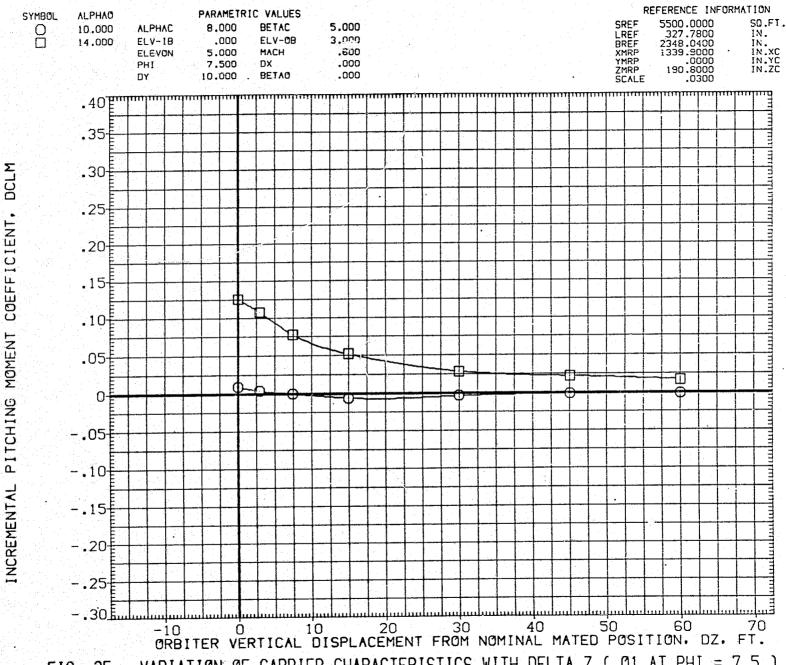


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 813

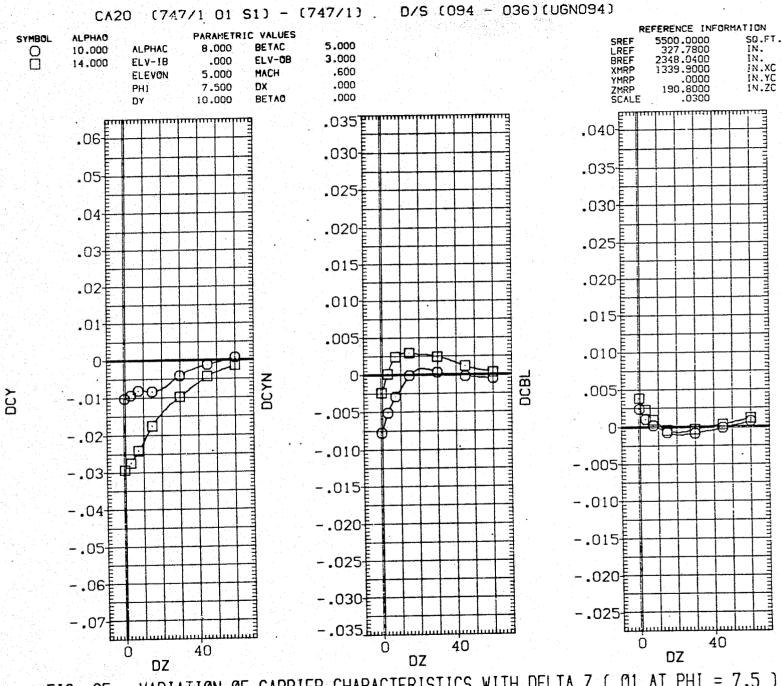


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 814

25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

FIG

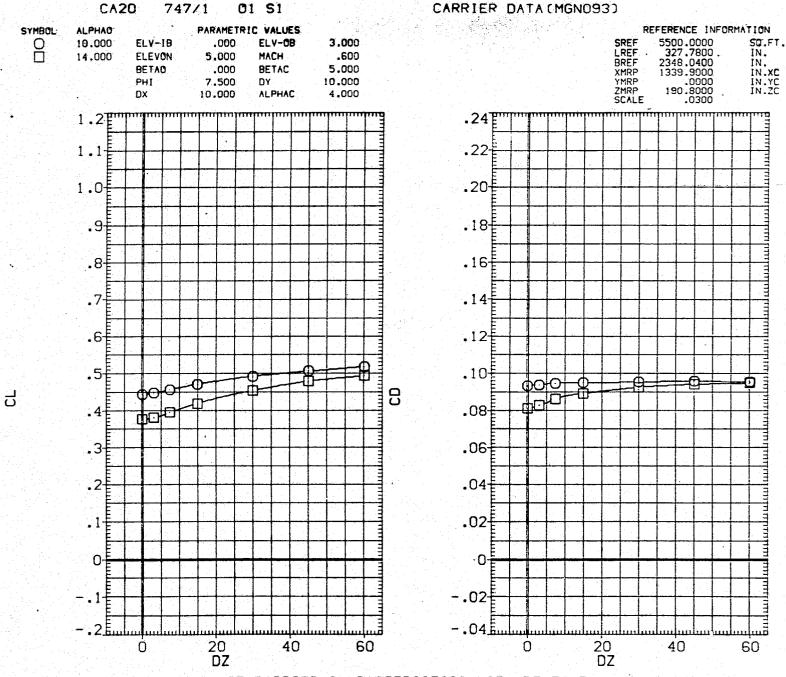
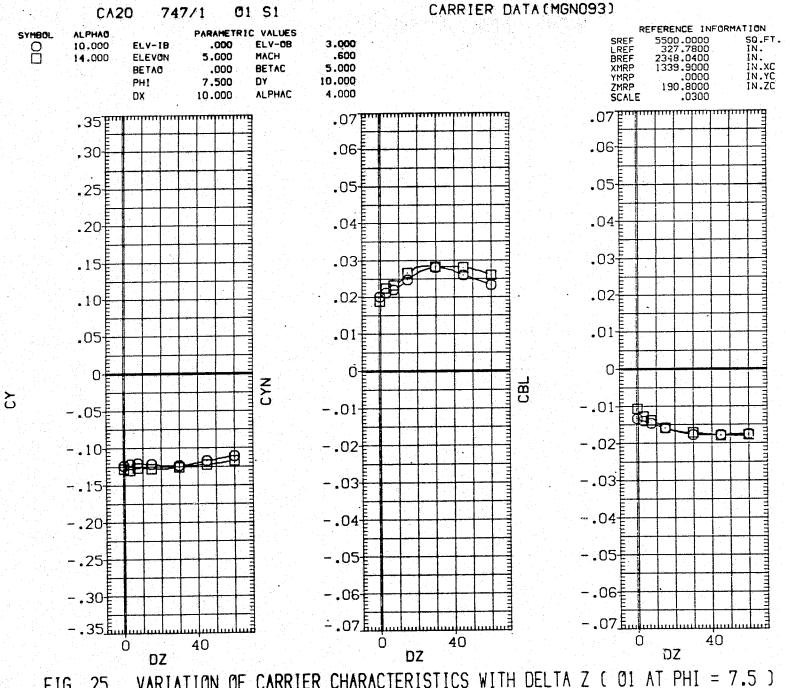


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG PAGE 818

FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 819

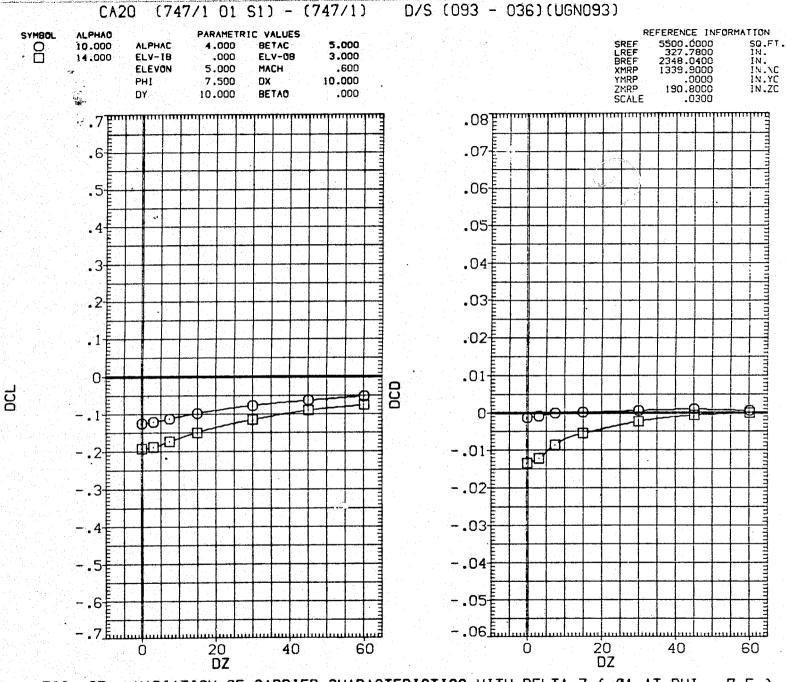
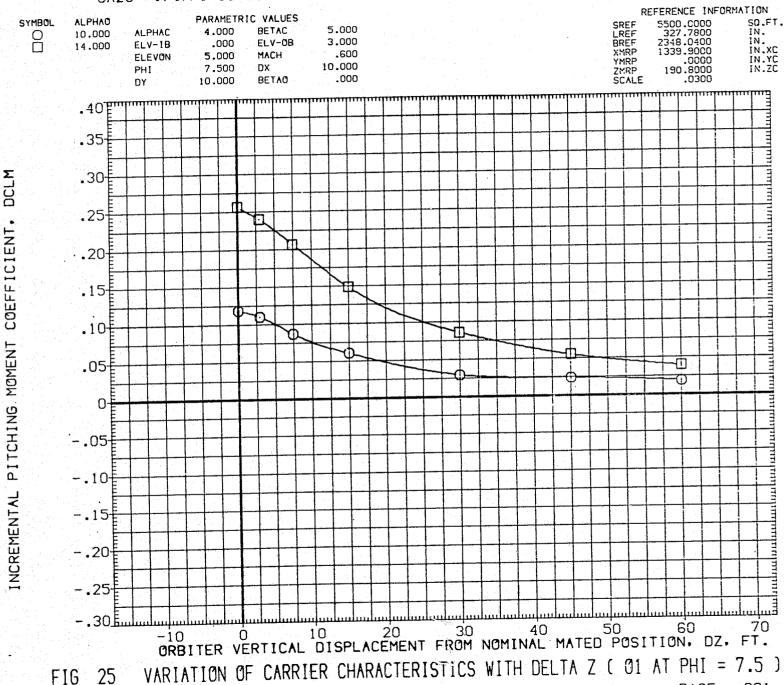
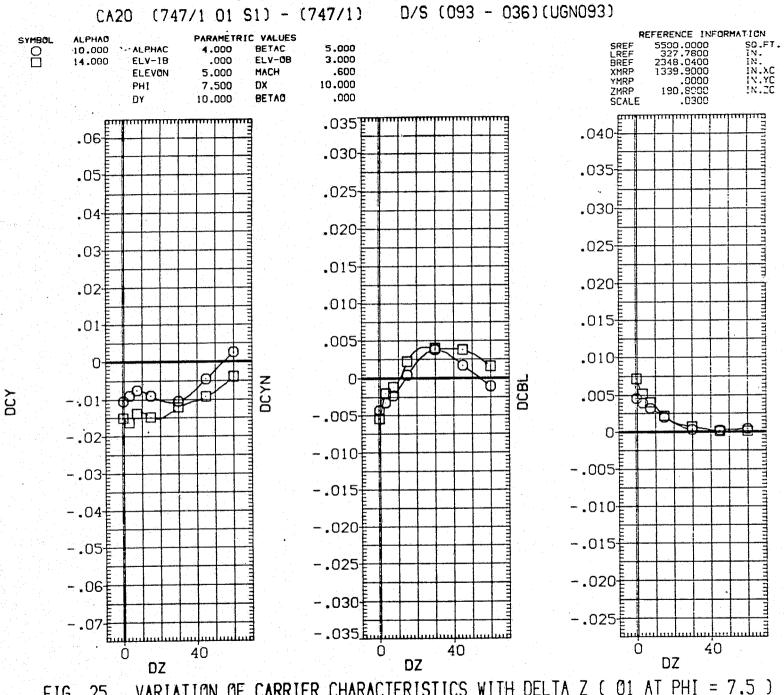


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
PAGE 820

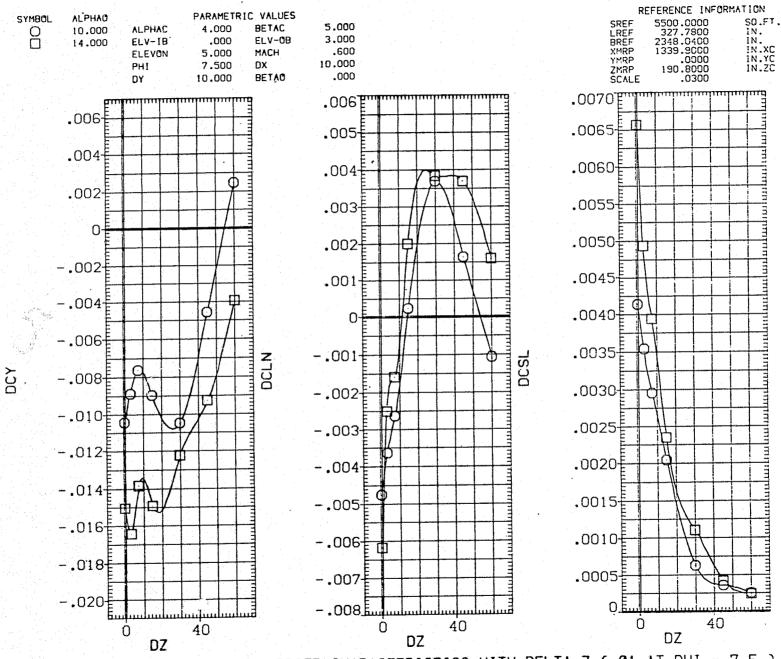


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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG 822 PAGE



VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)

FIG

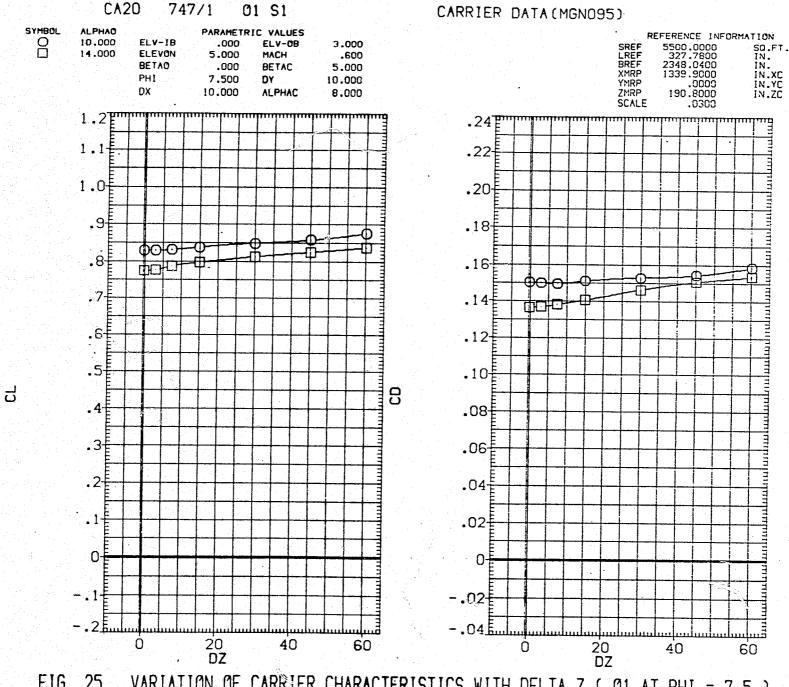
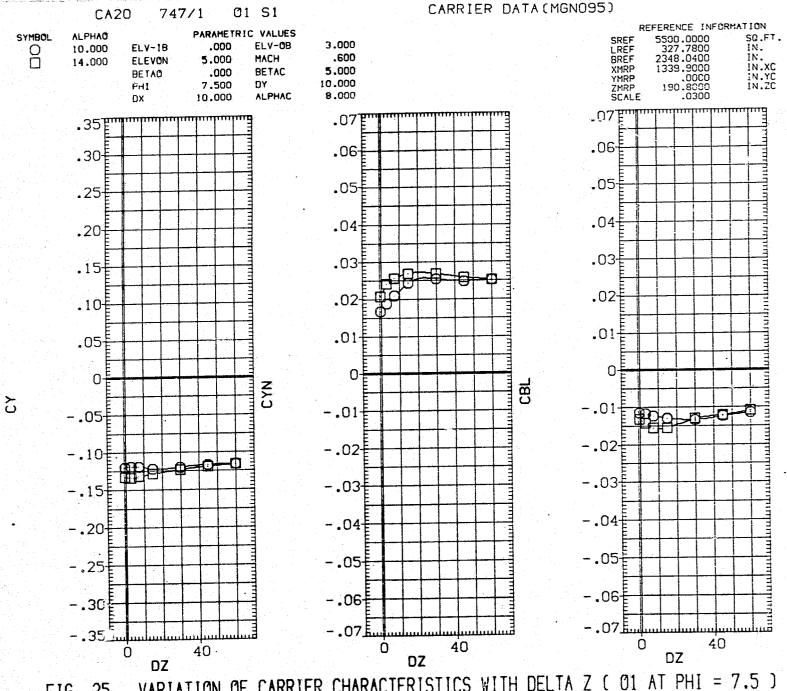
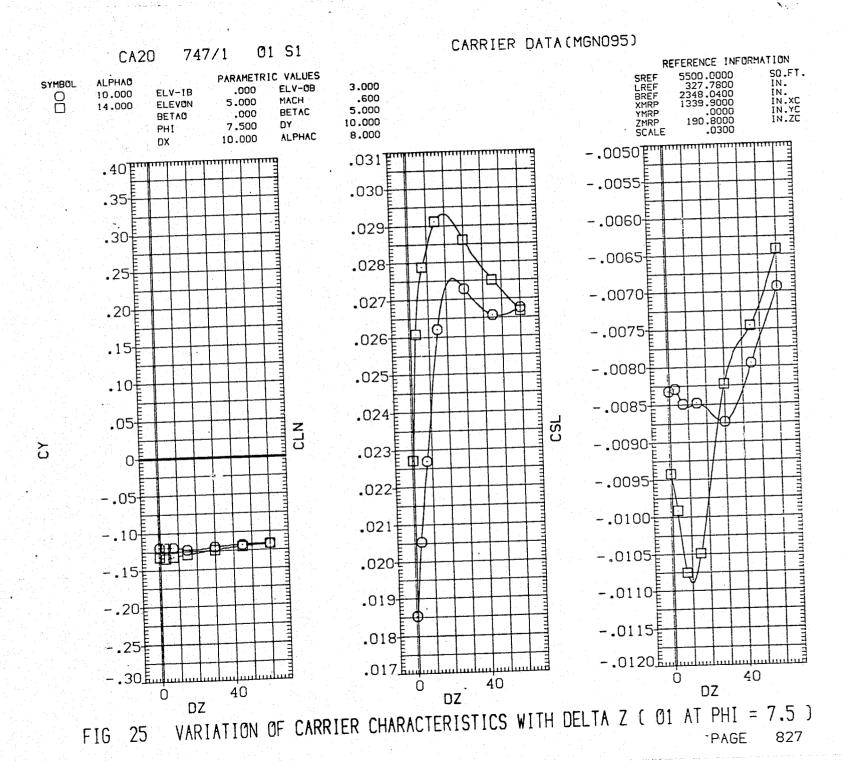


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) FIG PAGE 826



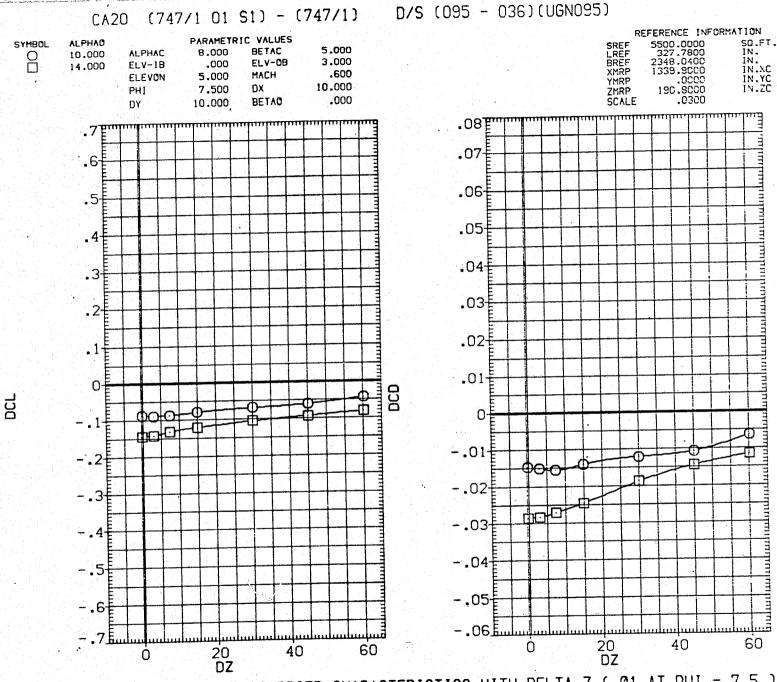
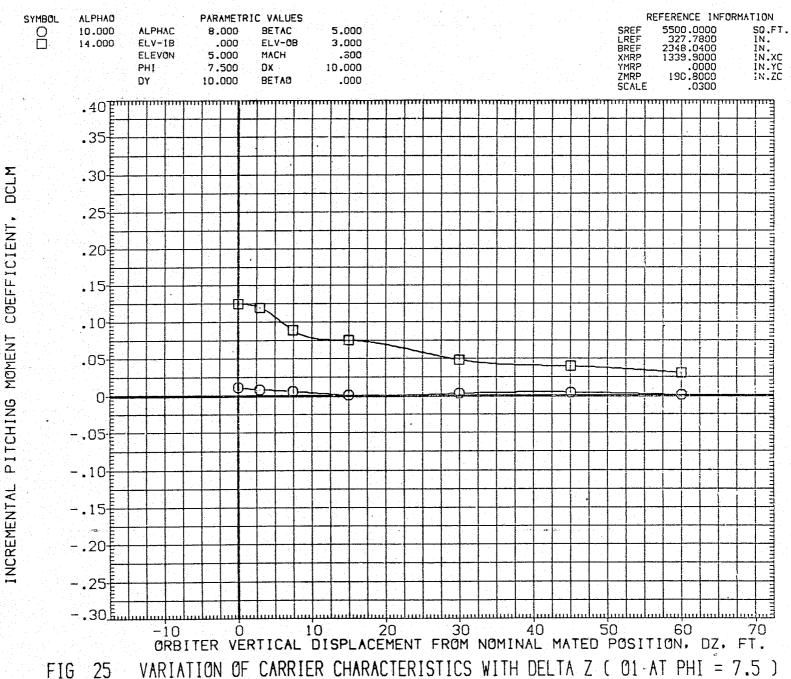
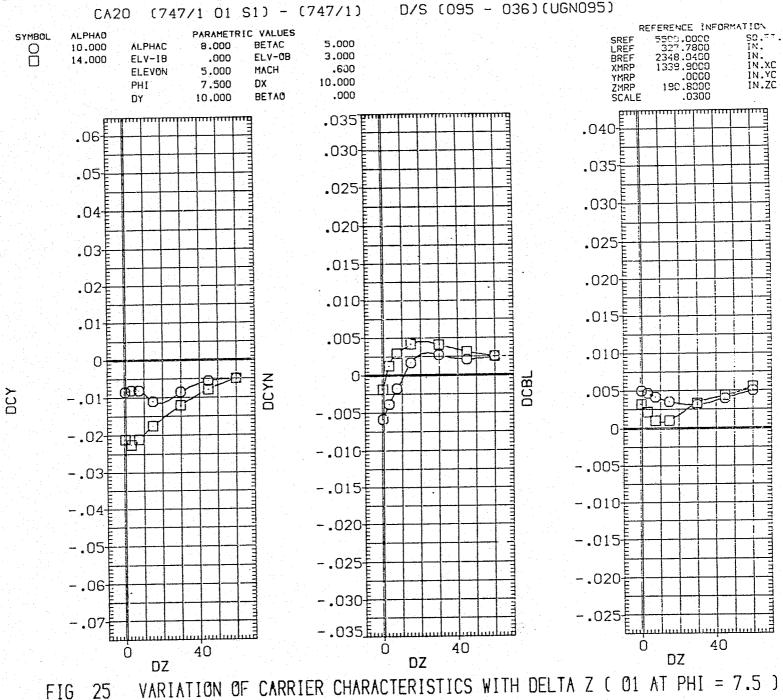


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5) PAGE 830

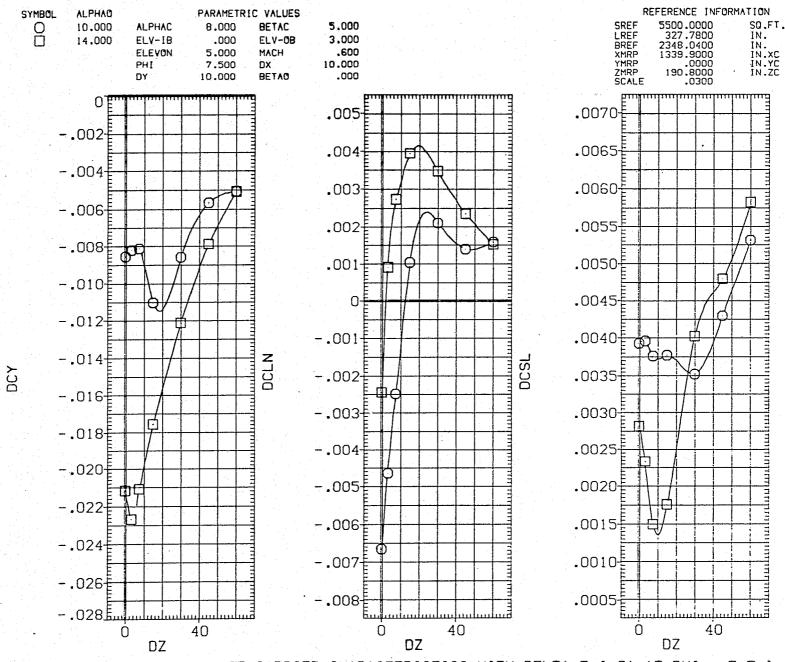


FIG 25 VARIATION OF CARRIER CHARACTERISTICS WITH DELTA Z (01 AT PHI = 7.5)
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